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# ABSTRACT <br> Errors in Constant-Velocity Shaft Couplings 

by<br>Philip M. Remington

A Multiloop spatial system of rotations is used to represent a shaft coupling, allowing a transmission plane which may deviate from the plane of symmetry to be specified. The plane of transmission, for intersecting input and output shaft axes, is the locus of intersections of screw axes that compose the system. By prescribing the transmission plane outside of the plane of symmetry (commonly called the "homokinetic plane" or "bisecting plane") to specified orientations, a phase shift beween the input and output shaft rotational displacements will be quantified. The rotational phase shift between the input and output shafts can be evaluated for a series of configurations to classify the critical deviations from constant-velocity transmission as a function of the transmission plane location.

# ERRORS IN CONSTANTVELOCITY SHAFT COUPLINGS 

by<br>Philip M. Remington

A Thesis<br>Submitted to the Faculty of New Jersey Institute of Technology<br>in Partial Fulfillment of the Requirements for the Degree of Masters of Science in Mechanical Engineering<br>October, 1992

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## APPROVALPAGE

## Errors in Constant Velocity Shaft Couplings

by
Philip M. Remington
$9 / 1 / 92$
Dr. Ian S. Fischer, Thesis Adviser
Associate Professor of Mechanical Engineering, NJIT
$9 / 1 / 22$
De. Rajesh N. Dave, Committec Member Assaciate Professor of Mechanical Engineering, NJTT
$\cdots+\operatorname{degt}_{1,1992}$

Dr. Anthony D. Rosato, Committee Member ,
Assistant Professor of Mechanical Engineering, NJTT

# BIOGRAPHICAL SKETCH 

Author: Philip M. Remington<br>Degree: Master of Science in Mechanical Engineering

Date: October 1992

## Date of Birth:

Place of Dirth:

# Undergraduate and Graduate Education: <br> - Master of Science in Mechanical Engineering, New Jersey Institute of Techology, Newark, N.J. 1992 <br> - Bachelor of Science in Mechanical Engineering. New Jersey Institute of Techology, Newark, N.J. 1990 

Madore Mechanical Engheering

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## CHAPTER 1 <br> INTRODUCTION

### 1.1 Overview

A constant-velocity shaft coupling, often refered to as a CV shaft coupling, ideally provides constant-velocity transmission between two non-colinear shafts. Because of wear and manufacturing tolerances, CV shaft couplings may deviate from the ideal geometry which produces constant output speed for a constant input speed. This study is concerned with calculating the fluctuation in the ouput-shaft speed, relative to the inpuit-shaft speed, when the dimensional differences between the ideal and actual geometry of the CV shaft coupling are known. A CV shaft coupling mathematical model is developed which allows the dimensional differences between the ideal and actual geometry to be specified. This model is then evaluated for fluctuations in the 口utpu-shaft speed relative to the inpur-shaft speed subject to various acual geometric possibilities. This mathematical model will allow the design engineer to directly correlate the CV shaft coupling dimensional colerances and the deviation from constani velocity performance. It will also lead to a better understanding of the wear patterns in the coupling components.


Fig. 1-1 The Cardan or Hooke joint is the most common shaft coupling for rotational transmission through two shafts whose axes of rotation intersect.

### 1.2 Definition of Shaft Coupling

Shaft Couplings are mechanisms used to transmit rotational displacement between two shafts whose axes of rotation are not collinear. The angle between the two shaft axes of rotation is refered to as the shaft angle. The most common shaft coupling in use is the Cardan or Hooke joint, shown in Fig. 1-1, which consists of four perpendicular revolute joints in series whose axes of rotation intersect at a point. The Cardan joint be more easily observed if redundant revolute joints are omitted, as represented in Fig. 1-2. The Cardan joint is nat a constant-velocity coupling, as there will be a cyclic variation in the output-shaft speed rolative to a constant inputshaft speed. The cyclic variation can be derived as a function of the frame angle, for a given shafi coupling.


Fig. 1-2 The Cardan or Hooke joint can be reduced to this form, remaining faithful to the kinematic design. This representation shows the essential kinematic structure, consisting of four perpendicular revolute joints in series. All four revolute joint axes must intersect at a single point.

## 13 Definition of Constant-Velocity Coupling

A constant-velocity shaft coupling is designed such that the output shaft always rotates at the same speed as the input shaft. The general characteristics of constant-velocity shaft couplings has been a topic of great interest to theoretical kinematics researchers and automotive engineers alike (Freudenstein and Maki 1979, Steeds 1937), A brief discussion of existing $C V$ shaft couplings will assist in understanding the design requirements. Constant-velocity shaft couplings, often called "constant-velocity umiversal joints", have been extensively used in front wheel automobile drives and this application has led to many advancements in CV shaft coupling design.


Fig. 1-5 This siliced view of a simpifitied Azeppa CV shaft coupling shows the application of bail-groove joints. Four other balls are siluated similarly about the shaft coupling. The center of curvature of ail spherical grooves meet at the intersection of shaft axes.

The most popular constan-velocity couplinge for shafts whose aves of rotation imergect at a point are those whth ball-groove jointe which have compactness prect sion and durability (Miller 1965). The ball-groove joint was first recognized as a viable option for constant velocity shaft couplings in the late 1920's. Although a large variety of CV shaft coupling designs came from F. E. Myard (1933), A. H. Rzeppa (1928) was recognized as the innovator of the ball-groove CV shaft coupling. A Rzeppa constant-
velocity shaft coupling contains six balls in spherical grooves. These grooves are made concentric about the intersection of the two shaft axes of rotation. The balls are forced against the grooves, which transmits torque between the input shaft and the output shaft (H. H. Mabie 1948).


Fig. 1-4 This sliced piece of a Rzeppa CV shaft coupling shows ine ball cage, difected by the spring loaded pilof lever, positioning a ball in a position symmetric to both the inpul shatt and the output shaft.

The success of the Rzeppa CV shaft coupling is contingent on the precise positioning of the balls in a plane symmetric to both shafts. This plane is called the homokinatic or bisecting plane. This was first accomplished by using a ball cage directed by a spring loaded pilot lever (Rzeppa 1928 ). This ball positioning mechanism can approximately position the balls in the homokinetic plane, which leads to the investigation of the deviation from consiant velocity due to errors in this approzimation.

Later patents on the Rzeppa CV shaft coupling involving eccentric and non. spherical ball grooves lead to ball positioning without need for a pilot lever (Rzeppa 1934,1935). For these designs the tolerances in the grooves must be very precise, to maintain acceptable constant-velocity transmission. These designs adhere to the
principle of the homokinetic plane so the mathematical model that follows remains valid. The deviation from constant-velocity transmission can be attributed to the deviation of the balls from the homokinetic plane.


Fig. 1-5 The double Cardan joint configured with equally opposite shait angles of can produce a sum rotational phase shift of zero between the input shaft and the output shaft.

There are two basic classes of constantvelocity shaft couplings. One class includes the double-Cardan joint driveline which has been extensively studied by Fuscher and Poul (1987) and the double-pode joint driveline by Akbil and Leee (1983). These drivelines operate on the premise that two non-constant-velocity couplings configured in series at equally opposite angles will produce a sum rotational phase shift of zero.

The other class of CV shaft couplings, which is of primary interest in this study, produces constant-velocity through shafts, whoge axes of rotation intersect at a point. This class of shaft couplinge operate on the principle of the homokinetio plane. The homokinetic plane is the plane symmetric to both the input and output thafte, where the location of transmission between the-shafts must occur for constant-velocity
transmission (Surges 1947). Most of all known mechanisms which comprise this class have been derived by Freudenstein and Maki (1979) using a graph-theory synthesis method.

This graph-theory synthesis method consisis of reducing all mechanisms to fundamental components, where rigid bodies are represented by vertices(points) and joints which connect the rigid bodies are represented by edges(lines) which connect the corresponding vertices. The variety of constant-velocity couplinga derived, includes the Tracta, Clemens, Altman, Myard and Reeppa plus some of which have not been implemented into practice. This graph-theory synthesis method revelled several possible combinations of joints, all of which follow the principle of the homokinatic plane.

The principle of the homokinetic plane more specifically includer kinematio requirements in addition to the fundamental geometric requirementa, As noted in Hum (1973), the theory of shaft couplinge designed to produce constantevelocity fransmission, requires an odd number of joints aprayed equally about a central axis. A central joint in the coupling must be loeated in the plane of symmery betweent the imput and outputshafte commonly called the "homokinericplane" ar "blisecting flane", 10 aehieve constant-velocity iransmission. This plane of eymmerry must also contain the intersection of the axes of the joints configured about it. Two examples of ov shaft couplings which explieitly how these requirement discused later, are ahown in Fie. 1.\%. As the homokinetic plane is the ideal location of tran mimsion, the notual plane of fansmissian will be pefered in fe the ransmission plane. The transmisalon plane ideally coincider with the homokinetic plane, however in actuality infer not the case. The effect on constant-velocity.transmission due to the deviation of the transmission plane from the homokinetic plane is developed here.


Fig. 1-6 Spherical construction of CV shaft coupling with extemal locating device positioning the central axis joint in the plane of transmission.

### 1.4 Description of Model

As noted in Freudenstein and Maki (1979), the sum of the freedoms of the joints in a spatially derived CV shaft coupling must be seven to maintain the needed symmerry and freedoms. This note is relevant only for spatial mechanisms. The mathematical model is developed in spherical space and need possess five rotational freedoms in the shaft coupling with specialized constraints for positioning the transmission plane. This configuration was alluded to by Hum (1973) in an approach to syntheaize a highoprecision general CV coupling composed of lower pairs with the minimum number of freedoms. This configuration, like the well-known Rzeppa (Reeppa 1953) or Bendix-Weiss (Surges 1947) ball groove type CV shaft couplings, operates on the prineiple of the homokinetle plane.


Fig. 1-7 (a) \& (b) These mechanisms consist of seven revolute joints arrayed equally about a central axis symmetric to the input and output shafts, which satisfies the principle of the nomokinetic plane.

The 7 R (seven revolute joints in series) spatial mechanisms, shown in Fig. 1.7. suitable for constant-velocity transmission, originally introduced by Myard (193,3), are tharoughly discussed by Hunt (1973), Spherical space, which simplifies the systom, can be achieved by projecting the spatial system of orthogonal rotations in Fig. 1.7 (a), declaring all axes intersect at a point, on that point, where all distances degenerate from the mechanism. This transformation, accomplished by representing the iwa intemediate joints on each side of the mechanism as single linearly independent Joints, satisfies the mobility criteria in spherical space. More simply, the transforma" Hon between any two axes in spherical space can be represented by a single rotation about an axiellinearly independent to both axes. This holds irue for the ransformation between the axes of the central jaint and each maft. This kinematic rule helpe describe the reduction of a one degree-affreedam, 7 R spatial mechanism 10 a wo degree of freedom, sR spherical mechanism with out violating the principle of the homokinetic plane, when the central joint is externally constrained to the homokinetic plane. The additional degree-of-freedom will be utilized to locate the transmission plane independent of the shaft locations. The prospect of utilizing this construction
as a means of analyzing the errors in couplings that operate on the principle of the homokinetic plane does prove to be effective. The deviation from constant-velociry transmission of the CV shaft coupling when the orientation of the transmission plane leaves the bisecting plane or homokinetic plane is the focus of this study.

### 1.5 Motivation

For Rzeppa CV shaft couplings the motion of the pilot lever and clearance of the ball grooves, determine the position of the transmission plane. Contact between the balls and ball grooves in these couplings are the primary location of fatigue (Sutherland 1976, Macielinski 1970). Wear and manufacturing-tolerance deviations from ideal geometry induce output speed fluctuation in what was intended as a constant-velocity coupling. These limitations inspire a need to investigate in detail the effect of the manufacturing tolerances and wear on the ability of the CV shaft coupling to transmit constant-velocity.

By developing a technique, to relate geometric aspecs like bail-groove clearances and the pilat lever motion accuracy to the inputoutput shaft rotational displacement, a better evaluation of designs will be possible. This model is developed on the premise that the trangmission plane can be located independent of the shaft locations. This will allow the model to evaluate the effect of specific locations of the transmission plane on the phase shift befween the input and output shaft rotational displacement at different frame angles. This knowledge will provide a reference for exising shaft couplings io be evaluated and mpdified to optimize shaft transmission toleranoes as a function of the ball groove clearances and pilot lever motion accuracy. The type of CV shaft couplings, where the axes of rotation of the shafts intersect will be investigated in this study and the errors which occur when the homokinetic-plane requirement is not satisfied will be presented and quantified. The relationship between the
ball-groove contact location and the deviation from constant-velocity of the coupling (when the transmission plane is not coincident with the homokinetic plane) is the focus of this study.

### 1.6 Description of the Topic of each Chapter*

Chapter 2 will provide the theoretical background and the development of the CV shaft coupling model. Chapter 3 explains the mathematical procedure which trans. forms the CV shaft coupling model into an explicit solution. Chapter 4 implements the model for various geometric configurations. A brief discussion will also be offered describing characteristics of particular interest found in the results. Chapter 5 will summarize the study and offer insight as to how this study provides essential information for the improvement of $C V$ couplings.

## Chapter 2 DEVELOPMENT OF THE MODEL

### 2.1 General discussion of approach



4

Fig. 2-1 General Black Box Coupling

A conceptual discussion of model development is presented in this chapter. The coupling model will be a multiloop system consisting exclusively of successive orihogonal revolute joins, which have been chosen as fundamental building blocks. To develap the model, a modified graph theory will be used io visualize the construction. Modified graph theory is an extension of the mechanism-categorizing: type-synthesia, technique developed by freudenstein (1978). While in the mechanism type-symthesis graph theory the relationship between links through joints was specified, graphs will be used here to specify the relationships between coordinate frames fixed on the links and the base through coordinate transformations. The constraints that describe this
will be individually contended with as the model is formulated. The criteria such as the degrees-of-freedom and the general principle of the homokinetic plane will be incorporated into with a general "black box" shaft coupling.

### 2.2 Degree-offreedom criteria

A generalized shaft coupling is a one degree-of-freedom mechanism which will provide the first formulation. The following degree-of-freedom equation can be be utilized to implement the first constraint for the generalized shaft coupling.

$$
\begin{gather*}
F=-\lambda L_{\mathrm{ind}}+\sum f_{\mathrm{i}} \\
L_{\mathrm{ind}}=j-l+1 \tag{2-2}
\end{gather*}
$$

where variables are defined such that:

$$
\begin{aligned}
& F=\text { degree of freedom of the system } \\
& \lambda=\text { degree of freedom of the space } \\
& l=\text { number of links in the system } \\
& j=\text { number of rotations in the system } \\
& \sum f i=\text { number of freedoms in the system } \\
& \sum f=\text { number of independent clased loops }
\end{aligned}
$$

This equation will be used evaluate the shafi coupling where all the axes of rotation intersect at a single point, which justifies projecting the system onto spherical space which is a three freedoms space, $\lambda=3$, characterized by $X, Y$ and $Z$ rotations. A simple shafi coupling is defined as having a single independent loop. Lind $=1$; and a single degree of freedom, $F=1$, sa that

$$
\begin{gather*}
1=-3(1)+\Sigma f  \tag{2-3}\\
4=\Sigma f_{1} \tag{2-4}
\end{gather*}
$$

This value corresponds to the Cardan joint, which with four revolute joints, has the minimum number of freedoms possible for a shaft coupling. It is important to note
that for the Cardan joint all axes of rotation of these revolute joints in series are perpendicular to their neighbors, except for the terminal joints. The terminal joints represent the rotation of the input and output shafts. This lack of symmetry of all couplings with an even number of freedoms in spherical space is the reason non-constant velocity transmission occurs. An additional degree of freedom in the system is needed to comply with the odd number of rotations requirement described by Hunt (1973). Since constant velocity is desired, the freedoms of the system will be set to $\Sigma \mathrm{fi}$ $=5$. This result is valid for a double Cardan joint where the intermediate shaft is reduced to zero length. Considering a coupling with five revolute joints, eq (2-1) is evaluated as eq (2-5). Five revolute joints, whose axes intersect at a point, meets the odd number of freedoms requirement, yet represent a two degree-of-freedom system. An external device to align the homokinetic plane will account for the second degree-of-freedom.

$$
\begin{equation*}
-3(1)+5=2 \tag{2.5}
\end{equation*}
$$

To satisfy the symmetry relationship between the shafts, the axes of rotation of the joints configured about the central joint must intersect in the homokinetic plane which creates indererminant configurations of the centrat joint. Indeterminate configHrations occurs at some point during the ariculation of the shaft coupling, when the two neighboring revolute joints axes of rotation become collinear. The indeterminacy of the central joint can only be eliminated by locating it with an extermal device. Thus It is the inemtion of this study folocate the central joint with an external device. The addinional degree of freedom will be allocated to the locating device which will specily the position of the central joint.

## 23 Application of modified graph theory



Fig. 2.2 The central axis joint in an indelerminant configuration, where its iwo nelghboring joints axes are collhear and redundant.

To develop the coupling model, a modified graph theory will be emplayed since the parameters thai describe the fixed frames play an imporant role in this construction and would be excluded with the exising graph theory. Graph theory is a method of reprasenting mechanisms with a combination of verices ( 1 ) and edges (e). The verices, as poins, will represent frames and the edges, as lines, will represemt rolations.

The joint rotations symbolized in Fig. 2-3 can be combined by condition of


Fig. 2.3 Modified graph theory diagram of underconstrained CV coupling.
loop closure as

$$
\begin{equation*}
Z_{1} x_{1} y_{2} x_{3} Z_{3} x_{4}=1 \tag{2.6}
\end{equation*}
$$

where:

$$
\begin{aligned}
& Z_{1}=\text { Input Shaft Rotation } \\
& X_{1}=\text { Imput Shaft Alignment Angle } \\
& Y_{2}=\text { Rotation with axis in ransmission plane } \\
& X_{3}=\text { Output Shaft Alignment Angle } \\
& Z_{3}=\text { Oupput Shaft Rotation } \\
& X_{4}=\text { Prame Angle }
\end{aligned}
$$

The choce of rotation axes were selected to mainain symmery benwern the shafts and adhere to the common convention of shaft rotations about Z axes. By defining the frame angle as a parameter, the system satisfies the five joint freedoms constraint.

This system must be expanded to allow the location of the central joint axis to be specified. An additional relationship between the central joint axis and the fixed frame will offer a means of prescribing the orientation of the transmission plane. The central joint axis lies in the transmission plane according to the principle of the homokinetic plane. Consider expanding the central joint rotation into two separate angles about the same axis. This expansion of the system will not effect the freedoms of the system since the new rotations, $Y_{1}$ and $Y_{3}$, shown in Fig. 2-4, are redundant with the original rotation $Y_{2}$. The frame between the redundant transmission angles is located in the transmission plane. The magnitude of the rotations $Y_{1}$ and $\gamma_{3}$ are proportioned to define a frame located in the transmission plane so that


Fig. 2-A CV coupling with frame in iransmission plane.

$$
\begin{equation*}
\gamma_{2}=r_{1} \gamma_{3} \tag{2.7}
\end{equation*}
$$

The frame angle will also be split into two redundant angles at this stage in the development of fhe madel, The new frame developed between the $Y_{1}$ and $\gamma_{3}$ rolation will be conneced to the new fransmission frame by an additional motation. This rotation between a frame in the transmission plane and the flxed frame will have its axis perpendicular to the transmission plane, and will articulate the central-joint axis about the prescribed transmission plane so that the rotation through the frame angle, which is the sum of the rotations $X_{21}$ and $X_{23}$, shown in Fig. 2-5, is


Fig. 2-5 CV Coupling with Rotation in Transmission Plane

$$
\begin{equation*}
X_{4}=X_{21} X_{23} \tag{2-8}
\end{equation*}
$$

The $Z_{2}$ transmission rotation, shown in Fig. $2-5$, whose axis is perpendicular to the axes of the $\gamma_{2}$ and $X_{4}$, rotations, defines an axis perpendicular to the transmission plane. The transmission plane is the locus of positions of the central-joint through articulation of the coupling. Each location of the central-joint axis will be constrained about this rotation which defines the transmission plane. The transmission rotation axis can only achieve positions perpendicular to the frame angle axis in this formulation, as one single rotation is installed to locato the transmission plane about the frame angle.

By expanding the iransformation specifying the ground link (frame angle) with a set of two $Y$ orotations, $Y_{21}$ and $Y_{23}$,shown in Fig. 2-6, it will be possible to avaluate all transmission plane location possibilities. By prescribing the two Vrotations with a product equal to the identity marix, they will not offect the original configuration. It is critical to maintain characterisics of the original system to mainain the frame angle about the $X$-axis. This is the final configuration which can will be developed into a mathematical model.

The mechanism can be considered as a two-loop structure, represented by the modified graph theory diagram shown in Fig 2-6, where,


Fig. 2.6 Moditled Graph Theory Diagram of Solution Model

Input Loop:

$$
\begin{gather*}
Z\left(\theta_{1}\right) X\left(\alpha_{1}\right) Y\left(\eta_{1}\right) Z\left(\theta_{2}\right) Y\left(\eta_{21}\right) X\left(\alpha_{21}\right)=1  \tag{2-9}\\
\text { Output Loop: } \\
Z\left(\theta_{3}\right) X\left(\alpha_{3}\right) Y\left(\eta_{3}\right) Z\left(\theta_{2}\right) Y\left(\eta_{23}\right) X\left(\alpha_{23}\right)=1 \tag{2.10}
\end{gather*}
$$

where,

$$
\begin{aligned}
Z_{1,} Z_{3} & =\text { Input and Output Shaft Rotations (variables) } \\
Y_{1}, Y_{3} & =\text { Central Axis Joint Rotations (variables) } \\
X_{1}, X_{3} & =\text { Outer Joints Rotations (variables) } \\
Z_{2} & =\text { Rotation in Transmission Plane (variables) } \\
X_{21}, Y_{21} & =\text { Input Shat Frame Rotations ( parameters ) } \\
X_{23}, Y_{23} & =\text { Ouput Shaft Frame Rotations ( paramelers) }
\end{aligned}
$$

These rolations will be thoroughly explained in the mathematical model that follows. To confirm the validity of the modifications the model, the degree of freadom equation will be used. For this model must maintain I degree of freedom in the system, $F=1$, through two independent loops, $L_{\text {ind }}=2$. The number of freedoms is predicted to be $\Sigma f i=7$. The number of freedoms corresponds to the number of variable rotations in the system which confirms this system is valid.

## CHAPTER 3

## DESCRIPTION OF MATHEMATICAL MODEL

### 3.1 General discussion of mathematical techniques

This model is developed on the premise that the all the axes of rotation intersect at a single point in space, which allows the model to be projected onto spherical space. This configuration is particularly relevant to the ball-groove CV joime as the balls of the central-joint maintain a circular orbit in the transmission plane. Although the central-joint of a CV joint is ideally constained to the bomokinetic plane, it is not particular to the instantaneous radius, only to the instantaneous angular position. From this principle the central-joint need not perform a circular orbil in the transmis. sion plane. This model will assume a circular orbit which maintains the exact instantaneous angular position. This provision simplifies the model, yet does not effect the imput-output relationship. The static forces and torques on the balls in the ball-grooves of a CV joint was accomplished by Bellomo (1975), which exchudes atl non-constant-velocity configurations due to mechanical blorances and wear, yel confirms the circular orbit of the central-axis joint.

The term "I" denotes the identity transformation and for spherical space represents a diagonalized $3 \times 3$ unity matrix. Each of the rotational transformations are 3-dimensional proper orthogonal rotation matrices in this solution scheme, which can be denoted by

$$
\begin{aligned}
& Z(\theta)=\left[\begin{array}{ccc}
C_{\theta} & -S_{\theta} & 0 \\
S_{\theta} & C_{\theta} & 0 \\
0 & 0 & 1
\end{array}\right] \quad Y(\eta)=\left[\begin{array}{ccc}
C_{\eta} & 0 & S_{\eta} \\
0 & 1 & 0 \\
-S_{\eta} & 0 & C_{\eta}
\end{array}\right] \quad X(\alpha)=\left[\begin{array}{ccc}
1 & 0 & 0 \\
0 & C_{\alpha} & -S_{\alpha} \\
0 & S_{\alpha} & C_{\alpha}
\end{array}\right](3-1) \\
& C_{\theta}=\cos (\theta), \quad S_{\theta}=\sin (\theta), \quad \text { and } T \theta=\tan (\theta) .
\end{aligned}
$$

### 3.2 Application of mathematical techniques to model

The rotational transformations are concatenated in a double-closed chain to represent the CV shaft coupling model. This is achieved using a closed form solution method. In this context "closed form" means a solution method based on analytic expressions, such that non-recursive calculations suffice to arrive at a solution (Craig 1986). Double-closed chain implies the use of loop closure for a system of rotations which achieve closure with two interdependent loops. Each of the elements of the matrix product of a closed chain represent a structure equation descriptive of the coordinate ransformathon in that loop, which comes to $n^{2}$ equations for $n$-space. The result is a constrain on the allowable values of the joint rotations in the chain, which maintain the closure of the chain (McCarhy 1990).

The input-output rotational displacement relationship can be solved from the two closed loop equations. The relationship beween the input shaf and the transmission rotational displacement is solved in the first loop. The corresponding transmission rotational displacement is applied to solve for the output-shaft rotational displacement in the second loop. The total system at this point has one degree of freedom with four parameters to describe the locations of the rotations of links
connected to the fixed-frame. As was previously mentioned, the $Y$-axis rotations are equally opposite which reduces the fixed frame description to three parameters. The single degree of freedom will be the rotation of the input shaft to obtain the corresponding output shaft rotational displacement.

To maintain the customary convention of the imput and output shaft axes as Z-axes pointing outward, some modifications will be employed. The rotations in the transmission plane for each of the loops will be supplementary. The necessary changes can be seen in Fig. 3-1. The $\theta_{2}$ and $\theta_{3}$ values in the output loop will have signs opposite from the sense of the input loop. The $\alpha_{1}$ and $\alpha_{3}$ values are now the supplement of the frame angle also called the driveline angular offset. These modifications produce a genuinely simplified model as both sides of the shaft coupling are now modeled identically.

This model maintains essential characteristics of the original configuration while incorporating the transmission plane lacating device. The first closed loop of rotations relating the input shaft to the transmission plane takes the form $Z X Y Z Y X=1$. The second closed loop of rotations relating the transmission plane to the oulput shaft and takes the identical form $\mathcal{Z X Y Z Y X}=1$.

The model is represented in Fig. $3-1$ as successive coordinate frames. These coordinate frames are characterized by the rotation at that location in the closed loop chain. This figure makes the mathematical model easier to understand and visualize.


Fig. 3-1 General Coupling with Fixed Transmission Plane.

The loop equations are
Input Loop:

$$
\begin{gathered}
Z\left(\theta_{1}\right) X\left(\alpha_{1}\right) Y\left(\eta_{1}\right) Z\left(\theta_{2}\right) Y\left(\eta_{21}\right) X\left(\alpha_{21}\right)=1 \\
\text { Outpui Loop: } \\
Z\left(\theta_{3}\right) X\left(\alpha_{3}\right) Y\left(\eta_{3}\right) Z\left(\theta_{2}\right) Y\left(\eta_{23}\right) X\left(\alpha_{23}\right)=1
\end{gathered}
$$

where

$$
\begin{array}{ll}
Z_{1}\left(\theta_{1}\right), Z_{3}\left(\theta_{3}\right) & =\text { Input and Output Shaft Rotations (variables ) } \\
Y_{1}\left(\eta_{1}\right), Y_{3}(\eta 3) & =\text { Central- Axis Joint Rotations (variables) } \\
X_{1}\left(\rho_{1}\right), X_{3}\left(\alpha_{3}\right) & =\text { Outer Joint Rotations (variables ) } \\
Z_{2}\left(\theta_{2}\right) & =\text { Rotation in Transmission Plane (variables) } \\
X_{21}\left(\alpha_{21}\right), Y_{21}\left(\eta_{21}\right) & =\text { Input Shaft Frame Dimensions (parameters) } \\
X_{23}\left(\alpha_{23}\right), Y_{23}\left(\eta_{23}\right) & =\text { Output Shaft Frame Dimensions (parameters ) }
\end{array}
$$

The fixed frame is parameterized by the driveline offset angles, $\alpha_{21}$ and $\alpha_{23}$ for the input and output loops, respectively. The corresponding fixed frame angles, $\eta 21$ and $\eta 23$ will occur about the axis perpendicular to the driveline offset angles in the transmission plane. The input-shaft rotation is input-variable angle $\theta_{1}$ and the output shaft rotation is output-variable angle $\theta_{3}$. These loops both have the same transmission rotation through intermediate variable angle $\theta_{2}$, in the transmission plane which connects them. The sign of the transmission rotation will be opposite from one loop to the other for convienient notation. Let it be noted that the $Z_{2 \text {-transmission rotation }}$ is the rotation of the locating device which constrains the Y-rotation, central-joint axis to the transmission plane.

The input loop is evaluated for function $\Theta_{2}=\Phi\left(\theta_{1}, \eta_{21}, 0_{21}\right)$, the transmission rotation, by expanding the rotation transformations in the form,

$$
\begin{equation*}
X\left(a_{1}\right) Y\left(\eta_{1}\right) Z\left(\theta_{2}\right)=\left(Y\left(\eta_{21}\right) X\left(a_{21}\right) Z\left(\theta_{1}\right)\right)^{\mathrm{T}} \tag{3.2}
\end{equation*}
$$

to obrain intermediate variable $\Theta_{2}$, element ( 1,2 ) is divided by $(1,1)$,

$$
\begin{equation*}
T_{\theta_{2}}=\frac{-S_{\theta_{1}} C_{\alpha_{21}}}{C_{\theta_{1}} C_{\eta_{21}}+S_{\theta_{1}} S_{\eta_{21}} S_{\alpha_{21}}} \tag{3-3}
\end{equation*}
$$

The outpui loop is evaluated for function $\Theta_{3}=\Phi(\Theta 2,723,023)$ by expanding the equation in the form,

$$
\begin{equation*}
Z\left(\theta_{3}\right) X\left(a_{3}\right) Y\left(\eta_{3}\right)=\left(Z_{( }\left(\Theta_{2}\right) Y\left(\eta_{23}\right) X\left(\alpha_{23}\right)\right)^{\mathrm{T}} \tag{3-4}
\end{equation*}
$$

and to abtain output variable $\mathrm{O}_{3}$, alement (1,2) is divided by (2,2),

$$
\begin{equation*}
T_{\Theta_{3}}=\frac{-S_{\Theta_{2}} C_{\eta_{23}}}{C_{\theta_{2}} C_{\alpha_{23}}+S_{\Theta_{2}} S_{\eta_{23}} S_{\alpha_{23}}} \tag{3-5}
\end{equation*}
$$

It is essential to take make provisions for quadrant changes, which the general arctangent function does not, in refining the solution. The function, $\operatorname{ATAN} 2(\sin (\theta), \cos (\theta))$, a FORTRAN notation, can be incorporated into the solution for both $\Theta_{2}$ and $\Theta_{3}$ to obtain the correct quadrants in the solution.

$$
\begin{align*}
& \Theta_{2}=\operatorname{ATAN} 2\left(-S_{\theta 1} C_{\alpha 21}, C_{\theta 1} C_{\eta 21}+S_{\theta 1} S_{\eta 21} S_{\alpha 21}\right)  \tag{3-6}\\
& \Theta_{3}=\operatorname{ATAN} 2\left(-S_{\Theta 2} C_{\eta 23}, C_{\Theta 2} C_{\alpha 23}+S_{\Theta 2} S_{\eta 23} S_{\alpha 23}\right) \tag{3-7}
\end{align*}
$$

These equations together make up the generalized input/output relationship for the CV coupling including the configurations where constant-velacity constraints are not satisfied. By substituting eq (3-6) into eq (3-7) can be obtained the generalized coupling input/output equation,

$$
\begin{gather*}
\Theta_{3}=\operatorname{ATAN} 2\left(\Psi_{1} S_{\theta 1}, \Psi_{2} C_{\theta 1}+\Psi_{3} S_{\theta 1}\right)  \tag{3-8}\\
\text { where } \\
\Psi_{1}=-C_{\alpha 21} C_{\eta 23} \\
\Psi_{2}=C_{\alpha 23} C_{\eta 21} \\
\Psi_{3}=S_{\eta 21} S_{a 21} C_{a 23}+C_{a 21} S_{a 23} S_{\eta 23}
\end{gather*}
$$

The focus of this study is the rotational phase shift between the input and output shaft rotational displacements for specified configurations so the following analysis will deal with the deviation,

$$
\begin{equation*}
\Theta_{\mathrm{dev}}=-\Theta_{3}\left(\theta_{1}, \eta_{21}, \alpha_{21}, \eta_{23}, \alpha_{23}\right)-\theta_{1} \tag{3-9}
\end{equation*}
$$

### 3.3 Refinement of mathematical model

A scheme to make the solution more amenable will simplify the variable angles into three functions of the original variables and the input shaft rotation. The new variables will be defined as the frame angle and two orthogonal locating-device transmission angles, which are illustrated in Fig. 3-2. The provision that angle $\eta \mathrm{dev}=\eta 21=\eta 23$ is incorporated to add simplicity to the system, without loss of generality, regarding the shaft couping. This provision reduces the five degrees of freedom system to four, although these two original variables were previously defined as equal. The frame angle used throughout the development can now be taken to full advantage by the transformation $X_{4}=X_{21} X_{23}$ which represents rotation through the angle $\alpha_{4}$ $=\alpha_{21}+\alpha_{23}$. Again the modification discussed and illustrated in the first section of this chapter leads to the use of a supplementary value. The angle, $\alpha_{4}$ is now redefined as the driveline offset angle, which is the supplement of the frame angle illustrated in Fig.3-2. The frame angle is the angle between the shafts about an axis perpendicular to both shafts. The frame angle between the input and output shafts is angle aframe $=\pi-\left(\alpha_{2} 1+\alpha_{23}\right)=\pi-a_{n}$, which uses the property that the frame angle is the supplement of the driveline offset. For this system the homokinetic plane is maintained whon $\alpha_{2}$ $=-\infty 3$ and the transmission plane under the influence of the alphatransmission angle deviates from the homokinetic plane by the difference between these angles now defined as angle $\alpha$ dev $=(\alpha 21-\alpha 3) / 2$.

These manipulations produce three desired functions which define anglesazt, a3, $\eta 21$ and $\eta 23$ in terms of angles $\eta d e v, \alpha$ dev and difame.

$$
\begin{gather*}
\eta_{21}=\eta_{23}=\eta_{\mathrm{dev}}  \tag{3-10}\\
\alpha_{21}=\left(\pi-\alpha_{\text {frame }}\right) / 2+\alpha_{\mathrm{dev}}  \tag{3-11}\\
\alpha_{23}=\left(\pi-\alpha_{\text {frame }}\right) / 2-\alpha_{\mathrm{dev}} \tag{3-12}
\end{gather*}
$$



Fig. 3-2 New variables for amerabie solution

Utilizing these new variables with some algebraic and trigonometric manipulation a more amenable solution will be developed. This is a concise generalized solution for the coupling input/output equation for specified transmission planes.

$$
\begin{gather*}
\Theta_{\mathrm{dev}}=-\operatorname{ATAN} 2\left(\Psi_{1} S_{\theta 1}, \Psi_{2} C_{\theta 1}+\Psi_{3} S_{\theta 1}\right)-\theta_{1}  \tag{3-13}\\
\Psi_{1}=S\left(\alpha_{\mathrm{dev}}-\alpha_{\text {frame }} / 2\right) \\
\Psi_{2}=S\left(\alpha_{\mathrm{dev}}+\alpha_{\text {frame }} / 2\right) \\
\Psi_{3}=T_{\eta d \text { devera frame }}
\end{gather*}
$$



Fig. 3-3 Locating Scheme for Era Transmission Axis.

The frame angle is comprised of a single $X$-rotation about an axis perpendicular to both the input and output shafts. The transmission angle $\alpha_{\mathrm{d}} \mathrm{dev}^{\text {occurs }}$ about the same axis and describes the angle between the transmission plane and the bisecting plane. The transmission angle $\eta$ dev occurs about an axis in both the transmission plane, determined by the transmission angle $\alpha_{\text {dev }}$, and the plane that contains both shafts. These two angles give the ability to locate the transmission in any orientation desired.


Fig. 3-4 Locating Scheme for Alpha Transmission Axis.
The transmission angle $\alpha_{\text {dev }}$ defines the location of the transmission plane with respect to the homokinetic plane about an axis perpendicular to both shafts. The
transmission axis $\eta$ dev is located at the intersection of transmission plane located by the angle $\alpha_{\mathrm{dev}}$ and the plane that contains both shafts. The angle $\eta$ dev and the angle adev are parts of the fixed frame that give the ability to achieve all possible transmission plane orientations.

## CHAPTER 4 RESULTS AND DISCUSSION

### 4.1 Discussion of Amenable Variable Scheme



Fig. 4. Alphatransmission angle about axis perpencticular to both shafts

The ransmission angles adev and ndev provide a means of referencing the angular phase shifi between the input and output shafts to the location of the tranemission plane. With the new variables devaloped in the previous chapter, this showe to be a more amenable scheme, as this equation shows.


Fig. 4-2 Eta transmission angle about axis in plane with both shafts

$$
\begin{gather*}
\Theta_{\mathrm{dev}}=\operatorname{ATAN} 2\left(\Psi_{1} S_{\theta 1}, \Psi_{2} C_{\theta 1}-\Psi_{3} S_{\theta 1}\right)-\theta_{1}  \tag{4-1}\\
\Psi_{1}=C\left(\left(\alpha_{\text {frame }}-\alpha_{d e v}\right) / 2\right) \\
\Psi_{2}=C\left(\left(\alpha_{\text {frame }}+\alpha_{\text {dev }}\right) / 2\right) \\
\Psi_{3}=T_{\text {idev }} S_{a \text { frame }}
\end{gather*}
$$

### 4.2 Presenation of Soliutions for Trial Parameters

It is important to note each of the deviaion parameters in this model are referenced the homokineuto plane. The frame angle determines the location of the homokinetio plane, implicitly. This allows the deviation of the iransmission plane from the homokinetic plane to be controlled exclusively by the angles $\alpha$ dev and $\eta \mathrm{dev}$.

The deviation through a complete articulation of the coupling for the case with angle $\eta$ dev $=0^{\circ}$ is shown in Figures 4-3,4-4.


Fig. 4-3 input/Oufput shaft deviation with $\alpha_{\text {frame }}=120$ degrees for various $\alpha_{\text {dov }}$ values.


Fig. 4-4 Input/Output shaft deviation with aframe $=150$ degrees for various $a$ dev values.

The deviation through a complete articulation of the coupling for the case with $a_{\mathrm{dev}}=0^{\circ}$ is shown in Figures 4-5,4-6.


Fig. 4-5 Input/Ouipur shaft deviation with atrame $=120$ degrees for various indev values.


Fig. 4-6 Input/Output shaft deviation with arrame $=150$ degrees for various $\eta$ dev values.

A general set of solutions incorporate combinations of the transmission angles $\alpha_{\text {dev }}$ and $\eta$ dev for frame angles, $\alpha$ frame $=120^{\circ}$ and $150^{\circ}$ as shown in Figures 4-7,4-8


Fig. 4-7 input/Ouiput shaft deviation with aframe $=120$ degrees and $\alpha$ dev $=1 / 2$ degree for various joev values.


Fig. 4-8 input/Output shaft deviation with $\alpha_{\text {trame }}=150$ degrees and $\alpha_{\text {dev }}=1 / 2$ degree for various $\eta \mathrm{dev}$ values.


Fig. 4-9 input/Output shaft deviation with aframe $=120$ degrees and $\alpha_{\text {dev }}=1$ degree for various ider values.


Fig. 4-10 Input/Output shaft deviation with $\alpha_{\text {frame }}=120$ degrees and $\alpha_{\text {dev }}=3 / 2$ degree for various $\eta_{\text {dev }}$ values.


Fig. 4- 11 input/Oufput shaft deviation with $\alpha$ frame $=120$ degrees and $\alpha$ dev $=2$ degree for various inder values.


Fig. 4-12 Input/Output shaft deviation with $\alpha_{\text {frame }}=150$ degrees and $\alpha_{\text {dev }}=2$ degree for various $\eta$ dev values.

## 43 Discussion of Results

The CV coupling deviation solution provides a complete relationship between the input and output shaft rotations for all possible configurations of the transmission plane. It is essential to delineate the effects $\alpha_{\mathrm{dev}}$ and $\eta \mathrm{dev}_{\text {, }}$ on the input/output relationship throughout various frame angles. These two angles give absolute control over the deviation of the transmission plane with respect to the homokinetic plane. The effects of these angles can be described by grouping the orientations of the central-axis joint into quadrants.

These quadrants represent sub-sets of all possible central-axis joint locations. The quadrants are characterized by the quarter-spaces whose boundaries are centralaxis joint locations which produce zero deviation between the input and output shafis. These boundaries correspond to the central-axis joint located in the homokinetic plane or the plane that contains both shafts. As either of these boundaries are crossed by the central-axis joint the deviation between the input and output shafts will swith signs. These ideas help the designer get a quick idea of the general effects of the location of the central-axis joint on the input/output relationship. The use of these quadrants will be elaborated upon, as this concept will assist in the discussion of the iwo deviation angle's effects upon the inpul/output relationship.

The adev deviation angle, when it is the only deviation of the ransmission plane from the homokinetic plane, will send the central-axis joint through all four quadrants. This will assure the impu/output relationship changing signs four times for one full rofation of the imput shaft. The adev deviation angle hag less effect on the input/output relationship per degree of deviation then does the \#dev deviation angle, although it is the most volatile with respect to the frame angle. The implicit locating of the homokinetic plane at one-half the frame angle directly effects the reference frame of the $\alpha_{\text {dev }}$. The $\alpha_{\text {frame }}$ axis coincides with the $\alpha_{\text {dev }}$ axis which shows the direct


Fig. 4-13 Views of adev io destinguish the fransmission plane passing through four quadranks
correlation between the aframe angle and the reference frame for the $\alpha$ dev angle. This deviation angle is the angle of greatest concern to designers as this angle contends with the axis that requites the most modification for a givenchange in the aframe angle, The $\alpha_{\text {dev }}$ reference changes by one-half the change in the frame angle as it is referenced from the homokinetic plane.


Fig. A-14 Views of $\eta$ dev to destinguish the transmission plane passing through iwo quadrants

The $\eta$ dev deviation angle rotates about the axis in the iransmission plane which is in the plane with both shafts. When this is the only angle of deviation the central axis joint will pass through two quadrants of the same sign. The location of the $\eta$ dev axis when $\alpha$ dev is zero is the line fouching all four quadrants. By passing through this line twice the inpu/output relationship will mainain the same sign. The ndev deviation angle has a greater effect on the input/output relationship per degree of deviation from the homokinetic plane yet, is nearly independent of the positioning of the locating device. The $\eta$ dev deviation axis is orthogonal to the axis of the frame angle and thus the $\eta$ dev reference is ideally unchanged by the bisecting angle variations.

The $\alpha_{\text {dev }}$ deviation angle, most dependent on the locating device, occurs about an axis perpendicular to both shafts. The $\alpha$ frame angle rotates about the same axis as the $\alpha$ dev angle. It should be noted that the locating of the homokinetic plane is at one-half the $\alpha_{\text {frame }}$ angle and in the plane with both shafts, so the $\alpha_{\text {dev }}$ angle as it is referenced to the homokinetic plane is most affected by the locating device. The variation in the position of the homokinetic plane about this axis during variations in the $\alpha$ frame angle, although implicitly defined in the solution, must be considered carefully. Although the locating devices in primarily positioning the transmission plane to accommodate the new frame angles, the input/output relationship may still develop deviation by the $\eta$ dev angle.

The input/output relationship, ©dev for one rotation of the imput shaft, passes through four quadrant boundaries when $\alpha_{\text {dev }}$ is non-zero. This implies the $\Theta_{\text {dev }}$ value will switch signs four times during one rotation of the input shaft. These solutions are double frequency periodic functions which are symmerric about zero when $\eta$ dev is zero. The larger the driveline offset angle the sharper the change and the larger the amplitude of the ©dev deviation values. The $\eta$ dev also produces a double frequency periodic function, although the $\Theta_{\text {dev value does not change signs. }}$ At the small deviation angles that can be atributed to tolerances and wear, the solutions appear to be sinusoidal. This is not completely true as extreme deviation angles and frame angles will show on interesting phenominon.

This phenominon becomes noticable for large deviation angles becomes particularly relevant when the central-axis joint is approaching or departing from a boindary. Note the twa sides of the coupling, about the transmission plane, oan be viewed as identical mechanisms. As the central axis joint approaches a boundary the $\Theta_{\text {dev }}$ values become smaller. This reducing of the $\Theta_{\text {dev value can be attributed to }}$ motions in the mechanism on each identical side of the mechanism. If the output side
of the mechanism is closer to the boundary configuration then the input side, a sharp change in the $\Theta_{\mathrm{dev}}$ value will occur. This occurs when the $\Theta_{\text {dev }}$ is appraoching zero from any quadrant. These function characteristics provide additional insight into the more volatile central-axis joint locations.

Although the amplitude of the $\Theta_{\text {dev }}$ deviation is small under the influence of the $\alpha_{\text {dev }}$ deviation angle, the locating device is most likely to produce greater errors about the $\alpha_{\text {dev }}$ deviation axis. The locating device is most apt to produce errors about this axis due to the previously discussed dependence on the $\alpha$ frame. The small $\eta$ dev, when $\alpha_{\text {dev }}$ is non-zero, makes a considerable change in the amplitude and phase of the $\Theta_{\text {dev }}$ angle. This finding localizes the most volatile central-axis joint locations and atributes the error in the CV coupling to specific characteristics of the locating device.

The $n$ dev deviation angle produces much larger magnitudes in the Edev deviation angle for a given cycle. The $\eta$ dev angle produces $\Theta_{\text {dev }}$ deviations $90^{\circ}$ out of phase from what the $\alpha_{\text {dev }}$ angle produces. The general shape of the corresponding input/onput relationships are quite similar between the two transmission plane deviation angles. The Idev deviation axis although generaliy less dependent on the locating device does produce a considerably larger amplitude in the imput/outpu relationship.

The most concerning variations in the input/output relationship accur when the nwo deviation angles are both non-zero. The slightest addition of the ndev to the adev con change the phase and increase the amplitude amplitude. The mechanical tolerances in the locaing device of a CV coupling should take into account the volatility of the $\eta$ dev as it can have grave effects at very minute values. As the locating device is generally concerned with positioning of the transmission plane about the
axis of the frame angle, perhaps additional concern should be made to accomodate the strong influence of the $\eta$ dev value.

## CHAPTER 5 CONCLUSION

The CV shaft coupling deviation solution provides a complete relationship between the input and output shaft rotations for all possible configurations of the transmission plane. The intent of this study is to give the designer needed insight into the relationship between the transmission plane and the deviation between the inpur and output shaft rotational displacements. The two deviation angles, $\alpha_{\alpha}$ dev and $\eta$ dev give absolute specification of the location of the transmission plane.

The possible central-joint axis locations are grouped into four quarter spaces. The quarter spaces are separated by the plane containing both shafts and the homokinetic plane. When the central-joint axis passes a quarter space boundary, the inputoutput ratationship, $\Theta_{\text {dev }}$ will switch signs. This implies, if the impul shafi rotation leads the output shaf rotation, then it will lag or visa-versa, As the centraljoimt axis moves deeper into a quarter-space the lead/ag magnitude will become larger.

The indev deviation angle per degree has a greater effect on the inpu/auput relationship then does the $\alpha_{\mathrm{dev}}$ deviation angle. The $\eta_{\mathrm{dev}}$ deviation axis is orthogonal to the axis of the frame angle and thus the $\eta$ dev reference angle is unchanged by the bisecting angle variations. The central-joint axis only passes through two quarter spaces under the influence of $\eta$ dev. By remaining in a single quarter space for $180^{\circ}$
the central-axis joint will reach much deeper and allow the lead/lag magnitudes to become exceedingly large. The axis of the deviation angle, $\eta$ dev is orthogonal to the axis of the frame angle, which shows that the deviation angle, $\eta \mathrm{dev}$ is least likely to be have errors due to the variations in the frame angle. These changes in the frame angle correspond to changes in the transmission plane locating device, ideally of one-half the frame angle.

The $\alpha_{\text {dev }}$ deviation angle, occurs about an axis perpendicular to both shafts. The $\alpha$ frame angle rotates about the same axis as the $\alpha$ dev angle. The location of the homokinetic plane is at one-half the $\alpha$ frame angle, so the $\alpha_{\text {dev }}$ angle as it is referenced 10 the homokinetic plane is most affected by the transmission plane locating device. The variation in the position of the homokinetic plane about this axis during variations in the $\alpha$ frame angle, although implicitly defined in the solution, must be considered carefuilly.

The central-joint axis for one rotation of the input shaft, passes through four quarter spaces boundaries when the dev is non-zero. This implies the inputoutput relationship, Oclev value will switoh signs four times during one rotation of the imput shaft. These solutions are double-frequency periodic funcions for specified deviation angles. The larger the driveline offsel angle the sharper the change and the larger the amplitude of the Edev deviation values.

Although the amplitude of the Odev deviation is small under the influence of the adev deviation angle, the imasmission plane locating device is mosi likely io produce greater errors about the aframe axis, The ramsmission plane locating device is most apt to produce errors about this axis. as the maximum variation of the homokinetic plane with respect to the changes in the frame angle will occur about this axis.

The most concerning variations in the input/output relationship occur when the two deviation angles are both non-zero. The slightest combination of the $\eta$ dev with the $\alpha_{\text {dev }}$ can change the phase and increase the amplitude of the lead or lag. The mechanical tolerances in the transmission plane locating device of a CV coupling should take into account the volatility of the $\eta$ dev as it can have grave effects at very minute values. As the locating device is generally concerned with positioning of the transmission plane about the axis of the frame angle, perhaps additional concern should be made to accomodate the strong influence of the $\eta$ dev value.

Combinations of the $\eta \mathrm{dev}$ and $\alpha \mathrm{dev}$ transmission angles prescribe all possible configurations of a general coupling with a specified transmission plane. The $\eta$ dev and $\alpha_{\text {dev }}$ transmission angles once established can be evaluated to obtain the input/output relationship. An efficient non-recursive formula for the relationship between the input and output shafts of a CV coupling, when constant-velocity constraints are not satisfied, has been presented here. A mathematical model has been derived to allow for all possible configurations of the CV joint to be evaluated.

This system offers insight on the CV Coupling's short comings and focuses on an approach to quantifying performance of specific CV coupling mechanisms. This sysiem provides information on how precise a CV Coupling will be subject to mechanism designs and rolerances. By quantifying the input/ourput relationship for a CV coupling a set of standards can be established.

## REFERENCES

Akivis, M. A., and V. V. Goldberg. 1972. "An Introduction to Linear Algebra \& Tensors. ${ }^{\text {" }}$ Dover Publications, Inc., New York, ISBN 0-486-63545-7.

Bellomo, N. 1975. "Theoretical analysis of static forces and torque transmission by a class of spatial systems of rigid bodies for constant velocity transmission." Mechanical Research Communications, Vol. 2249 -254.

Bottema, O., and B. Roth. 1979. "Theoretical Kinematics." North Holland Publishing Co., New York, 1979, ISBN 0-444-85124-0

Craig. J. J. 1986. "Introduction to Robotics." Addison-Wesley Publishing Co., Mass., ISBN 0.201-10326-5.

Duffy, J.s and M. J. Gilmartin. 1979. "Displacement Analysis of Spatial 7R Mechanisms Suitable for Constant Velocity Transmission between Parallel Shafts." Joumal of Mechanism Design ASME, Vol. 6 October 604-614.

Froudenstein, F., and E. R. Maki. 1979. "The creation of mechanisms according to kinematic structure and function." Environment and Plaming B, Vol. 6 $375-391$.

Him, K. H. 1973. "Consiant-Velocity Shaft Couplings: a General Theory." Transactions, Iownal of Engineering for Industry ASME, 95B 455464.

Mabie, H. H. 1948. "Consiant velocity Joints with particular reference to Constant-velocity Type." Machine Devign Vol.20, Na.5 101 -105.

Macielinski, J. W. 1070. "The Design and Selection of Universal Johnta," The Joumal of Automotive Engineering June, 8-14.

McCarthy, J. M. 1990. "Introduction to Theoretical Kinematics." MIT Press, Cambridge, Mass.

Miller, F. F. 1965. "Constant-Velocity Universal Ball Joints." Machine Design March, 184.

Myard, F. E. 1933. "Theorie Generale: Des Joints de Transmission de Rotation." Le Genie Civil April, 345-.

Myard, F. E. 1933. "Les Transmissions de Rotation a Couples D'emboitment." Le Genie Civil June, 539.

O'Neil, P. V. 1987. "Advanced Engineering Mathmatics 2nd edition." Wadsworth Publishing Co., California, ISBN 0-534-06792-1.

Rzeppa, A. II. 1953. "Universal Joint Drives." Machine Design April, 162-170.

Sandor, G. N., and A. G. Erdman. 1984. "Advanced Mechanism Design: Analysis and Synthesis Vol.2." Prentice-Hall, Inc., Englewood Cliffs, New Jersey.

Steeds, W. 1937. "Universal Joints." The Automobile Engineer, January, London, England, 10-12.

Sturges, E. B. 1947. "Constant.velocity Universal Joints." Product Engineering July, 120-123.

Suherland, G. H. 1976. "Finding bearing loads caused by Constant-Velocity U.Ioints." Machine Design April, 55-59.

Taniyama, K., S. Kubo, and T. Taniguchi. 1985. "Effecis of Dimensional Pactors on the Life of the Raeppa Universal Joint." SAE Technical Paper Series $\$ 850355$ Detroit, Mich, February 25-March 1, 25-32.

