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# AN ANALYSIS OF THE HEATING AND DENSIFICATION PROCESS DURING ROTATIONAL MOLDING OF A THERMOPLASTIC POWDER IN A UNIAXIALLY ROTATING CYLINDRICAL CAVITY

by Floyd S. Ribe

Dissertation submitted to the Faculty of the Graduate School of the New Jersey Institute of Technology in partial fulfillment of the requirements for the degree of Doctor of Engineering Science 1985

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Title of Thesis: AN ANALYSIS OF THE HEATING AND

DENSIFICATION PROCESS DURING ROTATIONAL

MOLDING OF A THERMOPLASTIC POWDER IN A

UNIAXIALLY ROTATING CYLINDRICAL CAVITY

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#### ABSTRACT

Title of Thesis: AN ANALYSIS OF THE HEATING AND

DENSIFICATION PROCESS DURING ROTATIONAL MOLDING

OF A THERMOPLASTIC POWDER IN A UNIAXIALLY

ROTATING CYLINDICAL CAVITY

Floyd Steven Ribe, Doctor of Engineering Science, 1985

Thesis directed by: Dr. Richard C. Progelhof
Professor of Mechanical Engineering

This dissertation presents the results of an experimental and theoretical investigation of the heating and densification portion of the rotational molding process in a uniaxial, cylindrical mold.

A thorough literature survey is included which reviewed past analysis of the rotational molding process and other areas that assisted in understanding of the process.

The results presented in this dissertation included an analysis of the densification process by use of Scanning Electronic Microscope (SEM) photography producing intermediate correlations between the physical properties of the densifying material and neck radius of adjacent coalescing spheres. In addition, a hybrid experimental procedure coupled with a small computer simulation was devised to determine the actual initial thermal conductivity and diffusivity of the powdered polymeric material.

Finally, a computer program was written to simulate the heating and densification process during the rotational molding process. Results showed good agreement with actual experimental findings.

#### ACKNOWLEDGMENTS

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#### I. INTRODUCTION

This dissertation presents the results of an experimental and theoretical investigation of a portion of the rotational molding process of a thermoplastic material in a cylindical mold. In rotational molding or rotomolding of thermoplastic powders, the process consists of the following steps:

- 1 Loading the mold with a fixed mass of resin
- 2 Simultaineously heating and melting the thermoplastic powder while the mold rotates
- 3 Cooling the mold
- 4 Unloading the mold

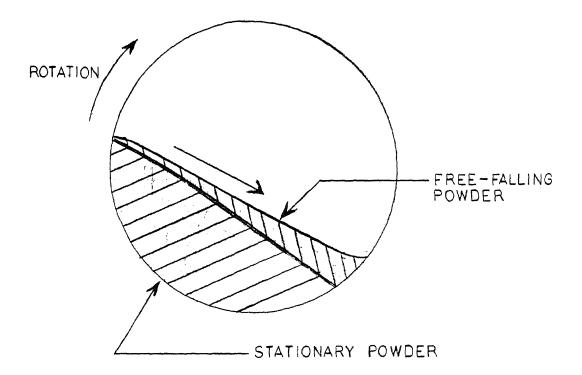
During loading, a premeasured mass of thermoplastic powder is placed in a two or three part split mold of which one section is bolted to a platform. The mold is closed and locked. The mold platform is bolted to an arm which moves the mold from station to station and biaxially rotates the mold at a predetermined rate. Rotational speed and directional speed ratios are adjusted by gearing.

After loading, the arm moves the mold into an oven and the biaxially rotation is started. At this point the powdered material is still cool and due to gravity the

powder forms a pool at the bottom of the mold. Figure 1 shows a typical pool flow of a uniaxial clockwise rotating mold and is the type used in this investigation to experimentally verify the theoretical analysis. As the mold rotates, friction causes the material to follow and remains stationary to the motion of the mold surface. When the powder particles reach a point where friction is overcome by gravity, the individual powder granules begin to free fall over the top surface of the following adjacent material of the stationary pool. The stationary and free fall zones are shown in Figure 1. Experiments have shown that the actual motion of the stationary and free falling powder is dependent upon particle geometry, air volume fraction and surface character of the powder.

This rotational process continues until the mold surface reaches a temperature that causes the material to begin to soften and adhere to the mold surface. This temperature is called the "stick temperature". With increasing mold temperature, more material adheres to the mold surface until the mold surface is covered. With a further increase in mold surface temperature, layers of powder will be built on the mold wall causing the pool to deplete. The material adhering to the surface of the mold is a porous powdered layer initially held together by point

Figure 1
TYPICAL POOL FLOW



contact of the individual powder particles.

During the period of time for which the particles are adhering to the wall of the mold, the material begins a densification process. During this process the material particles lose their individuality by joining themselves first at the contact points, then "melting" into a solid piece with minute air voids throughout the molten resin. The initial phase of this densification process is similar to the sintering, coalesence and fusion processes in the field of drop coalescense, paint technology, ceramics and glass.

When the heating process is completed, the mold is first air cooled allowing an even temperature distribution within the part and densification of the molded part to complete. Rapid cooing to room temperature is then usually accomplished by water spray on the exterior mold surface.

This investigation will analyze this rotational molding process excluding the cooling portion of the cycle. A literature survey will first be made to review past works. Finally two mathematical models will be developed to predict the material heating process which includes the time for complete densification.

#### II. LITERATURE SURVEY

The literature available of the rotational molding process and the analysis thereof is very limited. Because of this, the literature survey will first examine the available published works on the subject and then at other related articles that also provide additional insight toward understanding the rotational molding process. These areas include a survey of articles concerning sintering which help describe the densification process in rotational molding, a survey of the thermal conductivity of porous material used to predict the thermal conductivity of powdered material similar to that encountered in the initial phase of the rotational molding system, and a rewiew of previously published works that analyzed the mass flow in a rotating cylinder.

# Modeling of the Rotational Molding Process

One of the first to attempts to analytically model the rotational molding process was reported by Rao and Throne (31,32, and 14). They modeled the heat transfer to the mold and powder, the fluid flow of the powder, the sintering-melting and degradation during this process.

In their analysis, the authors' assume that an exponential internal mold surface temperature profile is resultant of a constant ambient oven temperature and

convective film coefficient. Assuming a polynomial temperature profile of the penetration thickness of the powdered material adjacent to the mold surface, they computed the amount of material adhering to the mold surface when the mold temperature reaches or exceeds the material stick temperature. By subtracting the predicted amount of powder that adhered to the mold surface, the new volume of the pool was determined. The analysis repeats until all material has left the pool.

Vanderbeck(30) modified Rao and Throne's models by improving some of the basic flow assumptions in the pool. As with Rao and Throne, Vanderbeck also assumes constant physical and thermal properties. Both authors disregard the fact of material that has traveled around the mold and has reentered the powdered pool and thus neglecting the insulating effect of the material adhering to the mold surface.

Throne(33) and Ahdout(34) modeled a rotational system which assumed the powdered material to be evenly distributed around the mold surface, neglecting the actual flow of the powder within the pool region and the mixing zone. Temperature dependent properties were calculated by using a linear interpolation between the known powdered state at the beginning and the final solid state at the end of the process.

The Throne, Rao, Vanderbeck and Adhout models all simulate in a very rough manner, a portion of specific phases of the rotational molding process. However, a complete simulation with temperature dependent properties to predict pool depletion and total densification has yet to be accomplished.

#### Mass Flow

In the study of mass flow in a rotating cylinder,
Lehmverg, Hehl and Schugerl(41) performed experiments using
color tracers placed in the pool of powdered material of a
rotating drum having transparent ends for visual
observations. As the drum rotated, the position of the
tracers were recorded. Their results showed an area where
most of the material remained stationary relative to the
wall during mold rotation and a thin layer of material on
top of the stationary pool that mixed as it rolled down the
incline of the surface of the material. These results as
well as results reported in references 53 and 54, fully
concur with the brief description of the typical mass flow
in rotational molding made in the introduction.

# Sintering

The sintering, coalescence and fusion of particles have been studied in the field of drop coalescence, paint technology, ceramics, glass and polymers. References

1-28,35,51,53,54, and 56 were reviewed with respect to the sintering process of polymers. The following is a brief summary of the major applicable works.

Frenkel(1) analyzed the phenomena of "Cold Welding" of two amorphous spheres. Based on thermodynamic relationships, a relationship to predict the neck radius, x, of two coalesing spheres (Figure 2), is given by:

$$x^2 = 3a\gamma t/2\eta$$

where  $\gamma$  is the surface tension;  $\eta$  , is the Newtonian viscosity; a, is the radius of the sphere; and t is time.

To illustrate the application of Frenkel's equation for isothermal sintering for polyethylene, the properties of which are listed in Table 1, the following sintering equations are generated:

For 105 °C 
$$\frac{x^2}{a} = 8.915 \times 10^{-6} \text{ t} \quad \text{where x and a are in}$$
 cm. and t is seconds

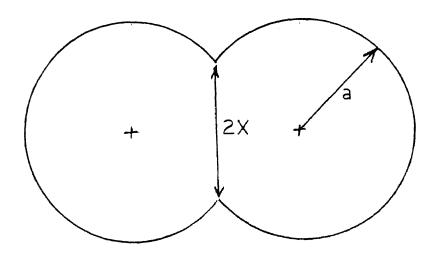
For 150 
$$^{\circ}$$
C  
 $\frac{x^2}{a} = 6.549 \times 10^{-5} \text{ t}$ 

For 
$$180 \, ^{\circ}C$$

$$\frac{x^2}{a} = 1.959 \times 10^{-4} t$$

The predicted results are plotted in Figure 3. Note the linear relationship between  $\frac{\chi^2}{a}$  as a function of time that

Figure 2
SINTERING OF TWO ADJACENT SPHERES



#### TABLE 1

#### PROPERTIES OF POLYETHYLENE

# Surface Tension(18)

$$\gamma(T) = \gamma_0 - (\partial \gamma / \partial T)(T - T_0)$$

For Polyethylene

$$\gamma(T)$$
 = 31 - (0.058)(T-105) Where:  
 $\gamma$  is in dynes/cm  
T is in degrees C.

#### Viscosity(6)

$$\eta = \eta_0 \exp \left\{ \frac{-E}{R} \frac{(T - T_0)}{(T) * (T_0)} \right\}$$

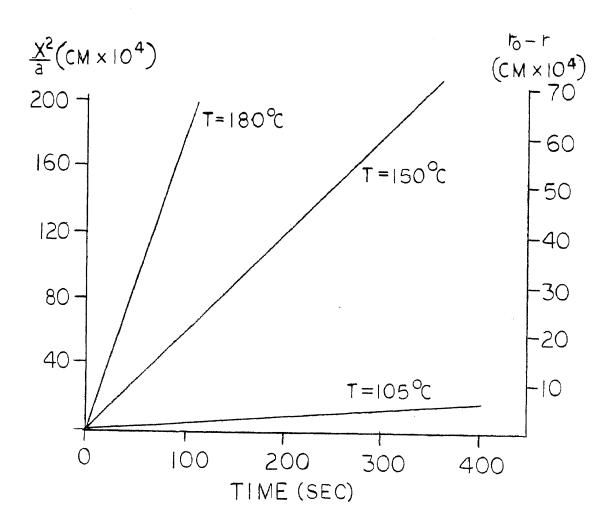
For Polyethylene

$$\eta = 5.22 \times 10^6 \exp \left\{-19.583 + \frac{7402.5}{T}\right\}$$

Where:  $\eta$  is in Poise T is in degrees K

Figure 3

PLOTS OF FRENKEL'S EQUATIONS FOR POLYETHYLENE



intersects at the origin.

Frenkel also derived the theory of densification of glass for the second stage of sintering where the voids become individual bubbles that slowly reduce in diameter. The equation for the collapse of an individual bubble is given by:

$$r_0 - r = \frac{\gamma}{2 \eta} t$$
 (eq. 2)

where  $r_o$  is the original pore radius at time zero and r is the radius at time t. Using the polyethylene properties in Table 1, the collapse of the voids for the three temperatures used are as follows:

For 105 °C

$$r_0 - r = 2.97 \times 10^{-6} t$$

For 150 °C

 $r_0 - r = 2.18 \times 10^{-5} t$ 

For 180 °C

 $r_0 - r = 6.53 \times 10^{-4} t$ 

These equations are plotted in Figure 3.

Frenkel's equation(eq 1) was experimentally confirmed by Kuczynski(2) where glass beads were heated on top of a glass plate having the same composition as the beads.

After heating, the samples were rapidly cooled and mounted in Bakelite. The dimensions were measured and plotted in

Figure 4

KUCZYNSKI'S RESULTS OF SINTERING GLASS SPHERES

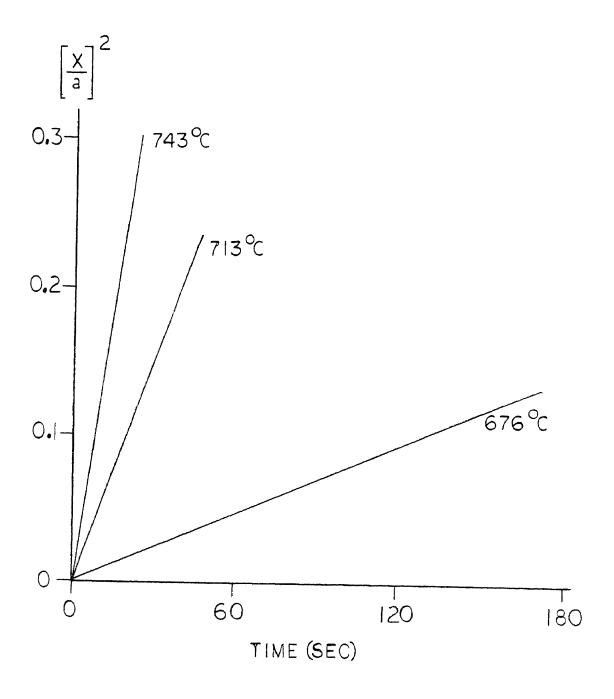


Figure 4. The experimental results confirmed the linear relationship of Frenkel's equation intersecting at the origin. Kuczynski along with Zaplatynskyj(13) confirmed Frenkel's colapsing theory, Equation 2, by heating capillary tubes and measuring the collapse of the internal diameter as a function of time.

Dillion, Matheson and Bradford(5) investigated the sintering of synthetic latex particles. Experimental results indicated that the coalescence process occurs by the same mechanism as was described by Frenkel equation.

Kuczynski, Neuville and Toner(3) performed the same type experiment as Kuczynski(2) using Poly(methy)

Methacrylate (acrylic). Figure 5 presents the experimental results. They found for this polymer, Frenkel's equation(eq. 1) is inadequate. The following empirical equation was then correlated:

where F is a function of temperature only and P is the slope of the curve in Figure 5. Lontz(4) replots Figure 5 using  $\frac{X}{a}$  for the Y axis (as opposed to  $\frac{\chi^2}{a}$ ) versus time(Fig. 6). It is obvious from this graph that the y intersection of the linear curves tend to increase with temperature giving an indication of a time delay to the

Figure 5
NECK GROWTH OF PMMA SPHERES

$$\frac{x^2}{a}$$
 vs TIME

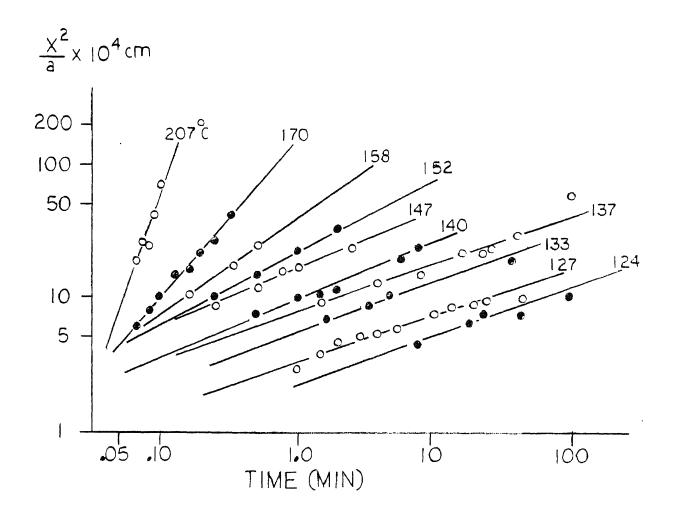
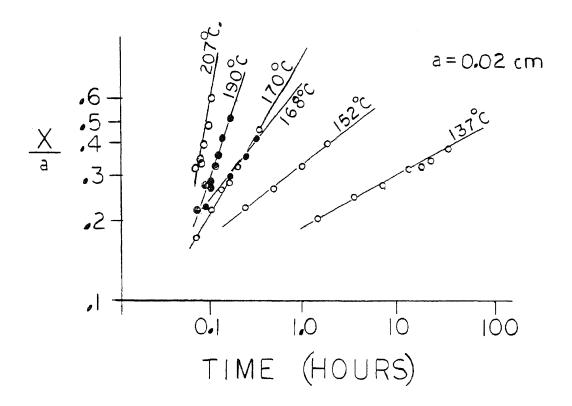


Figure 6

NECK GROWTH OF PMMA SPHERES

X
a vs TIME



sintering process. This is not unexpected due to the transient temperature response of the spheres.

Shonhorn, Frisch and Kwei(10), studying the kinetics of wetting of surfaces by polymer melts, showed that polymers at high temperatures exhibited a shifted Frenkel curve, as is evident in the Kuczynski, Neuville and Toner's plots(Figure 5).

Lontz(4) developed a model of sintering between two viscoelastic spheres yielding the equation:

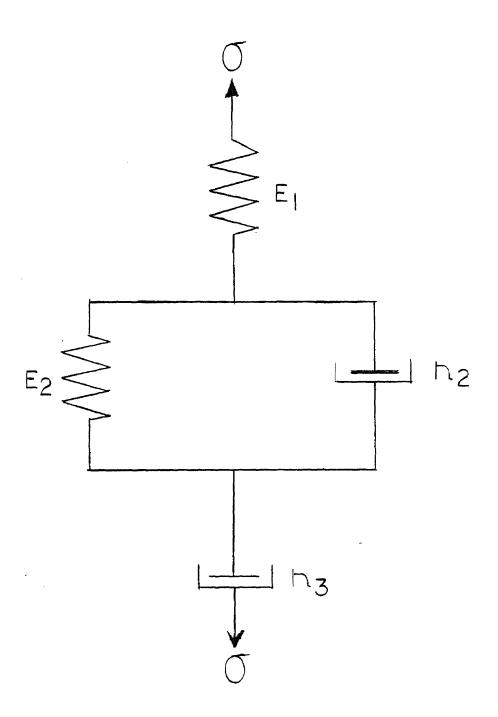
$$\frac{\chi^2}{a} = \frac{3t}{2\eta} \frac{1}{1 - \exp(-t/\lambda)}$$
 (Equation 4)

where the second right hand term is a correction factor to the Frenkel equation to account for viscoelastic effects. The term,  $\lambda$ , is a relaxation time constant. The relaxation time constant is determined experimentally for each resin and temperature. Its significance can be better understood by examining the four element Maxwell-Voigt model (Figure 7) for a viscoelastic body.

The spring  $E_1$  in Figure 7 act as the Hookean repsonse while the dashpot,  $\eta_3$ , corresponds to the Newtonian fluid response. The spring,  $E_2$ , and dashpot,  $\eta_2$ , represent the

Figure 7

FOUR ELEMENT MAXWELL - VOIGHT MODEL



retarded elastic response of the polymer molecules. When the stress,  $\sigma$ , is applied, E<sub>|</sub> and  $\eta_3$  react instantly while because of the physical arrangement, E<sub>2</sub> and  $\eta_2$  have a delayed response. Together the system has a viscoelastic effect. The ratio of  $\eta_2$  to E<sub>2</sub> is the relaxation time constant,  $\lambda$ , or:

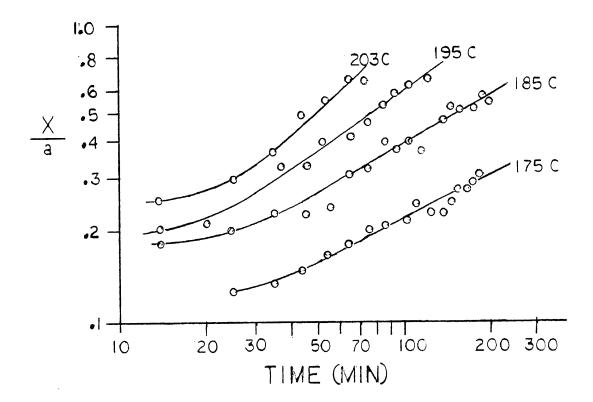
$$\lambda = {}^{\eta}2/E_2$$
 (Equation 5)

Narkis(8) also studied the sintering of closely packed Poly(methyl) Methacrylate spheres in a circulated air oven. Experimental results showing neck radius verus time for four temperatures are shown in Figure 8. The shapes of the curves were very similar to the theoretical curves given by Lontz's equation(Eq. 4)

An examination of these curves, show a non-linear portion is followed by a linear portion. The non-linear portion is due to the viscoelastic effect (E<sub>2</sub> and  $\eta_2$  of our model) after which the relaxation term approaches unity reverting the Lontz equation to a shifted Frenkel equation (the linear portion). Note that the Kuczynski, Neuville and Toner(3) curves only showed the linear portion of the curves. They neglected to realize any relaxation effect but only stated that Frenkel's equation was inadequate. Also note, the viscoelastic effect is shortened as would be expected due to greater molecular mobility as the sintering temperature is increasd.

Figure 8

SINTERING OF PMMA SPHERES BY NARKIS(8)



Steiner, Manson and Nippert(9), using the basic differential Frenkel equation, numerically integrated the equation using the exact value for the  $\sin\theta$  rather than Frenkel's assumption of  $\sin\theta=\theta$ . (Only valid for small values of  $\theta$ ). Numerical results of  $\frac{\chi^2}{d}$  versus time showed a better correlation to their experimental data for viscous sintering. A viscoelastic sintering equation which included a retardation time factor similar to that developed by Lontz was also derived.

Menges and associates (20) developed an isothermal growth correlation based on surface tension, neck curvatures and inner frictional forces. Theoretical results agreed with their experimental results.

Rosenzweig and Narkis(21) performed a study of dimensional changes of sintering particles. Based on the Frenkel's sintering model (Figure 9), the authors derived the following relationships of two sintering spheres:

$$X = (2aZ - Z^2)^{0.5}$$
 (Eq. 6)

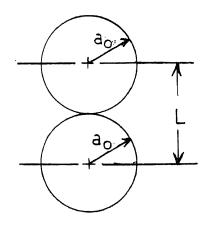
$$L = 2a - 2Z \tag{Eq 7}$$

$$4a^3 - 3z^2a + z^3 - 4a_0^3 = 0$$
 (Eq. 8)

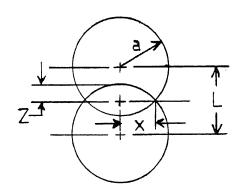
# Figure 9

# GEOMETRY OF TWO SINTERING SPHERES

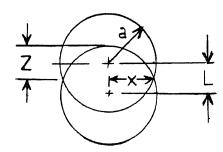
# (FRENKEL'S MODEL)

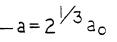


Initially - Two spheres with
 length L, between centers



During sintering neck radius x, is formed as lenth L, decreases





Final sintering results with sphere radius becoming equal to  $2^{1/3}$  a.

where: x = neck radius

a<sub>0</sub> = original sphere radius

a = sphere radius during sintering

L = length between the centers of the
two sintering spheres

These equations will be later used in this work to determine the density of the powdered material as a function of distance between centers of sintering spheres. The results were used to correlate the density of a group of spheres as a function of neck radius.

### Thermal Conductivity

Progelhof, Throne, and Ruetsch(33) performed an in depth investigation of published articles studing the thermal conductivity of foamed, powdered or composite materials. In addition to the articles reviewed by Progelhof, Throne, and Ruetsch, references 34-46 were also reviewed. In most correlations evaluated, the values of  $k_{\text{C}}$  and  $k_{\text{d}}$  (where  $k_{\text{C}}$  = thermal conductivity of the continuous material and  $k_{\text{d}}$  = thermal conductivity of the discrete material) must be determined. In the case of rotomolding, until the material starts to stick to the mold, it is assumed that the polymeric material is the discrete phase that is surrounded by a continuous phase of air. Upon the

commencement of the fusing process, the polymeric material is assumed to be the continous phase surrounding the discrete phase, air.

Figure 10 illustrates the Maxwell Model of thermal conductivity as reported by Progelhof, Throne and Ruetsch (35) of polyethylene and air. The top curve represents the predicted thermal conductivity with air as the discrete phase while the bottom curve represents polyethylene as the descrete phase. The dotted lines labled A', B' and C' represents possible routes where air originially the continuous phase changes over to the descrete phase as is expected to occur in the rotational molding process.

The original intent of this investigation was to use one of these equations to predict the thermal conductivity of the powdered material while in the pool. Using this value, new values as a function of temperature and neck growth would be estimated. However, in performing the thermal conductivity calculations, a great variance was found. Figure 11 shows six thermal conductivity estimate equations and the results using air as the discrete material and then as the continuous material. The results show a variance of two magnitudes. To avoid justifying one value over another, this investigation will use experimental data to determine the actual value. The experimental procedures are discussed latter.

Figure 10
MAXWELL'S MODEL FOR THERMAL CONDUCTIVITY

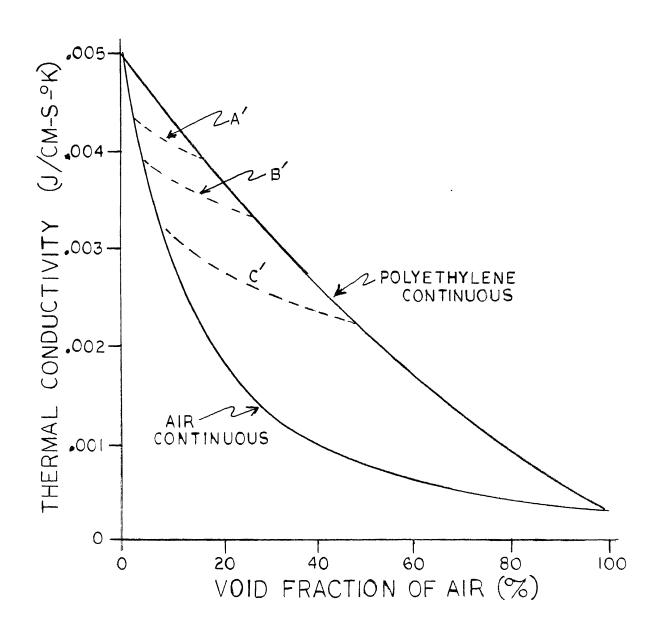


Figure 11

COMPARISON OF VARIOUS

VARIOUS COMPOSITE THERMAL CONDUCTIVITY EQUATIONS

## COMPUTED THERMAL CONDUCTIVITY ( J/cm-sec-K)

Descrete Material	Polyethylene	Air
Continuous Material	Air	Polyethylene
Fraction of Continuous Material	0.58	0.42
EQUATION		
Yagi - Kunni	9.90 E-05	3.24 E-04
Maxwell	4.98 E-04	2.52 E-03
Lewis and Nielson	2.76 E-03	1.10 E-02
Russel	1.08 E-03	2.66 E-03
Geometric Mean	1.39 E-03	1.39 E-03
Series	2.96 E-03	2.96 E-03

Thermal Conductivity of Polyethylene = 4.93 E-03 J/cm-s-K Thermal Conductivity of Air = 2.42 E-04

NOTE: Equations used in the computations of the thermal conductivity taken from Reference 35.

#### III. STATEMENT OF THE PROBLEM

All previously published investigations of rotational molding were based upon models that had limited restrictions. One assumed constant physical and thermal properties, while the other assumes the material already distributed around the mold. Investigations dealing with sintering(densification) of particles all involved an isothermal process. In rotomolding, the "sintering" process starts with polymer powder being placed in a mold at ambient temperature. The mold is then heated as it is rotated in an oven. As the temperature of the polymer reaches the stick temperature, it begins to adhere to the mold wall. This process is continued until all powder has adhered and then begins to densify.

This dissertation is an investigation of the rotational molding process. It includes an analysis of the material flow in the pool and an investigation of the heating and densification process. It does not include the cooling portion of the cycle. A theoretical simulation model of the rotational molding process was then derived to predict pool depletion and densification times using temperature and time dependent properties. In lieu of justifying the use of one of the many composite thermal conductivity equations (reviewed earlier) over another, an

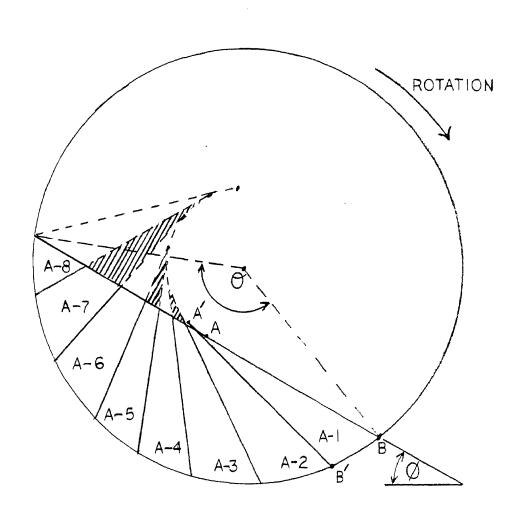
experimental procedure was developed in conjuction with a small computer program for determining the actual powder thermal conductivity. In the conculsions, the theoretical results predicted by this new model are compared to the actual measured rotational molding pool depletion times and with the other previous simulation results. The densification times between the simulations developed are then compared with the earlier simulation results of Throne and Adhout.

#### IV. MASS FLOW IN ROTATIONAL MOLDING

In order to properly simulate the rotational molding process, an understanding of the mass flow within the mold cavity during that process is essential. Consider a simple cylindical mold cavity as shown in Figure 12 rotating in a clockwise direction. The angle theta  $(\theta)$  is the included angle of the powdered material in the stationary zone of the pool. The angle phi  $(\phi)$  is the angle that the stationary zone maintains as the mold rotates. This angle  $(\phi)$  is a function of the material characteristics, the rotational speed, and the coeficent of friction of the mold surface. Based on Vanderbeck's experimental results, it has been shown that the angle  $(\phi)$  remains constant during the depletion of the pool.

Consider a line segment A-B which begins at the center of the cord formed by the surface of the stationary pool of powdered material and ends at the mold surface. If the mold is rotated for a time interval(DT), the line segment is also rotated to the postion A'-B' shown in Figure 12. The area(volume) included by the two line segments and the mold surface is the amount of material that has entered the stationary pool leaving the free-falling zone during that time interval. This section is labled A-1.

Figure 12
ENTERING MASS FLOW IN ROTATIONAL MOLDING



Allowing the mold to rotate for another time increment will move Area A-1 to the location of Area A-2 as shown in Figure 12. Note that the shift of A-1 has caused a portion of the area to protude past the surface line B-C of the stationary region. This outside area represents the amount of material of Area A-1 that has left the stationary pool and reentered the free-falling region during that time increment.

Rotating the mold by another time increment will move Area A-2 to the position of A-3. Again, a portion of the area, in the movement from A-2 to A-3, has exited Area A-2 of the stationary pool region and reentered the free fall zone during that time increment.

Continuing the rotation of the mold will cause the Area A-3 to move to A-4, then to A-5 and so on to A-8. After A-8 all material in the area that started at A-1 has completely exited the stationary pool. The amount of material that exits the stationary pool at each time interval for each area is equal to the amount of area passing through the pool surface Line B-C during that time increment.

From the preceding analysis, it is evident that the time length that the material remains in the pool depends on its location or position while entering the pool.

Material entering near the center surface of the pool (point A) remains in the pool for a short time duration while material entering near the mold surface remains longest in the pool.

The total mass leaving at each time interval is the amount that passes through line segment B-C in Figure 13 to join the free-falling powder. Moving line segment C-A in the opposite direction of the rotation for one time interval of rotation results in the postioning of that line labled C'-A' in Figure 13. The area bounded by line segments C-A, C'-A' and the mold surface is the material exiting the stationary pool in the next time interval. Continuing back at one time increment movement intervals, the areas shown are the areas leaving in the succeding time intervals. Some areas have a portion of their areas above the pool surface line meaning that this portion of material has not yet entered the stationary pool. Notice that the area shown leaving the stationary pool in the Figure is the same amount of area that had entered in Figure 12.

Figure 14 combines Figures 12 and 13 into a complete illustration of the mass flow within the stationary pool zone. In this figure an arc time length of eight time increments(intervals) were used. It can be seen that the material entering in the center(area labled 1) remains in

Figure 13
LEAVING MASS FLOW IN ROTATIONAL MOLDING

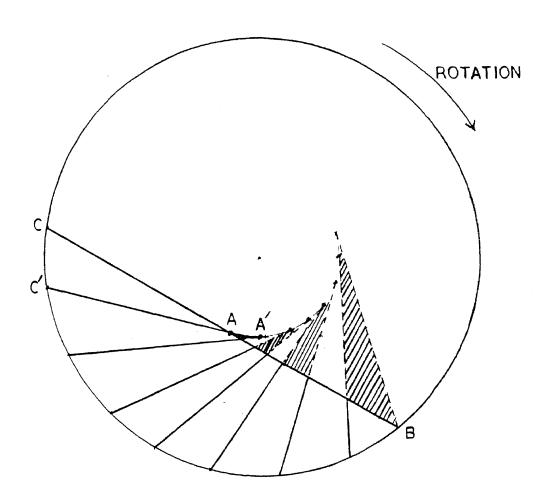
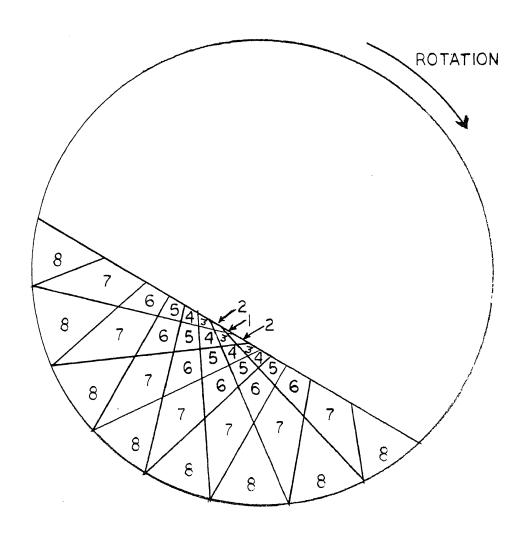


Figure 14

TOTAL MASS FLOW IN ROTATIONAL MOLDING



the pool for only one time increment while the areas labled 8 remain for all eight time increments. Those particles in between remain according to their relative position from the center.

It is assumed that after the material leaves the stationary pool, the material begins to fall down and mixes physically and thermally until it again rejoins the stationary pool and then repeats the cycle.

In addition to being rotated, the material is also being heated by the mold surface. When the mold reaches or exceeds the "stick temperature", the portion of material that is at or above that particular temperature will stick to mold wall, thus deminishing the amount of material in the pool. This process will continue until all material in the pool has left the pool by sticking to the mold surface.

# V. ANALYSIS OF THE HEAT TRANSFER FOR SIMULATION ASSUMPTIONS

In the rotational molding process, the mass of material in the stationary pool is assumed to be heated by conduction from the mold surface. The time increment that an individual element is in contact with the mold surface is given by:

$$t(ms) = \theta / \omega$$
 (Eq. 9)

 $\theta$  = angle theta(Figure 13)

 $\omega$  = angular velocity

Using typical values of an angle theta of 90 degrees, 7 RPM will result in a element contact time of 2.142 seconds.

During this contact time, the thermal penetration depth can be approximated by the results for a semi-infinite solid having a constant initial temperature and being subjected to sudden change in surface temperature:

$$T(x,t) = erf \left\{ \frac{X}{2 * SQR(\alpha * t)} \right\} * (Ti-Ts) + Ts (Eq. 10)$$

Ts = New surface temperature

x = distance from surface

 $\alpha$  = thermal diffusivity

t = time

Rearranging the above equation and solving for the distance, x will result in the equation:

 $x = 2* SQR(\alpha *t)*erfc((T(x,t)-Ts)/(Ti-Ts))$  (Eq. 11) The penetration depth will be at the position where T(x,t)

= Ti. The inverse error function has the numeric value of 3.60. The equation reduces to :

$$x = 7.2 * SQR(\alpha *t)$$
 (Eq. 12)

Using the numerical value of the thermal diffusivity of the High Density Polyethelyne powder as 1.7E-03 sq cm/sec, a contact time of 2.142 seconds will result in a penetration depth of 0.434 cm.

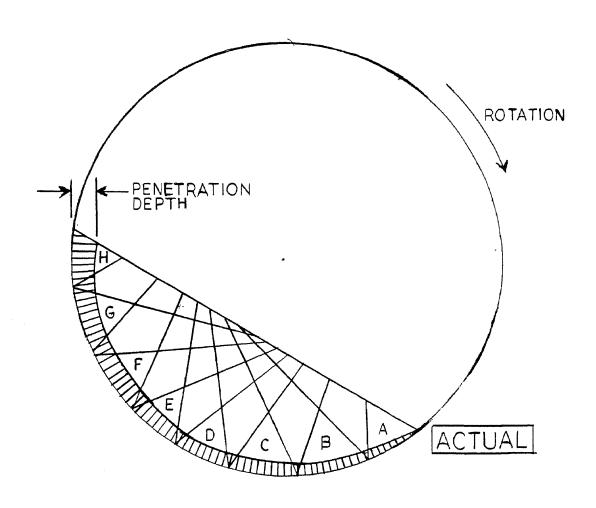
Since this penetration depth is very small compared to the depth of the pool, and that this penetration occurs only in the subareas adjacent to the mold surface, the computer simulation can use a rectangular system having the same number of columns as wedges in the actual system. This approximation will be in error when the pool is in the last phases of depletion. A comparison of the two systems is shown in Figure 15.

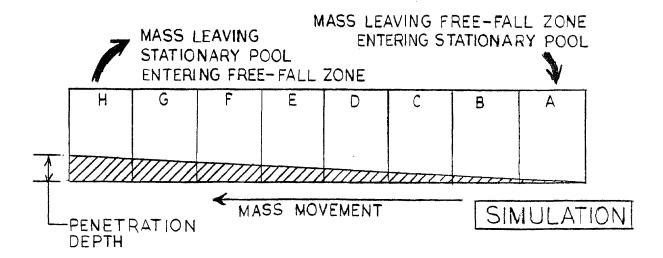
The rectangular system used in the computer simulation

Figure 15

COMPARISON BETWEEN THE ACTUAL ROTATIONAL MOLDING

AND THE COMPUTER SIMULATION MODEL





has a total area(or volume) equal to the area in the actual system and also a column width equal to the arc length of one wedge.

In this simulation, the material enters at position A in Figure 16 at a temperature equal to that of the free-falling powder region. The temperature profile for one time increment is then calculated. The material is then shifted over one position (labled B) simulating rotational movement and a new temperature profile is calculated for the next time interval using the old profile as its initial temperature.

This process is continued for all postions(columns A-H). After the last position (H), the material is assumed to leave the stationary pool and join the free-falling powder region. It is assumed in this phase of the model that all of the particles are thermally mixed. The average or equilibrium temperature of the powder in the free fall zone is computed by using a mass weighted average of the average temperature of the mass entering the free-falling region with the remaining material in this region, or:

$$\overline{T} = \frac{\overline{T_e * M_e + \overline{T_{FF}} * (M_{FF} - M_L)}}{M_{FF}}$$
 (Eq. 13)

where:  $\overline{T}$  = new ave temperature of free falling region.

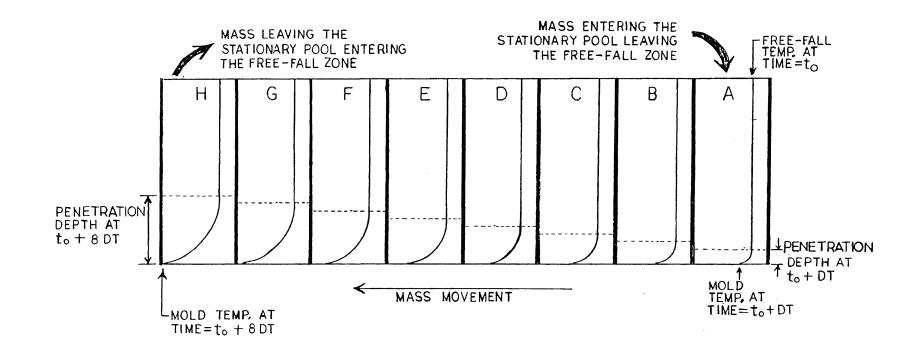


FIGURE 16

TEMPERATURE PROFILE DURING 8 TIME INTERVALS

Te = new ave, temperature of mass entering

Me = mass entering

TFF = previous average temperature of free falling region

 $M_{FF}$  = previous amount of mass in free falling region

 $M_L$  = amount of mass leaving which is equal to the mass entering,  $M_e$  .

Simplifing:

$$\overline{T} = \overline{Te} * \frac{Me}{MFF} + T_{FF} * \left[ I - \frac{Me}{MFF} \right]$$
 (Eq. 14)

The average temperature of the mass entering the free falling region,  $\overline{T_e}$ , is determined by the same mass weighted formula using the temperatures and mass within the penetration depth and that outside the penetration depth.

After the free falling region, the mass now re-enters the stationary pool at the present mixed mean free-falling temperature and the heating cycle described earlier is repeated.

When the polymer powder in contact with the mold surface attains the stick temperature, material above this temperature will adhere to the mold wall causing the powdered pool to become smaller in size. The simulation

will still use the same amount of wedges (columns), however, the time increment (interval) will become smaller to compensate for the smaller contact time.

The cycle is repeated until all material has left the powdered pool and adheres to the mold.

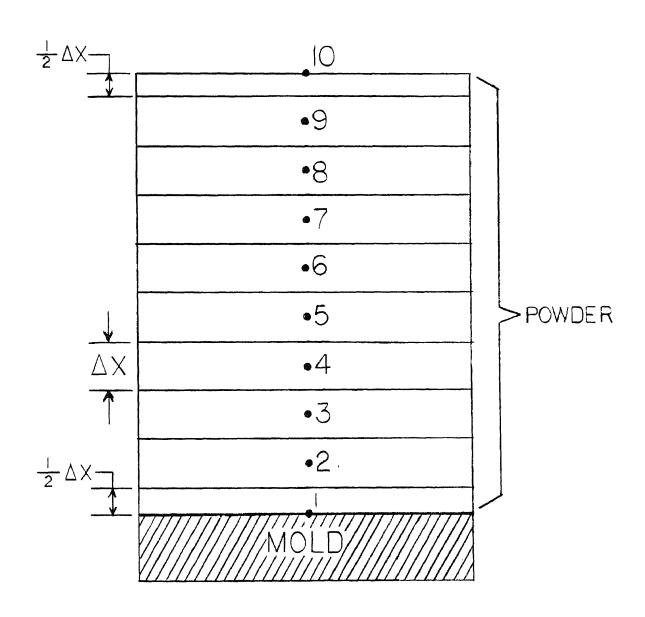
#### VI. THEORETICAL MODEL USING NODAL ANALYSIS

In the theoretical analysis presented in this dissertation, a numerical method will be used. The geometry of the specimen is subdivided into small but finite subvolumes of thickness of  $\Delta X$ . For each subvolume there is located a center nodal point which has been assigned a reference number, Figure 17. Note that the exterior subvolumes has a thickness of 1/2  $\Delta X$  with its node located at the surface.

In this actual transient heat transfer process, the temperature profile within a subvolume varies with position and time. However for this simplified model it was assumed that the temperature of each subvolume can be denoted by a nodal "mixed mean" or "equilibration" temperature. Thus the temperature within a node is assumed to be only a function of time. The mixed mean or equilibration temperature of the element is defined as the temperature the element will attain if all the internal energy of the element was distributed evenly throughout that element.

The temperature of each subvolume is assumed to be represented by the temperature of the node. It is further assumed that the rate of energy transfer between adjacent nodal points is approximated by the steady-state conduction

Figure 17
NODAL CONVENTION



equation using the nodal temperature values as the descriptive temperature between the nodes. Thus in this model, a discontinuous temperature profile is being used to approximate the actual temperature profile (Figure 18). It is obvious that as the subvolumes become smaller, the approximate temperature profile will approach the actual temperature profile.

Writing an energy balance on node i results in the following equation:

$$q(in) - q(out) + q(generated) = q(stored)$$
 (eq 15)

Or:

$$\sum q_{i} + q(generated) = m_{i} C_{i} \frac{dT}{dt}$$
 (eq 16)

where:  $q_{i} = heat flow entering node i$ 
 $m_{i} = mass of subvolume i$ 

 $\frac{dT}{dt}$  = the time derivative of temperature of subvolume i

The initial temperatures of all nodes were constant room temperature and the mold surface was subjected to a changing temperature. Because the temperature change will only occur within the penetration depth, nodal equations must then be made to include this depth. Temperatures of nodal volumes outside the penitration depth will not change, thus performing computations of these nodes are needless and a waste of valuable computer time.

Figure 18

APPROXIMATION OF THE TEMPERATURE PROFILE

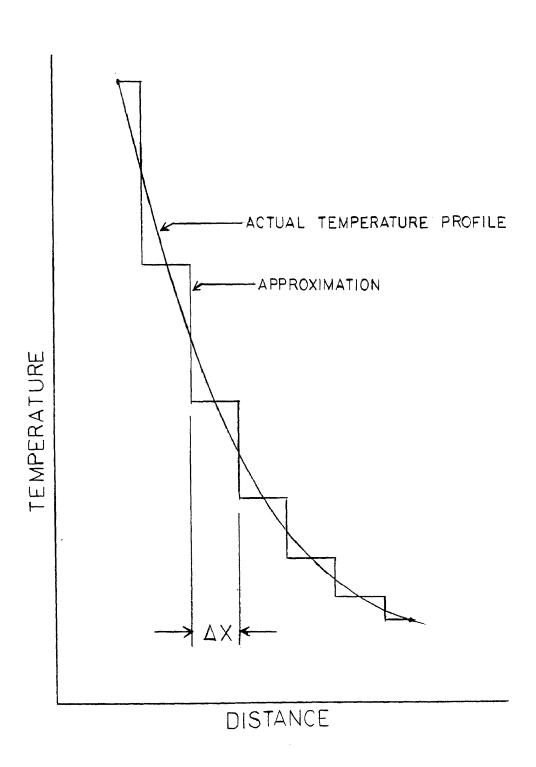


Figure 19 shows a typical stationary pool that is divided into eight wedges with each wedge divided into N nodal volumes. The nodal volumes have a fixed thickness such that the total distance extend slightly past the final penetration depth.

For this geometry, three types of nodal equations must be used. These equations are for: the surface nodes, the interior nodes and the final interior nodes outside the penetration depth.

The energy balance for the interior nodes of the column, Figure 19, is given by:

$$q(i-1:i) + q(i+1:i) = m_i C_i \frac{dT}{dt}$$
 (eq 17)

where: q(i-1:i) = heat flow from node i-1 to node i

= 
$$k(i-1:i) \frac{A}{\Delta X} (T_{i-1} - T_i)$$
 (eq 18)

k(i-1:i) = the average thermal conductivity

between node i-1 and node i

$$= 1/2 [k(i-1) + k(i)]$$
 (eq 19)

q(i+1:i) = heat flow from node i+1 to node i

= 
$$k(i+1:i) \frac{A}{\Delta X} (T_{i+1} - T_{i})$$
 (eq 20)

k(i+1:i) = the average thermal conductivity between

node i+l and node i

$$= 1/2 [k(i+1) + k(i)]$$
 (eq 21)

 $m_i$  = the mass of node i

$$= \rho A \Delta X \qquad (eq 22)$$

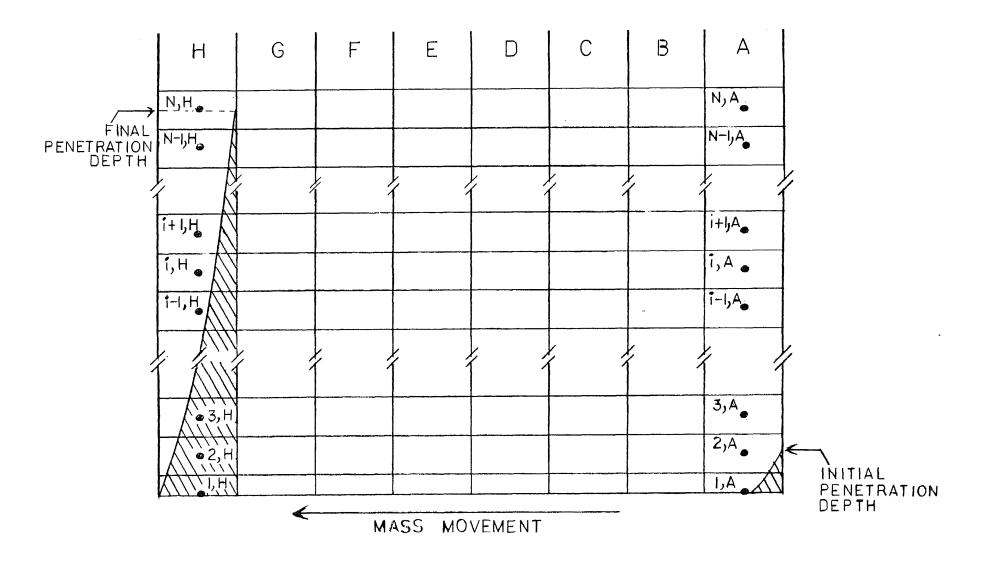


FIGURE 19

NODAL ASIGNMENTS FOR COMPUTER SIMULATION

which results in the relationship:

$$k(i-1;i) \xrightarrow{A} (T_{i-1}i-T_i) + k(i+1;i) \xrightarrow{A} (T_{i+1}-T_i) = \rho AAX C \frac{dT}{dt}$$
(eq 2.3)

In this slab geometry, the cross-sectional area A is constant. Solving for the time rate of change of temperature:

$$\frac{\mathrm{d}\mathbf{T}}{\mathrm{d}\mathbf{t}} = \frac{\mathbf{k}(\mathbf{i}-\mathbf{1}:\mathbf{i})}{\rho \, C \, \Delta \, \mathsf{X}^2} \, (\mathbf{T}_{\mathbf{i}-\mathbf{1}} - \mathbf{T}_{\mathbf{i}}) + \frac{\mathbf{k}(\mathbf{i}+\mathbf{1}:\mathbf{i})}{\rho \, C \, \Delta \, \mathsf{X}^2} \, (\mathbf{T}_{\mathbf{i}+\mathbf{1}} - \mathbf{T}_{\mathbf{i}})$$
 (eq 24)

The surface node will be assigned the same temperature as the mold surface. Since the mold temperature changes as a function of time, then:

$$T(surface node) = T(mold) = T(t)$$
 (eq 25)

The last interior node, which being outside the penetration depth, is always at a temperature of the pool temperature for that particular wedge, or:

$$T_{j, \cap} = T(pool)_j$$
 (eq 26)

Where:  $T_{j, \cap} = temperature of node n in wedge j$ 
 $T(pool)_j = pool temperature outside the penetration depth for wedge j

 $T_{j, \cap} = T_{j, \cap} = temperature of node n in wedge j$$ 

There are three basic methods of approximating the time derivative, the Pure Explicit or Euler's Method, the Pure Implicit Method, and the Implicit Crank-Nicolison Method.

Euler's Method estimates the temperature T' at the nodal point one time increment,  $\Delta \theta$ , later by computing the time derivative of the present temperature T, multiplying it by the time increment between T and T', and then adding this to the present temperature T, or:

$$T' = T + \frac{dT}{dt} \bigg|_{t} \Delta \Theta$$
 (eq 27)

This is shown graphically in Figure 20.

The Pure Implicit Method estimates the future nodal temperature T' in a similar fashion as the Euler's Method. But instead of using the time derivative of the present temperature T, it uses the derivative of the future temperature T', or:

$$T' = T + \frac{dT}{dt} \Big|_{t'} \Delta \Theta \qquad (eq 28)$$

This is shown in Figure 21.

In the Implicit Crank-Nicolson Method, the arithmetic mean value of the derivatives at the beginning and at the end of the time interval is used to determine the future tmeperature. Or:

$$T' = T + \left[ \frac{\frac{dT}{dt} \Big|_{t} + \frac{dT}{dt} \Big|_{t}}{2} \right] \Delta \Theta \qquad (eq 29)$$

This is shown in Figure 22.

Figure 20
EULER'S METHOD OF APPROXIMATION

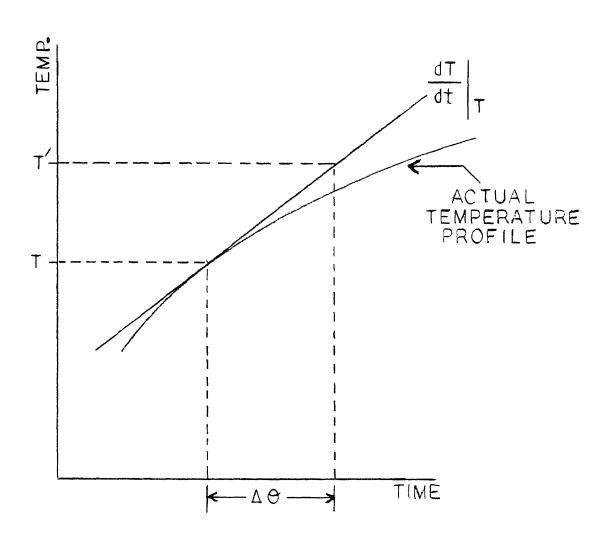


Figure 21
THE PURE IMPLICIT METHOD OF APPROXIMATION

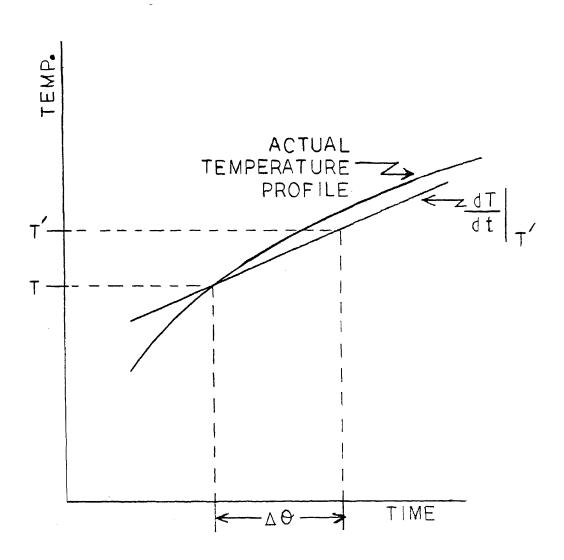
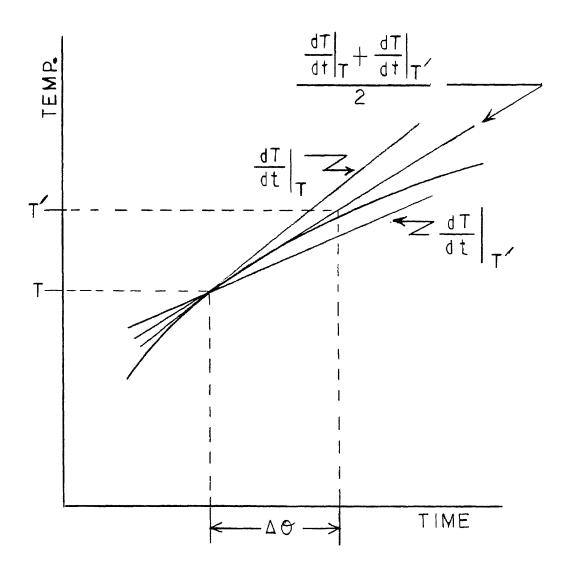


Figure 22

THE CRANK-NICOLSON METHOD OF APPROXIMATION



If the Pure Implicit or the Implicit Crank-Nicolson Method is used to approximate the time derivative, the result will be a set of N simultaneous equations with N unknowns. This would require the computer to solve a tri-diagonal matrix for each time increment for each column.

Euler's Method, however, results in a set of N equations with one unknown per nodal energy balance equation. Consequenty, the future temperature T' can be found directly by solving a simple algebraic equation. Therefore, in this dissertation, Euler's Method of approximation was used.

Substituting for the time derivative for an interior nodal equation results in:

$$\frac{T_{\mathbf{i}}'-T_{\mathbf{i}}}{\Lambda \theta} = \frac{k(\mathbf{i}-1:\mathbf{i})}{\rho C \Delta X^2} (T_{\mathbf{i}-1}-T_{\mathbf{i}}) + \frac{k(\mathbf{i}+1:\mathbf{i})}{\rho C \Delta X^2} (T_{\mathbf{i}+1}-T_{\mathbf{i}})$$
 (eq 30)

Solving for the future temperature T':

$$T_{i}' = T_{i} \left(1 - \frac{\Delta \theta}{\rho C \Delta X^{2}} \{k(i-1:i) + k(i+1:i)\}\right) + \frac{\Delta \theta}{\rho C \Delta X^{2}} \{k(i-1:i)T_{i-1} + k(i+1:i)T_{i+1}\}$$
 (eq 31)

In the use of the Euler's Method of approximation, it must be noted that the equations become unstable when the expression  $\frac{1-K\Delta\theta}{\rho\,C\Delta X^2}\Sigma^k$  becomes negative (K is some constant).

Often the k's, K,  $\rho$  , C and  $\Delta X$  are predetermined. Therefore it is necessary to compute the value of  $\Delta \theta$  such

that the value of the expression is greater than or equal to zero. Or:

$$\frac{1 - K\Delta\theta \Sigma k}{\sigma C\Delta X^2} \ge 0.0$$
 (eq 32)

Or:

$$\frac{\text{K}\Delta\theta \Sigma k}{\rho \text{C}\Delta X^2} \le 1.0$$
 (eq. 33)

Solving for 
$$\Delta\theta$$
: 
$$\Delta\theta \leq \frac{\rho C\Delta X^2}{K \Sigma k} \tag{eq 34}$$

Usually the maximum allowable value of will be used, or:

$$\Delta\theta_{\text{MAX}} = \frac{\rho C \Delta X^2}{K \Sigma k}$$
 (eq 35)

If the values of  $\rho$ , C, and k are not constant, the minimum values of  $\rho$  and C and the maximum value of k must be used to determine the stability constants.

Since the nodal thickness, AX in the simulation developed, becomes smaller as the material densifies, the maximum time increment ( $\Delta\theta$ ) will be recalculated at each iteration assuring that all nodal equations remain stable.

The second simulation model developed includes the effect of convection heating by oven air. The exterior surface nodal equation of the mold is given by:

$$T_{eM}' = T_{eM} + \left[1 - \frac{2 \Delta \theta}{\rho_{M} C_{M} \Delta X_{M}} \left(\frac{k_{M}}{\Delta X_{M}} + H_{A}\right)\right] + \frac{2 \Delta \theta}{\rho_{M} C_{M} \Delta X_{M}} \left[\frac{k_{M}}{\Delta X_{M}} * T_{IM} + H_{A} * T_{OVEN}\right]$$
 (eq. 36)

where:  $T^{\bullet}_{eM}$  = Future temperature of the external mold surface one time increment (  $\Delta\theta$  ) latter.

T<sub>eM</sub> = Present temperature of the external mold surface.

 $\rho_{M}$  = Density of the mold material

 $K_M$  = Thermal Conductivity of mold material

 $C_{M}$  = Specific heat of mold material

 $\Delta X_{M} = Mold thickness$ 

 $H_{A}$  = Coefficient of Convection of the air in the oven

Toven = Oven temperature

T<sub>IM</sub> = Temperature of mold's internal surface

The maximum allowable time increment for this nodal equation is:

$$\Delta\theta_{\text{MAX}} = \frac{\rho_{\text{M}} C_{\text{M}} \Delta X_{\text{M}}}{2\left(\frac{K_{\text{M}}}{\Delta X_{\text{M}}} + H_{\text{A}}\right)}$$
 (eq 37)

The mold's internal surface nodal temperature equation is:

$$T_{IM}' = \left[1 - \frac{2 \Delta \theta k_M}{\rho_M C_M \Delta X_M^2}\right]^T_{IM} + \frac{2 \Delta \theta K_M}{\rho_M C_M \Delta X_M^2} * T_{eM}$$
 (eq 38)

where:  $T'_{IM}$  = Future temperature of the internal mold surface one time increment (  $\Delta\theta$  ) later.

T<sub>IM</sub> = Present temperature of the internal mold surface.

The maximum allowable time increment for this nodal equation is:

$$\Delta\theta_{\text{MAX}} = \frac{\rho_{\text{M}} C_{\text{M}} \Delta X_{\text{M}}^{2}}{2 K_{\text{M}}}$$
 (eq 39)

When using more than one type of nodal equation, the minimum  $^{\Delta\theta}_{MAX}$  must be the largest time increment used in the numerical analysis. This will assure stability for all equations.

The temperature of the first node of the powdered material now becomes equal of  $\mathbf{T}_{\text{TM}}$  .

The equations developed in this chapter were used to predict the powder temperatures during the rotational molding process.

#### VII. INVESTIGATION INTO THE DENSIFICATION PROCESS

To accurately simulate the rotational molding process from powder rotation to the formation of molten mass, the powder densification process must be understood. Experimental studies were performed to determine the basic densification phenomena and the results were used to describe the densification in the simulation model.

Because the individual powder particles used in rotational molding have such irregular shape (see Figure 23), it becomes very difficult to obtain experimental observational data that can be used to describe the densification process . Thus, it becomes necessary to experiment with a much simpler particle geometry and then hypothesize how the process occurs with much more complex geometry. However, problems still occur in attempting to freeze the densification process at a particular time frame while the material is in an actual rotational mold. The solution was to model the rotational mold in an environment with a flat plate apparatus, Figure 24. The powdered material was heated through the base plate and at a particular instant of time the heating process was stopped by removing the plate from the heater and placing the plate on dry ice.

FIGURE 23

TYPICAL ROTATIONAL MOLDING MATERIAL DURING DENSIFICATION

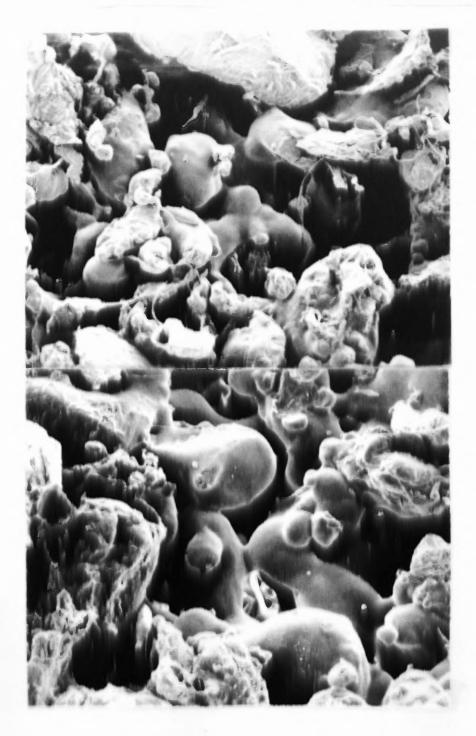


Figure 24
FLAT PLATE APPARATUS USED IN DENSIFICATION STUDY

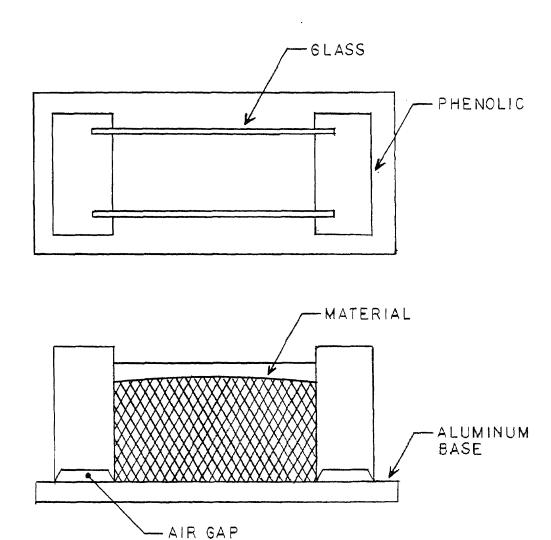


Figure 25 compares a photo of a rotational molded part of DuPont acrylic microspheres made in the uniaxial rotational mold in the NJIT laboratory and a photo of a part molded on the flat plate apparatus. Both molding processes were conducted under similar temperature time histories. (material data listed in Appendix C) Notice the similarities; both have a totaly densified region, a region with slight necking, and a region where spheres are attached but no appreciable or noticeable radius at the points of contact. Hence, the flat plate apparatus is a viable means of analyzing and understanding the densification process in rotational molding.

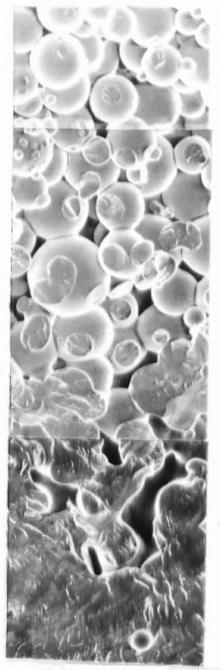
An analysis of the densification process was performed after a review of a flat plate experiment using significantly larger spheres illustrating a non-necking densification process. This experiment involved using one-eighth diameter High Density Polyethylene(HDPE) spheres (Appendix B for data) packed in a body centered cubic array using an apparatus shown in Figure 24. This apparatus was heated through its base, simulating the non-isothermal heating that occurs in the rotational molding process.

As energy was applied to the base plate , the lower portion of the bottom spheres became less viscous and was forced down by the weight of the spheres above. The less

#### FIGURE 25

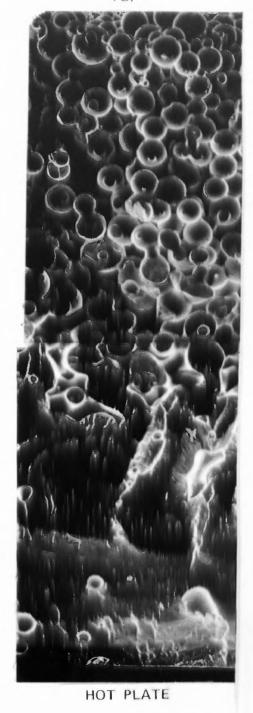
# COMPARISON BETWEEN ROTATIONAL MOLDING AND FLAT PLATE MOLDING OF ACRYLIC MICROSPHERES

TOP



HOT PLATE

TOP



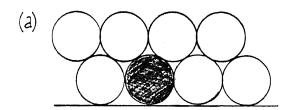
viscous material flowed between the spheres, filling the voids(Figure 26a-c).

The upper portion of the bottom layer next became soft allowing the pressure of the harder spheres above to deform the lower level spheres (Figure 26d). The final shape of a bottom layer spheres appeared to be flat on the bottom due to the flat plate and dimpled on the top due to the more viscous upper level spheres. With continued heating of the plate, the interior spheres deformed next. As seen in the bottom layer analysis, the lower portion of the more viscous upper layer retained its spherical shape while the lower layer was deformed (Figure 27-a).

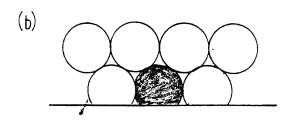
As the upper portion of the interior sphere increases in temperature, the harder, more viscous next upper layer spheres deformed the interior sphere below(Fig. 27a-d). The final shape is spherical in the lower region and dimpled on top (Figure 27e). Duplication of the experiments using one-eighth diameter acrylic spheres produced the same results as those obtained with the HDPE spheres.

The experimental results indicated a different densification process that was described in the literature survey. Unlike sintering, the one-eighth diameter spheres

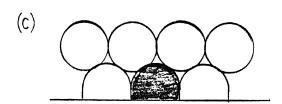
Figure 26
THE MELTING OF BOTTOM LAYER SPHERES



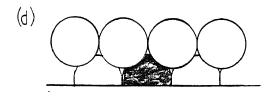
Original set-up



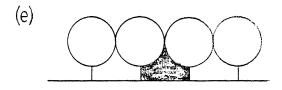
As heat in applied, the lower portion of the bottom spheres become less viscous (melts) and is forced down by the weight above.



The displaced material begins to fill the voids



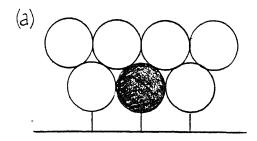
The heat continues and causes the upper portion of the bottom layer to become soft allowing the pressure of the more viscous, harder spheres to deform the lower level spheres.



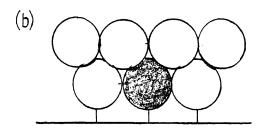
The final shape is flat on the bottom(caused by the plate) and dimpled on top (caused by the harder upper level spheres)

#### Figure 27

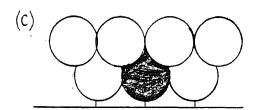
#### THE MELTING OF MIDDLE LAYER SPHERES

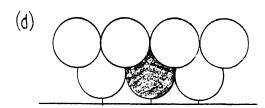


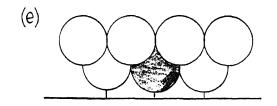
As was seen in the Bottom Layer Analysis, the lower portion of the middle layer retains its spherical shape during melting because it is more viscous(harder) than the layer below causing the lower level to be deformed by the spheres above it.



As the upper portion of the sphere increases in temperature, it allows the harder, more viscous upper layer spheres to deform it.







The final shape is spherical on the bottom and dimpled on top.

showed no necking but rather filled the voids between the individual spheres by viscous flow.

The spheres used is this experiment were approximately one hundred times greater in size than normal rotational material. The only useful conclusions that were drawn from this study was that material flow caused by lower viscosity may also be a significant factor in the mechanism that occurs in the densification process of rotational molding.

The next set of experiments were performed with the DuPont acrylic microspheres using the Flat Plate apparatus. Because the size distribution of the microspheres was very close to that of a rotational molding grade polyethylene powder, the results obtained using the microspheres gave considerable insight into the actual densification mechanism.

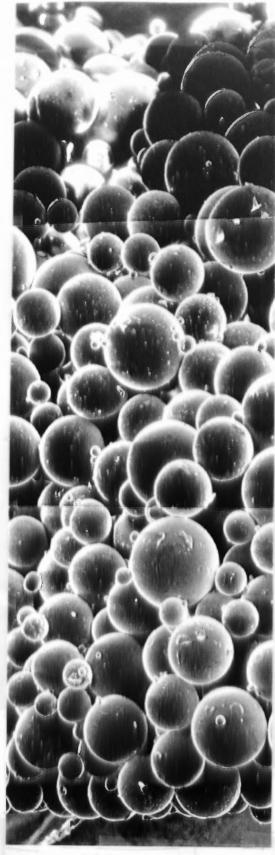
Several experiments were conducted terminating the densification process at different time intervals. The specimens were then mounted and photographed with a Scanning Electronic Microscope. The temperature — time profiles used in these experiments were identical to those used by Vanderbeck(30) in his experiments. A direct comparison between results of the computer simulation developed in a latter section in this dissertation using a model of the densification process observed here and the

actual experiments results of Vanderbeck.

Figure 28 shows the photographs of six simulated rotomolding experimental runs of the acrylic microspheres that were frozen at various time frames. Though difficult to see in the photographs, all the spheres have adhered together at the points of contact and the densification process was initiated.

As the temperature rises, the spheres nearer to the heat source began the necking process by sintering. With increasing time and mold temperature, neck radius of the spheres near the high temperature surface continue to increase. In addition, the necking process starts to occur with the interior spheres.

The latter photographs show that as the neck growth reaches a maximum of between 10% and 20%, there appears to be a dramatic change occuring. Necking is overcome by viscous flow from melting. This process continues until all the spheres have lost their individuality and have melted into one homogenous part. This drastic change is a characteristic of a material where by the temperature causes the material to flow faster than the slow viscous flow of the sintering process. This temperature at which material flow overcomes the sintering process will be referred to as the melt interface temperature (Tm).

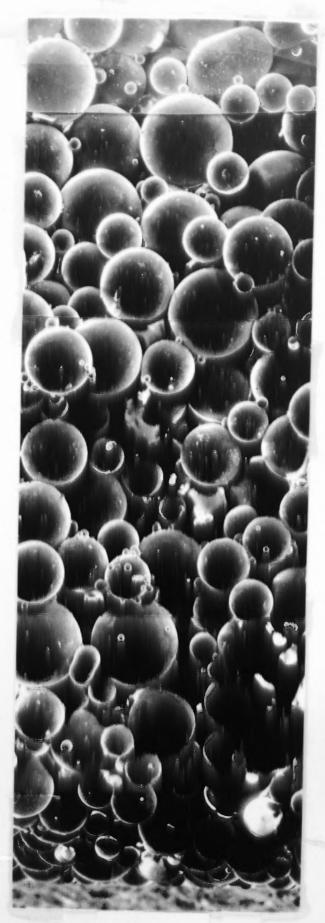


HOT PLATE

FIGURE 28

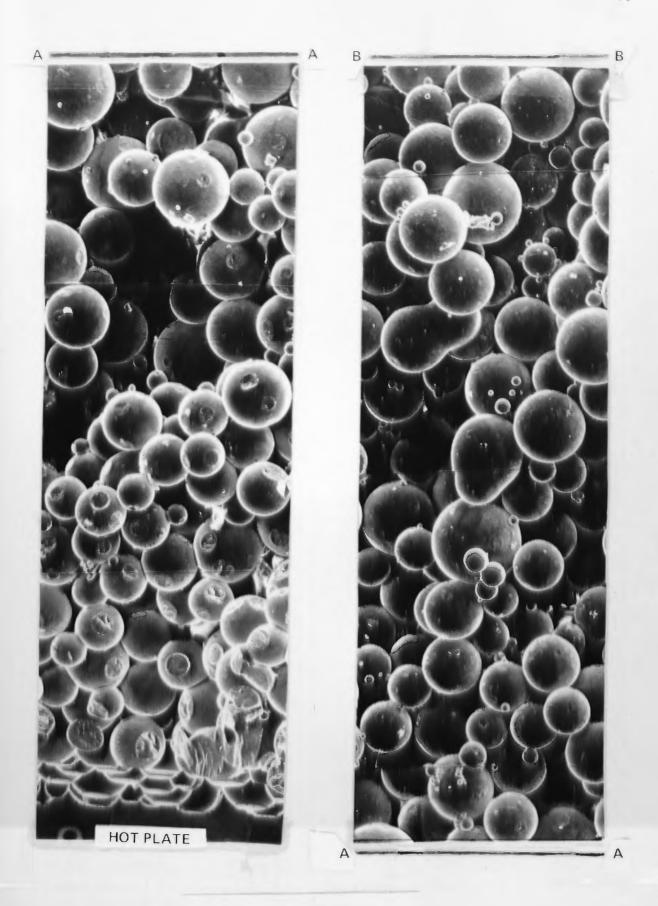
SCANNING ELECTRONIC
MICROSCOPE PHOTOGRAPHS OF
ACRYLIC MICROSPHERES FOR
VARIOUS TIME INTERVALS

8 MINUTES



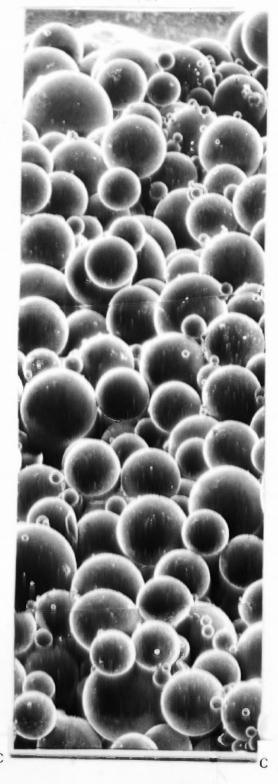
10 MINUTES

HOT PLATE

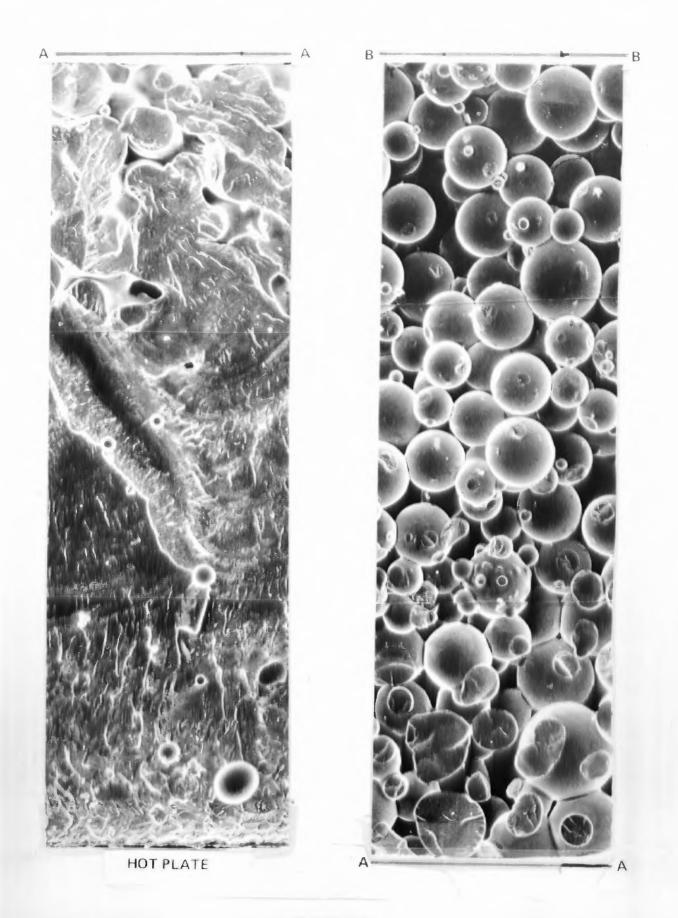


12 MINUTES

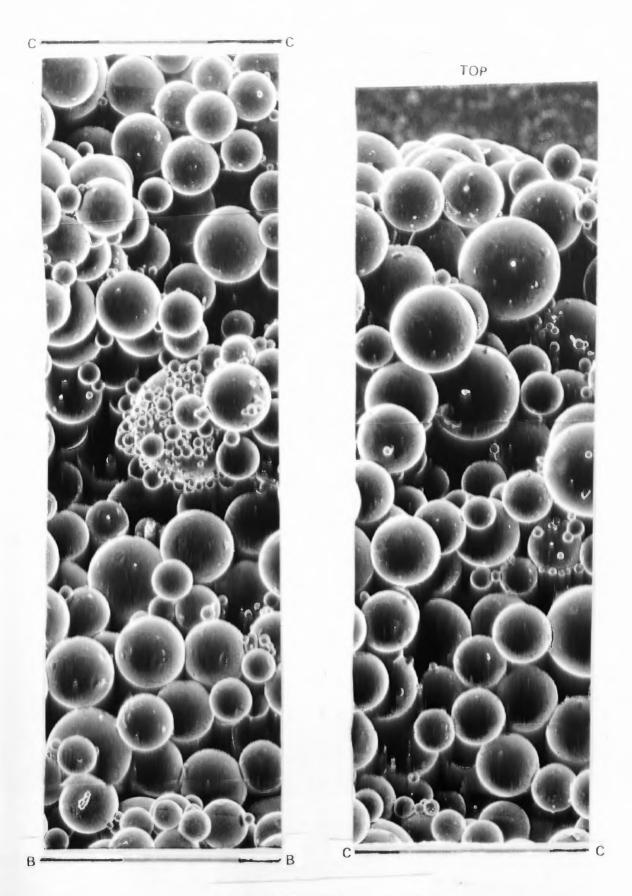




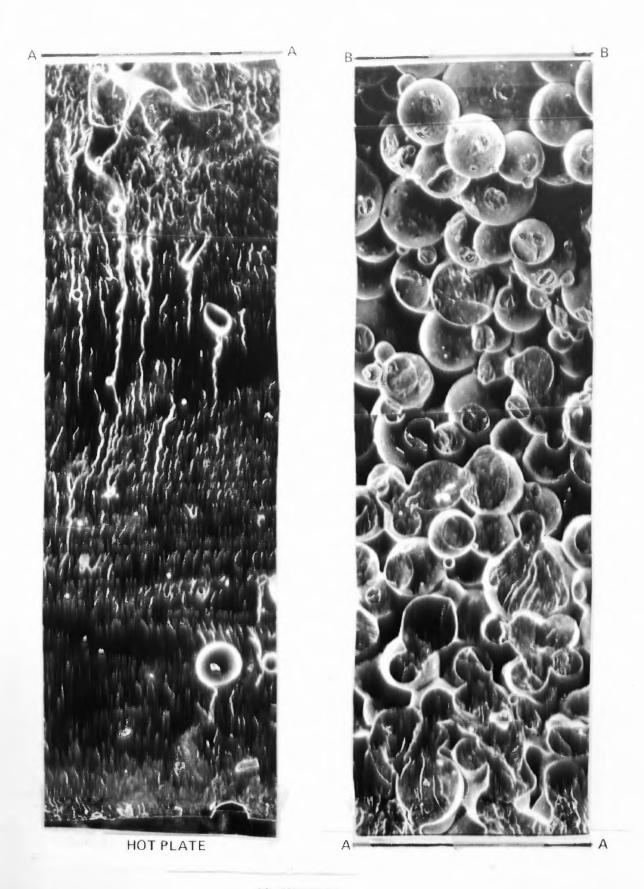
12 MINUTES (CONTINUED)



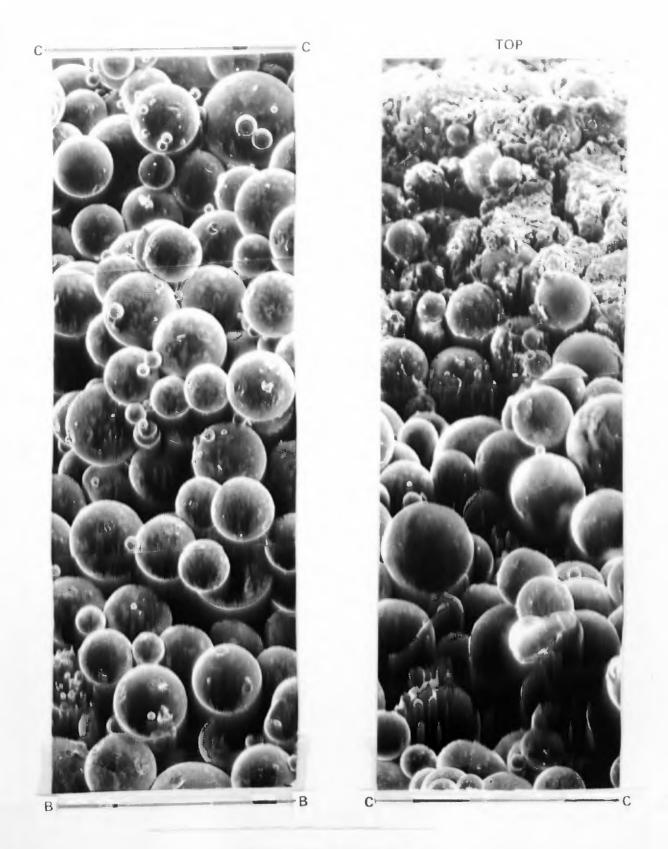
16 MINUTES



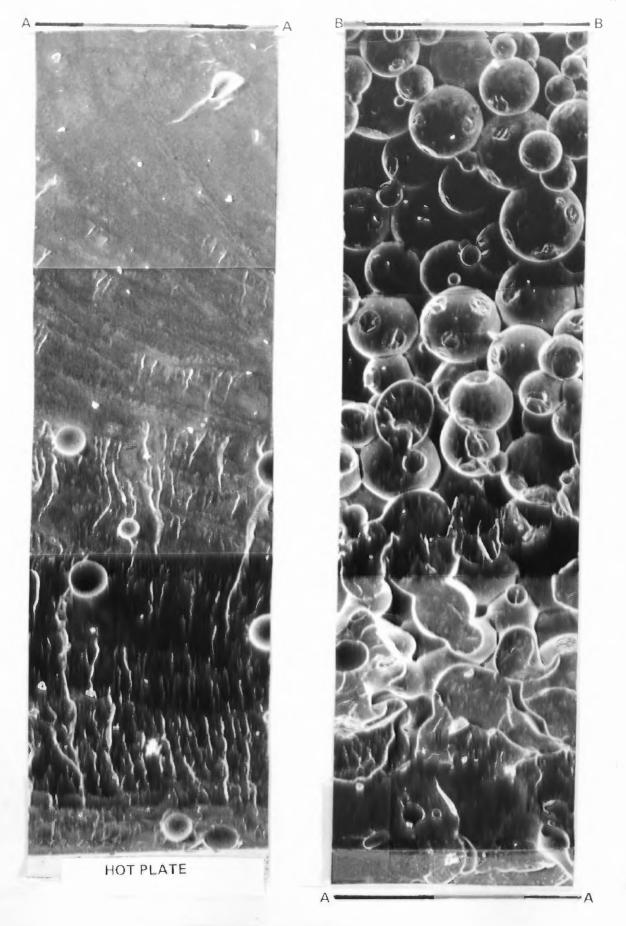
16 MINUTES (CONTINUED)

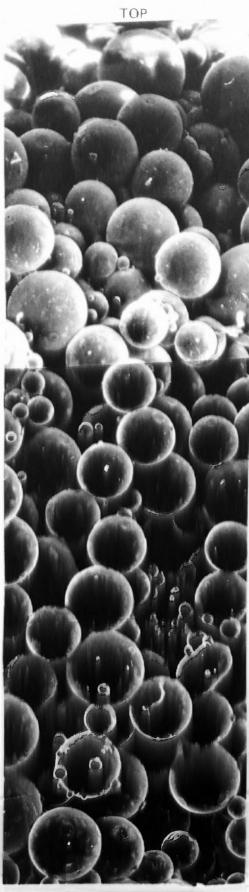


20 MINUTES



20 MINUTES (CONTINUED)





24 MINUTES (CONTINUED)

#### VIII. DETERMINATION OF THE PHYSICAL PROPERTIES

As discussed previously, the simulation model will use nodal analysis in which the average neck size for a representative particle will be computed at each node as a function of time. The value of the thermal conductivity of each nodal subvolume at any particular time will be a direct function of the neck size within the nodal volume at that instant of time. The equation used to estimate the instantaneous thermal conductivity showing the effect of necking due to sintering is:

$$K_{E1} = K_P + \left[\frac{X_i}{a}\right]^N (K_S - K_P)$$
 (eq 40)

where: KE; = thermal conductivity at node i

Kp = thermal conductivity of initial powder

 $K_S$  = thermal conductivity of the solid

polymer

X; = neck radius of node i

a = original sphere radius

N = power law exponent (value = 1)

Even though the actual polyethylene powdered material is not spherical, the value of the original sphere radius, a, is the average sift size radius.

The thermal conductivity of the powdered material (  $K_{\mbox{\it P}}$ ) was to be initially be determined using the equations

mentioned earlier in the literature survey for predicting the thermal conductivity of composite system. Figure 11 compared the results of using various composite material equations for determining thermal conductivity based on the material data in Appendix A. The calculated values of equivalent thermal conductivities(KE;) can be seen to vary greatly. Therefore it becomes necessary to experimentally determine, with use of a simple computer simulation, the actual thermal conductivity of the powder. The experiment and the subsequent determination of the powder thermal conductivity will be described in the next chapter.

The neck size for each node is calculated in the computer simulation by adding the calculated change in neck size during that time increment to the previous neck size of the node. Using Frenkel's Equation:

$$x = \left[\frac{3at\gamma}{2\eta}\right]^{0.5}$$
 (eq 1)

and differentiating eq. 1 with respect to time will give the change of neck radius as a function of time, or:

$$\frac{\partial X}{\partial t} = \left[ \frac{3 \, a \, \gamma}{8 \, \eta \, t} \right]^{0.5} \tag{eq 41}$$

Converting the equation to a small finite change in time ( $\Delta t$ ), the change in neck radius for a time  $\Delta t$  is:

$$\Delta X = \left[\frac{3 \, \text{ar}}{8 \, \text{nt}}\right]^{0.5} \Delta t \qquad (eq. 42)$$

When the neck radius exceeds 50% of the sphere radius, Frenkel states that the collapse of the bubble formed is:

$$r_0 - r = \frac{\gamma}{2 \eta} t \qquad (eq 2)$$

differenting with respect to time:

$$\frac{\partial \Gamma}{\partial t} = -\frac{\gamma}{2\eta} \tag{eq 43}$$

for a small finite change in time:

$$\Delta r = -\frac{\gamma}{27} \Delta t \qquad (eq 44)$$

since the negative change in bubble radius ( $\triangle \Gamma$ ) is equal to the increase of neck radius, the change in sphere radius above 50% necking is:

$$\Delta X = \frac{\Upsilon}{2\eta} \Delta t \qquad (eq. 45)$$

As the temperature of a node exceeds the melt temperature, material flow overcomes necking in the densification process. Hence, the neck radius for that particular node instentaniously becomes equal to the radius of the spheres. In other words, the material has completely densified into a homogenious part for that nodal subvolume.

Density is computed in the similar manner as to that of thermal conductivity. The original powder density was found by experimentally by weighing a volume of the

polyethylene material and comparing it to the same volume of water. The density measured for the polyethelyne powder is listed in Appendix A. The density at the complete melt is that of the solid polymer. Density and other physical data of the base resin was supplied by the manufacturer.

The intermediate density is a assumed to be linearly proportional to the distance between the centers of two sintering spheres as was shown in Figure 9. As length L decreases, the density increases proportionally. Using Frenkel's model, the length L can be correlated to the neck radius, x, thus allowing a direct coorelation of density to neck radius. Figure 29 is a plot of length L vs neck radius using equations 6 through 8 and solving them simutaneously with a = 0.0125 cm. The curve between 0% and 40% neck radius/ sphere radius ratio can be approximated with a straight line fit with only a 2.5% variance. Therefore, it is possible to state that density is proportional to neck radius, or:

$$\rho_{Ei} = \rho_{P} + \frac{x_{i}}{a} (\rho_{s} - \rho_{p})$$
 (eq 46)

where:

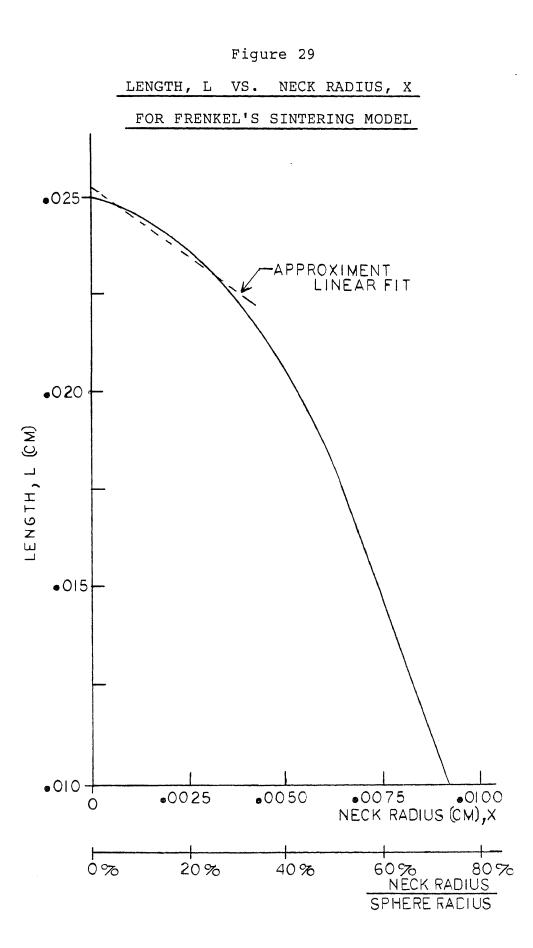
 $\rho_{E_i}$  = Equivalent density at node i

 $\rho_{\rm p}$  = initial powder density

 $\rho_s$  = density of solid polymer

x; = neck radius at node i

a = sphere radius



Because of the density change, the distance between nodes in the simulation also changes. The distance between node i and node i-l is one half the thickness of node i added to the half the thickness of node i-l. These distances are a function of their relative nodal densities. Therefore the nodal distance between node i and node i-l is proportional to the ratio of the original density to the new average density between node i and node i-l, or:

$$Dx_{i,i-1} = \frac{\rho_p}{(\rho_{E_i} + \rho_{E_{i-1}})^{*\frac{1}{2}}} * Dx$$
 (eq 47)

where:  $Dx_{i,i-1} = Nodal$  distance between node i and node i-1

 $\rho_n$  = Density of original powdered material

 $\rho_{r}$  = Equivalent density at node i

 $\rho_F$  = Equivalent density at node i-1

Dx = original nodal distance

Since specific heat is energy per amount of mass to raise the temperature of the mass one degree, the specific heat of the powder is assumed equal to that of the solid since the mass of air in the powder is neglible. Hence the specific heat remains constant in the simulation developed.

### IX. DETERMINATION OF INITIAL THERMAL CONDUCTIVITY

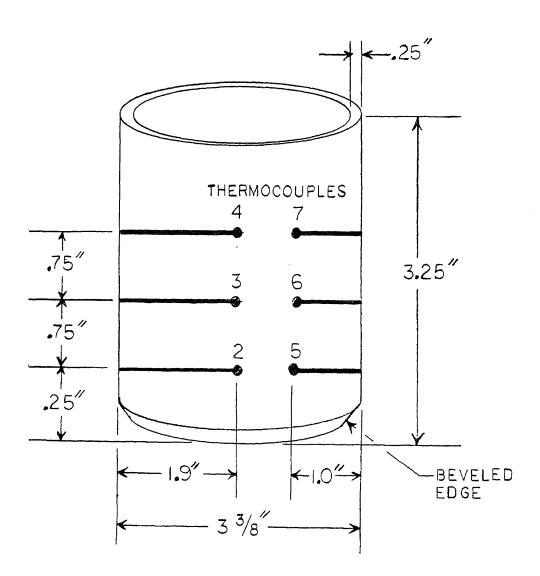
Since the consistency of the thermal conductivity equations of composite materials vary too greatly, it becomes necessary to determine the conductivity of the initial state of the polymeric powder by a combination of an experiment and a computer simulation. The experiment will entail heating the powder and measuring temperature—time histories at specific locations while the computer program will simulate the experiment in an attempt to duplicate the temperature—time histories and thus determining the thermal diffusivity which includes the thermal conductivity.

The experimental set-up used in shown in Figure 30. The apparatus consists of a cardboard tube 3.25 inches long, 3.375 inches inner diameter with a 0.25 inch wall thickness. The tube is beveled at the bottom to minimize heat transfer up the tube wall. Aluminum foil was then epoxied to the beveled edge, bottom of the tube, to contain the powdered material.

Six thermocouples were positioned as shown in Figure 30. To maintain their correct height level, sewing thread was positioned through the tube at the different levels.

Figure 30

THERMAL CONDUCTIVITY EXPERIMENTAL SET-UP



The thermocouple was then wrapped along the taught thread.

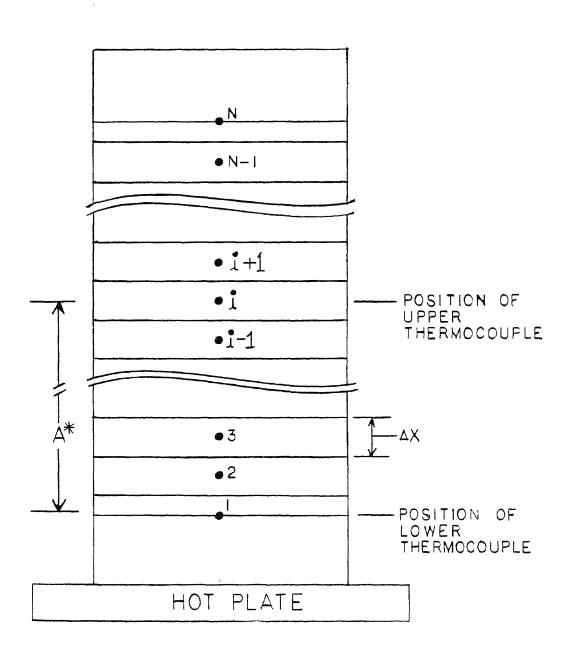
The experimental proceedure used in these experiments were as follows:

- 1. The tube with the powdered material was placed on a hot plate and is heated slowly.
- 2. The temperature at each thermocouple location was then recorded every two minutes for seventy minutes.
- 3. This data was then used in a computer simulation to determine the thermal conductivity of the powdered material.

Since the experiment was conducted with six thermocouples, the computer simulation can estimate six average thermal conductivities (actually diffusivity) for each experimental run. They are the diffusivities from position 2 to 3, 2 to 4, 3 to 4, 5 to 6, 5 to 7 and 6 to 7. Using two consecutive runs at the same heating rate, twelve estimates of the diffusivity were made.

The computer program used was a simple flat plate simulation having a nodal geometry as is shown in Figure 31. The simulation allowed input of an actual measured temperature - time history at node 1 and predicted the temperature - time history of one of the other two thermocouples located at node i in the simulation. In chosing the correct diffusivity in the simulation, the predicted temperature - time history duplicated the actual.

Figure 31
NODAL GEOMETRY



A\* = DISTANCE BETWEEN THE TWO THERMOCOUPLES USED IN THE SIMULATION

Six predictions of diffusivities per experimental run were obtained in using the simulation. They included predicted diffusivities between positions stated in the previous paragraph. The method of selecting the correct diffusivity was through trial and error. If the diffusivity used was too low, the predicted temperature time history was lower than the actual. If the diffusivity used was too high, the predicted history was higher.

The extra nodes from i+1 to n were added to negate end effects. A technique to check that this assumption was valid was to perform the simulation with more nodes. If the predicted temperature time history at node i was unchanged then node i was uneffected by the end effects. In the actual experiment, end effects were negated by adding an additional quantity of powder, approximately one and a half inches above the last thermocouple.

There is essentially one nodal equation used in the flat plate simulation. For the internal nodes ( node 2 to node n-1), the equation is:

$$T_{j}' = T_{j} * \left[1 - 2 \frac{DT * AL}{DX^{2}}\right] + \frac{DT * AL}{DX^{2}} * \left[T_{j-1} + T_{j+1}\right]$$
 (eq. 48)

where  $T_{j}$ ' = the temperature of node j one time increment(DT) latter.

 $T_i$  = the present temperature of node j

 $T_{i-1}$  = the present temperature of node j-1

 $T_{j+1}$  = the present temperature of node j+1

DT = time increment

AL = thermal diffusivity =  $k/\rho C_{D}$ 

k = thermal conductivity

c<sub>p</sub> = specific heat

 $\rho$  = density

The last node n is assumed adiabatic. The equation for this node is the same as equation 41 except  $\mathbf{T}_{n+1}$  will have the same value as  $\mathbf{T}_{n-1}$ , or:

$$T_n' = T_n * \left[ 1 - 2 \frac{DT * AL}{DX^2} \right] + 2 \frac{DT * AL}{DX^2} * T_{n-1}$$
 (Eq 49)

The computer simulation performs the following steps:

- 1 Sets values for nodal distance(DX ) and diffusivity(AL) and then determines the maximum time increment.
- 2 Reduces the time increment to a division of a one minute interval so that temperatures will be calculated for each minute.
  - 3 Sets all nodes to the initial temperatures.
- 4 Increments time by one time increment and sets node 1 to the temperature of the lower thermocouple at that time.

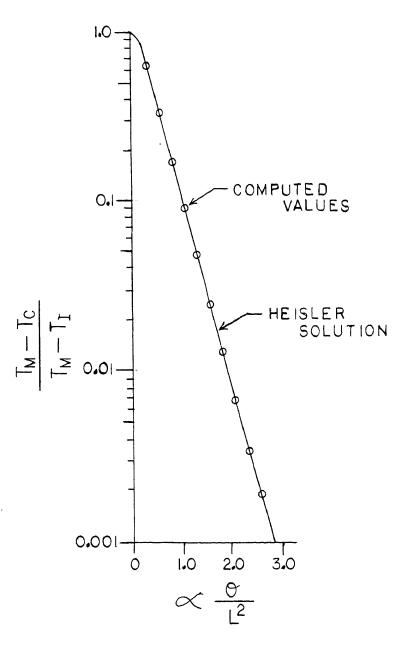
- 5 Calculates the new nodal temperature for node 2 through node n.
- 6 If the time is at a minute interval, then print the predicted temperature of the other thermocouple.
- 7 Change the old nodal temperature values to the new predicted values.
  - 8 Go back to step 4 until the simulation is over.

To check the validity of this simulation, results of a trail simulation are compared to the Heisler Charts in Figure 32. Note the exact fit proving the simulation.

In the performance of this simulation, if the temperature time history of node i is identical to the experiment, the correct diffusivity must be ajusted accordingly and the simulation repeated. An example of the simulation - experiment matching for the Phillips Petroleum polyethylene(Appendix A) is shown in Figure 33.

Figure 34 shows the values of the diffusivity found with the corresponding thermal conductivity. This spread in values will be used in the rotational molding simulation to determine the effects in molding time. Appendix D lists the computer program used in this flat plate simulation.

Figure 32
COMPARISON BETWEEN SIMULATION AND HEISLER CHARTS



T<sub>M</sub> = MOLD TEMP,

T<sub>C</sub> = CENTERLINE
TEMP,

T<sub>I</sub> = INITIAL
TEMP,

Figure 33

COMPARISON BETWEEN MEASURED AND COMPUTER

PREDICTED TEMPERATURES

RUN NO. 1 o-measured data

•-COMPUTER RESULTS

DIFFUSIVITY USED = 6 x 10<sup>-3</sup> FT<sup>2</sup>/HR

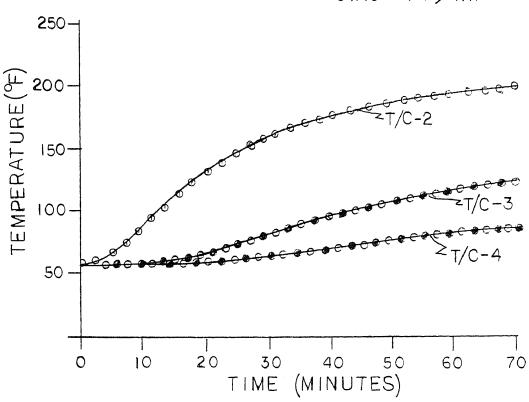


FIGURE 34

EXPERIMENTALLY AND COMPUTER SIMULATED DETERMINATION

OF THERMAL CONDUCTIVITY

Run No.	Thermo-	Diffusivity		Thermal Conductivity	
		(Ft2/Hr)	(Cm2/sec)	(B/hr-ft-F)	(J/cm-s-K)
1	2 - 3	6.0E-03	1.55E-03	0.105	1.82E-03
	2 - 4	6.0E-03	1.55E-03	0.105	1.82E-03
	3 - 4	6.0E-03	1.55E-03	0.105	1.82E-03
	5 <b>-</b> 6	7.5E-03	1.94E-03	0.131	2.27E-03
	5 <b>–</b> 7	6.5E-03	1.68E-03	0.113	1.96E-03
	6 - 7	7.5E-03	1.94E-03	0.131	2.27E-03
2	2 - 3	6.0E-03	1.55E-03	0.105	1.82E-03
	2 - 4	6.0E-03	1.55E-03	0.105	1.82E-03
	3 - 4	6.0E-03	1.55E-03	0.105	1.82E-03
	5 <b>-</b> 6	5.0E-03	1.29E-03	0.087	1.51E-03
	5 - 7	7.0E-03	1.81E-03	0.122	2.11E-03
	6 - 7	9.5E-03	2.45E-03	0.160	2.77E-03

Average Thermal Conductivity = .155 B/Hr-ft-F Standard Deviation = 0.020 B/Hr-ft-F

NOTE: Thermal Conductivity calculated computed diffusivity using a density of 31.8 lb/ft3 and a specific heat of 0.55 B/lb-F (Appendix A).

## X. THE COMPUTER SIMULATION FOR ROTATIONAL MOLDING

The rotational molding computer simulation (Appendix F) consists of the main program named ROTOTP, supported by the following subroutines: PRINT and CONST, and functions: VIS, SU, RHO, RADX, COND, and CP. The main program ROTOTP performs the simulation, PRINT is the printing routine and CONST computes the constants used in the nodal temperature equations. In addition CONST compute the neck growth for each node during every time increment. The functions VIS, SU, RHO, COND, and CP computes the viscosity, surface tension, density, thermal conductivity and specific heat respectively for each node. Function RADX computes the distance between adjacent nodes based on the density at those nodes.

The main program is divided into ten parts as is labeled in the program listing. Preceding Section I is a glossary of all variables used in the main programs. Section I initializes all the variables and sets all program flags. The variables include: the mold stick temperature(SS), the complete melt temperature(SM), the printing interval(PT), the number of wedges(NC), the material radius(RAD), the initial temperature(IT), the mold radius(RO), the radius from the center of the mold to the surface of the stationary pool material(RI), acceleration

due to gravity(G), rotational speed of the mold(RPM), the angle of response(BA), and the coefficients used in the mold temperature-time history equation(A, B, and YY).

Section II performs initial computations using the initial values in Section I. They include computation of the pool angle theta(TH), cross-sectional area of stationary pool(AA), radial mold velocity(W), maximum penetration thickness(AD), maximum time increment(DT), the nodal distance(DX), and the number of nodes per column(N). This section then sets each node to the initial temperature and each nodal neck radius to zero. Then all major values are printed.

Section III begins the iteration process. The mold temperature(MT) based on time is determined. For each iteration the maximum time increment(DT) and time increment used(TS) is determined using the maximum diffusivity and the minimum nodal distance( all usually located at the surface node). Cross sectional areas(AT and AS), penetration thickness(AD) and free-fall time(TF) is also calculated for each iteration.

Section IV computes the nodal temperatures and nodal neck size at each nodal subvolume. The values Cl, C2, C3, DB, and DA in the temperature equation and the new neck size is determined in the subroutine CONST. The program

also checks each node to determine if its temperature is equal to or has exceeded the complete melt temperature(SM). When reached, the nodal neck radius becomes equal to the original radius of the material simulating full densification of the material.

Section V controls the printing of nodal temperatures and neck sizes. The printing time interval(PT) was selected in Section I. In addition, the temperature and neck size are printed at the point in time when there is no material remaining in the stationary pool and when all material has completely densified. At the end of the simulation, the neck size of each node location is printed showing the amount of neck growth that occured before complete melt had overcome the sintering process.

If mold sticking has not yet occured in the simulation, the program enters Section VI where it calculates the new free-fall section's average temperature for the next time interval. It calculates the temperature energy of the last column of the stationary pool that enters the free-fall zone during the next time interval and adds it to that remaining in the pool after a portion of it re-enters the stationary pool.

The program then rotates the stationary pool nodes

over one column distance simulating one time increment of travel. For example, the values of the nodes in the next to the last column move over to the last column. The nodal values in the second to the last column are shifted to the next to the last column and so on.

Next, the nodes in the first column are then initialized to the temperature of the material that had just exited from the free-fall zone and re-entered the stationary pool. The computer simulation then returns to Section IV and begins another iteration process for the next time interval.

If there is material sticking to the mold, the program by-passes Section VI and goes on to Section VII where it determines the amount of material in the last column that will stick to the mold wall and the amount that enters the free-fall zone during the next time interval.

This routine then sets up new nodal columns to simulate the stuck material. Since the time interval(TS) decreases as the cross-section of the stationary pool decreases, the simulation maintains, for each stuck column, its width(DC), position in the mold(PS), the number of nodes(N3), its height(H2), the distance between the last two nodes of the column(H3), and the nodal neck sizes (ANECK). It must be noted that the stuck height(H2) is

very seldom an integer of the nodal distance(DX), therefore all nodal distances are the length DX except between the last two nodes which has a distance larger than DX making up the extra distance.

This section then calculates the energy of the material in the stationary pool column not sticking but entering the free-fall zone.

Section VIII simulates the rotation of the mold when sticking has occured. The stationary pool and then the stuck material is rotated one column width in the same manner described in the discussion of Section VI. The values of the stuck column's width(DC), number of nodes(N3), height(H2), and the distance between the last two nodes(H3) is also shifted along with the column. The position(PS) is updated to reflect its new position in the mold.

Section IX checks if the rotation causes any of the stuck material to re-enter the stationary pool. If no re-entering occurs, the computer simulation goes to Section X. If re-entering occurs, the simulation determines the portion of column width(EN) that re-enters the pool as well as the portion remaining outside(OT).

The portion of the column remaining outside (OT) is combined to the adjacent column to make one equivalent column having a combined average nodal temperatures with a width equal to OT plus the width of the adjacent column.

The nodal temperatures and neck size of the stationary pool's first column is assigned the same values as the re-entering material(EN). Since the nodal distance of the last two nodes is usually larger than the normal nodal distance(DX), the last node is replaced by two nodes. The next to the last node will have the same neck size and temperature as the last node of the stuck material re-entering the pool. The remaining material is to be combined with the re-entering powder from the free fall zone to form an equivalent node. Additional nodal subvolumes are added representing the re-entering powder from the free fall zone adding an additional height equal to the penetration thickness.

If there is not enough mass in the stationary pool to allow adding the height of a penetration thickness, then fewer subvolumes are used having an equivalent cross sectional area equal to that of the remaining pool material. The last node will then be modeled as an adiabatic edge. The temperatures and neck radius nodal values of these added subvolumes are equal to that of the re-entering material from the free-fall zone.

Section X calculates the new angle theta(TH) based on the amount of material in the stationary pool zone. The program then returns to Section III and starts the iteration again for the next time increment.

When the amount of material in the stationary pool becomes zero, angle theta (TH) becomes zero. Therefore, the rotational simulation portion of this program is no longer needed making this simulation a simple one-dimensional heat transfer simulation having the number of columns equal to that of the number of columns of stuck material. The simulation then consists of the iteration process in Sections III and IV until the temperature of the last nodes reaches or exceeds the complete melt temperature(SM). At that point, the simulation calls SUBROUTINE PRINT and then terminates the simulation by printing the neck size of each node height level at the point when complete melt had overcome the sintering process.

#### XI. COMPUTER SIMULATION WITH OVEN CONVECTION

To simulate heating by oven convection in rotational molding, a few significant modifications were made to the original nodal model and to the computer program earlier developed (as listed in Appendix F.).

The original model assumed a temperature-time relationship which was described by a polynomial equation for the mold's internal surface, NODE 1, that is in direct contact with the powdered material. The temperature values of the other nodes are then calculated during each iteration based on the temperature time history of NODE 1.

With oven convection, two additional nodes are used in the model accounting of the effect of the mold itself and the resistance between the heated air and the mold's surface. Figure 35 shows the position of these extra nodes. The node labled A in the figure represents the oven air and has a temperature value equal to the oven temperature, or:

T(A) = T(OVEN)

NODE B represents the exterior surface of the mold.

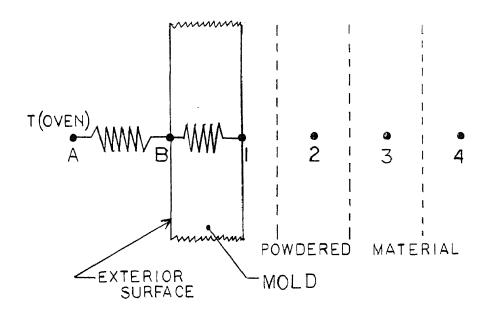
Its nodal temperature equation (eq 36) is listed in Chapter

VI. NODE 1 represents the interior surface of the mold

having equation 38 as its nodal temperature equation. The

Figure 35

OVEN CONVECTION SURFACE NODAL CONVENTION



distance between NODE B and NODE 1 is equal to the thickness of the mold.

With the additional nodal equations, there are two additional stability equations, each having a maximum allowable time increment. The computer simulation calculates for each iteration the three maximum time increments and then allows the time increment used to be no greater than the minimum value of the three.

The future temperature values of NODE B and NODE1 as well as the remaining nodes are computed for each iteration using the nodal temperature equations developed. Appendix H is the listing of the Rotational Molding Simulation having oven convection.

## XII. DISCUSSION OF RESULTS

Using the computer simulation program earlier developed, a number of runs were performed in order to predict pool depletion and total densification times with variations in temperature-time histories, rotational speeds, material amount, and thermal properties of the powdered material. The results of the rotomolding simulations as well as results of the Vanderbeck and Throne/Ahdout simulations are compared in Table 2 to the actual experimental results performed by Vanderbeck(30). The simulations reported in Table 2 used the polynominal temperature-time equations discussed earilier.

Section I of Table 2 compares each simulation at various power input levels having 50 grams of powdered material in the mold cavity. The different power inputs correspond to a particular mold temperature-time history as described by Vanderbeck(30). Results show a 40% to 55% error in the Vanderbeck simulation, a -3.6% to -10.2% error in the Throne/Ahdout simulation and a -2.1% to -11.1% error in the Rotational Mold Simulation developed here.

In Section II and III, the thermal diffusivity of the powder used in Section I was decreased by 20% and increased by 20% respectively. Simulations using the 500 watt and

TABLE 2 POOL DEPLETION TIMES

		I				*····	
TUO	ON % ERROR	-3.6 -7.4 -7.3 -10.2	-1.8 -10.2	-5.2 -12.8	-11.2 -10.5 -7.8 -8.2	-7.9 -7.9 -8.2	45.1 48.1 49.4 49.1
THRONE/AHDOUT	SIMULATION PREDICTED (SEC) E	369 310 278 246	376 246	363 239	246 246 246 246	278 278 278 278	737 647 596 546
	% ERROR	-2.1 -5.9 -7.9 -11.1	-1.2	-3.0	-15.1 -16.7 -15.0 -16.0	-11.2 -13.2 -14.5	. 7.20 7.120
SIMULATION	PREDICTED (SEC)	375 315 276 244	378 247	372 240	235 227 227 225	268 262 259 258	508 459 424 392
Ą	IHEKMAL DIFFUSIVITY (CM2/SEC)	1.70E-03 1.70E-03 1.70E-03 1.70E-03	1.36E-03 1.36E-03	2.04E-03 2.04E-03	1.70E-03 1.70E-03 1.70E-03 1.70E-03	1.70E-03 1.70E-03 1.70E-03 1.70E-03	1.70E-03 1.70E-03 1.70E-03 1.70E-03
	% ERROR	44 40 46 53	# T		488 888 748 7	40 34 28 28	. 22 22 23 24 25 25
	PREDICTED (SEC)	553 470 439 411	     	) 1       1	392 368 354	424 403 388 373	655 573 516 486
1011110	HCIUAL TIME (SEC)	383 335 300 274	383 274	383 274	277 275 267 268	302 302 303 292	508 437 399 366
-	H RI	4.796 4.796 4.796 4.796	4.796	4.796	4.796 4.796 4.796 4.796	4.796 4.796 4.796 4.796	1.361 1.361 1.361 1.361
i i	MAIEKIAL AMOUNT (GRMS)	0.00 0.00 0.00	50.0	50.0	0.000	0.000 0.000 0.000	884. 84.33 84.33
-	RPM M	~~~	~~	~ ~	10 13 25 25	10 15 20 20 20	~~~
i	FUWER INPUT (WATTS)	500 600 700 800	500 800	500 800	800 800 800 800	700 700 700 700	500 600 700 800
		₩	11	III	>1	>	, v

For Temperature Equation Used:  $T=At+Bt^2+IT$ 

IT = 20 C					
œ		-0.324	-0.430	-0.531	-0.711
Œ		18.2	22.0	25.0	0.60
Watts	1	500	009	700	BOO

800 watt temperature-time histories resulted in the predicted pool depletion times shown for both the Throne/Ahdout and the Rotational Mold Simulations. For the 20% change, the Rotomolding Simulation showed a change between 0.8% and 1.6% while the Throne/Ahdout Flat Plate simulation resulted in a 1.6% to 3.3% change.

Section IV and V in Table 2 show the results of varying the rotational speed of the mold. The Vanderbeck simulation yielded a difference of 27% to 42% between the predicted and actual experimental results. Because the Throne/Ahdout simulation is a flat plate analysis with no rotational parameter, the pool depletion times are constant for power input reguardless of the actual rotational speed. The Throne/Ahdout simulation produced a 4.8% to 11.2% difference between actual and experimental results. The Rotational Mold Simulation had a 11.2% to 16.7% difference.

Note that the Rotational Mold Simulation developed here predicts a time difference of 10 seconds between the 10 RPM and 25 RPM pool depletion times of the 800 watt power input while the actual experimental difference was 9 seconds. For the 700 watt power input, the predicted and actual experimental time differences were identical with 10 seconds.

Section VI is the actual and predicted pool depletion

time results of a mold containing 84.5 grams of powder. The Throne/Ahdout has a difference ranging from 45.1% to 49.1%. The Rotomold Simulation produced a 0.0% to 7.1% difference.

Table 3 is a comparison of pool depletion and total densification times with the Rotomold and Throne/Ahdout simulation containing a 50 gram charge using oven convection coefficients in lieu of the polynomial temperature equations. Using a typical oven temperature of 371 degrees Centigrade, an aluminum mold thickness of 0.655 cm yielded results showing a close agreement between the two simulation for typical air convection coefficients.

The previous comparison of the 50 gram charge simulations show that the Rotomold Simulation developed here and the Throne/Ahdout Simulation will predict pool depletion and total densification times within a satisfactory tolerance. The Vanderbeck Simulation however, showed between 27% and 57% error. It also showed at those power inputs an insensitivity to changes in thermal conductivity.

It is confusing however, that for the 50 gram charge simulations, the Throne/Ahdout Simulation which is a flat plate simulation with no rotation or thermal mixing has as

TABLE 3

## DEPLETION AND DENSIFICATION TIMES

Convection	Coefficient	Ribe Simulation Depletion	on (sec)	Throne/Al Simulation Depletion	
(B/Hr-Ft2-F)	(J/sec-cm2-K)	Deprecion	cation	Depiceron	cation
2	1.1356E-03	442	577	431	572
4	2.2712E-03	241	300	232	295
6	3.4068E-03	174	208	167	204
8	4.5424E-03	141	162	134	158
10	5.6780E-03	126	135	125	135

Material Stick Temperature (deg C) = 110 Material Melt Temperature (deg C) = 138 Oven Air Temperature (deg C) = 371.0 Amount of material in mold (grams) = 50.0

## Properties of Aluminum Mold

Thermal Conductivity (J/cm-sec-K) = 2.025 Density (Kg/Cm3) = 2.707E-03 Specific Heat (J/Kg-K) = 8.7085E-02 Thermal Diffusivity (Cm2/sec) = 8.588

good as or even better results then the Rotational Mold Simulation developed. It would seem logical that the flat plate simulation temperature profile would show a maximum temperature at the mold's surface dropping rapidly to a minimum temperature at the inner surface. The difference between the minimum and maximum temperature would then increase with time causing a dramatic increase in predicted pool depletion and densification time for the powdered material. The flat plate simulation did not predict this. In fact, its predictions for the temperature profiles were very close to the Rotomold Simulation.

This seeming discrepency can be explained. In the Vanderbeck experiments, the pool depletion times for 500 through 800 watts are shown(Table 3) equivalent to an oven convection coefficient of approximently 1.137E-03 to 2.271E-03 J/Sec-cm2-K (2.0 to 4.0 B/hr-ft2-F). The ratio of the convection coefficient to the thermal conductivity of the powder divided by the thickness( h/(k/L) ) determines the temperature profile. Krieth (55) states that if the ratio, h/(k/L) is less than 0.1 then the temperature throughout the material can be assumed constant for transcient thermal analysis. In other words, the heat input is so slow that it allows the material to distribute its internal energy evenly.

Pre-distributing 50 grams of powdered material around the mold will result in a material height of 0.4 cm, the ratio, h/(k/L) varies between 0.22 and 0.46. Although these values are not below 0.1, they are close enough to produce similar results. In addition, because these ratios are low, a 20% change in either direction of the thermal conductivity (as was reported in Section II and III of Table 2) would produce very little change in the pool depletion and total densification times.

With 80 grams of material charge, the material height (the value L) becomes 1.67 cm increasing the ratio four fold. Section VI of Table 2 shows the dramatic change in prediction times of the Throne/Ahdout simulation as compared to the actual results. The error now increases to a range of 45% to 49%.

By increasing the thermal conductivity, Table 4 shows again the limitation of the Throne/Ahdout simulation. Notice that as the conductivity, k, becomes smaller (making h/(k/l) larger), the deviation between the Rotational Molding Simulation and the Throne/Ahdout Simulation becomes extremely large.

Similar results are found if the value of the coefficient of convection becomes large as may occur in

TABLE 4

## DEPLETION AND DENSIFICATION TIMES AS A FUNCTION OF THERMAL CONDUCTIVITY

Thermal Conduc- tivity (J/Cm-sec-K)	Thermal Diffu- sivity (cm2/sec)	Ribe Simulat (sec) Depletion	1	Throne/A Simulat (sec Depletion	ion)
1.99E-02	1.7E-02	95	84	116	113
1.99E-03	1.7E-03	126	126	135	135
1.99E-04	1.7E-04	154	245	200	258
1.99E-05	1.7E-05	215	873	418	919

Material Stick Temperature (deg C) = 110 Material Melt Temperature (deg C) = 138 Oven Air Temperature (deg C) = 371.0Amount of material in mold (grams) = 50.0Oven Convection Coefficient (J/cm2-sec-K) = 5.678E-03

Properties of Aluminum Mold

Thermal Conductivity (J/cm-sec-K) = 2.025Density (Kg/Cm3) = 2.707E-03Specific Heat (J/Kg-K) = 8.7085E-02Thermal Diffusivity (Cm2/sec) = 8.588 radiant and forced Fluid heating of the mold. Table 5 illustrates the effect of changing convection coefficients by factors of magnitude. Again, high errors (19% to 34%) occur in the prediction of the pool depletion and densification times.

In summary, the Throne/Ahdout Simulation is accurate only when using certain combinations of the simulation parameters, otherwize large errors will occur in pool depletion and total densification predictions. The Rotational Molding Simulation developed in this dissertation does predict accurate times for the pool depletion and densification as was verified by the experimental results of Van der Beck(30). It models closely the material flow, mold heating of the powdered material, rotation of the stationary pool and material melted to the mold wall, and the thermal and physical mixing during the free-fall zone.

TABLE 5

# DEPLETION AND DENSIFICATION TIMES AS A FUNCTION OF CONVECTION COEFFICIENT

Convection Coefficient (J/Cm2-S-K)	Deplet: Ribe Sim.	ion Time T/A Sim.	es (sec) % Diff.	Densif: Ribe Sim.	ication T/A Sim.	Times % Diff.
3.407E-03	331	445	34	355	468	32
3.407E-02	122	151	24	137	178	30
3.407E-01	101	121	20	109	130	19
3.407	99	118	19	105	125	19

Material Stick Temperature (deg C) = 110 Material Melt Temperature (deg C) = 138 Oven Air Temperature (deg C) = 371.0 Amount of material in mold (grams) = 80.0 Thermal Diffusivity (Cm2/sec) = 1.7E-03

Properties of Aluminum Mold

Thermal Conductivity (J/cm-sec-K) = 2.025Density (Kg/Cm3) = 2.707E-03Specific Heat (J/Kg-K) = 8.7085E-02Thermal Diffusivity (Cm2/sec) = 8.588

NOTE: T/A = Throne/Adhout

### XIII. CONCLUSIONS AND REMARKS

This dissertation has presented an investigation of the densification process in rotational molding of a thermoplastic powder in a cylindrical cavity.

First, a thorough literature survey was performed to review past works of rotational molding analysis, as well as the study of sintering, the effects of temperature on viscosity and surface tension for polymeric material, a review of thermal conductivity equations of composite materials and a review of other areas that would assist in the analysis of the densification process in rotational molding.

Next, an in depth study of the mass flow in a rotating cylinder together with an analysis of the heat transfer during rotational molding were performed to attain the understanding necessary to model the densification process.

The densification process in rotational molding was then modeled mathematically using Nodal Analysis. Nodal temperature equations were derived for typical oven convection heating and a special case where the mold's temperature-time history is known. The special case was modeled to provide a comparison and validation of the

simulation developed with previous experimental research.

The research for this dissertation included an exhaustive investigation and analysis into the densification process (neck formation) by use of Scanning Electronic Microscope(SEM) photography. Because of the large depth-of-field of the SEM, photographs which were almost impossible to obtain before, allow the analysis to be performed. Based on this analysis, the intermediate physical property correlations needed in the nodal temperature equations were derived.

Lacking an agreement between results of published composite thermal conductivity equations, a hybrid experimental procedure coupled with a computer simulation was devised to determine the actual initial thermal conductivity of the powdered polymeric material.

Finally, a computer program was written to simulate the heating and densification during the rotational molding process. Results showed agreement with actual rotational molding experimental findings. In addition, results of other simulations were compared showing their shortcomings.

It is important to include in this summary a discussion of one logistical drawback in the use of this simulation; that being the amount of computer time required.

Because of the intricate modeling, the simulation will require from 800 to 6000 seconds of computer time. This equates to a maximum of \$650.00 per simulation when used on a Control Data Corp. mainframe system(CDC 6500).

If, because of cost, the use of the Rotational Molding Simulation becomes prohibitive, modifications to the Throne/Ahdout Simulation will improve its accuracy. A complete examination and modification to the Throne/Ahdout Simulation is beyond the scope of work for this dissertation. However let it be noted that the major modification would encompass a thermal mixing routine to simulate the free fall thermal/physical mixing zone for the powdered material that has a temperature below the stick temperature.

Since miminal computer cost was not an original requirement, this dissertation has performed and completed all its objectives. It has investigated the heating and densification portion of the rotational molding process. Included, was an analysis of the heat transfer, mass flow and neck formation colmenating into a valid and proven simulation.

APPENDIX A

NOMINAL PHYSICAL PROPERTIES OF MARLEX LX470 (POLYETHYLENE)

PROPERTY <sup>1</sup>	ASTM	Metric Units	Value
Density	D1505	g/cm3	0.943
Melt Index	D1238	g/10 min	3.0
Flow Index, CIL, 190C 10.4 MPa2		g/10 min	3.0
Brittleness Temperature	D746	Deg. C	-118
Specific Heat @ 90C(ref 58)		cal/g-C	0.55
ROTATIONAL MOLDED PROPERTIES	3		
ESCR, Condition A, F50	D1693	h	200
Tensile Strength at Yield 2"(50.8 mm) per min.	D638 Type IV Spec	MPa	22.1
Elongation 2"(50.8 mm) per min.	D638 TYPE IV Spec	ર્લ	350
Flexural Modulus	D790	MPa	965
Impact ARM Standard 4 at -28.9 deg C		J	68

<sup>1</sup> Physical properties reported herein were determined on compression molded speciments prepared in accordance with Procedure C of ASTM D1928 (Ref 57).

<sup>&</sup>lt;sup>2</sup> Data obtained using a gas extrusion plastometer based on design by Canadian Industries, Ltd., with a die having an orifice diameter of 0.49 mm and a land length of 4.48mm.

<sup>&</sup>lt;sup>3</sup> Physical properties are based on parts molded at optimum conditions (Ref 57).

<sup>4</sup> Ten pound dart with 0.5 inch point in center of 0.125 inch thick unsupported 3.5 inch diameter area.

## APPENDIX A (continued)

U. S. STANDARD SIEVE SIZE DISTRIBUTION OF MARLEX LX470

Size	Inches	mm 	Percent of material stopped by sieve
30 35	0.0232 0.0197	0.590 0.500	0.0007 4.40
40	0.0197	0.420	16.41
45	0.0138	0.350	20.81
50	0.0117	0.297	19.19
60	0.0088	0.250	5.96
80	0.0070	0.177	18.10
100	0.0059	0.149	7.78
120	0.0049	0.125	3.56
170	0.0035	0.088	2.32
200	0.0029	0.074	1.05
230	0.0025	0.063	0.27
270	0.0021	0.055	0.11
PAN			0.14
			100.00 %

Measured Specific Gravity of Powder = 0.5096 Specific Gravity of Resin = 0.9430

Void Fraction of Air = 45.96%

APPENDIX B

# PHYSICAL PLASTIC PROPERTIES OF ONE-EIGHT DIAMETER SPHERES MADE OF POLYETHYLENE AND ACRYLIC

PROPERTY	UNITS	POLY- ETHYLENE	ACRYLIC
Impact Strength, Notched Izod	Ft-lb/in	1 - 10	0.4 - 0.6
Tensile Strength	PSI x 1000	2.5 - 5	7.9
Tensile Modulus	PSI x 1000	85 -160	350 - 450
Thermal Conductivity	Cal/cm2/sec/ C/cm x 10000	8	4 - 6
Specific Gravity		0.94 - 0.96	1.18 - 1.19
Elongation	8	5 - 10	2 - 10
Flexural Strength	PSI x 1000	2 - 3	14 - 16
Flexural Modulus	PSI x 1000	90 - 150	350 - 450

#### APPENDIX C

# PHYSICAL PROPERTIES OF MICROSPHERES COMMERICALLY NAMED "ELVACITE" ACRYLIC RESINS 2021 by DuPont

Density(resin) 1.196 Kg/M3
Glass Transition Temperature 100 Deg C
Tukon Hardness, Knoop No. 20
Tensile Strength (23 Deg C., 50% RH) 106 MPa 15kPsi
Elongation at Break (23 Deg C., 50% RH) 4%

U. S. STANDARD SIEVE SIZE DISTRIBUTION

Sieve Size	Inches	mm	Percent of material stopped by sieve
30	0.0232	0.590	0.48
35	0.0197	0.500	0.16
40	0.0164	0.420	0.18
45	0.0138	0.350	0.27
50	0.0117	0.297	3.26
60	0.0088	0.250	18.87
80	0.0070	0.177	65.58
100	0.0059	0.149	4.83
120	0.0049	0.125	3.28
170	0.0035	0.088	1.92
200	0.0029	0.074	0.92
230	0.0025	0.063	0.10
270	0.0021	0.055	0.08
PAN			0.07
			100.00 %

Measured Specific Gravity of Powder = 0.730 Specific Gravity of Resin = 1.196

Void Fraction of Air = 38.96%

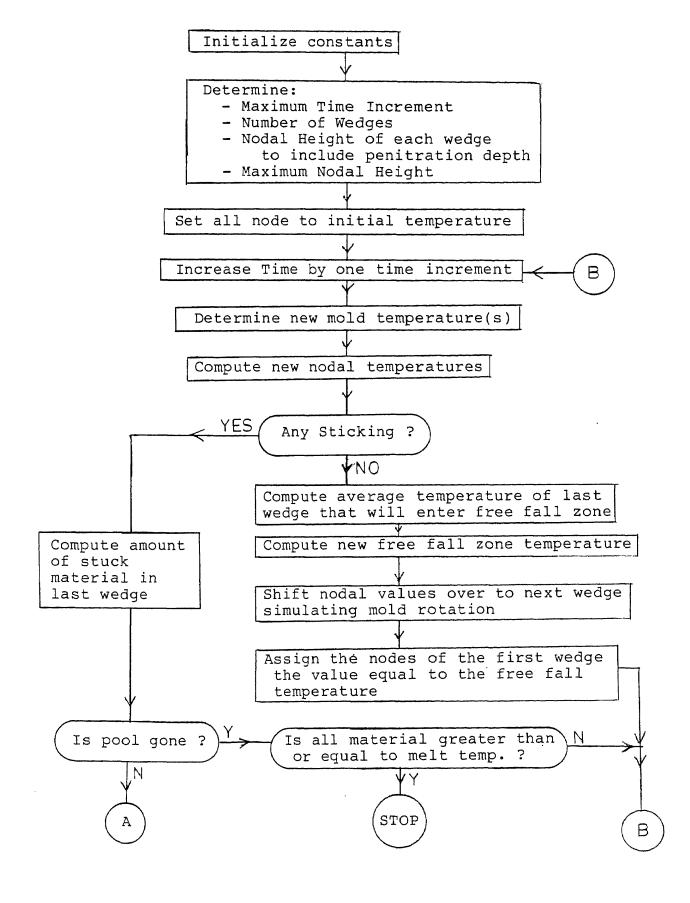
#### APPENDIX D

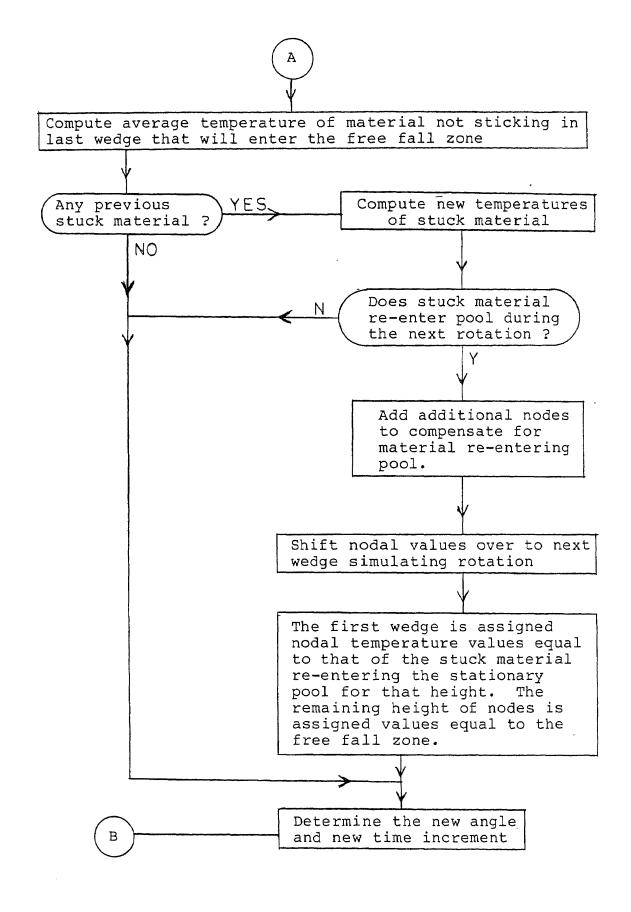
#### FLAT PLATE SIMULATION

```
90 REM SLAB TEMPERATURE TIME HISTORY SIMULATION
95 DIM T(30),TN(30),TT(15),TP(15)
97 OPEN1,4,2
100 AL=6.200E-03:REM FT2/HR
105 PRINT#1, "AL = "; AL
110 AL=AL*0.04 : REM IN2/SEC
120 DX=0.09375
                :N=17 :REM DX IN INCHES
125 \text{ TM} = 0.0 : NN = 0
130 TS=DX©2/(2*AL)
140 M = INT(60/TS) + 1
150 DT=60/M
160 FORJJ=1TO11:READ TT(JJ):NEXTJJ
170 FORJJ=1TO11: READ TP(JJ):NEXTJJ
180 FORI=1TON:T(I)=TP(1):NEXTI
200 TM=TM+DT:NN=NN+1
210 GOSUB 1000
220 T(N+1)=T(N-1)
230 FOR I=2TON
240 TN(I) = T(I) * (1-2*DT*AL/DX©2) + DT*AL/DX©2*(T(I-1)+T(I+1))
245 IFTN(I)<T(I)THEN STOP
250 NEXTI
260 FORI=2TON:T(I)=TN(I):NEXTI
265 PRINTTM; T(9)
270 IF INT(NN/M)=NN/M THEN PRINT#1,TM;TM/60,T(1),T(9),T(16)
280 IF TM/60>60 THEN PRINT#1:CLOSE1:STOP
290 GOTO200
1000 REM SUBROUTINE TO DETERMINE 1ST NODE TEMPERATURE
1010 FOR JJ=2TO11
1020 IF(TM/60 <TT(JJ)) THEN 1030
1025 NEXTJJ
1030 RA=(TM/60 - TT(JJ-1))/(TT(JJ) - TT(JJ-1))
1035 IFRA=OTHEN T(1)=TP(JJ-1):RETURN
1040 T(1) = TP(JJ-1) + RA*(TP(JJ) - TP(JJ-1))
1050 RETURN
2000 DATA 0,2,5,10,15,20,30,40,50,60,100
```

2010 DATA 64,67,86,127,153,171,195,210,220,227,227

APPENDIX E
FLOW DIAGRAM FOR THE ROTATIONAL MOLDING SIMULATION





## ROTATIONAL MOLDING SIMULATION

 PROGRAM R485 (I NPUT, OUTPUT, TAPES=INPUT, TAPE6=OUTPUT, TAPE8)
DIMENSION TN(100),T(200,100),OC(200),PS(200),N3(200),H3(200)
 6 , H2(200) , ANECK(200, 100) , BNECK(100) , ANEK(100)
 COMMON SS, SM, AL, AM, DX, N, TM, AA, AT, AD, TH, DL, N2, CT, NC, N3, RAD, TS
COMMON TAIR, 34T, MT
 INTEGER CT,KL
RATA ANECK RATCH (200007 P. D. 1007)
 DATA ANECK, BYECK/20000* 0.0, 100* 0.0/
GLOSSARY
 SLUSSART
A COEFFIEIENT IN TEMPERATURE EQUATION
 AA TOTAL IROSS SECTIONAL POOL AREA (CM2)
AD PENITRATION THICKNESS (CM)
 AL INITIAL THERMAL DIFFUSIVITY (CM2/SEC)
AM THERMAL DIFFUSIVITY AT COMPLETE MELT (CM2/SEC)
 ANECK NECK RADIUS AT EACH NOTE (CM)
ANEK NECK RADIJS OF EACH NODAL AREA AT THE TIME
 MELTING OVERCOMES THE SINTERING PORCESS (CM)
ANK DUMMY VARIABLE = ANECK(1,1)
 AR DUMMY VARIABLE USED IN CALCULATING REPAINING
CROSS SECTIONAL POOL AREA
 AS TOTAL AREA ENCOMPANSING NODAL ANALYSIS
( HEIGHT OF NODES X TOTAL WIDTH)
 AT SROSS-SESTIONAL AREA OF ONE WEBGE (CH 2)
AV DUMMY VARIABLE USED TO CETERMINE NEW ANGLE THETHAT
 A3 - 3 ROSS-SECTIONAL A FEA OF STUCK COLUMN CF MATERIAL THAT
THAT RE-ENTERS THE FOOL IN THE NEXT TIME INCREMENT.
 A4 THE DROSS SECTIONAL AREA OF THE STUCK COLUMN OF
MATERIAL THAT REMAINS OUTSIDE THE POOL
 DURING THE NEXT TIME INCREMENT (C)2)
B COEFFICIENT IN TEMPERATURE EQUATION
 BA ANGLE OF RESPONSE OF THE POOL (RAD)
BB DUMMY VARIABLE USED IN TEMPERATURE EQUATION
 BE DUMMY VARIABLE USEC SURFACE NOBAL NECK SIZE EQUATION
 BNECK NEW COMPUTED NECK SIZE FROM SUBROUTINE (CM)
CC HALF LENGTH OF POOL CORD (CM)
 CT COUNTER USED TO COUNT THE NUMBER OF STUCK COLUMNS
CI CONSTANT USED IN TEMPERATURE EQUATIONS DETERMINED IN
 SUBROUTINE CONST
CZ CONSTANT USED IN NODAL TEMPERATURE EQUATIONS
 C3 CONSTANT USED IN NODAL TEMPERATURE EQUATIONS
DB AVERAGE DIFFUSIVITY BETWEEN NODES I+1 AND NODE I (CM2/SEC:  DB AVERAGE DIFFUSIVITY BETWEEN NODES I-1 AND NODE I (CM2/SEC
 DO AVERAGE STATE OF STUCK NOTAL COLUMNS (RAD)
DL ARC LENGTH PER TIME INCREMENT (RAD/SEC)
 DT MAXIMUM TIME INCREMENT (SEC)
DX DISTANCE BETWEEN NODES (CM)
 EN PORTION OF SOLUMN ARE WIDTH RE-ENTERING THE
STATIONARY POOL (RADIANS)
 FT AVERAGE TEMPERATURE OF MATERIAL IN FREE FALL ZONE (C)
G ACCELLERATION DUE TO GRAVITY (CM/SEC2)
 H HEIGHT OF STUCK MATERIAL LEAVING POOL (CM)
HT DUMMY VARIABLE USED IN COMPUTING POOL ANGLE THETHA
THE POSSITE AND THE COSE OF THE CONFORTING LEGENMENTS THE LINE

}	TIME INGREMENT (CM)
	H2 HEIGHT OF STUCK MATERIAL COLUMN OUTSIDE POOL
}	H3 DISTANSE BETWEEN LAST TWO NOBES OF EACH COLUMN OF
	STUCK MATERIAL OUTSIDE POOL (CM)
}	I COUNTING VARIABLE USED IN CO-LOOPS
2	IFLAG FLAG SHOWING THAT NODAL HEIGHT HSD REACHED
}	
2	II COUNTING VARIABLE USED IN DC-LOOPS
<del>}</del>	III COUNTING VARIABLE USED IN DO-LOOPS
3	IREM FLAG SHOWING POCL MASS IS GONE WHEN = 0
<del>)</del>	IT INITIAL TEMPERATURE OF POOL POWDER (C)
C	12 JUMMY VARIABLE USED WITH THE PRINT SUERCUTINE
<del>C</del>	
C	J COUNTING VARIABLE USED IN DO-LOOPS
9	JI COUNTING VARIABLE USED IN DO-LOOPS
0	JJ COUNTING VARIABLE USED IN DO-LOOPS
<del></del>	- JJI COUNTING VARIABLE USED IN DO-LOOPS
C	JK DUMMY VARIABLE
<del>0</del>	JL =LAG
3	JTVAL DUMMY VARIABLE
6	K DJMMY VARIABLE
C	KK FLAG VARIABLE
<del>c</del>	KL JUMMY VARIABLE .
C	K4 DUMMY VARIABLE
<del>c</del>	LK COUNTING INTEGER USED WITH PRINT SUBROUTINE
С	LT DUMMY VARIABLE
<del>c</del>	M CROSS SECTIONAL OF MASS LEAVING POCL AND STUCK TO-HALL
C	MT MOLD TEMPERATURE (C)
<del>G</del> -	N - NUMBER OF NODES IN EACH STATIONARY POOL COLUMN
C	NC NUMBER OF POOL GOLUMNS OF NODES
<del>C</del>	NEC GOUNTING VARIABLE
C	NMAX MAXIMUM NODE HEIGHT BASED ON VOLUME OF POOL (CM)
<del>\$</del>	NNAX EQUALS N+1
C	NNTP DUMMY VARIABLE WHICH EQUALS N6+N4+1
<del></del>	N2-GOLUMN YUMBER OF STUCK MATERIAL
C	N3 NUMBER OF NODES IN STUCK COLUMN MINUS 2
<del>c</del>	NSTEAR DUMMY VARIABLE
С	N4 DJMMY VARIABLE
<del>c</del>	NO NUMBER OF NODES IN FENTIRATION THICKNESS
C	OT PORTION OF RE-ENTERING STUCK COLUMN NOT ENTERING POOL (RAD
<del>-</del> C	PS POSITION OF A PARTICULAR STUCK COLUMN IN MOLD (RAD)
Č	PT PRINTING INTERVAL (SEC)
- <del>Č</del>	RA RATIO OF RIVRO
Č	RAD INITIAL RADIUS OF MATERIAL (CM)
<del>-c</del> -	RB JUHHY VARIABLE
C	RI RADIUS FROM MOLD CENTER TO SURFACE OF STATIONARY POOL (CM)
-C	RM DUMMY VARIABLE
C	1 PO TAMES SAUTHS OF MOID ICM!
·- <del>C</del>	RO TAMES SABIOS OF MEED TOTAL SPEED OF MOLO (RPM)
C	R2 DJMMY VARIABLE
<del>-c</del> -	
C	SM COMPLETE MELT TEMPERATURE (BEG C)
	SR RATIO OF FREE-FALL TIME TO TIME INTERVAL TS
<del>-с</del>	SS TEMPERATURE THAT MATERIAL BEGINS STICKING TO THE MOLD (C)
-6	S1 DUMMY VARIABLE USED IN WEDGE AREA EQUATION  S2 DUMMY VARIABLE USED IN WEDGE AREA EQUATION
C	T NODAL TEMPERATURE
	T NODAL TEMPERATURE

	TH ANGLE (THETHA) FROM CENTIR OF MOLD TO JUNCTION OF
С	POOL SURFACE AND MOLD (RAD)
C	TH TIME
C	TN NEW COMPUTED NODAL TEMPERATURE (C)  TP COMBINED NODAL TEMPERATURE USED IN DETERMINING
C	TP COMBINED NODAL TEMPERATURE USED IN DETERMINING  A VERAGE TEM FERATURE (C)
<del>- c</del>	TS TIME INTERVAL USED (SEC)
C	TW CONTACT TIME OF MATERIAL TO WALL IN POOL REGION (SEC)
<del>- č</del>	W ROTATIONAL SPEED OF MOLD (RAD/SEC)
Č	X5 POSITION OF LEADING AREA OF STUCK MATERIAL (RAD)
<del>č</del>	YY CONSTANT IN TEMPERATURE EQUATION
C	
<del></del>	** SECTION I ****
C	
	EN-0.0
	IREM=1
	LK=8
	NEC=1
	7FLAG=0 PT=30.0
	TM=0.0
	NC=8
	RAD=0.0125
	RI=1.360 E
	₹0=6.6675
	RPM=10.0
	BA=53
	SS=110.0
	SM=138.0
£.	IT=20.0
	A=22.0
	B=-0.430
	YY=20.0 G=980.0
	9-700.0
C **	*** SECTION II *****
<del></del>	
	RA=RI/RO
	TH=(3.14159/2.0-AT44(RA/(-RA*RA+1.0)**0.5))*2
C	DETERMINE AREA IN STATIONARY POOL
	44-1.3.40.K0-4.14-214(14)
	AL=COND(0)/(CP(0)*RHO(0)) -AM=COND(RAD)/(CP(RAD)*RHO(RAD))
	W=0 • 10472*RPM
ki-i-i-i-i-i-i-i-i-i-i-i-i-i-i-i-i-i-i-	TW=TH/W TS=TW/NC
	ITM=0
	TTMY=1
	TM=TM+TS
С	DETERMINE MAXIMUM PENITRATION DISTANCE
	AD=7.2*(AL *TW) **0.5
C	DETERMINE MAXIMUM TIME INCREMENT
	DT=TW/NS
C	DETERMINE NODAL DISTANCE
	DX=(DT*2*AM)**0.5
	IF(AM.LT.AL) DX=(DT*2*AL)**0.5
C	DETERMINE NUMBER OF NODES PER COLUMN

```
-N=(INT(4D/DX)+1)+2---
       NNAX=N+1
       N2=NC+2
       CT=N2-1
       SET INITIAL TEMPERATURE AND NECK RADIUS
       DO 125 I=1,NC
       DO 126 J=1, NNAX
       TI=(U,I)T
       ANECK (I, J) =0.0
  125 CONTINUE
 -1-25 CONTINUE-
       BA=0.0174533*3A
       WRITE(6:15)
   10 FORMAT (1H1,294ROTATIONAL MOLDING SIMULATOR )
       FORMAT(5X, 26+POWDER HEIGHT RADIUS(CM)= ,F10.8,/,6x,17HMOLD RADIUS(
      15M)=-,F10.34/+6X421HINTIAL NO. OF NOEES=-,13./,-----
      26X,19HC)EF. A OF TEMP EQ=,F10.3,/,6X,20HCOEF. 3 OF TEMP EQ= ,F10.3
      3, /, 6x, 20 HNODE THICKNESS (CM)= ,FIO. 4,/, 6x, 38HSTICK TEMP(C) AND DIFF
      4USIVITY(CM2/S) = ,F10.3,3X,E10.3,/,6X,46HCOMPLETE MELT TEMP(G) AND
     5DIFFUSIVITY(242/S)= .F1(.3,3x,E10.3,/,6x,6HRP+= .F10.3)
C
0
    C
<del>-1000-FW=FH/W-------</del>
       CC=2*RO*SIN(T4/2)
       ANK=ANE3K(1.1)
       IF (CT.GT.N2) ANK=ANECK(CT.1)
       AM=COND(ANK)/(RHO(ANK)*CP(ANK))
       BE=RADX(ANK)
       RECHECK MAXIMUM TIME INCREMENT EACH ITERATION ---
       DT=(BE*0X) **2/(2.0*AM)
       IF (AM. LT. AL) - DT= (B= * DX) **2/(2.0*AL)
       TM=TM-TS
       IF ( (IRE4 . EG. U) . OR . (ITM. GT . 0) ) GOTO 109
       TS=TW/NC
 ITMX=1
       IF+TTL.GT.DT+ TTL=DT-
       IF(TTL.GT.TS1) TTL=TS1
    ----IF(TTL.EQ.TS) G3T0 109 ----
       ITMX=INT(TS/TTL)+1
       TS=ITL/IIXX
  109 IF (IREM. EQ.0) TS=OT
       OL=W* TS*RO
      IF (TM.EQ.0.0) WRITE (6,20)DT, TS, DL
   29 FORMAT(1H +/+5X+23+MAX TIME INCREMENTS(S)= +F10+5+/+6X+21 HTIME INC
      1REMENT USED= ,F10.5,/,6x,18HARC/TIME INC(CM)= ,F10.5 )
       TM=TM+TS ... - ...
       IF (IPEM. EQ. 0) GOTO 540
       <del>NMAX= INT (AA/(NC*DL*DX)) + 1</del>
       AD=7.2*(AL *TW) **0.5
       TF=(1+3333*GG/(G*GDS(BA)))**9+5-
       SR=TF/TS
       IF(SR.LT.1.0) SZ=1.0
C
       DETERMINE AREA OF EACH WEDGE
       51=00*SIN(TH/2.0)
```

```
52=R0*C35(TH/2.0)*TAN(W*TS/2.0)
      AT=0.5+(S1+52)+(S1-S2)+SIN(W+TS)+0.5+RC+RO+(W+TS-SIN(W+TS))
     AS= (N=0.5) *DX*DL---
      IF (AS.GT.AT) 4S=AT
      IF (N.LT. NMAX) GO TO 560
      N=NMA X
     IFLAG=1
  540 CONTINUE
     00 550 I=I,NJ
      T(I_{+}N+1)=T(I_{+}N-1)
     ANECK (I, N+1) = ANECK(I, N-1)
  550 CONTINUE
 -<del>560-CONTINUE</del>
     NNAX=N+1
     NE=INT(AD/DX)+1
C
   **** SECTION IV ***
      <del>- COMPUTE NODAL TEMPERATURES AND NECK RADIUS -</del>
 580 4T=A*TM/60.0+3*(TM/60.0) 4*2+YY
     DO 620 I=1,NC
     NODAL TEMPERATURE EQUATIONS FOR STATIONARY POOL
     1(1,1)=41
     TN(1) = 4T
     CALL CONSTICT.C2,C3,EB,DA,MT,MT,MT,ANECK(I,1),ANECK(I,2),
    1 BNECK(1))
     ANECK(I, 1) = BNECK(1)
     V, S = U 0 0 0 0 0
     GALL CONST (C1, C2, C3, CB, DA, T(I, J), T(I, J-1), T(I, J+1), ANECK (I, J),
    1 ANECK(I, J+1), BNECK(J))
     1DA/G2*T(I,J+1))
 63 CONTINUE
     DO 640 JJ=1.N
     TtI,JJ)=TN(JJ)
     IF ((JJ.EQ.NEC).AND. (TN(JJ).GT.SM)) ANEK(NEC) = BNECK(JJ)
     IF ((JJ.EQ.NEC).AND.(TN(JJ).GT.SM)) NEC=NEC+1
     IF(TN(JJ).GT.SM) BNECK(JJ)=RAD
   ANECKII, JJ)=BYESKIJJ)
 640 CONTINUE
 620 CONTINUE -
     IF(CT.LT.N2) SO TO 890
     NODAL TEMPERATURE EQUATIONS FOR STUCK MATERIAL
     00 880 I=N2,CT
     T(1,1)=41
     -CALL CONST (C1, C2, C3, DB, DA, MT, MT, MT, ANECK (I, 1), ANECK (I, 2),
   1 BNECK(1)
     ANEGK(I, 1) = 3NEGK(1)
     IF(N3(I) EG. 0) GO TO 820
     IF(N3(I).EQ.1) GO TO 780
     JTVAL=N3 (I)
     DO 770 J=2,JTVAL
     CALL CONSTICT, C2, C3, DB, DA, T(I, J), T(I, J-1), T(I, J+1), ANECK(I, J),
    1 ANECK(I,J+1),BNECK(J))
    1DA/C2 *T(I,J+1))
 770 CONTINUE
```

```
788 J=N3(I)+1
                    CALL CONST (C1, C2, C3, DB, DA, F(I, J), T(I, J-1), T(I, J+1), ANECK (I, J),
                  1 ANECK(I,J+1),BNECK(J))
                    BB=TS/C3*(DB/C1*T(I,J-1)+DA/(H3(I)*C2/DX)*T(I,J+1))
                    TN(J)=T(I,J)*(1-TS/G3*(DB/C1+DA/(H3(I)*C2/DX)))+9B
           823 CONTINUE
                    J=N3(I)+2
                    CALL CONST(C1,C2,C3,DB,DA,T(I,J),T(I,J-1),T(I,J ),ANECK(I,J),
                 1 ANECK(I,J ), BNECK(J))
                    BB=DB*TS/(H3(I)*(G3/DX)**2*(H3(I)~DX/2))
                    TN(J)=T(I,J)*(1-88)+88*T(I,J-1)-----
                    DO 878 JJ=1.J
          TO STORY IN THE PROPERTY CONTRACTOR AND THE PROPERTY CONTR
                    IF((JJ.EQ.NEC).AND.(TN(JJ).GT.SM)) ANEK(NEC)=BNECK(JJ)
                    IF ((JJ.EQ.NEC) .4NO. (TN(JJ).6T.SM)) NEC=NEC+1
                    IF (TN (JJ).GT.SM) BNECK(JJ) = RAD
                 375 CONTINUE
       -880 CONTINUE
         890 CONTINUE
                    IF((T(NC,N).3E.SS).AND.(IREM.EQ.1).AND.(N.EQ.NMAX)) GO TO 891
                  -IF (T(NC+N) -GE - SM) - 50 F0 891 ----
                    IF(TM.LT.LK*PT) GOTO 892
             * * * * *
     C
                            SECTION V ****
     4
                   PRINT ROUTINE
----
          891 I3=0
   I3=1
                   12 = 0
                   CALL PRINT (ANECK, 12, 13)
         ---- IF (I (NC, N) -GI - SML 301029
                   IF((T(NC,N).LT.SS).OR.(IREM.EQ.0)) GOTO 892
        ---- IF (N. LT. NMAX) GO TOB92--
                   WRITE (6, 30) TH
          -30 FORMAT(LH ,///,5X,34HMASS COMPLETELY GONE - FIME(SEC) = ,F19.3) ----
                   IREM=0
     ----AT=0.0--
                   AA = 0.0
                  TH=0.0
                  LK=LK-1
                 <del>-6010 892</del>
            29 CONTINUE
            -- WRITE (6, 31)
            31 FORMAT(1X,///,5X,29H NECK RADIUS AT COMPLETE MELT ,/,10X,
                1 4HNODE, 6X, SHRADIUS -----
                  00 33 J=1.N
                  WRITE (6, 32) J, ANEK(J)
            32 FORMAT(10X,15,5X,F10.6)
           37 CONTINUE
                  STOP
       -392 CONTINUE
                  ITM=ITM+1
                  IF ((IREM. EQ. 0) .OR . (ITM. L. T. ITMX) ) TM=TM+TS
```

```
IF ( TREY. EG. 0) . ) R. ( TH. LT. ITMX ) ) GO TO 1000
       ITM=0
       IF ( F(NC, 1) . ST. SS ) 50 FO 1200
           SECTION VI ****
C
0
       CALCULATE MIX MEAN TEMPEFATURE WHEN NO MATERIAL HAS STUCK
C
 1109 TP=0.0
       DO 1110 I-2.N
       TP=TP+T(NC.I)
 1110 CONTINUE
       TP=(TP+4T/2)/(N-0.5)
       <del>FT=(&S*TP+(&T-&S)*IF}/&T</del>---
       DO 1140 III=2, NC
      -I<del>=NC-III+2-----</del>
        DO 1142 J=1, NNAX
      T(F,J)=T(F-1,J)
       ANECK (I, J) = ANECK (I-1, J)
1142 CONTINUE --
 1140 CONTINUE
       DO 1150 I=1. NNAX
       T(1,I)=IT
       ANECK(1, I) =0.0
  1150 CONTINUE
       IT= ((SR-1) +IT+FT)/SR
       TM=TM+TS
       60 TO 530
  1200 CONTINUE
--<del>-----</del>
    **** SECTION VII
      DETERMINE THE AMOUNT OF STICK AND NEW ANGLE
C
       00 1220 I=2. N
       IF(T(NC,I).LE.SS) GO TO 1221
1226 CONTINUE -----
  1221 CONTINUE
     H=(RA+(I-2))*OX
      --00-1250-II=<del>2.N</del> -----
       IF(T(NC-1, II).LE.SS) GO TO 1251
1252 CONTINUE
  1251 CONTINUE
       RB=(SS-TINC, II -1))/(TINC, II)-TINC, II-1))
       H1 = (RB + (II - 2)) *DX
     . IF (INT (H/DX) -1 .LT.0) GO TO 1109
       CT=CT+1
      -<del>00(</del>N2<del>-</del>1)=X*T3---
       PS(N2-1) = TH/2.0
       W3(N2=1)=1NT(H/)X)-1-
       H3(N2-1) = H-DX*N3(N2-1)
       H2(N2=1)=H
       JTVAL=N3 (N2-1) +1
       00 1350 J=1, JT VAL
       T(N2-1,J)=T(N2+J)
       ANECK (NZ-1,J) = ANECK (NC, J)
```

102	CONTINUE	
	T(N2-1,N3(N2-1)+2)=SS	
`	ANECK (N2-1, N3 (N2-1)+2) = ANECK (NC, N3 (N2-1)+2)	
C <del>C</del>	COMPUTE MIX TEMPERATURE OF MATERIAL NOT STICKING	AH #
C	ENTERING FREE-FALL ZONE	
<u> </u>		and the second s
	TP=0.0	
	00 1380 J=1,N	ay amanana asama dasayi da'inadalahadi bashar bar abas ar amanana asami si si in ay ari
	TP=TP+T(NG,J)	
1367	CONTINUE	
	K=RA-0.5  IF (K.GE. 0.0)	
	IF (K.LT. 0. 0) TP=(TP-T(NC, I-1)*K)/(N-I-K+1)	
-	M=H+BL	الله المراجع والدينية المراجع المراجعة اليومية المراجعة ( عالم المراجعة المراجعة المراجعة ( ) 4 - المراجعة
	FT=((AS-M)*TP+(AT-AS)*IT)/(AT-M)	
<del>C e</del> (	TATE POWDER	destruction and the second section of the sect
	DO 1440 III=2.NC	
	I=NC-III+2	appropriate groups they are parties for the control of the control
	00 1445 J=1, NNAX T(I,J)=F(I-1,J)	
	ANECK(I, J) = ANECK(I-1, J)	
1445	GONTINUE	
1440	CONTINUE	<u></u> .
	<del>K4=1</del>	
_	IF(CT.LT.N2) 50 TO 1900	
<del></del>		·
<del>C</del>	*** SECTION VIII ****	and processing a company of the first state of the company of the
C	STUCK MATE SHIFT	
	DO 1478 JJI=N2, CT	and the state of t
	JI=CT-JJI+N2	
	DC (JI)=3C (JI-1)	
	PS(JI)=PS(JI-1)+W*TS	
	N3 (JI)=N3 (JI=1)	the handle state of the same and the same an
	H3(JI)=H3(JI-1) H2(JI)=H2(JI-1)	and the second section of the second section of the second section of the second section secti
	N3TEMP=N3(JI)+2	
	00 1530 J=1, V3TEMP	and the second s
	T(JI,J)=T(JI-1,J)	
	ANECK(JI, J)=ANECK(JI-1+J)	anner de de la company de la c
1536	CONTINUE	
	CONTINUE X5=6.28318-TH/2	<ul> <li>I aya yangangi sa gang telah dengangan sa Bay a semenah dengan sebagai dan berasa dan</li></ul>
	EN=0.0	
	KL=0	
	IF (PS (GT) *LT *X 5) · GD - TO 1900	and the second s
C		
_	*** SECTION IX ****	en المحتفظة وويدناه و المتعلقات المحتفظة المحتفظة المحتفظة المحتفظة المحتفظة المحتفظة المحتفظة المحتفظة المحتفظة
C 	STUCK MATERIAL RE-ENTERING PCCL ROUTINE	and the second seco
-	00 1570 JI=N2,CT	
45-5	IF (PS (JI ) · GE · X5) - 60 TO - 1575	والمراجعة
1570 1575	CONTINUE	
1777	THE THOU	

IF(RM.LE.0.0) GO TO 1840

IF(IFLAG.EG.1) GOTO 1870 IF(N6+N4.LE.N) GO TO 1870

DO 1862 J=NNAK, NNTP

ANECK(1,N4) = ANECK(JI,N3(JI)+2)

T (1,N4)= (RM+T(JI,N3(JI)+2)---+(DX-RM)+IT)/DX----

N4=N4+1

NNTP=N6+N4+1 NNAX=N+1 DO 1860 I=2,NC

-1<del>842--CONTINJE</del>--

```
74/825 0°T=0.ROUND= 4/ S/ M/-D,-DS FTN 5.1+601 134
MITINE PRINT
NG/-OT,APG=-COMMON/-FIXED,CS= USER/-FIXED,OB=-TB/-SB/-St/-ER/-ID/-PMB/-ST,PL-
           SUBROUTINE PRINT (T. LK.K)
           DIMENSION T(200, 100), N3(200)
           <del>COMMON SS, SN 34 L</del>, 4 M, D X, N <del>, TM , AA 3 A T , AD , TH , DL y N2 , CT , NC , N</del>3 <del>y</del> RAD <del>, TS -----</del>
           COMMON TAIR, BMT, MT
           REAL MT
           INTEGER CT
           TK=[K+1
           BB=TH*57.29578
           IF (K. EQ. 1) GOT023
           WRITE (6, 20) TM, AA, AT, AD, BB, DL, NC
        29 FORMAT(//, -11H F1ME(S{C)=,F10.3,/,6X,16HTOTAL AREA(C2)= ,F10.3,
          1/,6x,12HWEDGE AREA= ,F10.4,/,6x,28HORIGINAL PENITRATION DEPTH= ,
          2F10.4,7,6X,124ANGLE(DEG) = ,F10.3,7,6X,27HARC LENGTH/TIME INCREMENT
          3= ,F10.5,/,6X,24HNO. OF WEDGES IN POOL = ,I3,/)
          WRITE(6,10) TAIR, BMT, MT ---
        10 FORMAT(6X, 24HTEMP. OF AIR (DEG. C) = .F10.3./.6X_{*}
         1 44HTEMP. OF MOLD+S EXTERNAL SURFACE (DEG. C) = FID. 3,
          2 /,6x,44HTEMP. OF MOLD'S INTERIOR SURFACE (DEG. C) = ,F10.3,///)
           WRITE (6, 21)
        21 FORMAT(6X,30HTEMPERATURE HISTORY OF POWDER )
        23 IF (K. EQ. 1) WRITE (5,24)
        24 FORMAT(//, 12H NECK RADIUS
          ICOL=15
           IREP=((NC-1)/ICOL)+1
          <del>-IREM=NC-(IREP-1)*ICOL</del>---
           00 100 KK=1, IREP
          -IF (KK.E) . IREP) I COL = IREM ---
           KJ = (KK-1) * ICOL +1
           KJ1=KJ+(ICOL-1)
           IF (KK.EQ.IREP) KJ1=KJ+IR EM-1
          WRITE (6, IUB) (J, J=KJ, KJI)
      108 FORMAT(1X,6HAREA ,15(13,5X))
           00 104 J=1.N---
           IF(K.EQ.O) WRITE(6,102)J,(T(I,J),I=KJ,KJ1)
        IF (K. EQ. 1) WRITE (6, 103) J, (T(I, J), I=KJ, KJ1)
      102 FORMAT(1X, I3, 3X, 15(F7.3, 1X))
      103 FORMAT(1x, 13, 3x, 15(F7, 5, 1x))
      104 CONTINUE
       ----<del>\RITE(6,105)----</del>
      105 FORMAT(1X,//)
   -- 103 CONTINUE
       97 CONTINUE
          IF(CT.LT.N2) RETURY
          ICOL=15
         - IF(K.EQ.0) WRITE(6,98) ---
        98 FORMAT(1X,//, 31H TEMPERATURE OF STUCK MATERIAL 1/)
       IF (K. EQ. 1) WRITE (8, 99)
        99 FORMAT(//, 30H NECK RADIUS OF STUCK MATERIAL
                                                             )
          <del>-IREP=INF((CT-(N</del>2-<del>-))/ICOL)+1</del>
           IREM=CT-N2+1-(IREP-1)*ICOL
          <del>-00-200-KK=1.I</del>ZEP---
           KJ = (KK - L) * ICOL + N2
          IF(KK.EQ.IREP) ICOL=IREM
           KJ1=KJ+(ICCL-1)
```

CTION CP(X) CIFIC HEAT  2302.7  TURN  CTION RADX  MENSION N3(A  MON SS, SM, A  S FUNCTION  OF MOLD TE  (/RAD  (X.GT. 0.6)  (X=1.0  TO 1  (X=RHO(0.0)  TO 1  (X=RHO(0.0)  TINUE  TURN  )	(Y) 230) AL, AM, DX  DETERMINEMP  30TO 10  30TO 20 /RHO(Y)	,N,TM,	ΑΑ,ΑΤ,	AD, TH,				
CCIFIC HEAT  2302.7  TURN  CTION RADX  ENSION N3(3  MON 35.5M.  S FUNCTION  OF MOLD TE  (/RAD  (X.GT. 0.6)  (X=1.0  TO 1  (X.GT.1.6)  (X=RHO(0.0)  ITINUE  FURN	(Y) 230) AL, AM, DX  DETERMINEMP  30TO 10  30TO 20 /RHO(Y)	,N,TM,	ΑΑ,ΑΤ,	AD, TH,				
2302.7  FURN  ENSION N3(3  IMON 35.5M,  S FUNCTION  OF MOLD TE  (/RAD  (X.GT. 0.6)  (X.GT. 1.6)	(Y) 200) 4L, AM, DX  OETERMI EMP  30TO 10  30TO 20 /RHO(Y)	,N,TM,	ΑΑ,ΑΤ,	AD, TH,				
2302.7  FURN  ENSION N3(3  IMON 35.5M,  S FUNCTION  OF MOLD TE  (/RAD  (X.GT. 0.6)  (X.GT. 1.6)	(Y) 200) 4L, AM, DX  OETERMI EMP  30TO 10  30TO 20 /RHO(Y)	,N,TM,	ΑΑ,ΑΤ,	AD, TH,				
CTION RADX  ENSION N3()  ENSION N3()  S FUNCTION  OF MOLD TE  //RAD  X.GT. 0.6)  X=1.0  TO 1  X=RHO(0.0)  TO 1  X=RHO(0.0)  ITINUE  FUR N	200) AL, AM, DX  DETERMINEMP  3070 10  5070 20  / RHO (Y)	NES THE						
CTION RADX  ENSION N3()  ENSION N3()  S FUNCTION  OF MOLD TE  //RAD  X.GT. 0.6)  X=1.0  TO 1  X=RHO(0.0)  TO 1  X=RHO(0.0)  ITINUE  FUR N	200) AL, AM, DX  DETERMINEMP  3070 10  5070 20  / RHO (Y)	NES THE						
CTION RADXI SENSION N3() IMON SS, SM, I SENSION N3() IS FUNCTION OF MOLD TE  //RAD IX.GT. 0.6) IX=1.0 TO 1 IX.GT. 1.6) IX=RHO(0.0) ITO 1 IX=RHO(0.0) ITINUE FURN	200) AL, AM, DX  DETERMINEMP  3070 10  5070 20  / RHO (Y)	NES THE						
CTION RADX( IENSION N3() IMON SS, SM, I IS FUNCTION OF MOLD TO IX.GT. 0.6) IX=1.0 TO 1 IX.GT.1.G) IX=RHO(0.0) ITINUE FURN	200) AL, AM, DX  DETERMINEMP  3070 10  5070 20  / RHO (Y)	NES THE						
MENSION N3 ()  IMON 35, SM,    S FUNCTION  OF MOLD TE  (/RAD  (X.GT. 0.6)  (X=1.0  TO 1  (X-GT. 1.6)	200) AL, AM, DX  DETERMINEMP  3070 10  5070 20  / RHO (Y)	NES THE						
TMON SS, SM, (S FUNCTION OF MOLD TO A CONTROL OF MO	AL, AM, DX  DETERMINEMP  SOTO 10  SOTO 20  /RHO(Y)	NES THE						
S FUNCTION OF MOLD TO (/RAD (X.GT. 0.6) (X=1.0 TO 1 (X.GT.1.6) (X=RH0(0.0) (X=RH0(0.0) (X=RH0(0.0)	0ETERMI EMP 30TO 10 30TO 20 /RH0(Y)	NES THE						
OF MOLD TO  //RAD  (X.GT. 0.0)  (X.GT. 1.0)	30TO 10 30TO 20 4RHO(Y)		ERATI	O USEO	TO C	ALCULA	TE DX	AS A
OF MOLD TO  //RAD  (X.GT. 0.0)  (X.GT. 1.0)	30TO 10 30TO 20 4RHO(Y)		ERATI	O USEU	10 6	ACUUCA	TE UX	
(/RAD (X.GT.0.6) (X=1.0 TO 1 (X.GT.1.6) (X=RHO(0.0) TO 1 (X=RHO(0.0) (TINUE	30TO 10 30TO 20 7RHO(Y)							· ·
X.GT.0.6) X=1.0 TO 1 X.GT.1.6) X=RHO (0.0) TO 1 X=RHO (0.0) ITI NUE	50T0 20 <del>∕RH0(Y)</del>							· · ·
X.GT.0.6) X=1.0 TO 1 X.GT.1.6) X=RHO (0.0) TO 1 X=RHO (0.0) ITI NUE	50T0 20 <del>∕RH0(Y)</del>	-						
TO 1 X.GT.1.G) ( X.GT.1.G) ( X=RHO (8.0) TO 1 X=RHO (8.0) ITINUE	50T0 20 <del>∕RH0(Y)</del>	-						<del></del>
X.GT.1.G) ( )X=RHO (0.0) TO 1 )X=RHO (0.0) ITINUE FURN	<del>/RH0 (Y)</del>	-						
X=RHO (8.0) TO 1 X=RHO (8.0) ITI NUE TUR N	<del>/RH0 (Y)</del>	-						<u></u>
TO 1 )X=RHO(B.0) HTINUE FURN		•						
X=RHO (D.0) ITI NUE FUR N	<u> </u>	-						•
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S=5.22E06*E	XP(-19.5	583 + 7	402.5	/ <del>(T+27</del> 3	311			
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	=31.0-0.058 TURN  NCTION VISC  VISCOSITY  S=5.22E06*6 TURN	=31.0-0.058*(T-105) TURN  NCTION VIS(T)  VISCOSITY FUNCTION  S=5.22E06*EXP(-19.1	=31.0-0.058*(T-105) TURN  NCTION VIS(T)  VISCOSITY FUNCTION (FOI  S=5.22E06*EXP(-19.583 + 7) TURN  D	=31.0-0.058*(T-105) TURN  NCTION VIS(T)  VISCOSITY FUNCTION (FOISE) S=5.22E06*EXP(-19.583 + 7402.5; TURN  D	=31.0-0.058*(T-105) TURN  NCTION VIS(T)  VISCOSITY FUNCTION (FOISE)  S=5.22E06*EXP(-19.583 + 7402.5/(T+27) TURN  D	TURN  NCTION VIS(T)  VISCOSITY FUNCTION (FOISE)  S=5.22E06*EXP(-19.583 + 7402.5/(T+273))  TURN  D	=31.0-0.058*(T-105) TURN  NCTION VIS(T)  VISCOSITY FUNCTION (FOISE)  S=5.22E06*EXP(-19.583 + 7402.5/(T+273)) TURN  D	=31.0-0.058*(T~105) TURN  NCTION VIS(T)  VISCOSITY FUNCTION (FOISE)  S=5.22E06*EXP(-19.583 + 7402.5/(T+273)) TURN  U

## SAMPLE SIMULATION OUTPUT

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TIME (SEC) -
             30.121
     TOTAL AREA(C2)=
                        11.927
     WEDGE AREA=
                    2.0527
    ORIGINAL PENITRATION DEPTH-
                                     .4293
     ANGLE (DEG) =
                    86.005
    ARC LENGTH/TIME INCREMENT=
    NO. OF WEDGES IN POOL . 8
    TEMPERATURE HISTORY OF POWDER
AREA
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      32.667 32.667 32.667 32.667 32.667 32.667 32.667
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                      29.646 29.899 30.084 30.221 30.329 30.414
      26.494 27.153 27.527 27.841
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                                     28.076
                                             28.265 28.417 28.542
      26.494 26.405
                                     26.882 27.019 27.140 27.245
                      26.576 26.727
      26.494
              26.405 26.316 26.318
                                     26.341 26.382 26.427 26.475
      26.494 26.405 26.316 26.228
                                     26.171 26.131 26.107 26.092
                                     26.140 26.063 25.995 25.937
      26.494 26.405 26.316 26.228
      26.494 26.405 26.316 26.228
                                     26-140 26-052 25-968 25-889
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              26.405 26.316 26.228 26.140 26.052 25.964 25.878
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TOTAL AREA(CZ) =
                        11.927
     WEDGE AREA-
                    2-0527
     ORIGINAL PENITRATION DEPTH-
                                     .4293
     ANGLE (DEG) =
                    88.005
     ARC LENGTHATINE INCREMENT
                                  1.28014
     HO. OF WEDGES IN POOL - 8
     TEMPERATURE HISTORY OF POWDER
AREA
       1 ... .... 2
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       45.067 45.067 45.067 45.067 45.067 45.067 45.067 45.067
       40.381 41.072
                      41.550 41.847 42.066 42.228 42.356 42.457
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      37.871 38.642 39.080 39.449 39.726 39.950 40.131 40.280
      37.871 37.767 37.968 38.146 38.329 38.490 36.634 38.759
      37.871 37.767 37.663 37.666 37.692 37.741 37.795 37.851
 5
      37.871 . 37.767 . 37.663 . 37.559 . 37.492 . 37.445 ... 37.416 ... 37.398
      37.871 37.767 37.663 37.559 37.455 37.364 37.284 37.215
      37.871. 37.767. 37.663. 37.559. 37.455. 37.351. 37.252. 37.158.
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      37.871 37.767 37.663 37.559 37.455 37.351 37.247 37.145
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      37.871 ... 37.767 ... 37.663 ... 37.559 ... 37.455 ... 37.351 ... 37.247 ... 37.144
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      37.871 37.767 37.663 37.559 37.455 37.351 37.247 37.144
      37.871 .. 37.767 .. 37.663. .. 37.559. .. 37.455 ... 37.351 ... 37.247 ... 37.144 ...
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       37.871 37.767 37.663 37.559 37.455 37.351 37.247 37.144
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TIME (SEC) =

60.241

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TIME (SEC) -
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    TOTAL AREA(C2) .
                   11.927
    WEDGE AREA-
                2.0527
    ORIGINAL PENITRATION DEPTH=
                             . 4293
    ANGLE (DEG) =
                88.005
    ARC LENGTH/TIME INCREMENTS 1.28014
    NO. OF WEDGES IN POOL . 8
    TEMPERATURE HISTORY OF POWDER
AREA.
   57.095 57.095 57.095 57.095 57.095 57.095 57.095
 1
     52.369 53.067 53.549 53.849 54.070 54.234 54.368 54.466
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     49.839 50.617 51.058 51.431 51.711 51.937 52.120 52.271
     49.839 49.734 49.937 50.116 50.301 50.464 50.609 50.736
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     49.839 49.734 49.629 49.631 49.658 49.707 49.762 49.819
     49.839 ... 49.734 ... 49.629 ... 49.524 ... 49.456 ... 49.408 ... 49.379 ... 49.361 ....
     49.839 49.734 49.629 49.524 49.419 49.327 49.245 49.176
     49.839 49.734 49.629 49.524 49.419 49.313 49.213 49.118
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     49.839 49.734 49.629 49.524 49.419 49.313 49.208 49.105
     49.839 49.734 49.629 49.524 49.419 49.313 49.208 49.103
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     49.839 49.734 49.629 49.524 49.419 49.313 49.208 49.103
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     49.839 49.734 49.629 49.524 49.419 49.313 49.208 49.103
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T1ME (SEC) = 120.220
    TOTAL AREA(CZ)=
                      11.927
    WEDGE AREA-
                  2.0527
    ORIGINAL PENITRATION DEPTH-
                                  .4293
    ANGLE (DEG) =
                   88.005
    ARC LENGTH/TIME INCREMENT=
    NO. OF WEDGES IN POOL - 8
    TEMPERATURE HISTORY OF POWDER
AREA
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 1
      68.962 68.962 68.962 68.962 68.962 68.962 68.962 68.962
      64.314 65.000 65.475 65.770 65.987 66.149 66.276 66.377
      61.825 62.590 63.025 63.392 63.667 63.890 64.070 64.218
      61.825 . 61.722 . 61.921 . 62.098 . 62.280 . 62.440 . 62.583 . 62.708 .
 5
      61.825 61.722 61.618 61.621 61.647 61.695 61.749 61.806
      61.825 61.722 61.618 61.515 61.449 61.402 61.373 61.355
 7
      61.825 61.722 61.618 61.515 61.412 61.321 61.241 61.173
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      61.625 61.722 61.618 61.515 61.412 61.308 61.205 61.101
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      61.825 61.722 61.618 61.515 61.412 61.308 61.205 61.101
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TOTAL AREA(C2) =
                   11.927
    WEDGE AREA-
                2.0527
   ORIGINAL PENITRATION DEPTH=
   ANGLE (DEG) =
                88.005
   ARC LENGTH/TIME INCREMENT 1.28014.
   NO. OF WEDGES IN POOL . 8
   TEMPERATURE HISTORY OF POWDER
                            AREA .. 1 .. ..... 2 .... 3 .... 4
     80.461 80.461 80.461 80.461 80.461 80.461 80.461
     75.913 76.584 77.049 77.338 77.550 77.709 77.838 77.932
     73.477 74.226 74.651 75.010 75.280 75.496 75.674 75.819
 3
     73.477 73.376 73.571 73.744 73.922 74.079 74.219 74.341
     73.477 73.376 73.275 73.277 73.303 73.350 73.403 73.456
     73.477 73.376 73.275 73.173 73.108 73.062 73.034 73.017
     73.477 73.376 73.275 73.173 73.072 72.984 72.906 72.838
     73.477 73.376 73.275 73.173 73.072 72.971 72.874 72.782
     73.477 73.376 73.275 73.173 73.072 72.971 72.670 72.770
                             73.072 72.971 72.670 72.769
     .73.477 ..73.376 .. 73.275
.10...
                       73.173
11
     73.477 73.376 73.275 73.173 73.072 72.971 72.670 72.769
     73.477 ...73.376... 73.275... 73.173... -73.072.... 72.971 .....72.870 ....72.769 ...
     73.477 73.376 73.275 73.173 73.072 72.971 72.670 72.769
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TIME (SEC) = 150.079

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TIME (SEC) =
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         TOTAL AREA(C2) -
                                             11.927
         WEDGE AREA-
                                     2.0527
         ORIGINAL PENITRATION DEPTH-
                                                                    .4293
         ANGLE (DEG) .
                                     88.005
         ARC LENGTH/TIME INCREMENTS 1.28014
         NO. OF WEDGES IN POOL . B
         TEMPERATURE HISTORY OF POWDER
AREA
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            91.795 91.795 91.795 91.795 91.795 91.795 91.795
            87.351 88.007 88.461 88.743 88.951 89.106 89.227 89.324
            84.971 85.703 86.119 86.469 86.733 86.946 87.118 87.260
            84.971... 84.872. 85.063 85.232 85.406... 85.559... 85.696.. 65.816...
            84.971 84.872 84.773 84.776 84.801 84.847 84.899 84.953
            84.971 84.872 84.773 84.674 84.611 84.566 84.538 84.522
            84.971 84.872 84.773 84.674 84.576 84.489 84.413 84.347
            84.971 84.872 84.773 84.674 84.576 84.477 84.382 84.292
            84.971 84.872 84.773 84.674 84.576 84.477 84.378 84.280
            84.971 84.872 84.773 84.674 84.576 84.477 84.378 84.279
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TIME (SEC) =
           210.058
                      11.927
    TOTAL AREA(C2)=
    WEDGE AREA=
                  2.0527
    ORIGINAL PENITRATION DEPTH=
                                  .4293
    ANGLE (DEG) -
                  88.005
    ARC LENGTHITINE INCREMENTS
                             1.28014
    NO. OF WEDGES IN POOL = 8
    TEMPERATURE HISTORY OF POWDER
AREA
     1
              2 . .
                    . . 4
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     102.766 102.766 102.766 102.766 102.766 102.766 102.766
      98.427 99.067 99.511 99.766 99.989 100.141 100.259 100.354
      96.102 96.817 97.223 97.566 97.823 98.031 98.199 98.338
      96.102 96.006 96.192 96.357 96.527 96.677 96.811 96.928
      96.102 96.006
                    95.909
                           95.912
                                  95.936 95.981 96.032 96.085
     96.102 96.006 95.909
                          95.813 95.751 95.707 95.680 95.664
      96.102 96.006 95.909
                           95.813 95.716 95.632 95.557 95.493
      96.102 96.006 95.909
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                           95.813 95.716 95.620 95.523 95.426
 13
      96.102 96.006
                    95.909
                           95.813 95.716 95.620 95.523 95.426
NECK RADIUS
ARE A.
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T1ME (SEC) = 240.145
     TOTAL AREA(C2) = ...
                    11.225
     WEDGE AREA=
                  1.9385
     ORIGINAL PENITRATION DEPTH=
     ANGLE (DEG) ..
                  86.090
    ARC LENGTHATIME INCREMENTS
                             1.25229
    NO. OF WEDGES IN POOL . 8
    TEMPERATURE HISTORY OF POWDER
AREA
     1 2 ... 3 4
                                  113.555 113.555 113.555 113.555 113.555 113.555 113.555 113.555
     108-853 109-602 110-102 110-413 110-644 110-820 110-961 111-078
     106.417 107.213 107.697 108.083 108.382 108.629 108.835 109.013
     106.417 106.387 106.626 106.829 107.037 107.229 107.406 107.571
     106.417 106.387 106.344 106.374 106.433 106.517 106.611 106.714
     106.417 106.387 106.344 106.278 106.247 106.240 106.256 106.290
     106.417 106.387 106.344 106.278 106.214 106.166 106.134 106.120
     106.417 106.387 106.344 106.278 106.214 106.155 106.105 106.068
     106.417 106.387 106.344 106.278 106.214 106.155 106.101 106.057
     106.417 106.387 106.344 106.278 106.214 106.155 106.101 106.055
10
     106.417 106.387 106.344 106.278 106.214 106.155 106.101 106.055
11
12
     106-417 106-387 106-344 106-278 106-214 106-155 106-101 106-055
     106.417 106.387 106.344 106.278 106.214 106.155 106.101 106.055
TEMPERATURE OF STUCK MATERIAL
                          13
                                 14
                                              16
                                                     17
 1 ....113.555 .113.555 113.555 113.555 113.555 113.555 113.555 113.555 113.555 113.555 113.555
 2 110.824 111.503 112.037 112.434 112.797 113.061 113.239 113.349 113.412 113.449 113.474 113.495
MECK RADIUS
AREA
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      .00000 .00001 .00001 .00001 .00002 .00002 .00002
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      .00000 .00001 .00001 .00001
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NECK RADIUS OF STUCK MATERIAL
AREA
                   12
                          13
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                                .00004 .00005
                                             70000 .00005 .00006 .00006 .00006 .00007
                           22001
                                 70000 -00005 -00005 -00006 -00006 -00006 -00007
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T1ME (SEC) = 270.061
                                                 3.261
             TOTAL AREA(C2)=
             WEDGE AREA-
                                             .3601
             ORIGINAL PENITRATION DEPTH-
             ANGLE (DEG) = 46.973
             ARC LENGTH/TIME INCREMENT=
                                                                      .68328
             NO. OF MEDGES IN POOL = 8
             TEMPERATURE HISTORY OF POWDER
   AREA 1 2 3 4 5 6 7
              124.018 124.018 124.018 124.018 124.018 124.018 124.018 124.018
              122.028 122.041 122.063 122.063 122.076 122.092 122.111 122.107
              120.216 120.244 120.290 120.289 120.317 120.340 120.374 120.354
              118.702 118.741 118.799 118.788 118.797 118.799 118.813 118.758
              117.497 117.544 117.579 117.496 117.436 117.361 117.315 117.216
              116.587 116.426 116.274 116.052 115.886 115.730 115.633 115.525
              114-870 114-420 114-183 114-018 113-888 113-778 113-720 113-679
              111.016 111.359 111.526 111.685 111.708 111.760 111.813 111.874
     9 109,433 109.648 109.828 110.040 110.153 110.278 110.386 110.495
    10 109.287 109.302 109.334 109.406 109.462 109.538 109.614 109.696
             109.287 109.283 109.275 109.276 109.279 109.295 109.320 109.355
    12 109.267 109.263 109.274 109.264 109.253 109.245 109.243 109.249
13 109-287 109-283 109-274 109-264 109-251 109-239 109-231 109-226
    14 109.287 109.283 109.274 109.264 109.251 109.239 109.229 109.223
   15 109.287 109.283 109.274 109.264 109.251 109.239 109.229 109.222
   16 109.287 109.283 109.274 109.264 109.251 109.239 109.229 109.222
17 109.287 109.283 109.274 109.264 109.251 109.239 109.229 109.222
   TEMPERATURE OF STUCK MATERIAL
                                                                     14 15 16 17 18 19 20 21
              124.018 124.018 124.018 124.018 124.018 124.018 124.018 124.018 124.018 124.018 124.018 124.018 124.018 124.018
              122-109 122-089 122-052 122-059 122-037 122-028 122-020 122-020 121-988 121-987 121-976 121-968 121-969 121-950 121-961
              120.352 120.303 120.213 120.219 120.174 120.154 120.135 120.134 120.093 120.074 120.054 120.040 120.045 120.013 119.998
 4 118.725 118.595 118.434 118.432 118.368 118.338 118.308 116.304 118.261 118.238 118.217 118.204 118.217 118.183 118.170
              117.133 116.886 116.665 116.659 116.593 116.564 116.533 116.533 116.519 116.504 116.492 116.491 116.516 116.495 116.487
             113.416 115.073 114.864 114.853 114.815 114.808 114.798 114.819 114.873 114.887 114.894 114.916 114.968 114.973 114.990
  7 113.607 113.268 113.105 113.154 113.176 113.219 113.259 113.329 113.468 113.513 113.559 113.615 113.697 113.739 113.791 8 111.921 111.699 111.660 111.777 111.882 111.998 112.117 112.258 111.400 111.552 111.706 111.859 112.009 112.136 112.305
   9 110-197 110-453 110-715 110-945 111-153 111-378 111-612 111-852 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 
               121.939 | 121.929 | 121.927 | 121.914 | 121.908 | 121.900 | 121.802 | 121.821 | 121.823 | 121.823 | 121.823 | 121.823 | 121.823 | 121.823 | 121.823 | 121.823 | 121.823 | 121.823 | 121.823 | 121.823 | 121.823 | 121.823 | 121.823 | 121.823 | 121.823 | 121.823 | 121.823 | 121.823 | 121.823 | 121.823 | 121.823 | 121.823 | 121.823 | 121.823 | 121.823 | 121.823 | 121.823 | 121.823 | 121.823 | 121.823 | 121.823 | 121.823 | 121.823 | 121.823 | 121.823 | 121.823 | 121.823 | 121.823 | 121.823 | 121.823 | 121.823 | 121.823 | 121.823 | 121.823 | 121.823 | 121.823 | 121.823 | 121.823 | 121.823 | 121.823 | 121.823 | 121.823 | 121.823 | 121.823 | 121.823 | 121.823 | 121.823 | 121.823 | 121.823 | 121.823 | 121.823 | 121.823 | 121.823 | 121.823 | 121.823 | 121.823 | 121.823 | 121.823 | 121.823 | 121.823 | 121.823 | 121.823 | 121.823 | 121.823 | 121.823 | 121.823 | 121.823 | 121.823 | 121.823 | 121.823 | 121.823 | 121.823 | 121.823 | 121.823 | 121.823 | 121.823 | 121.823 | 121.823 | 121.823 | 121.823 | 121.823 | 121.823 | 121.823 | 121.823 | 121.823 | 121.823 | 121.823 | 121.823 | 121.823 | 121.823 | 121.823 | 121.823 | 121.823 | 121.823 | 121.823 | 121.823 | 121.823 | 121.823 | 121.823 | 121.823 | 121.823 | 121.823 | 121.823 | 121.823 | 121.823 | 121.823 | 121.823 | 121.823 | 121.823 | 121.823 | 121.823 | 121.823 | 121.823 | 121.823 | 121.823 | 121.823 | 121.823 | 121.823 | 121.823 | 121.823 | 121.823 | 121.823 | 121.823 | 121.823 | 121.823 | 121.823 | 121.823 | 121.823 | 121.823 | 121.823 | 121.823 | 121.823 | 121.823 | 121.823 | 121.823 | 121.823 | 121.823 | 121.823 | 121.823 | 121.823 | 121.823 | 121.823 | 121.823 | 121.823 | 121.823 | 121.823 | 121.823 | 121.823 | 121.823 | 121.823 | 121.823 | 121.823 | 121.823 | 121.823 | 121.823 | 121.823 | 121.823 | 121.823 | 121.823 | 121.823 | 121.823 | 121.823 | 121.823 | 121.823 | 121.823 | 121.823 | 121.823 | 121.823 | 121.823 | 121.823 | 121.823 | 121.823 | 121.823 | 121.823 | 121.823 | 121.823 | 121.823 | 121.823 | 121.823 | 121.823 | 121.823 | 121.823 | 121.823 | 121.823 | 121.82
               118.162 118.153 118.169 116.152 118.151 118.143 118.137 118.138 118.166 118.167 118.175 118.180 118.181 118.156 118.160
               116.472 116.478 116.511 116.509 116.525 116.532 116.540 116.556 116.604 116.620 116.645 116.666 116.685 116.685 116.726
              115.002 115.034 115.093 115.115 115.158 115.188 115.221 115.261 115.332 115.369 115.419 115.467 115.518 115.566 115.627
               113.842 113.906 113.996 114.050 114.105 114.166 114.229 114.299 114.393 114.456 114.534:114.615 114.708 114.828 114.909
      8 112.482 112.647 112.814 112.920 113.054 113.182 113.315 113.452 113.584 113.695 113.835 113.997 114.193 114.486 114.594
              40 41 42 43 44 45 46 47 48 49 50 51 124.018 124.018 124.018 124.018 124.018 124.018 124.018 124.018 124.018
     AREA 40
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2 121.860 121.873 121.884 121.898 121.908 121.925 121.933 121.946 121.959 121.973 121.989 122.017
3 119.901 119.925 119.938 119.944 119.967 120.002 120.020 120.046 120.074 120.104 120.138 120.194
4 118.218 118.258 118.269 118.275 118.313 118.366 118.396 118.483 118.483 118.531 118.584 118.668
5 116.787 116.852 116.862 116.893 116.950 117.023 117.067 117.126 117.190 117.259 117.336 117.491
6 115.713 115.807 115.808 115.807 115.946 116.039 116.098 116.174 116.259 116.352 116.453 116.582

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NECK RADIUS
                                                           AREA
      1
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     .00022 .00023 .00023 .00021 .00021 .00024 .00024 .00024
                 .00012 .00012 .00012 .00012 .00012
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MECK	RAD IUS_DE	STUCK	MATERIAL												
AREA	10	11	12	13	14	15	16	17	16	19	20	21	2.2		
1	00050.	00040	.00040	.00040	-00040	-00040							22 	23	24
2	- 00045	.00039	.00039	.00039	.00039	.00039	.00040							00042 00041	
3	.00037	-00037	.00037	.00038	. 00038	.00038									.00041 00040
4	.00025	.00025	.00025	.00025	.00025	.00026	.00026		.00026	.00027			.00027	.00027	.00028
5	00024_	-00024	00024	.00025	-00025	-00025	-00025						-00025	-00025	-00028
6	.00013	.00013	.00013	.00013	.00013	.00013	.00014		.00014				.00015	.00015	.00015
7	.00012	.00012	.00013	.00013	00013	-00013	00013					-00014	-00015	00015	-00015
8	.00002	.00002	.00002	.00002	.00002	•0000z	.00003	.00003	.00003				.00004	.00004	.00004
	00002	00002.	. 00002	00002		•0000 <b>2</b> .	00003.		0.00000	0.00000		-0.00000	0.00000	0.0000	0-00000
AREA	25	26	27	28	29	30	31	32	33	34	35	36	37	3.6	39
	-00042	00043						-00044	-00044	00044	-00044	-00045	-00045	-00045	00045
2	• 00042	.00042	.00042	.00042	.00043	.00043	.00043		.00044	.00044	.00044	.00044	.00044	. 00045	.00045
3	00040.	.00041	.00041	.00041	00041		00042			00042	00043	00043_	.00043		.00043
7 8	.00028 00016	.00028	.00028	.00028	.00029	.00029	.00029	.00030	.00031	.00031		.00033	.00036	.00042	.00042
	.00015	.00016	.00017	.00017	-00017	00017	00017	.00016	-00018			-00019	- 00019	00019	00019
7	-00015	-00016		.00016	-00016	-00017	.00017		.00017	.00017	.00018	.00018	.00018	.00019	.00019
A	.00004	.00005	.00005	.00005	.00005	-			-00006		00007		00007	-00007	80000
AREA	40	41	42	43		-00005 -45	46	• 00006 • 47	.00006	.00006		:.00007	- 00007	• 00007	.00007
1	.00046	-00046	.00046	00047	.00047	.00047	.00047	-00047	- 48		50				
ž	.00045		. 00046	.00046			00047	- 00047	.00048	•00048					
3	.00043	.00042	.00040	.00033	.00033	.00034	.00034	.00034	.00034	-00035	00048.				
4	.00042	-00041	-00039	.00032	00032		00033		.00033	-00034	.00035	•00035 •00033			
5	.00019	.000ZO	.00020	.00020	.00020	.00021	.00021	.00021	.00021	.00022	.00022	•00022			
6	.00019	.00019	.00019	.00020	.00020				.00021	.00021	00021				
· 7	.00008	.00008	.00008	.00008	.00009	.00009	.00009	.00009	.00010	.00010	.00010			The second second second	
. 8 .	.00008		0.00000	0.00000	0.00000					0.00000	0-00000	.0.0001			4
										~~~ ~ ~ ~ ~ ~ ~ ~ ~ ~ ~ ~ ~ ~ ~ ~ ~ ~					a

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TIME (SEC)= 276.381
    TOTAL AREA(C2)=
                 .0294
    WEDGE AREA.
    ORIGINAL PENITRATION DEPTH=
                             -2050
                20.060
    ANGLE (DEG) =
    ARC LENGTH/TIME INCREMENT
                            .29180_
    NO. OF WEDGES IN POOL - 8
    TEMPERATURE HISTORY OF POWDER
AREA
     1 . 2
                . 3
                              126.195 126.195 126.195 126.195 126.195 126.195 126.195 126.195
    123.981 123.981 123.993 123.993 123.004 123.004 123.004
    121.986 121.986 121.986 122.009 122.009 122.012 122.012 122.012
    120.209 120.209 120.209 120.245 120.245 120.247 120.247 120.246
    118.709 118.709 118.709 118.760 118.760 118.754 118.749 118.743
    117.403 117.403 117.402 117.545 117.537 117.406 117.464 117.440
    116,580 116,564 116,536 116,571 116,521 116,311 116,248 116,187
    115.811 115.637 115.483 115.365 115.234 114.785 114.703 114.630
    113.274 113.222 113.181 113.019 113.006 112.704 112.709 112.708
    110.264 110.480 110.666 110.765 110.884 110.845 110.939 111.012
10
    109.999 110.039 110.092 110.136 110.132 110.158 110.216 110.266
```

## TEMPERATURE OF STUCK MATERIAL

AREA	10	11	12	13	14	15	16	17	18	19	20	21	22	23	24	
 	126.195	126.195	126,195	126.195	126.195	126.195	126.195	126.195	126.195	124.195	126.195	126.195	126.195	126.195	126.195	
2	124.001	124.001	124.008	124.008	124.017	124.017	124.027	124.027	124.032	124.032	124.042	124.041	124.051	124.050	124.058	
3	122.025	122.025	122.039	122.039	122.056	122.055	122.076	122-074	122-064	122.081	122.101	122.097	122.116	122.111	122.126	
4	120.266	120.264	120.286	120.283	120.307	120.302	120.330	120.323	120.336	120.327	120.352	120.341	120.367	120.354	120.371	
5	118.767	118.758	118.782	118.769	118.793	118.777	116.801	118.782	118.789	118.768	118.783	118.760	118.768	118.744	118.755	
6	117.453	117.423	117.435	117.401	117.412	117.376	117.386	117.349	117.343	117.305	117.311	117.273	117.270	117.234	117.236	
 	116.169	116.109	116.093	116.036	116.025	115.970	115.959	115.910	115.891	115.845	115.438	115.795	115.780	115.743	115.736	
8	114.590	114.532	114.502	114.453	114,435	114.392	114.373	114.339	114.324	114.293	114.286	114.258	114.242	114.224	114.223	
9	112.710	112.721	112.727	112.739	112.755	112.761	112.755	-112.766	112.775	112.763	112.794	112.800	112.803	112.821	112.846	
10	111.101	111.193	111.259	111.331	111.388	111.437	111.431	111.483	111.515	111.560	111.597	111.641	111.683	111.740	111.815	
11	110.028	110.056	110.089	110.122	110.161	.110.210	110.363				.110.738	110.875	111.059	111.160	111.370	
AREA	25	26	27	28	29		31		33				37	38	39	
 _1											126.195					
2											124.068					
3											122.125					
4											120.328					
5											.118.647					
6											117.087					
 											115.615					
8											114.301					
9											113.233					
10											111.931					
11	111.485	0.000	0.000	0.000	0.000	0.000	0.000				0.000	0.000	0.000	Q000	0.000	
AREA	40	41	42	43	44		46					51	52	53	54	ı
 _1_											126.195					1 
2											123.995					7
3											121.952					
4	120.312	120.303	120.302	120.283	120.237	120.239	120.221	120.208	120.174	120.158	120.092	120.067	120.083	120.071	120.079	
5.	118.623	118.607	118.600	118.579	118.512	_118.516	118.498	118-484	118.548	118.434	118.361	118.344	118.369	118.360	118.375	
6	117.071	117.053	117.042	117.025	116.944	116.953	116.941	116.931	116.899	116.892	116.823	116.820	116.857	116.857	116.883	
 	115.638	115.625	115.617	115.611	115.528	115.547	115.547	115.546	115.531	115.537	115.486	115.504	115.557	115.571	115-611	
 		*** ***	114.402	114.412	114.349	114.382	114.396	114.413	114.427	114.450	114.434	114.478	114.551	114.580	114.636	
					-		112.562	113.582	113.641	113.681	113.713	113.789	113.884	113.928	114.000	

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 12.02   12.02   12.02   12.02   12.02   12.02   12.02   12.02   12.02   12.02   12.02   12.02   12.02   12.02   12.02   12.02   12.02   12.02   12.02   12.02   12.02   12.02   12.02   12.02   12.02   12.02   12.02   12.02   12.02   12.02   12.02   12.02   12.02   12.02   12.02   12.02   12.02   12.02   12.02   12.02   12.02   12.02   12.02   12.02   12.02   12.02   12.02   12.02   12.02   12.02   12.02   12.02   12.02   12.02   12.02   12.02   12.02   12.02   12.02   12.02   12.02   12.02   12.02   12.02   12.02   12.02   12.02   12.02   12.02   12.02   12.02   12.02   12.02   12.02   12.02   12.02   12.02   12.02   12.02   12.02   12.02   12.02   12.02   12.02   12.02   12.02   12.02   12.02   12.02   12.02   12.02   12.02   12.02   12.02   12.02   12.02   12.02   12.02   12.02   12.02   12.02   12.02   12.02   12.02   12.02   12.02   12.02   12.02   12.02   12.02   12.02   12.02   12.02   12.02   12.02   12.02   12.02   12.02   12.02   12.02   12.02   12.02   12.02   12.02   12.02   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116-186 116-218 116-251 115-632 115-355 116-399 115-428 115-476 115-573 115-632 115-727 115-763 114-757 114-848 115-141 115-289 115-609 119-486 100050 .00050 .00051 .00051 .00051 .00051 .00050 .00050 .00051 .00051 .00051 .00059 .00035 .00036 .00036 .00037 .00039 .00039 .00037 .00034 .00034 .00037 .00039 .00039 .00012 .00022 .00022 .00022 .00023 .00023 .00011 .00011 .00011 .00011 .00011 .00011 .00000 .00000 .00000 .00000 .00000 .00000	63 116.186 116.218 116.251 116.274 116.355 116.399 95 115.428 115.476 115.573 115.632 115.727 115.783 07 114.757 114.848 115.141 115.289 115.409 115.486 07 114.757 114.848 115.141 115.289 115.409 115.486 07 00050 .00050 .00051 .00051 .00051 .00051 080 .00050 .00050 .00050 .00051 .00051 080 .00048 .00050 .00050 .00051 .00053 090 .00048 .00034 .00034 .00035 .00035 090 2 .00052 .00052 .00052 .00052 090 2 .00052 .00052 .00052 .00052 090 11 .00011 .00011 .00011 .00011 00 .00000 .00000 .00000 .00000 .00001	63 116.186 116.218 116.251 116.274 116.355 116.399 95 115.428 115.476 115.573 115.632 115.727 115.783 07 114.757 114.848 115.141 115.289 115.409 115.486  50 .00050 .00050 .00051 .00051 .00051 .00051 50 .00050 .00058 .00050 .00059 .00053 50 .00035 .00048 .00048 .00048 .00048 .00048 35 .00035 .00022 .00022 .00022 .00023 .00035 22 .00022 .00022 .00022 .00022 .00023 .00035 11 .00011 .00011 .00011 .00011 .00011 .00011 11 .00010 .00000 .00000 .00000 .00001  10 .00000 .00000 .00000 .00000 .00001	63 116.186 116.218 116.251 116.274 116.355 116.399 95 115.428 115.476 115.573 115.632 115.727 115.783 07 114.757 114.848 115.141 115.289 115.409 115.486  20 00050 00050 00059 00051 00051 00051 50 00050 00058 00058 00059 00055 22 00022 00022 00022 00022 00023 24 00021 00011 00011 00011 00011 00011 11 00011 00001 00000 00000 00000 000001 00 00000 00000 00000 00000 00000 00001 00 00000 00000 00000 00000 00000	63 116.186 116.218 116.251 116.274 116.355 116.399 95 115.428 115.476 115.573 115.632 115.727 115.763 07 114.757 114.848 115.141 115.289 115.409 115.486 07 114.757 114.848 115.141 115.289 115.409 115.486 50 00050 00050 00051 00051 00051 00051 50 00050 00050 00050 00051 00051 00051 50 00050 00050 00050 00051 00052 00052 52 00052 00052 00052 00052 00052 00052 52 00052 00052 00052 00052 00052 11 00011 00011 00011 00011 00011 10 00010 00000 00000 00000 00000	4040141 414144	117.246 117	~	117,407		
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07 114.757 114.848 115.141 115.289 115.409 115.406  2	07 114.757 114.848 115.141 115.289 115.409 115.406  2	07 114.757 114.848 115.141 115.289 115.409 115.406  2	07 114,757 114,848 115,141 115,289 115,409 115,406  2 3 4 6 5 5 6 6 6 7 7 6 7 7 6 7 7 7 114,848 115,141 115,289 115,409 115,406 115,406  50 00050 00050 00051 00051 00051 00051  50 00050 00050 00050 00051 00051 00054  34 00034 00034 00034 00034 00035 00035  22 00022 00022 00022 00022 00022 00023  23 00011 00011 00011 00011 00011 00011  11 00011 00010 00000 00000 00001  00 00000 00000 00000 00001	07 114.757 114.848 115.141 115.289 115.409 115.406  2 3 4 6 5 5 6 6 6 6 7 7 6 6 7 7 6 7 7 114.848 115.141 115.289 115.409 115.406  50 .00050 .00050 .00051 .00051 .00051 .00051  50 .00050 .00050 .00050 .00050 .00053 .00053  35 .00034 .00034 .00034 .00034 .00035 .00035  22 .00022 .00022 .00022 .00022 .00022 .00023 .00023 .00023 .00023 .00023 .00023 .00023 .00023 .00023 .00023 .00023 .00023 .00023 .00023 .00023 .00023 .00023 .00023 .00023 .00023 .00023 .00023 .00023 .00023 .00023 .00023 .00023 .00023 .00023 .00023 .00023 .00023 .00003 .00000 .00000 .00000 .00000 .00000 .00000 .00000 .00000 .00000 .00000 .00000 .00000 .00000 .00000 .00000 .00000 .00000 .00000 .00000 .00000 .00000 .00000 .00000 .00000 .00000 .00000 .00000 .00000 .00000 .00000 .00000 .00000 .00000 .00000 .00000 .00000 .00000 .00000 .00000 .00000 .00000 .00000 .00000 .00000 .00000 .00000 .00000 .00000 .00000 .00000 .00000 .00000 .00000 .00000 .00000 .00000 .00000 .00000 .00000 .00000 .00000 .00000 .00000 .00000 .00000 .00000 .00000 .00000 .00000 .00000 .00000 .00000 .00000 .00000 .00000 .00000 .00000 .00000 .00000 .00000 .00000 .00000 .00000 .00000 .00000 .00000 .00000 .00000 .00000 .00000 .00000 .00000 .00000 .00000 .00000 .00000 .00000 .00000 .00000 .00000 .00000 .00000 .00000 .00000 .00000 .00000 .00000 .00000 .00000 .00000 .00000 .00000 .00000 .00000 .00000 .00000 .00000 .00000 .00000 .00000 .00000 .00000 .00000 .00000 .00000 .00000 .00000 .00000 .00000 .00000 .00000 .00000 .00000 .00000 .00000 .00000 .00000 .00000 .00000 .00000 .00000 .00000 .00000 .00000 .00000 .00000 .00000 .00000 .00000 .00000 .00000 .00000 .00000 .00000 .00000 .00000 .00000 .00000 .00000 .00000 .00000 .00000 .00000 .00000 .00000 .00000 .00000 .00000 .00000 .00000 .00000 .00000 .00000 .00000 .00000 .00000 .00000 .00000 .00000 .00000 .00000 .00000 .00000 .00000 .00000 .00000 .00000 .00000 .00000 .00000 .00000 .00000 .00000 .00000 .00000 .00000 .00000 .00000 .00000 .00000 .00000 .00000 .00000 .00000 .00000 .00000 .00000 .00000 .00000 .00000 .00000 .00000 .00000 .00000 .00000 .00	115.428 115.476	115.632 119		115.886		•
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NECK AREA	RADIUS E	F STUCK	MATERIAL 12	L . 13	14					• •						
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3	. 00049															
<u>.</u>	-00036					00036	-00037	-00031						-00037	-00037_	
. <b>5</b>	•00035					.00035	.00034	.00034	.00033	.00033	.00030	.00030	.00025			
7	.00023					-00023		00024	00024			00024				
i	-00011							.00023	.00023	.00023	.00023	.00023	-00024			
9	.00011					.00012	• 00012		.00012		00012				00013	
10.	. 00001							-00012	.00012			.00012	.00012	.00012	.00013	
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AREA	. 25	26	27	2.8	29	30	31	32		34	35	*0000S	• 0000Z	.00002	.00002	
1	• 00053	.00053					.00054			.00055	.00055	.00055			39	
2	-00052	-00052				00053	-00053	-00053	-00053			00054	- 00055	• 00055 • 00056	.00055	
4	•00050 •00037	.00051			.00051		.00051	- 00051	-00052	.00052	-00052	-00052	- 00052	. 00052	- 00052	
5	•00025	.00038	-00038		00038		•00038	- 00038	00039	.00039	-00039	-00040			00042	
6	-00024	•00025	.00026			45000	.00026	•00026	.00026	•00027	.00027	.00027	-00027	.00027	- 00027	TO SECULAR CO. LOS AND MINISTRALLINGS
7	.00024	.00024	.00023		.00014	.00014		+ DOOZ5	00025				00026	00026	-00026	-
	-00013	-	.00013	.00013				•00014	.00014	.00014	.00014	.00015	. 00015	-00015	- 0001 5	
9	.00013	.00013	.00013	.00013	.00013	-00013	+00014	•00013		00014		-00014		-00014	-00015	
10	-00002		00002				-00017	.00013	00003	.00014	• 00014	.00014	.00014	.00014	.00004	
11	.00002	0.00000	0.00000	0.00000	0.00000	0.00000	0-00000	0.00000	0.00000					00004.		
AREA	.40	41	42	43	.44	45	46	47	48	40	50	<b>61</b>	0.00000	-53	0.00000	
1	.00055	.00055	.00055	.00055	.00056	.00056	.00056	-00056	.00056	A 00056	-00057	.00057		.00057		
2		00054		.00055	00055	-00055	-00055	- 00055	00056	00056		.00056		-00056	.00058	
3	.00052	.00052	.00053	,00053	.00053	-00052	.00052	.00052	.00050	.00050	.00048	.00043	. 00043		.00044	
		00043	00047.		00051		00051	00051	00049	00049		.00041		00042.	-00042	
5	•00027	.00027	.00028	-00028	.00028	.00028	.00025	-00028	.00028	.00028	.00029	.00029	.00029	.000Z9	.00030	
7	.00015	.00027	.00027	.00027		▲00027				00028	.00028	00028	00029	.00029	-00029	
***** <b>B</b>	-00015			.00016	.00016	.00016	.00016		.00016	.00016	.00017	.00017	.00017	.00017	.00017	
9	.00004	.00004	.00004	.00005	.00005	-00005	- DODOR	*00005	-00016		-00614				-00017	·
10	400004	.00004					400005	-00005	.00005	-00005	- 00005	+00006	.00006	. 00006	.00006	
AREA	55	56	57	58	59	60	61	62	63	64	65	99	67			
1	-00058	00058	.00056		0005B	-00056	.00059		00059		00059			68 - 00060	69 - 00060	
2	.00057	.00057	.00057	-00057	.00058	.00058	.00058	-00058	.00058		.00058	.00059	. 00059	+00059	.00059	
3	.00044 .00042	.00044	00044	.00044	.00044	-00044		-00045	.00045	.00045	`` <b>```` D</b> 0045``	-00046	. 00046	. 00046	.00046	espec
5	.00030	.00030	.00042	.00043	.00043	.00043 .00030	+00043	.00043	.00043		.00041	.00037	.00037	.00033	.00033	
6	.00029	.00029	.00030	.00029	.00029	-00029	.00028	-00031 -00019		.00031		-00032	.00032	00032_		
7	.00018	.00017	.00018	.00018	.00018	00018			.00019	.00020	.00020	.00020	.00020	.00020	.00021	
8	.00017	.00017	.00017	.00018		.00018	.00018		.00018	.00019	.00019	-00019	.00019		-00020	
9	-00006	.00007		.00007	-00007	-00007	-00007	- 00000R	. DODDA	. 00000	00000	00000		• 00009	.00009	
10	•00006	.00006	0.00000	0.00000	0.00000	0.00000	44 00000	<b>V.</b> UU UU U	0.00000	0.00000	0.00000	0.00000	0.00000	0.00000	0-00000	
AREA .	70		. 12		or Programs	£2	/ Q			and acceptant of the same					-144400	
1	.00060	.00060	.00061	.00050		-00050	. 00050	.00050								· · · · · · · · · · · · · · · · · · ·
2	.00057	.00057	.00055	. 00049		00049		00050.							-	
3	00034		.00047	.00047		.00047	.00047									
	.00033	•00033	.00033	.00033	.00033	.00034	.00035	.00035 .00034								<u></u>
ś	.00021	.00033	.00021	.00021		100022		- 00022								5
ž	.00020	.00020	.00020	.00021		.00021	.00021	-00022		A CONTRACTOR OF STREET				···	Programme - physical -	
	.00009	.00009	.00010	.00010		.00010	.00010				,					
9	. 00009	.00009	.00009	.00010	.00010		.00010		The second State of the second	and the same of the same of				The first are always upon approximate		Charles and an appropriate little

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TIME (SEC.) - 300.008
      TOTAL AREA(CZ)-
                    0.0000
      WEDGE AREA-
     DRIGINAL PENITRATION DEPTH-
                                     -2050
     ANGLE (DEG) -
                     0.000
     ARC LENGTH/TIME INCREMENT=
                                  1.57144
     TEMPERATURE HISTORY OF POWDER
AREA
       1
               2
                       3
      134.228 134.228 134.228 134.228 134.228 134.228 134.228 134.228
      132.091 132.091 132.091 132.092 132.091 132.079 132.079 132.079
      130-056 130-057 130-057 130-058 130-057 130-033 130-033 130-032
      128.205 128.206 128.206 128.208 128.207 128.172 128.171 128.170
      126.586 126.587 126.588 126.590 126.588 126.542 126.541 126.539
      125.179 125.100 125.101 125.103 125.101 125.125 125.125 125.122
     124.021 124.022 124.023 124.025 124.023 123.958 123.958 123.955
     123.099 123.100 123.101 123.103 123.101 123.029 123.028 123.025
     122-438 122-439 122-440 122-442 122-440 122-363 122-362 122-359
 10
     122.033 122.034 122.035 122.038 122.035 121.955 121.954 121.951
11
     121.898 121.899 121.900 121.902 121.900 121.619 121.818 121.815
TEMPERATURE OF STUCK MATERIAL
AREA
                                                   <u> 16 17 18 19 20 21 </u>
                     12...
                           13 14 15
                                                                                                  22
     134.228 134.228 134.228 134.228 134.228 134.228 134.228 134.228 134.228 134.228 134.228 134.228 134.228 134.228
     131.957_131.955_131.959_131.959_131.964_131.969_132.008_132.008_137.024_132.028_132.037_132.047_132.065_132.068_132.086
     129.786 129.780 129.788 129.788 129.798 129.810 129.888 129.888 129.921 129.926 129.946 129.966 130.002 130.009 130.044
 4_ 127.790, 127.782, 127.793, 127.794, 127.809, 127.826, 127.947, 127.948, 127.998, 128.009, 128.036, 128.066, 128.122, 128.132, 128.186,
     126.018 126.007 126.021 126.022 126.042 126.066 126.228 126.229 126.295 126.311 126.339 126.380 126.440 126.454 126.528
     124.445 124.430 124.449 124.450 124.474 124.506 124.719 124.720 124.806 124.826 124.864 124.917 126.998 125.016 125.111....
     123.110 123.091 123.113 123.115 123.144 123.183 123.451 123.453 123.560 123.585 123.633 123.699 123.802 123.624 123.940
     121.994 121.971 121.997 121.999 122.033 122.082 122.410 122.412 122.542 122.573 122.632 122.712 122.839 122.866 123.005
     121.126 121.097 121.127 121.129 121.169 121.228 121.622 121.624 121.780 121.817 121.886 121.982 122.134 122.165 122.329
     120.496 120.461 120.495 120.497 120.543 120.614 121.080 121.082 121.265 121.309 121.390 121.502 121.681 121.718 121.908
10
     119.755 119.696 119.745 119.748 119.814 119.930 120.640 120.643 120.900 120.961 121.070 121.223 121.459 121.506 121.744
11
AREA
                     27
                             28
                                    29 30 31 32 33 34 35 36 37 38 39
     134.228 134.228 134.228 134.228 134.228 134.228 134.228 134.228 134.228 134.228 134.228 134.228 134.228 134.228
     132.090.132.095 132.103.132.106.132.111.132.115.132.120.132.121.132.120.132.135.132.135.132.137.132.141.132.148.132.140.132.160...
     130.054 130.064 130.080 130.085 130.096 130.104 130.114 130.116 130.129 130.144 130.148 130.154 130.169 130.169 130.194
      128.201 128.217 128.240 128.248 128.265 128.277 128.292 128.296 128.314 128.338 128.344 128.355 128.379 128.379 128.420 ...
      126.548 126.570 126.602 126.613 126.634 126.651 126.672 126.676 126.701 126.732 126.740 126.756 126.787 126.787 126.842
     125.136 125.165 125.206 125.219 125.246 125.207 125.209 125.209 125.331 125.369 125.379 125.398 125.438 125.438 125.506
     123.972 124.007 124.053 124.070 124.085 124.111 124.143 124.149 124.188 124.234 124.247 124.269 124.318 124.317 124.400
      <u> 123.043 123.086 123.141 123.162 123.181 123.213 123.250 123.258 123.303 123.357 123.372 123.308 123.455 123.656 123.552</u>
      122.374 122.425 122.491 122.515 122.539 122.577 122.619 122.627 122.681 122.742 122.759 122.788 122.854 122.853 122.955
     121.960 121.620 121.723 121.761 121.798 121.857.121.916: 121.928.122.007.122.093 122.119 122.156 122.269 122.265 122.396....
10
     121.810 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000
11
                      42
                          43 44 45 46 47 48 49 50 51 52 53 54
AREA.
     134.228 134.228 134.228 134.228 134.228 134.228 134.228 134.228 134.228 134.228 134.228 134.228 134.228 134.228
      <u> 132.162 132.166 132.173 132.174 132.189 132.193 132.194 132.201 132.221 132.224 132.243 132.257 132.269 132.272 132.280</u>
     130.197 130.205 130.218 130.221 130.250 130.257 130.259 130.272 130.304 130.311 130.340 130.354 130.378 130.385 130.400
                                                                                                                             Α.
     126.424 128.441 128.471 128.476 128.531 128.542 128.544 128.563 ND28.610 128.622 128.662 128.676 128.712 128.722 128.746 1
     126.847 126.869 126.906 126.913 126.985 127.001 127.004 127.030 127.099 127.114 127.177 127.207 127.264 127.268 127.299
     125.513 125.540 125.584 125.594 125.685 125.705 125.708 125.708 125.835 125.835 125.834 125.939 125.985 126.044 126.062 126.100
     124.408 124.441 124.492 124.504 124.617 124.641 124.645 124.688 124.805 124.828 124.938 125.001 125.072 125.093 125.138
     <u>123.561 123.600 123.659 123.673 123.810 123.836 123.843 123.895 124.038 124.065 124.202 124.281 124.364 124.389 124.442</u>
     122.966 123.011 123.079 123.095 123.259 123.291 123.296 123.357 123.529 123.560 123.726 123.623 123.916 123.947 124.006
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70 102-838 122-838 122-884 122-884 122-970 123-211 123-251 123-484-123-614-123-729-123-764-123-834

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AREA
    55
            56
                   57
                          56
                                 59
                                                     62
                                                            63
                                                                                66
                                                                                       67
                                                                                                     69
    134.228 134.228 134.228 134.228 134.228 134.228 134.228 134.228 134.228 134.228 134.228 134.228 134.228
     132.285 132.286 132.295 132.303 132.313 132.316 132.327 132.336 132.339 132.344 132.352 132.364 132.368 132.382 132.389
    130.410 130.417 130.431 130.446 130.467 130.474 130.495 130.514 130.519 130.530 130.546 130.571 130.578 130.607 130.622
     128.761 128.770 128.792 128.815 128.846 128.856 128.888 128.914 128.921 128.937 128.958 128.984 128.995 129.028 129.052
    127.319 127.332 127.362 127.393 127.434 127.448 127.491 127.525 127.536 127.557 127.586 127.625 127.639 127.688 127.720
    126.125 126.141 126.180 126.219 126.269 126.286 126.338 126.364 126.377 126.404 126.442 126.493 126.511 126.575 126.617
    125.168 125.188 125.236 125.263 125.363 125.364 125.428 125.461 125.477 125.510 125.555 125.619 125.641 125.721 125.772
    124.475 124.499 124.558 124.613 124.683 124.707 124.782 124.824 124.842 124.880 124.934 125.010 125.036 125.121 125.180
    124.043 124.070 123.727 123.809 123.911 123.947 124.061 124.125 124.151 124.205 124.284 124.397 124.434 124.568 124.652
    123.877 123.910 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000
10
                                AREA
            71
                   72
                          73
 1
    134,228 134,228 134,228 134,228 134,228 134,228 134,228 134,228
    132.392 132.396 132.398 132.422 132.445 132.456 132.464 132.476
    130.634 130.641 130.653 130.723 130.770 130.792 130.807 130.832
    129-072 129-083 129-105 129-222 129-293 129-327 129-350 129-388
    127.750 127.763 127.795 127.962 128.060 128.104 128.135 128.185
    126.654 126.671 126.714 126.933 127.058 127.118 127.151 127.214
    125.817 125.838 125.891 126.164 126.318 126.384 126.430 126.505
    125.234 125.258 125.322 125.653 125.838 125.913 125.967 126.054
    124.727 124.759 124.852 125.333 125.584 125.677 125.745 125.853
NECK RADIUS
AREA
      1
                                               7
 1
     .00092
           ..00092
                  .00092 .00093
                                .00093 .00093 .00093
     .00091
            .00091 .00091 .00091 .00091 .00091 .00091
     .00087 .00087 .00087 .00087 .00087 .00087 .00087 .00087
      .00072 .00072 .00072 .00073
                                .00073 .00073 .00073
     .00057 .00057 .00057
                         .00057
                                -00057 -00057 -00057 -00057
     .00055 .00055 .00055 .00056 .00056 .00056 .00056 .00056
     .00043 .00043 .00043 .00044 .00044 .00044 .00044 .00044
     <u> -00042 -00042 -00042 -00043 -00043 -00043 -00043 -00043</u>
     .00031 .00032 .00032 .00032 .00032 .00032 .00032
10
     400031 400031 400031 400031 400032 400032 400032
11
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AREA	10	11	12	13	14	15	16	17	18	10	20	44	2.2	4.6	0.1	
1	.00093					-00094	-00094	00094	- 00094	19	20	21	22	23	24	
2	.00091	.00091			.00092	.00092		,00092				.0000		.00094		
<b>' 3</b>	.00087	.00087						00088	100072	000072					.00093	
4	.00073						.00074							.00089		
5	.00070	.00070							.00074	.00074	.00074					
6	.00057							- 00069			00066	00066	00061	.00061	.00061_	
7	- 00055	. 00055									.00058		.00059	.00059	.00059	
8	.00043				.00044				.00056		00057	00057	*00057	400057	-00057_	
9	_ QQQ43		.00043			E 2000.	.00044			.00044	.00045				.00045	
10	.00031	.00031			.00032	EFUUV.	20022		.00044					.00044		
11	.00031	.00031				00032	• 00032	.00032	.00032	.00033	.00033	.00033	• 00033	.00033	.00033	
AREA	25	26	27			* 00021	. 00032		00032	00032	00032	00033		.00033	00033	
1	.00095	_		28	29	30	3 I	32	33	34	35	36	37	38	30	
ž	.00093	.00093			.00095						.00096	.00096	49000	.00096	49000	
3		-00090					-00094	.00094	.00094	LOODER	.00095	.00095	.00095		.00095	
K	• 00075	.00075			0009.0	00090	00090	.00090	00090	-00091	.00091	.00091	.00091	.00091	-00091	
5	.00061	•00075			.00075			•00076	.00076	.00077	.00077	•00077	.00078	.0007A	400079	
6	.00059				.00062			. 00062	00062	.00062		00063	£8000 a	£4000	. 00063	
7	.00057	.00059			.00060					.00060	.00060	.00061				
,		.00057	00055		.00047			00048		-00048						
•	.00045	.00045	.00046					.00046	.00046	-00047	-00047	-00047	. 00 04 7	00047	00040	
9		00045			-00045	00045_	-00046	-00046	-00046	-00046	.00046	-00046	- 00046	-00046	-00036	
10	.00034	.00034	.00034	.00034	-00034	-00024	- 77774	. 00034	00038	0000E	AAA A = E	~~~~				
11	.00033	0.00000	0.00000	0.00000	0.00000	0.0000	0-00000	0.00000	0.00000	0.00000	0.00000	0.00000	0.00000	.00000	0.00000	
AREA			• ••	1.5	77	72	70	7/			50	K 3	83		<b>*</b> /	
1 .		.00097	.00097	. 00097	.00097	.00097	-00097	-00098	-00098	-000ga	-0000	. 00000	- 00000	_ • 00099_	00000	
2	.00095	.00095	.00096	.00096	.00096	.00096	-00096	.00096	.00096	.00096	-00097			•00097	• DODAA	
3	00091	.00091	.00092	00092	.00091	.00091	-00091	-00091	-00089	-00089	.00087	.00097	00097	-00047	• 000048	
4	.00079	.00081	.00085	.00085	.00089	.00088	-00088	.00088	.00086	.00086	.00084	00000				
5	.00063	.00063	. 00 0 6 4		.00064	-00064	-00064	44000	44000	44000	00004	000079	00079	.00079 .00065_	.00080	
6	.00061	.00061	.00062		-00062	-00062	- 00062	-00062		. 00042	00063			.00063	00000	
7	. 00049	. 00049	. 00049	. 00049	-00049	-00050	-00050	- 00050	.00003	-00005	00000	.00003	.00003	.00051	.00064	
8	.00048	.00048	.00048	.00048	.00048	.00049	.00049	.00049	.00049	. 00.000		~ OCO5.C		~ 00031_	00051	
9	.00036	-00037	-00037		.00037				- 00044	000049	00039	.00050	• 00090	.00050	.00050	
10	.00036	.00036	.00036	. 00036	.00037	.00037	-00037	-00037	- 00035		00030	00038	<u> </u>	.00035	-00039	
AREA	55	56	57	58	59	60	61	62	63	44	4.5	.00038	• 00038	.00038	.00039	
1	.00099	.00099	.00099	.00100		.00100				00101	65			.00102	_69	
2	. 00098	00098	.00098	.00098	.00098	.00098	-00099	-00099	00000	-00101	.00101	.00101	* 00101	00102	.00102	
3	.00083	.00083	.000B3	.00083												
4	.00080	-00080	.00080	.00080	.00080	00000	.00080	00000	-00080	-00004	- 00004	00000	00000	.00085 .00071	.00085	
5	.00066	.00066	.00066	• 00000	• 00000 1	•0000	.00007	• DO D 6 7	-00067	-00067	_ 00048	- 00044	. 00049	00040	20012	
6	.00064	-D0064	.00065	.00064	.00064	.00064	.00063	.00054	.00054	-00055	- 00055	.00055	.00068	.00068 .00056	• 00069	
7		.00052	.00052	• 000052	• 00072	•UUUJZ	• 00005	.00053	<b>-00023</b>	-00053	-00053	- 00084		AAAE.		
8			.00051		.00051	.00051	.00052	.00052	.00052	-00052	- 00053	00054	* UU U J *	00054	.00054 .00043	
9	.00039	.00039	.00039													
10	.00039	.00039	0.00000	0.00000	0.00000	0.00000	0.00000	0.00000	0.00000	0-00000	200041	******	• 000+2	.00042 Q.00000_1	•00042	
AREA	70	71	72	73	74	75	76	77		MENNAMA.	WALMANDE	MUNICULAR	Nº NO NO O	n* 00000	00000	
1	.00102		.00103	.00092		D0092										
2	.00098	.00098	.00096	.00090	.00090		.00090									
3	.00086	.00086	48000	.00086												
4	.00071	.00071	.00072		.0007Z						*****					
ĸ	.00069	00069	00069				.00072							•		
2				00070_	00070_		00070_									
5	.00056	.00056	.00056	.00057	.00057	.00057	.00057									
.7	.00055	00055	. 00055	.00055	00056	00056	. 00056	00056								
		000/0	00043	00044	00011	4 4 4 4 4	BAAAR									
£	- 00043	.00043	.00043	.00044	.00044	.00044	.00045	.00045								

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TIME (SEC) = 330.036
    TOTAL AREA(CZ)-
                   0.000
    WEDGE AREA-
               0.0000
    ORIGINAL PENITRATION DEPTH-
                          . 2050
    ANGLE (DEG) =
                0.000
    ARC LENGTH/TIME INCREMENT=
                          .42824
    NO. OF WEDGES IN POOL - 8
    TEMPERATURE HISTORY OF POWDER
AREA
           2
                3
                     . . 4.
                             5
                              6 7 8
    144.199 144.199 144.199 144.199 144.199 144.199 144.199
. . 2
    143.243.143.243 143.243 143.243 143.286 143.286 153.286.
 3
    142.371 142.371 142.371 142.372 142.372 142.315 142.315 142.315
    141.374 141.375 141.375 141.375 141.375 141.514 141.514 141.514
    140-641 140-641 140-641 140-642 140-642 140-491 140-491 140-491
    139.714 139.714 139.716 139.715 139.824 139.824 130.824
    139.093 139.093 139.093 139.095 139.095 138.994 138.993 138.993
    136.965 136.965 136.965 136.968 136.968 136.890 136.890 136.889
    135.594 135.594 135.594 135.596 135.597 135.536 135.536 135.535
    134.833 134.834 134.834 134.838 134.838 134.793 134.792
10
    134.591 134.591 134.591 134.596 134.595 134.553 134.553 134.552
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## TEMPERATURE OF STUCK MATERIAL

AREA				13	14	15	16	17	18	19	20	21	22	23	24
															144.199
4											143.219				
3											142.289				
2											141.410				
3.											_140.635				
9											139.931				
											137.780				
ā											136.009				
9 10											134.704				
	1320707	121 217	1324410	136.416	132.731	132.070	133.380	133.302	133.074	133.737	133.856	134.029	134.644	134.377	134.040
AREA	25	26	27		29		31				35				134.360
AREA											144.199		37	38	39
2	143.281	142.244	142.274	142.222	142.225	142 220	142 740	143 270	142 128	143 340	143.270	144 444	143 348	744+TAA-	147+144
<b>a</b>											142.314				
Ţ.,											141.515				
5											140.660				
6	139.800	139.737	139.866	139.812	139.618	139.844	139.929	139.935	130.012	1 39. 995	140.002	140.015	140.007	140-002	140.048
7	138-965	139-122	139.124	139.201	139-208	189-241	139.239	139.246	139-318	139.363	139.351	139-366	130.431	139.425	139.506
8											137.515				
9											136.231				
10											134.984				
11_	134.461	0.000	0.000	0.000	0.000	0.000	0.000	0.000	0.000	0.000	0.000	0.000	0.000	0.000	0.000
AREA	40	41						47					52	53	54
1	144.199	144.199	144.199	144.199	144.199	144.199	144,199	144.199	144,199	144.199	144.199	144.199	164.199	144.199	144.199
2	143.256	143.281	143.270	143,271	143.305	143.308	143,308	143.304	143.324	143.298	143.325	143.330	143.328	143.331	143.339
3	142.385	142.359	142.406	142.408	142.423	142.429	142.429	142.463.	142.426	142.471	142.458	142.472	142.509	142.502	142.502
4	141.507	141.569	141.552	141.556	141.652	141.659	141.659	141.585	141.675	141.603	141.698	141.718	141.695	141.718	141.725
5	140.791	140.774	140.844	140.849	140.909	140.919	140.773	140.903	140.838	140.929	140.944	140.932	141.030	140.988	141.021
6	140.072	140.138	140.154	140.159	140.009	140.023	140.170	140.025	140.240	140.170	140.339	140.342	140-288	140.357	140.353
7	139.510	_139.562	139.606	139.613	139.462	139.478.	139.293	139.498	139.557	139.649	139.540	139.622	139-801	130.744	139.777
8	137.766	137.841	137.928	137.939	138.677	138.698	138.827	136.848	139.113	139.127	139.135	139.224	139.234	139.210	139.335
0	136.525	136.617	136.730	.136.745.	137.137	.137.167	137.219	137.376	137.764	137.843	.138,560	138.785	138.731	138-916	138-956
				1 7 7 7 1 2	136.187	136.234	136.244	136.409	136.920	137.020	137.618	138.025	138.643	138.665	138.886

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65 66 67 68 69
          144.199 144.199 144.199 144.199 144.199 144.199 144.199 144.199 144.199 144.199 144.199 144.199 144.199 144.199
          143.327 143.340 143.368 143.379 143.383 143.385 143.394 143.396 143.390 143.398 143.394 143.405 143.406 143.420 143.426
          142.526 142.511 142.583 142.586 142.612 142.616 142.618 142.621 142.624 142.615 142.640 142.640 142.640 142.647 142.693
          141.698 141.744 141.852 141.882 141.898 141.885 141.901 141.910 141.894 141.919 141.915 141.951 141.951 141.962 142.005 142.027
          141.062 141.011 141.223 141.236 141.244 141.251 141.249 141.254 141.263 141.258 141.307 141.327 141.334 141.417 141.453
          140.304 140.417 140.659 140.702 140.655 140.615 140.650 140.680 140.667 140.712 140.731 140.802 140.825 140.920 140.962
          139.876 139.784 140.212 140.085 140.078 140.097 140.145 140.178 140.200 140.226 140.299 140.368 140.391 140.527 140.583
          139.255 139.459 139.529 139.568 139.602 139.613 139.730 139.797 139.815 139.879 139.947 140.055 140.091 140.246 140.311
          139.119 139.051 138.317 138.566 138.881 138.981 139.220 139.332 139.381 139.470 139.579 139.728 139.776 139.981 140.064
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          138.851 139.055 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000
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          143.438 143.441 143.449 143.520 143.555 143.563 143.573 143.584
          142.703 142.709 142.740 142.878 142.943 142.969 142.982 143.007
          142.064 142.072 142.100 142.311 142.413 142.438 142.467 142.499
          141.482 141.492 141.553 141.828 141.953 142.000 142.028 142.074
          141.022 141.035 141.090 141.434 141.594 141.636 141.679 141.730
          140.638 140.654 140.738 141.137 141.315 141.376 141.417 141.480...
          140.386 140.404 140.489 140.941 141.144 141.202 141.254 141.320
          140.153 140.175 140.284 140.818 141.049 141.119 141.174 141.251
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61 ....

AREA

AREA	RADIUS D	F STUCK	MATERIAL 12	13	14	15	14	17	1.0	10	20	21	22	24	74	
1			.01250	01250	-01250	01250		17	. IQ	4.4	- 49			68	24 0 .01250	
Z	01250	.01250	.01250	01250	01250	-01250	01250	01250	.01250	•01270	.01250	.01250	.01250	.01250	01250	
3	.01250	.01250	.01250	-01250	-01250	-01250	AULERU	01250	~~*NTE3N	01.230	401230	01230	-01 250	-01250	.01250	
4	-01250	.01250	.01250	-01250	-01250	-01250	* O1230	01250	.01250	•01 250	.01250	.01250	-01250	•01250	.01250	
5		.01250	.01250	-01250	-01250	-01250	-01280	01250	**************************************	•01230		• D1250	01250	01250	01250.	
6		.01250			.01250	01250	01250	.01250	.01250	.01250	.01250	.01250	.01250	.01250	•01250 •01250	
7		.00106														
8	-00092	-00092	- 00000	00100	00100	*00100	.00101	.00107	.00107							
9	.00000	.00090	.00090			0.VVV.L.2.		FAUNDS	* DDDA*	_ MANU&			~~~~	8888		
10						+		• OO OAT	.00092	-00042	. 00097	.00092	_ 00002		00000	
11		-00078														
	• 00077	.00077	.00077	.00077	•00077	•00077	4000/9	600079	.00079	-00079	.00080	.00080	- 00080	-00081	-00082	
		26	- ·		<b>47</b>	Ju	. 31	32	44	34	28	34	37	3.6		
1	.01250	.01250	.01250	.01250	.01250	-01250	-01250	- 01 260	. A12EA	01 750						
2	01250_	-01250	.01250				401230	AUI / 3U		-01250	. กาวสถ	A125A	21282			
3	.01250	.01250	.01250	.01250	.01250	-01250	-01250	-01250	.01250	A1 25A	M1280	U1.Z2U		-01230	.01250	
4	. 01250	-01250		.01250	.01250	.01250	.01250	-01250	-01250	001520	+01270	•01520	•01250	• 01 250	.01250	
5		.01250	.01250	.01 250	.01280	. 61 284	. 01 254					U1Z50			.01250	
6		.01250		****	OLLOU	• 01 520	*01520	* OT 520	•01250	-01270	<b>-01230</b>	-01250	-01250	. 01280		
7				901270	* 71530		OT 230	UL 250		·01250		.01250	01250	.01250		
<u> </u>	00000	.01250	01 5 20	*U127U	•01270	*O1230	-01250	-01750	. 01250	-01280	01280	01 7E 0	A4 4 E A	01250	01 5 5 6	
		46000	95000	PUUNAD	auuuya.	auuuyb	00097	-00097	-00097	. 00007	- በበስロን	- 6669	4444	22222	00000	
.,																
10	00082	-00082	-00082													
11			*****		~ + ~ ~ ~ ~ ~	0.0000	4.00000	UAUUUUU	O_ HORODB	0.00000	n.nnnan	A AAAAA	A AAAAA			
AREA			7	T. #1	📆 🏋 🚅			. 3 (	30	• •	50	K 1	8.2	4.5	<b>#</b> 4	
1	.01250	.01250	* AT 5 2A	*01520	• 01270	• 01250	•01250	a 01 2 3 0	-01250	A01250	_01750	-01250	- 01 280	A1 2 EA	61356	
2	.01250	-01250	-01250	-01250	-01250	-01250	-01250	.01250	-01250	-01 250	01250	01250	01250	01230	-01250	
3	.01250	.01250	-01250	.01250	-01250	-01250	-01250	.01250	01250	01 250	01250		01 250			
4	.01250	.01250	-01250	.01250	-01250	.01250	-01250	-01250	01250	01250	401270	01270	.01270	.01230	.01250	
5	.01250	-01250	- 01 250	.01250	-01250	.01280	- 01 250	.01250	01250	*AT \$20.	4 41224.		91229.	OIZ20.	01230_	
6	.01250	-01250	-01250	- 01 250	.01250	.01280	01250	01250	*01520	•01230	•01250	.01230	. 01 250	.01250	.01250 .01250	
7	.01250	-01250	-01250	- 01 250	-01250	-01280	. 01250	.01250		**************************************	01420.			01250		
8	.00098	200096	.00009	.00000	.01250	01250	01250	+01230	.01230	•01250	-01250	.01250	.01250	.01250	-01250	
9	-00086	.00086	-00087	- 00087	.00007	00007	00000	.01250	-4D1530	NT 520_	-01230	01250	01250	01250	01250_	
10.	-00085	- 00085	- 00 085	- 00001	-000097	000001	*****	.00000	.00088	.00088	.01250	.01250	.01250	.01250	.01250	
AREA	55	56	57	100000. 58	59	60				0008 /		01250_				
1							61	62	63	64	65	66	67	68	69	
<b>.</b>		.01250 .01250	VI 3EV	01 250	AULEZU.	~~************************************		• ULZZQ		OY 23 B			01250	01250	.01250	
4			• OT 720	.01250	•01530	*01520	•01230	.01250	.01250	• 01 Z 3 O	.01250	.01Z50	.01250	.01250	.01250	
		01250	•01.230	DT 52D.	•01720	- <u> 0147D</u> .	-01520	- OLZ50 .		- 01.250	D1250.	01250	. • OL 250 <sub>-</sub>	01210	01250	
3		.01250	* AT 5 20	-01250	•01250	-01250	-01250	-01750	-01250	_01250	_ <b>01280</b>	. 01 260	01 250	01250	A1 3 F A	
	.01250		UL Z DU	- + UT 520	0125U	OLZ3Q_	O125D	D1 250	-01250	_01250	・ハリラモハ	01350	01 250	A1 2 EA	A1 AFA	
9		.01250	• OT 520	********	•01230	•01230	*U1230	-01250		_ 01 750	- A325A	· 01380	01000	44		
7	.01250		AULE34		- AUL 23U.		-01230	- 01 7 90		. กา 780	21382	A124A	~ ~ ~ ~ ~			
5	•01250	•01250														
9			4.U.L.C.7.U.				01230	- 01740	. กาวแก	N1 38A	<b>6135</b> 0	A 1 4 5 A	~ ~ ~ ~ ~			
10	• 01 5 3 O	01230	0.00000	0.0000	<b>0.0000</b>	V. 00000	0.00000	0.00000	0.00000	0.00000	0.00000	<b>ሲ_ሰሰሰሰ</b> ሲ	A. AAAAA			
AREA .			12	- / 5			[ D	7 /				erene analas anticonaras e co			0.00000	
<b>1</b>		.01250		.01250	•01250	· .01250~		. <del>4</del> 1.250	the section of							
2	.01250	-01250	-01250	.01250	.01250		02250	.01250		in a suite and I	Bid and Dates				elisate e e tipo e como que do ser contrato e con actual de la companya della com	
_	.01250	.01250	.01250	.01250	.01250	.01250	.01250	.01250		,				•	and the second of the second o	the
3	01280	01250	01 250	01 250	01250	.01250	-01250	.01280								
3						~~~~~~~~~								~~~~~		<u>_</u>
			- 01 250	. 01 280	_01280	. 61 2 2 6										
5	.01250	.01250	.01250		.01250	.01250	+01Z50	.01250								
5 5	•01250 •01250	.01250 .01250	.01250	.01250	.01250	.01250	01250	-01250		and the second second second second second				All many to represent the record magnetic	the same and the s	
5	•01250 •01250	.01250	.01250	.01250	.01250 .01250 .01250	.01250 .01250	.01250 .01250	. 01250 .01250	enemana a company		Commence and Application Commence of the Comme			Marine of the section		

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TIME (SEC) = 333.366
                          0.000
     TOTAL AREA(CZ) =
                     0.0000
     WEDGE AREA-
     DRIGINAL PENITRATION DEPTH-
                                      .2050
     ANGLE LDEGI - 0.000
                                    .42824
     ARC LENGTH/TIME INCREMENT=
     NO. OF WEDGES IN POOL . 8
     TEMPERATURE HISTORY OF POWDER
AREA
               2
                       3
      145.288 145.288 145.288 145.288 145.288 145.288 145.288 145.288 145.288
     144.339 144.323 144.323 144.324 144.323 144.336 144.336 144.344
     143.410 143.428 143.426 143.429 143.429 143.401 143.401 143.366
     142,557 142,512 142,512 142,513 142,512 142,549 142,548 142,569
     141.734 141.773 141.773 141.774 141.774 141.715 141.715 141.693
     141.028 140.969 140.969 140.970 140.970 141.013 141.013 141.032
     140,345 140,390 140,390 140,392 140,392 140,320 140,320 140,329
     139.607 139.573 139.573 139.575 139.574 139.503 139.503 139.509
     139.160.139.195 139.195 139.197 139.197 139.131 139.131 139.131
     136.643 136.652 136.652 136.654 136.654 136.612 136.612 136.604
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    .. 138.337.138.365 138.365 138.368 138.368 138.305 138.305 138.291
TEMPERATURE OF STUCK MATERIAL
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AREA	10	_11	12	13	14	15	_16	17	18	19	20	21	22	23	26
1	145.288	145.288	145.288	145.288	145.288	145.288	145.288	145.288	145.288	145.288	145.288	145.288	145.288	145.288	145.288
2	144.256	144.253	144.255	144.255	144.248.	144.266	144.271	144.271	.144.302	144.278	144.306	144.309	144.311	144.326	144.330
3	143.237	143.230	143.235	143.235	143.261	143.261	143.329	143.329	143.294	143.340	143.310	143.332	143.387	143.371	143.398
4	142.324.	142.314	142.321	142.321	142.311	142.355	142.327	142.328	142-433	142-377	142.453	142.471	142.491	142.531	142.529
5													141.646		
6													140.900		
7													140.061		
8													139.495		
9													138.868		
10													137.441		
11													136.770		
AREA	25	26						32							39
1	145.288	145.288	145.288	145.288	145.288	145.288	145.288	145.288	145.286	145.288	145.288	145.288	145.288	145.288	145,288
. 2	144.332	144.336	144.346	145.345	144.346	144.354	144.359	144.365	144.364	144.367	144.369	144.371	144.382	144.381	144.387
3													143.499		
4													142.697.		
5													141.929		
6													141.267		
7													140.629		
8													140.129		
9													139.707		
10	138.550	137.459										139.041	139.260	139.211	139.550
11	137.662	0.000						0.000					0.000	0.000	0.000
AREA	<b>4</b> D	41						47					52	53	54
1	145.288	145.288	145.288	145.288	145.288	145.200	145.288	145.288	145.288	145.288	145.288	145.288	145.288	145.288	145.288
2	144,366	144.395	144.392	144.397	144.422	144.431	144.434	144-438	144.473	144.479.	144.516	144.532	144.543	144.548 :	144.557.
3	143.519	143.519	143.552	143.551	143.608	143.602	143,599	143.634	143.699	143.709	143.772	143.810	143.839	143.845	L434860
4	142.713	142.735	142.735	142.749	142.836	142.861	142.870	142.866	142.990	143.011	143.119	143.168	143.201	143.214	143.239
5	141.976	141.987	142.052	142.053	142.186	142.161	142.177	142.243	142.377	142.399	142.525	142.597	142.653	142.666 1	142.695
<b>6</b>	141.306	1	141.384	151.405	141.578	141.612	.141.626	141.662	141.841	141.874	142.048	142.127	152-182	142.202	42.242
7													141.816		
8	140.259	140.346	140.426	140.450	140.726	140.766	140.776	140.847	141.086	141.129	141.356	141.464	141.538	141.564.1	41.614
-			140.130	140.157	140.482	140.508	140.513	140.612	140.875	140.920	141,160	141.283	141.372	141.398 1	41.449
													141.297		
															a V W

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AREA
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     145.288 145.288 145.288 145.288 145.288 145.288 145.288 145.288 145.288 145.288 145.288 145.288 145.288 145.288
     144.556 144.567 144.566 144.576 144.587 144.590 144.603 144.610 144.612 144.618 144.625 144.637 144.640 144.657 144.664
     143.874.143.874 143.882 143.898 143.921.143.928.143.953.143.967.143.971 143.982 143.999.144.020 144.027.144.075
     143.238 143.270 143.268 143.295 143.327 143.337 143.875 143.395 143.401 143.419 143.441 143.474 143.485 143.534 143.555
    142.720 142.722 142.735 142.768 142.812 142.826 142.874 142.900 142.909 142.931 142.962 143.004 143.017 143.082 143.110
     142.243 142.289 142.286 142.329 142.382 142.397 142.457 142.489 142.499 142.527 142.563 142.615 142.632 142.710 142.744
     141.908 141.917 141.932 141.979 142.041 142.061 142.130 142.166 142.179 142.210 142.253 142.313 142.331 142.422 142.461
    141.622 141.673 141.672 141.727 141.796 141.817 141.895 141.937 141.950 141.985 142.033 142.100 142.121 142.222 142.266
    141.480 141.500 141.357 141.426 141.511 141.538 141.635 141.687 141.704 141.746 141.805 141.889 141.914 142.037 142.091
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     143.139 143.147 143.185 143.379 143.466 143.493 143.514 143.543
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    142.502 142.514 142.567 142.839 142.962 142.998 143.027 143.068
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        REA 70 71 72 73 74 75 76 77
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NECK RADIUS AT		
1	.001145	
	.001219	
3	.001216	
	.001104	
5	.001108	
	.001002	
7	.001007	
9	•000900	:
10	688000	
11	.000856	
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WITH OVEN CONVECTION HEATING
       <del>PROGRAM RH485(INPUT.OUTPUT.TAPES=INPUT.TAPE6=OLTPUT.TAPE8)</del>
       DIMENSION TN (100),T (200,100),DC(200),PS(200),N3(200),H3(200)
     C , H2(20)), AYECK(200, 100), BNECK(100), ANEK(100)
       C OM MON SS.SM.A L.AM.DX.N.TM.AA.AT.AD.TH.DL.NZ.GT.NC.N3.RAD.TS
      COMMON TAIR, BMT, MT
       INTEGER CT.KL
       <del>२EAL-11+13+112+K+M+IF+11T+LT+MT+111+PTN-</del>
       DATA ANECK, BNECK/20000*0.0,100*0.0/
C
    GLOSSARY
C
C
          AA TOTAL TROSS SECTIONAL POOL AREA (CM2)
          ACNST -- DUMMY VARIABLE-
          ACONST DUMMY VARIABLE
C
-€-
          AD PENITRATION THICKNESS (CH)
 C
          AH CONVECTION DOEFFICIENT OF OVEN AIR (J/Ch2-SEC-K)
          AL-INITIAL PONDER THERMAL DIFFUSIVITY (CM2/SEC)
·C-
          ALCP SPECIFIC HEAT OF ALUMINUM MOLD (J/KG-C)
          ALDIFF THERMAL DIFFUSIVITY OF ALUMINUM MCLC (CM2/SEC)
 C
          ALK THERMAL CONDUCTIVITY OF MOLD (J/CM-SEC-K)
 ·C
          ALRHO DENSITY OF ALUMINUM MOLD (KG/CM3)
          AM THERMAL DIFFUSIVITY AT COMPLETE MELT (C12/SEC)
 r.
-С-
          ANECK - NECK-RADIUS AT EACH NOCE (CM)
                NECK RADIUS OF EACH NODAL AREA AT THE TIME
 C
                    MELTING OVERCOMES THE SINTERING PORCESS (CM)
 C
               DUMMY VARIABLE = ANECK(1,1)
          AR - JUMMY VARIABLE USED IN CALCULATING REHAUNING --
                    DROSS SECTIONAL POOL AREA
 C
C
          AS TOTAL AREA ENCOMPANSING NODAL ANALYSIS
--C
                    ( HEIGHT OF NODES X TOTAL WIDTH)
          <del>AT - 3 ROSS - SECTIONAL - A REA - OF - ONE - WEDGE - 1CM</del> 2+
 C
          ATEST DUMMY VARIABLE
t
              DUMMY VARIABLE USED TO DETERMINE NEW ANGLE THETHA
              CROSS SECTIONAL AREA OF STUCK COLUMN OF MATERIAL THAT
 C
          A3
 C
                    <del>THAT RE-</del>ENTERS THE POOL IN THE NEXT TIME INCREMENT (1
 C
              THE CROSS SECTIONAL AREA OF THE STUCK COLUMN OF
 C
                    MATERIAL THAT REMAINS OUTSIDE THE POOL
 C
                    DURING THE NEXT TIME INCREMENT (C)2)
          BA ANGLE OF RESPONSE OF THE POOL (RAD) ----
              DUMMY VARIABLE USED IN TEMPERATURE EQUATION
          88
          BE DUMMY - VARIABLE USEB - SURFACE NOBAL NECK SIZE EQUATION --
 €
          BMT EXTERIOR TEMP OF MCLD (DEG C)
 T.
          BMIN EXTERIOR TEMP OF MULD DURING
 C
                    THE NEXT TIME INCREMENT (DEG C)
---
          SNECK - NEW -COMPUTED NECK SIZE FROM SUBROUTINE (CH)
          CC HALF LENGTH OF POOL CORD (CM)
          CT - SOUNTER USED TO COUNT THE NUMBER OF -STUCK COLUMNS
 C
              CONSTANT USED IN TEMPERATURE EQUATIONS CETERMINED IN
          C1
 -∙€
              ----SUBROUTINE CONST
 С
              CONSTANT USED IN NODAL TEMPERATURE EQUATIONS
 0
          C3 CONSTANT USED IN NODAL TEMPERATURE EQUATIONS
             AVERAGE DIFFUSIVITY BETWEEN NODES I+1 AND NODE I (CM2/SEC)
          AG
          DB -4-VERAGE DIFFUSIVITY BETHEEN NODES I-1 AND NODE I -1-10M2/SECT
 C
              ARC WIDTH OF STUCK NODAL COLUMNS (RAD)
          DL - ARC LENGTH DER-TIME INCREMENT (RADISEC)
```

WM RH485	74/825 OPT=0,ROUNC= 4/ S/ M/-D,-DS FTN 5.1+601 163
<del></del>	DT MAXIMUM TIME INCREMENT (SEC)
C	DX DISTANCE BETWEEN NODES (CM)
-6	- BXMOLO THICKNESS OF MOLB(CM)
Č	EN PORTION OF COLUMN ARC WIDTH RE-ENTERING THE
<u> </u>	
С	FT AVERAGE TEMPERATURE OF MATERIAL IN FREE FALL ZONE (C)
<del></del>	G ACCELLERATION DUE TO GRAVITY (CM/SEC2)
C	H HEIGHT OF STUCK MATERIAL LEAVING POOL (CM)
<del></del>	HT DUMMY VARIABLE USEE IN COMPUTING POOL ANGLE THETHA
C	HI HEIGHT OF STUCK MATERIAL LEAVING POOL DURING THE NEXT
<del></del>	TIME INGREMENT (CM)
С	H2 HEIGHT OF STUCK MATERIAL COLUMN OUTSIDE POOL
<del></del>	- H3 DISTANCE BETHEEN LAST THO NODES OF EACH COLUMN OF
C	STJCK MATERIAL OUTSIDE POOL (CM)
	-I - COUNTING - VARIABLE - USED IN EO-LOOPS
C	IFLAG FLAG SHOWING THAT NODAL HEIGHT HSD REACHED
	ITS MAXIMUM WHEN #-1-
Č	II COUNTING VARIABLE USED IN DO-LOOPS
-	III - COUNTING- VARIABLE - USED IN DO-LOOPS
C	IREM FLAG SHOWING POOL MASS IS GONE WHEN = 0
<u> </u>	
C	ITM COUNTER
	ITMX -NUMBER OF ITERATIONS -REFORE MOLD ROTATION
C	
C	J COUNTING VARIABLE USED IN EO-LOOPS
C	JJ COUNTING VARIABLE USED IN DO-LCOPS
<del>-</del> 6	JJICOUNTING -VARIABLE -USED IN DO-LOOPS
C	JK DUMMY VARIABLE
<del>-</del>	
C	JTVAL DUMMY VARIABLE
<del>. č</del>	- K DUMMY VARIABLE
Ċ	KK FLAG VARIABLE
<del> c</del>	- KL-DUMMY VARIABLE
C	K4 DUMMY VARIABLE
<del></del>	
C	LT DUMMY VARIABLE
C	- M - CROSS SECTIONAL OF MASS LEAVING POCK-AND STUCK TO WALL
С	MT INTERIOR MOLD SURFACE TEMP (DEG C)
C	MIN-INTERIOR-MOLD SURFACE TEMP-ONE-TIME-INGREMENT
C	LATTER (DEG C)
t	N NUMBER OF NODES IN EACH STATIONARY POOL COLUMN
C	NC NUMBER OF POOL COLUMNS OF NODES
	NEC GOUNTING VARIABLE
C	NAX MAXIMUM NODE HEIGHT BASED ON VOLUME OF POOL (CM)
	NNAX E GUALS N+1
C	NNTP DUMMY VARIABLE WHICH EQUALS N6+N4+1
Ç	N2 COLUMN NUMBER OF STUCK MATERIAL
C	N3 NUMBER OF NODES IN STUCK COLUMN MINUS 2
~ · · · · · · · · · · · · · · · · · · ·	NSTEMP DUMMY VARIABLE
C	N4 DJMMY VARIABLE
<b>.</b>	NE HUMBER OF NODES IN PENITRATION THICKNESS
C " C -	OT PORTION OF RE-ENTERING STUCK COLUMN NOT ENTERING POOL (RAD)
1	PS POSITION OF A PARTICULAR STUCK COLUMN IN MOLD (RAD)
ç	PT PRINTING INTERVAL (SEC)

TAIR=371.0 HT=II BMT=IT

```
C ***** SECTION II ****
    -RA=RI/R9----
  220 TH=(3.14159/2.0-ATAN(RA/(-RA*RA+1.0)**0.5))*2
    - DETERMIYE-AREA-IN-STATIONARY-POOL---------
     AA=0.5*RO*(TH-SIN(TH))
     -<del>AL=COND(3)/(CP(3)*R!IO(3)</del>}-----
     AM=COND(RAD)/(CP(RAD)*RHO(RAD))
     H=0.10472*RPM----
     TW= TH/W
     <del>-TS=T#7N3 ------</del>
     ITM=0
  ACNST=2/(DXMOLD*ALRHO*ALCP)*(AH*ALK/DXMOLD)
     -TS1=1 /ACNST
     TM=TM+TS
CONTRACTOR DETERMINE WAXIMUM PENITRATION DISTANCE
     AD=7.2*(AL*TW) **0.5
····C···········DETERMINE :MAXIMUM TIME--ITOREMENT ·······--
     DIT=TW/NC
C DETERMINE NODAL DISTANCE
     DX= (DT*2*AM) ** 0.5
DETERMINE NUMBER OF NODES PER COLUMN
----N=(INT(40/DX)+1)+2
     NNAX=N+1
    -- N2=NC+2 -- --
     CT=N2-1
--C--<del>---SET</del>-INITIAL TE<del>MP</del>ERATURE-AND NECK RADIUS -------
     DO 125 I=1,NC
  - - 00 126 J=1 NVAX
      T [ = ( L, I ) T
   ----ANEGK(I,J)=0.6---
  126 CONTINUE
   125 CONTINUE
      BA= 0.017 4533 + 9 A
     15 FORMAT(1H1,23HR)TATIONAL MCLDING SIMULATOR )
     WRITE (6,12) TAIR , AH, DXHOLD, ALDIFF
   12 FORMAT(1X, //, 6X, 24HTEMP. OF AIR (DEG. C) = ,F10.3./, 6X,
  -- 133HCONVECTION CO EF. (J/CK2-SEC-K) = ,E10.3./.6X.
     222HMOLD THICKNESS (CM) = .F10.5, /.6X,
   <del>---3-27HMGLD-DIFFUSIVITY-(CM2/S)----F10.5</del>}----
      WRITE(6,11)RI,RO,N,DX,SS,AL,SM,AM,RPM
   <del>-11-FORMAT(6x+25HPOWDER-HEIGHT-RADIUS(CM)=-</del>,F10.3,/,6x,17HMOLD-RADIUS(
     1CM) = ,F1C.3,/,6X,21HINTIAL NO. OF NODES= ,I3,/,
   -3---6X,20HNODE-THICKNESS(CM)= -,F10-4,/,6X,38HSTICK-TEMP(C)-AND-DIFF
     4USIVITY(CM2/S) = ,F10.3,3X,E10.3,/,6X,46HCOMPLETE MELT TEMP(C) AND
 501FFUSIVITY(042/S)= -,F-1 (.3,3X, E 10.3,/,6X,6HRPM= -,F10.3)
 C
1900 TW=TH/W
      GG=2*R0*SIN(T4/2)
    - ANK=ANESK(1,1)
      IF (CT.GT.N2) ANK = ANECK (CT,1)
```

RH 4 85	74/825 OPT=0, ROUND= A/ S/ M/-D,-DS FTN 5.1+601 166
	SE=RADX(ANK)
С	RECHECK MAXIMUM TIME INCREMENT EACH ITERATION
	DT= 1BE*D X) **2/(2.0*AM) IF(AM.LT.AL) DT=(BE*DX)**2/(2.0*AL)
	TM=TM=TS
	IF((IREM.EQ.0).OR.(ITM.GT.0)) GOTO 109 TS=TW/NC
	TTL=TS
	ITMX=1 IF(TTL.GT.DT) TTL=DT
	IF(TTL.ST.TS1) TTL=TS1
	IF(TTL.EQ. TS) GOTO 109
	ITMX=INT(TS/TTL)+1 TS=TTL/ITMX
	IF (IREM. EQ.0) TS=DT
	DL=W*TS*R0
	IF(TM.EQ.0.0) WRITE(6.20) DI, TS, DL
	FORMAT(1H ,/.6X,23HMAX TIME INCREMENTS(S)= ,F10.5,/,6X,21HTIME INC
	REMENT-JSEB=-,F10.5,/,6X,18HARG/TIME-INC(CH) = ,F10.5 )
	TM=TM+TS
	IFTIREM. EQ. 0) GOTO 540  NMAX=INT (AA/(NC*DL*DX))+1
	AD=7.2*(4L*TW)**0.5
	TF=(1.3333*CC/(G*COS(BA)))**n.5
	SR=1F/13
	IF(SR.LT.1.0) SR=1.0
	DETERMINE AREA OF EACH WEDGE
	S1=R0*SIN(TH/2.8) S2=R0*C0S(TH/2.8)*TAN(**TS/2.8)
	AT=0.5*(S1+S2)*(S1-S2)*SIN(W*TS)+0.5*P0*R0*(W*TS-SIN(W*TS))
	AS=(N=0.5)*DX*DL
	IF (AS.GT.AT) AS=AT
•	IF(N.LT. NMAX) GO TO 560
	N=NMAX TFLAG=1
	IFLAG=1 CONTINUE
	00 550 I=1,NC
	T(I, N+1) = T(I, N-1)
	ANECK (I, N+1)=ANECK(I,N-1)
	CONTINUE
<del>- 55</del> 2-	CONTINUE
	NNA X= N+1 N6= <del>INT (4 0 / 0 x ) + 1                                </del>
С	NO-INTRADUCTOR
	** SECTION IV ****
C	
C	COMPUTE NODAL TEMPERATURES AND NECK RADIUS
€	
<u>с</u>	MPUTE EXTERIOR AND INTERIOR MOLD SURFACE TEMPERATURES
587	CONTINUE
	ACONST=2*TS/(3XMOLD*ALRHO*ALCP)
	BMTN=BMT *(1.0-ACONST * (AH+ALK/DXMOLD)) + ACONST*(AH*TATR+ALK*MT/DXMOLD)
· · · · · · · · · · · · · · · · · · ·	. ACONST*(AH*TAIR+ALK*MT/DXMOLD) MTN=MT*(1, ŭ-ACONST*ALK/DXMCLD)+ACONST*ALK*BMT/DXMOLD
	いしがーロー・チ チャガニガ ひんえつ しょせ にいくい ヤロ ぐにつきょせ ぐいはつきょうせい しはまた カソけっとう

M RH485

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C NODAL TEMPERATURE EQUATIONS FOR STATIONARY POOL
          T(I,1)=MT
          TN(1) = 4T
          CALL CONST(C1,C2,C3,D8,DA,MT,MT,MT,ANECK(I,1),ANECK(I,2),
     - 1 BNECK(1))
          ANECK(I,1)=BNECK(1)
 DO 638 J=2,N
          CALL CONST (C1, C2, C3, DB, DA, T (I, J), T (I, J-1), T (I, L+1), ANECK (I, J),
        1 ANECK(I,J+1), BNECK(J))
          TN(J) = T(I,J) + (1-TS/C3+(DB/C1+DA/C2)) + TS/C3+(DB/C1+T(I,J-1)+
 630 CONTINUE
       (LL)MT = (LL,I)T
IF((JJ.EQ.NEC).AND.(TN(JJ),GT.SM)) NEC=NEC+1
ANECK(I.JJ)=84ECK(JJ)
--- 64: CONTINUE
    525 CONTINUE
  IF(CT.LT.N2)-50-T0-890------
          NODAL TEMPERATURE EQUATIONS FOR STUCK MATERIAL
T(I_{\bullet}1)=YI
1 BNECK(1))
        ANECK(I,1)=3NECK(1)
          IF(N3(I).EQ.0) GO TO 820
JTVAL=N3(I)
CALL CONST (G1, G2, C3, DB, DA, T (I, J), T (I, J-1), T (I, J+1), ANECK (I, J),
       1 ANECK (I, J+1), BNECK (J))
          TN(J) = T(I, J) + (1-TS/C3 + (DE/C1+DA/C2)) + TS/C3 + (DB/C1 + T(I, J-1) + I)
       19A/62*T(T, J+1)
  770 CONTINUE
-- -78: J=N3(I)+1
          CALL CONST(C1,C2,C3,EB,DA,T(I,J),T(I,J-1),T(I,J+1),ANECK(I,J),
BB=TS/C3*(DB/C1*T(I,J-1)+DA/(H3(I)*C2/DX)*T(I,J+1))
821 CONTINUE
                         Name of a final particular of the control of the co
CALL CONST (C1, C2, C3, DB, DA, T(I, J), T(I, J-1), T(I, J), ANECK (I, J),
- 1 ANECKII - J - BNECK(J))
          B8=DB*TS/(H3([)*(C3/OX)**2*(+3([)+0X/2))
IN(J)=I(I,J)*(1-BB)+BB*T(I,J-1)
          DO 870 JJ=1,J
 IF((JJ.EQ.NEC).AND.(TN(JJ).GT.SM)) ANEK(NEC) = BNECK(JJ)
IF (TN (JJ).GT.SM) BNECK (JJ)=RAD
         -ANECK (I, JJ)=34ECK(JJ)
    371 CONTINUE
    88 CONTINUE
 0
- 890 CONTINUE
```

CALCULATE MIX MEAN TEMPERATURE WHEN NO MATERIAL HAS STUCK

-G --\*\*\*\* -- SECTION VI -- \*\*\*\*\*

TP = (TP + MT/2) / (N - 0.5)

DO 1146 III=2, NC ==NC-IHI+2 DO 1142 J=1, NNAX T(I,J)=T(I-I,J)

FT=(AS\*TP+(AT-AS)\*IT)/AT-----

CCC

1109 TP=0.0

```
74/825 OPT=0, ROUND= A/ S/ M/-D,-DS FTN 5.1+601
  1 RH485
                        ANECK(I, J) = ANECK(I-1, J)
       1142 CONTINUE
      1140 CONTINUE ----
                        DO 1150 I=1, NNAX
                         <del>T(1+I)=IT-----</del>
                        ANECK (1, I) =0.0
      -1<del>150-CONTIN</del>UE-----
                        IT=((SR-1)*IT+FT)/SR
                        TM=TM+T3 ----
                        GO TO 580
   <del>-1200 CONTINUE</del> --
    C
   G **** SECTION VII ****
                        DETERMINE THE AMOUNT OF STICK AND NEW ANGLE
    C
                        DO 1220 I=2.N
          1225 CONTINUE
                                                                     The second of the second secon
-- 1221 CONTINUE
                         RA = (SS - T(NC, I - 1)) / (T(NC, I) - T(NC, I - 1))
                       #= <del>{R$+{F-2}}*</del>}X
                         DO 1250 II=2,N
 1251 CONTINUE
  RB=(SS-T(NC,II-1))/(T(NC,II)-T(NC,II-1))
                        H1=(RB+(II-2))*DX-----
                        IF (INT (H/DX) -1 .LT.G) GO TO 1109
       DC(N2-1) = W*TS
  N3(N2-1) = INT(H/DX)-1
                       H3 (N2-1) = H-BX* N3 (N2-1)
                        H2(N2-1)=H
                    -JTVAL=N3 (N2-1)+1----
                        DO 1350 J=1, JT VAL
                       T(N2-1,J)=T(NC,J)
                         ANECK (N2-1,J) = ANECK (NC,J)
 -135<del>: CONTINUE</del>
                         T(N2-1+N3(N2-1)+2)=SS
                        ANECK (N2-1, N3(N2-1) + 2) = A NECK (NC, N3(N2-1)+2)
     C
                      COMPUTE MIX TEMPERATURE OF MATERIAL NOT STICKING BUT
    C
                                   ENTERING FREE-FALL ZONE
                        TP=0.0
                 TP=TP+T(NC+J)
        1391 CONTINUE
                         K=RA-0.5
IF (K.LT. 3.6) TP=(TP-T(NC,I-1)*K)/(N-I-K+1)
                                                          and the second of the second o
       FT = ((\Delta S - M) + TP + (\Delta T - \Delta S) + TT)/(\Delta T - M)
                  POTATE - PONDER -----
- - C
                         DO 1446 III=2, NC
```

```
74/825 OPT=3, ROUNC= A/ S/ M/-D, -DS
                                         FTN 5.1+601
                                                       170
     DO 1445 J=1, NVAX
     T(I,J)=T(I-1,J)
     ANECK(I, J) = ANECK(I-I,J)
 1445 CONTINUE
 1440 CONTINUE
     K4 = 1
      IF (CT.LT.N2) 30 TO 1908
C
        SECTION VIII *****
C
     STUCK MATE SHIFT
C
      DO 1476 JJI=N2,CT
     JI=CT-JJI+N2
      DC(JI)=DC(JI-1)
     (1-1L) EV = (1L) EV
  ----H3 (JI)=H3 (JI-1)
      H2(JI)=H2(JI-1)
  DO 1530 J=1, N3 TEMP
      すてしまっしょうてしまー1・しょ
      ANECK(JI, J) = ANECK(JI-1, J)
 1532 CONTINUE
 1471 CONTINUE
 -1550-X5=6.28318-TH/2---
      EN=0.0
      KL = 0 -
      IF(PS(CT).LT.X5) GO TO 1900
---<del>C-</del>----
   **** SECTION IX
                  * * ** *
TO THE STUCK MATERIAL RESENTERING POOL ROUTINE
_.C
      00 1576 JI =N2 CT
      IF (PS (JI).GE.X5) GO TO 1575
1570 CONTINUE
 1575 CONTINUE
    --- EN=PS (JI ) - X5
      OT=DC (JI ) - EN
   PS(JI)=X5
   ---DC+JI)=3T
      GO TO 1770
IF (OT.LT.0.0) KL=1
      CT=JI=1
      KK = 3
   JL=0--
      JK=N3 (JI-1)+1
- 165; -IF(UK._E.N3(JI)+1 ) 60-10 1660
      JK=JK-1
  GO TO 1550
  1660 - CONTINUE
      IF ((H3(JI).GE. 1.5*)X).AN[.KK.EQ.1) JL=1
   IF (Jt. EQ.1) JK=JK+1
      DO 1676 I=2,JK
     T(JI=1,I)=(0T*T(JI,I)+DC(JI=1)*T(JI=1,I))/(0T+CC(JI-1))
```

M RH495

```
ANECK(JI-1,I)=(OT*ANECK(JI,I)+DC(JI-1)*ANECK(JI-1,I))/(OT+
        DC(JI-1))
  1670 CONTINUE
       JK=JK+1
       A3=DC(JI-1)*(H3(JI-1)-DX/2)
       A4=OT*(H3(JI)-DX/2)
       IF(JL.EQ.1) 44=07*(H3(JI)-1.5*DX)
       IF(JL \cdot EQ \cdot 1) T(JI, JK) = T(JI, N3(JI) + 2)
       IF (JL.EQ.1) ANECK(JI,JK)=ANECK(JI,N3(JI)+2)
       IF((JL.EQ.6).AND.(KK.EQ.1)) A3=DC(JI-1)*DX
       T(JI=1,JK)=(T(JI,JK)*A4+T(JI=1,JK)*A3)/(A3+A4)
       ANECK(JI-1,JK) = (ANECK(JI,JK)*A4+ANECK(JI-1,JK)*A3)/(A3+A4)
       OC(JI=1)=OC(JI=1)+OT
       IF(KL.NE.1) GD TO 1760
       PStJI-11=PStJI1
       JI = JI - 1
  ----<del>60 T0 155</del>0
  1760
        PS(JI-1) = X5
  1775 CONTINUE
       T(1,1) = YT
       N4=N3 (JI) +1------
       IF (H3 (JI ) • GE • 1 • 5 * DX) N4 = N4+1
    T(1,J)=T(JI,J)
     ANECK (1, J) = ANECK (JI, J)
  1810 CONTINUE
       RM=H2(JI)=(N4-0.5)+DX----
       IF(RM.LE.0.0) GO TO 1840
       N4=N4+1
       T(1,N4) = (RM*T(JI,N3(JI)+2) + (DX-RM)*IT)/DX
      - ANECK (1, N4) = A NECK(JI, H3 (JI)+2)----
-- 1840 CONTINJE
       IF (IFLAG. EQ. 1) 60 TO 1870 ----
       IF(N6+N4.LE.N) GO TO 1878
       NNTP=N6+N4+1
       NNAX=N+1
       DO 1866 I=2, NO
       DO 1862 J=NNAX,NNTP
       <del>T(I+H+I) T=(L+I) T</del>
       ANECK(I, J)=0.0
  186? CONTINJE
   1860 CONTINUE
      N=N6+N4---
   1870 CONTINUE
       NNAX=N+1
 190 CONTINUE
T(1,II)=IT
   ----ANECK (1, II)=0.0
   1905 CONTINJE
    \Delta \Delta = \Delta \Delta - M
 \Delta R = \Delta A
      -IF ( (CT. LT. N2). 02. (EN. LE. E. 6) + 60 TO -1950
```

74/825

```
-SUBROUTINE PRINT(T, LK, K)
     DIMENSION T(200,100),N3(200)
     COMMON SSOSMOR LOAMODXONO TMOAAOATOADOTHODLONZOCTONCONZORADOTS
     COMMON TAIR, BMT, MT
     REAL PT-
     INTEGER CT
     -K=LK+1--
     BB=TH*57.29578
     IF (K. EQ. 1) 60T023
     WRITE (6, 20) TM, AA, AT, AD, BB, DL, NC
1/,6X,12HWEDGE AREA= ,F10.4,/,6X,28HORIGINAL PENITRATION DEPTH= ,
     2F10.4,/,6X,12HANGLE(DEG) = ,F10.3,/,6X,27HARC LENGTH/TIME INCREMENT
     3= ,F10.5,/,6X,24HN). OF WEDGES IN POOL = ,I3,/)
-----<del>WRITE(6</del>;10) -<del>TAIR</del>,BMT,MT-----
   10 FORMAT(6X, 24HTEMP. OF AIR (DEG. C) = _{,}F10.3./.6X.
 2 /.6X.44HTEMP. OF MOLD+S INTERIOR SURFACE (DEG. C) = .F18.3,///)
  ----<del>\\RITE</del>(6,21)
   21 FORMAT(6x,30+TEMPERATURE HISTORY OF POWDER )
23 IF (K. EQ. 1) WRITE(6, 24)
   24 FORMAT(//, 12H NECK RADIUS )
----- ICOL=15 ----
     IREP= ((NC-1)/ICOL)+1
   --- IREM-NC- (IREP-1) * ICOL
     DO 100 KK=1. IREP
     "IF(KK:EQ:IREP) ICOL=IREM """
     KJ = (KK-1) * ICOL + 1
K-J1=KJ+(TCOL-1)
     IF(KK.EQ.IREP) KJ1=KJ+IREM-1
    WRITE(6, 108) (J, J=KJ, KJ1)
  108 FORMAT(1X,5HAREA ,15(I3,5X))
    00 164 J=1,N
     IF (K \cdot EQ \cdot O) WRITE (6, 102) \cdot J \cdot (T(I, J), I = KJ, KJ1)
   IF(K.EQ.1) WRITE(6,103) J. (T(I,J), I=KJ, KJ1)
   102 FORMAT(1X,13,3X,15(F7,3,1X))
104 CONTINUE
   ----<del>WRITE(6,105)</del>
   105 FORMAT(1X,//)
- 10: CONTINUE
   97 CONTINUE
    IF (CT.LT.N2) ZETURN
     ICOL=15
    9º FORMAT(1X,//, 31H TEMPERATURE OF STUCK MATERIAL ,/)
-----IF(K.EQ.1) WRITE(6.99)
   99 FORMATI//, 304 NECK RADIUS OF STUCK MATERIAL )
  IREP=INT((CT-(N2 -)) /ICOL)+1
      IREM=CT-N2+1-(IREP-1)*ICCL
   ·····00-200 KK=4▼I₹E2--- ···-
      KJ = (KK-1) * IC \cap L + N2
      IF(KK.EQ.IREP) ICOL = IREM
      KJ1=KJ+(ICCL-1)
 -----IF(KK.EG.IREP) KUL=-KU+IREM-1
```

PRINT	74/825 OPT=0, ROUNC= A/ S/ M/-D, -DS FTN 5.1+601 174
	WRITE(6,108)(J,J=KJ,KJ1)
	NTN=N3(KJ)+2
	00.204 J=1.NTN
	DO 203 II=KJ.KJ1
	NTMP=N3(II)+2
	IF (NTMP. EQ.NTN) GO TO 203
	IDIFF=NIN-NTMP
	DO 202 LL=1,IDIFF T(II,NTMP+LL)=0.0
<b>3</b> n 2	CONTINUE
	CONTINUE
203	IF (K. EQ. 0) WRITE (6, 102) J, (T(I,J),I=KJ,KJ1)
	IF(K.EQ. 1) WRITE(6, 103) J, (T(I,J), I=KJ, KJ1)
204	CONTINUE
	CONTINUE
305	RETURN
	END
Property of Special Company	
	0100017716 B0107104 00 03 00 04 74 70 73 410 414 D184
	SUBROUTINE CONSTICT, C2, C3, OB, OA, T1, T2, T3, ANB, ANA, RNK)
-с	DIMENSION N3(200)
C	
<u> </u>	
U	COMMON SS, SM, AL, AM, DX, N, TM, AA, AT, AD, TH, DL, NZ, GT, NC, N3, RAD, TS
-	A1=0.5*(ANB*ANA)
	C1=RADX(ANB) *PX
	DZ=RADX(ANA)*DX
	C3=RADX(A1)*DX
	DELB=(3.0*SU(11)*RAD/(8.0*VIS(T1)*(TM+0.5*TS)))**0.5*TS
•	IF(ANB/RAD.SE.0.5) DELB=0.5*SU(T1)*TS/(VIS(T1))
	BB=COND(ANB)/(RHO(A1)*CP(A1))
	DA=COND(ANA)/(RHO(A1)*CP(A1))
	BNK=ANB+DELB
	IF (BNK.GT.RAD) BNK=RAD
	IF (DELB. GT.O. 0) RETURN
	WRITE (6, 10) TM, AND, ANA, BNK, RAD, TS, SU(T1), VIS(T1), DELB
10	FORMAT(1X, 10 (E10.5, 1X))
	RETURN
	END
<del></del>	
	The second secon

	FUNCTION RHO (Y)
	DIMENSION N3(200)
<del>e                                     </del>	
<u>C</u>	DENSITY FUNCTION ( <g cm**3)<="" td=""></g>
<del>С —</del> С	R1=DENSITY AT SS
<del>c</del>	R2=DENSITY AT S4
C	
	COMMON SS, SM, AL, AM, DX, N, TM, AA, AT, AD, TH, DL, N2, CT, NC, N3, RAD, TS R1=5.0937E-04
	R2=8.8058E-04
	X=Y/RAD
	IF (X.GT.0.0) 50TO 10
	RH0=R1 
1	3 IF(X.GT. 1.0) 30TO 20
	<del>RH0=R1+{R2-R1} *X</del>
_	GO TO 1
	1 CONTINUE
	END
-с-	DIMENSION N3(200)
, c : c	CONDUCTIVITY FUNCTION (JCULE/(CM-SEC-DEG K))
Ċ	C1 = THERMAL CONDUCTIVITY AT TEMPERATURE SS
	A THE THE PART OF THE TAX A TA
<del>- c-</del>	C2 = THERMAL CONDUCTIVITY AT TEMPERATURE SM
C C	
	COMMON SS, SM, AL, AM, DX, N, 1M, AA, AT, AD, TH, DL, N2, CT, NC, N3, RAD, TS
	COMMON SS, SM, AL, AM, DX, N, 1M, AA, AT, AD, TH, DL, N2, CT, NC, N3, RAD, TS C1=1.99E-03 C2=4.9325E-03 X=Y/RAD
	COMMON SS, SM, AL, AM, DX, N, 1M, AA, AT, AD, TH, DL, N2, CT, NC, N3, RAD, TS C1=1.99E-03 C2=4.9325E-03 X=Y/RAD IF (X.GT, 0.0) 30TO 10
	COMMON SS, SM, AL, AM, DX, N, 1M, AA, AT, AD, TH, DL, N2, CT, NC, N3, RAD, TS C1=1.99E-03 C2=4.9325E-03 X=Y/RAD IF(X.GT.0.0) 50TO 10 COND=C1
С	COMMON SS, SM, AL, AM, DX, N, 1M, AA, AT, AD, TH, DL, N2, CT, NC, N3, RAD, TS  C1=1.99E-03  C2=4.9325E-03  X=Y/RAD  IF(X.GT.0.0) 50T0 10  COND=C1  G0 T0 1  13 IF(X.GT.1.0) 50T0 20
С	COMMON SS, SM, AL, AM, DX, N, 1M, AA, AT, AD, TH, DL, N2, CT, NC, N3, RAD, TS C1=1.99E-03 C2=4.9325E-03 X=Y/RAD IF(X.GT, 0.0) 50TO 10 COND=C1 GO TO 1 13 IF(X.GT.1.0) 50TO 20 COND=C1+X*(G2-G1)
C	COMMON SS, SM, AL, AM, DX, N, 1M, AA, AT, AD, TH, DL, N2, CT, NC, N3, RAD, TS  C1=1.99E-03  C2=4.9325E-03  X=Y/RAD  IF(X.GT.0.0) 50T0 10  COND=C1  60 T0 1  13 IF(X.GT.1.0) 50T0 20  COND=C1+X*(C2-G1)  GO TO 1
C	COMMON SS, SM, AL, AM, DX, N, 1M, AA, AT, AD, TH, DL, N2, CT, NC, N3, RAD, TS  C1=1.99E-03  C2=4.9325E-03  X=Y/RAD  IF(X.GT, 0.0) 50T0 10  C0ND=C1  G0 T0 1  13 IF(X.GT.1.0) 50T0 20  C0ND=C1+X*(C2-G1)  G0 T0 1  29 C0ND=C2  1 CONTINUE
C	COMMON SS, SM, AL, AM, DX, N, 1M, AA, AT, AD, TH, DL, N2, CT, NC, N3, RAD, TS C1=1.99E-03 C2=4.9325E-03 X=Y/RAD IF (X.GT. 0.0) 50T0 10 C0ND=C1 G0 T0 1 13 IF (X.GT.1.0) 30T0 23 C0ND=C1+X*(C2-C1) G0 T0 1 20 COND=C2 1 CONTINUE RETURN
C	COMMON SS, SM, AL, AM, DX, N, 1M, AA, AT, AD, TH, DL, N2, CT, NC, N3, RAD, TS  C1=1.99E-03  C2=4.9325E-03  X=Y/RAD  IF(X.GT, 0.0) 50T0 10  C0ND=C1  G0 T0 1  13 IF(X.GT.1.0) 50T0 20  C0ND=C1+X*(C2-G1)  G0 T0 1  29 C0ND=C2  1 CONTINUE
C	COMMON SS, SM, AL, AM, DX, N, 1M, AA, AT, AD, TH, DL, N2, CT, NC, N3, RAD, TS C1=1.99E-03 C2=4.9325E-03 X=Y/RAD IF (X.GT. 0.0) SOTO 10 COND=C1 GO TO 1 13 IF (X.GT.1.0) SOTO 20 COND=C1+X*(G2-G1) GO TO 1 20 COND=C2 1 CONTINUE RETURN END
C	COMMON SS, SM, AL, AM, DX, N, 1M, AA, AT, AD, TH, DL, N2, CT, NC, N3, RAD, TS C1=1.99E-03 C2=4.9325E-03 X=Y/RAD IF (X.GT. 0.0) SOTO 10 COND=C1 GO TO 1 13 IF (X.GT.1.0) SOTO 20 COND=C1+X*(G2-G1) GO TO 1 20 COND=C2 1 CONTINUE RETURN END

	RG=-COMMON/-FIXED.CS= USER/-FIXED.DB=-TB/-SB/-SL/ ER/-IB/-P
	FUNCTION RADX(Y)
	DIMENSION N3(200) <del>- Common SSysMyalyaMydxyNyTMyaayaTyaDyTHyDLyN2yGTyNGyN3yR</del> A <del>Dy</del>
С	
<del>c</del>	THIS FUNCTION DETERMINES THE RATIO USED TO CALCULATE DX AS
C 	OF MOLC TEMP
	X=Y/RAD
	- IF(X.GT.0.0) GOTO 10
-	7 AUX = 1 • U
	① IF(X.GT.1.0) GOTO 20
	GO TO 1
	5-RADX=RH3 (0.0) /RH0 (2AD)
	1 CONTINUE
	END
	ENU
	FUNCTION CP(X)
C	
	SPECIFIC HEAT (JOULE/(KG-DEG K))
C	CP=2302.7
	RETURN
	END
-	
	FUNCTION SU(T)
- C	
е	SURFACE TENSION FUNCTION (DYN ES /CM)
C	
	SU=31.0=0.058*(T-105)
	RETURN
	END
С	01101 113111
С	
	RETURN
	and the second s

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