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BORHANI-DARIAN, Mohammad Ali, 1942-
THERMAL STRATIFICATION AND CIRCULATION
OF WATER BODIES SUBJECTED TO THERMAL
DISCHARGE.

New Jersey Institute of Technology,
D.Eng.Sc., 1977
Engineering, mechanical

Xerox University Microfilms, Ann Arbor, Michigan 48106

THERMAL STRATIFICATION AND CIRCULATION OF
WATER BODIES SUBJECTED TO THERMAL DISCHARGE

BY

MOHAMMAD A. BORHANI

A DISSERTATION
PRESENTED IN PARTIAL FULFILLMENT OF
THE REQUIREMENTS FOR THE DEGREE
OF
DOCTOR OF ENGINEERING SCIENCE
AT
NEW JERSEY INSTITUTE OF TECHNOLOGY

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1977

ABSTRACT

A three-dimensional analytical model for large water bodies is presented. Time histories and spatial distribution of pressure, velocity and temperature in water bodies, subjected to thermal discharge, are determined employing a digital computer. The dynamic response is obtained for a rectangular water body by applying a finite difference method to the mass, momentum and energy balance equations. These partial differential equations are algebraically manipulated to obtain; 1) three parabolic differential equations integrated temporally to find the horizontal velocity components and temperature; 2) one algebraic integral equation to get the vertical velocity component; 3) one elliptic differential equation integrated spatially to find the pressure; and 4) one differential equation to get the water level. Numerical stability criteria are developed which facilitate the selection of space and time increments for stable numerical integration.

The distinctive feature of this analysis, as compared to previous studies, is the calculation of pressure and water level from equations of motion without simplifying assumptions such as hydrostatic pressure approximation and rigid-lid concept.

The mathematical formulation is verified by applying this analysis to cases where the final steady state flow patterns have been determined analytically or experimentally by others. In particular, the final steady state solution obtained from this

dynamic analysis is verified with existing flow measurements of laminar flow development in a square duct. Furthermore, the natural circulation flows developed by this analysis are verified with known flow patterns in partially heated ponds.

The problem of thermal discharge entering a river with known initial velocity and temperature distribution is then analyzed. The time histories of the velocity and temperature distribution as well as the velocity and temperature profiles are obtained. These results provide the values of temperature rise and the rate of temperature rise needed for the assessment of the extent of thermal pollution in water bodies.

APPROVAL OF DISSERTATION

THERMAL STRATIFICATION AND CIRCULATION OF
WATER BODIES SUBJECTED TO THERMAL DISCHARGE

BY

MOHAMMAD A. BORHANI

FOR

DEPARTMENT OF MECHANICAL ENGINEERING
NEW JERSEY INSTITUTE OF TECHNOLOGY

BY

FACULTY COMMITTEE

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Newark, New Jersey

March, 1977

ACKNOWLEDGEMENT

The author wishes to acknowledge and express his gratitude to Dr. Amir N. Nahavandi, whose patience and persistence encouraged the author in the successful completion of this dissertation. His continued guidance, willingness to spend time with the student on research and his high standards have made him the best possible choice as an advisor.

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1. INTRODUCTION

Thermal power and industrial processing plants use large quantities of cooling water from natural or artificial water bodies and return the water to the source at higher temperatures. Changes that occur, either in the temperature of the water or in temperature-related water quality constituents, may adversely affect the aquatic environment of the water source. When the thermal discharge from a plant, into a water body, raises the water temperature by such an extent as to damage aquatic life or other legitimate uses of the water source, some degree of thermal pollution exists.

To assess the extent of a thermal pollution, utilities, ecologists, federal and state agencies as well as the public are interested in determining the space and time distribution of temperature within a water body, (such as lakes, reservoirs and rivers) when it is subjected to a prescribed thermal load.

The temperature distribution within a water impoundment is intimately associated with the hydrodynamic behavior of the water mass. Fluid flow in the impoundment is, in turn, affected by the hydrodynamic boundary conditions, three-dimensionality of the water body and the degree of stratification imposed on the water mass. The problem is further complicated by the heat transfer process through the water surface, internal mixing, inflows and outflows. For these reasons, a quantitative prediction of the temperature field for a particular water source must consider the mass-transport,

momentum-transport, and heat-transport processes which produce the temperature changes.

The theoretical analysis of the dynamic behavior of a large water body, subjected to thermal discharge, is based on the solution of space- and time-dependent conservation and state equations. Conservation of mass, momentum, and energy for the water flow in the impoundment together with the equation of state, provide a sufficient number of partial differential equations and algebraic relations in space and time for the solution of the problem. Although, these equations have been known for over a century, due to their highly nonlinear nature, a direct closed form analytical solution is considered practically impossible for most general cases involving realistic geometries and boundary conditions. The available analytical solutions of problems involving thermal discharge is severely limited by the many simplifying assumptions made in order to achieve a solution. In fact, analytical solutions to the problems of thermal discharge are obtained only for problems involving simple geometry and boundary conditions. Solutions to more realistic geometries and boundary conditions can be sought only by the development of computer modeling in three-dimensions under transient conditions.

Current three-dimensional computer simulation of thermal discharge into water bodies are based on a number of simplifying assumptions discussed hereunder:

1. Compliance with the Conservation of Mass and Momentum

Generally, the three components of water velocity are found from the longitudinal, lateral, and vertical momentum equations. These three components of velocity should satisfy the conservation of mass equation. A major difficulty arises when simultaneous compliance with the principles of conservation of mass and momentum cannot be ensured. Under the Boussinesq approximation, the conservation of mass reduces to zero divergence for the water velocity (with no mass storage term) i.e., the sum of all inflows to and outflows from any fluid control volume in the flow field must vanish. Since the new velocity components, calculated by the integrations of the momentum equations do not necessarily satisfy the continuity equation, the mass balance would be disturbed, thereby yielding a small surplus or deficit inflow at certain control volumes. To resolve this dilemma, Brady and Geyer (2)¹ hypothesize that the surplus or deficit inflows redistribute in all directions by equal magnitude. Obviously, the disadvantage of this velocity adjustment technique is that the adjusted velocities no longer satisfy the momentum equations. Considering a water body, such as a river, flowing longitudinally with a laterally uniform velocity distribution, and applying the approximate boundary conditions on the lateral velocities at the thermal discharge location, they found that the out-of-balance surplus inflow to each adjacent cell is exceptionally large. When their model attempts to distribute this surplus inflow among the surrounding

¹Underlined numerals in parenthesis designate references listed in Section 8.

fluid cells, the resulting vertical velocity adjustment becomes excessively large which aborts the run. Brady and Geyer referred to this problem as the "vertical over-responsiveness of the model" and devised schemes to suppress the associated undesirable vertical fluctuations by either: 1) temporarily suppressing vertical redistribution such that each fluid layer achieves its own mass balance independently; or 2) temporarily enforcing a rigid lid at the fluid free surface such that a zero vertical velocity is maintained at the most upper layer of the model at every fluid cell. The rigid lid assumption is used by many authors to simplify the solution to the stratification and circulation in water bodies (18, 20, 32). Obviously, this situation calls for an improved approach for a simultaneous compliance with the conservation equations of mass and momentum.

2. Surface Flow Distribution

The treatment of the slope variation at the water free surface constitutes another major difficulty. The upwelling (or downwelling) flow at the surface creates positive (or negative) surge waves (36, 10). The mathematical modeling of these phenomena under unsteady flow conditions is complicated. To resolve this difficulty, Waldrop and Farmer (39) first assumed that the vertical component of velocity near the surface will be redirected in the horizontal direction (in x and y directions) and the surface will move. Later, Waldrop and Farmer (40), in a major effort to resolve the surface flow distribution problem, introduced a mass balance at the free surface and extended their horizontal momentum equations to elements near the water

surface. This situation calls for further studies of the behavior of the free surface under unsteady flow conditions.

3. Pressure Distribution

The velocity components in the flow field are extremely sensitive to values of nodal pressures. Slight changes in the pressure distribution would affect the circulation in the water body considerably. The calculation of the correct pressure distribution is, therefore, of paramount importance. Most authors (41, 42, 19, 4, 16, 23, 8) assume that the total dynamic pressure at each point in the flow is equal to hydrostatic pressure obtained under static conditions. This assumption, referred to as hydrostatic approximation, leads to inaccuracies in regions of severe upwelling and downwelling which will adversely affect the correctness of circulation velocities. Thus, a better technique for the computation of the pressure field is called for.

4. Numerical Stability

Calculations of space and time increments for obtaining a stable numerical solution of partial differential equations of mass, momentum and energy conservation is a difficult task. Most authors resort to overly simplified criteria for the establishment of the upper bound of the integration time step for an assumed set of space increments. Brady and Geyer (2) state that the maximum integration step appears to be limited by relationships between the velocities and the smallest cell dimensions in each direction. Waldrop and Farmer (39)

and Harlow and Welch (9) resort to a stability limit calculated on the basis of one-dimensional, unsteady, incompressible flow equations with a free surface. This limit imposes an upper bound on their integration time step equal to the ratio of the smallest cell size and the surface wave speed. More accurate means for the prediction of the space and time increments are needed.

Most authors believe that much work is still required to refine various features of the present unsteady three-dimensional computer models before the present models can be applied to situations involving significant stratification effects, or flow fluctuations, in water bodies.

1.1 History and Background

Wind-induced circulation in shallow waters has been studied by Liu and Perez (22), Liggett and Hadjithodorou (18, 19, 20), among others. These studies neglect thermal stratification and employ unsteady momentum and continuity equations to determine the pressure and velocity distribution as functions of space and time in large water bodies. The problem of circulation in stratified lakes idealized as a two layer lake, in which epilimnion and hypolimnion are considered homogeneous has been analyzed by Liggett and Lee (15, 21, 10). These studies neglect the existence of the thermocline and are unrealistic.

The aforementioned studies do not take into account the coupled hydrometeorological phenomena controlling the disposition of energy at

the surface of water bodies and neglect the mechanisms that are responsible for distributing energy internally throughout the fluid mass.

Orlob and Selna (27) developed a digital computer program for the calculation of temperature variations in deep reservoirs. Their analysis is limited by the assumption that all transfer of heat energy is accomplished along the vertical axis. Stefan (35) presented a two-dimensional mathematical formulation consisting of continuity, momentum and energy equations for the modeling of heated water over lakes. However, he made no attempt to solve the governing equations.

Brady and Geyer (2) provided a comprehensive review of analytical modeling techniques for the solution of problems involving thermal discharges and presented a general computer model for simulating thermal discharge in three dimensions. They established the feasibility of their technique by a sample application to a river site where density effects are relatively insignificant. The compliance with the conservation of mass and momentum is obtained by redistributing the surplus or deficit inflows in a control volume in all directions by equal magnitudes. This feature was responsible for their observed "vertical over-responsiveness of the model" leading to excessive vertical velocities, as discussed earlier. They recommended further development work for situations involving significant stratification effects and fluctuating flows.

Waldrop and Farmer (39, 40), in a three dimensional study of

buoyant plumes, examined the flow field resulting from the interaction of a stream with the much larger body of flowing water into which it debouches. They expressed the local density of the water as function of salinity (39) and temperature (40). Three dimensional unsteady conservation equations, used in their study to describe the interaction, included the effects of buoyancy, inertia, and the difference in the hydrostatic heads of the two currents. They solved the conservation equations with a time-dependent finite difference technique. The treatment of the slope variation at the water free surface is accomplished as discussed earlier in surface flow distribution. The pressure distribution is calculated from the vertical momentum equations after neglecting the fluid vertical inertia.

Leendertse et al (16), in a study similar to that of Waldrop and Farmer (40), developed a three-dimensional model for estuaries, bays, and coastal seas in which non-isotropic density conditions exist. They performed a numerical integration of the finite difference equations of motion, transport, and continuity and applied their computational method to the analysis of a number of basins with boundaries of increasing complexity. They employ the hydrostatic approximation discussed earlier and neglect the vertical acceleration altogether.

Mercier (24, 25) developed a predictor-corrector method for the solution of incompressible viscous fluid and applied this method to the transient flow in a density stratified reservoir. His solution

is restricted to two-dimensional and adiabatic flow. Rather than considering the density as a function of temperature, Mercier obtains the water density by setting the material derivative of the density equal to zero.

Spradley and Churchill (34) examined the problem of pressure- and buoyancy-driven thermal convection in a rectangular enclosure for unsteady laminar compressible flow at various reduced levels of gravity. The enclosure was heated on one side and cooled on the opposite side. The conservation equations of mass, momentum, and energy were solved numerically for a compressible, heat-conducting ideal gas employing an explicit finite-difference technique. Their solution show that the thermally induced motion is acoustic in nature owing to the sonic character of the induced pressure waves.

Further reviews of the state of art for the solution of the conservation equations of mass, momentum, and energy as well as the study of the behavior of water bodies subjected to thermal discharge are available (28, 33, 30, 29) and interested readers are referred to these references for additional information.

A careful examination of the above references demonstrates that a completely adequate mathematical technique for the combined analysis of thermal stratification and circulation of water bodies is still lacking. A rigorous mathematical simulation for the prediction of the thermal and hydraulic behavior of impoundments is needed.

1.2 Scope and Objectives of the Present Study

The main objectives of the present study are as follows:

- 1) To develop a three-dimensional model for the mathematical description of a large rectangular water body free from the major shortcomings described earlier.
- 2) To develop a numerical integration technique for solving the above mathematical model on a digital computer.
- 3) To develop a digital computer program for predicting the thermal stratification and circulation phenomenon in a given rectangular water body subjected to thermal discharge and to determine the time histories and spatial distribution of pressure, velocity and temperature fields within the water body.

The major contributions of the present study, as compared with the previous works in this general area, can be summarized as follows:

- 1) Compliance with the conservation of mass and momentum is obtained by the introduction of two flow regions in the entire flow field: a) the water-level region containing a portion of water near the free surface in which the water-level rises or falls, as the case may be, during the dynamic solution; and b) the sub-water-level region located under the water-level region which remains totally filled with fluid at all times during the transient. As will be seen later, the superimposition of a three-dimensional grid on these two regions facilitates the simultaneous compliance of the principles of conservation of mass and momentum for the entire flow field without any mass imbalance.

2) A different set of differential equations for the conservation of mass, momentum and energy is applied to the cells located in the water-level and sub-water-level flow regions. This is necessary because the cells in the water-level region have a variable height while the cells in the sub-water level region have a constant height. This feature facilitates the calculation of surface flow distribution.

3) The pressure distribution is obtained by combining the momentum and continuity equations. This feature eliminates the need for the hydrostatic pressure approximation. It also eliminates the need for neglecting the acceleration terms in the momentum equations in the vertical direction.

4) A detailed numerical stability analysis is performed which provides accurate criteria for the selection of the space and time increments.

2. MATHEMATICAL FORMULATION

2.1 Simplifying Assumptions

The mathematical formulation in this study is based on the following assumptions:

- 1) The equations of motion, energy and continuity are applied in their time-smoothed form to turbulent incompressible three-dimensional flow with Cartesian coordinates, as shown in Figure 1, with z positive upward.
- 2) It is assumed that Boussinesq's approximation applies. In other words, densities are treated as constants except in terms involving gravity. This allows natural circulation to take place. Furthermore, density in the body force term is considered to be a function of temperature only.
- 3) The effects of turbulence are modeled by using eddy transport coefficients. Horizontal and vertical momentum eddy viscosities and thermal eddy diffusivities are considered as constants throughout the body of water though different magnitudes for horizontal and vertical directions.
- 4) It is assumed that there are no internal heat sources and the heat exchange between the water body and the atmosphere takes place near the water free surface.
- 5) Loss of mass due to evaporation at the surface and conductive heat transfer through the impoundment solid boundaries are generally small and are neglected.
- 6) The Coriolis forces acting on the water body is considered to

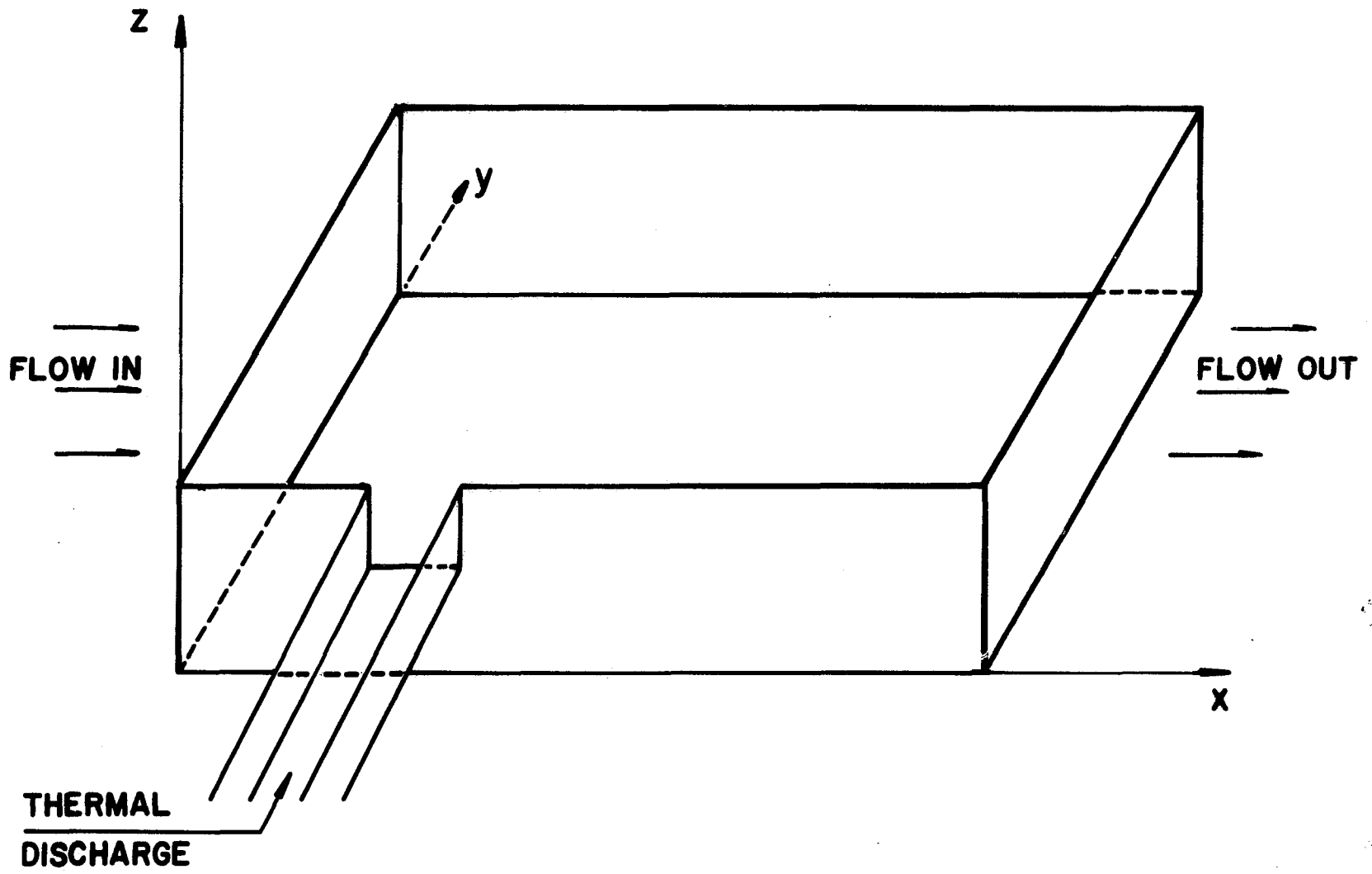


FIG I. SCHEMATIC DIAGRAM OF THE WATER BODY

be negligible.

The boundary conditions of the problem involve: 1) the geometry of the impoundment including the thermal discharge inlet, as well as the river inflow and outflow configurations; 2) the mass flow rates of the inflows across the boundaries; and 3) the level, pressure and temperature along the inlet boundaries. It is assumed that all of the above parameters are known.

The initial conditions of the problem includes the pressure, velocity, level and temperature distributions of the impoundment at time $t = 0$. These variables are computed by the present program by first performing a dynamic analysis under no thermal discharge conditions and then using the calculated pressure, velocity, level and temperature distributions, as initial conditions, in a dynamic analysis involving a thermal discharge.

2.2 Governing Equations

The mathematical formulation of the sub-water-level flow region consists of the following fundamental equations derived in Appendix 1.

Momentum equations:

$$\frac{\partial u}{\partial t} + u \frac{\partial u}{\partial x} + v \frac{\partial u}{\partial y} + w \frac{\partial u}{\partial z} = -\frac{1}{\rho_0} \frac{\partial p}{\partial x} + \nu_h \left(\frac{\partial^2 u}{\partial x^2} + \frac{\partial^2 u}{\partial y^2} \right) + \nu_v \frac{\partial^2 u}{\partial z^2} \quad (1)$$

$$\frac{\partial v}{\partial t} + u \frac{\partial v}{\partial x} + v \frac{\partial v}{\partial y} + w \frac{\partial v}{\partial z} = -\frac{1}{\rho_0} \frac{\partial p}{\partial y} + \nu_h \left(\frac{\partial^2 v}{\partial x^2} + \frac{\partial^2 v}{\partial y^2} \right) + \nu_v \frac{\partial^2 v}{\partial z^2} \quad (2)$$

$$\frac{\partial w}{\partial t} + u \frac{\partial w}{\partial x} + v \frac{\partial w}{\partial y} + w \frac{\partial w}{\partial z} = - \frac{1}{\rho_0} \frac{\partial p}{\partial z} + \nu_h \left(\frac{\partial^2 w}{\partial x^2} + \frac{\partial^2 w}{\partial y^2} \right) + \nu_v \frac{\partial^2 w}{\partial z^2} - g \frac{\rho}{\rho_0} \quad (3)$$

Continuity equation:

$$\frac{\partial u}{\partial x} + \frac{\partial v}{\partial y} + \frac{\partial w}{\partial z} = 0 \quad (4)$$

Energy equation:

$$\frac{\partial T}{\partial t} + u \frac{\partial T}{\partial x} + v \frac{\partial T}{\partial y} + w \frac{\partial T}{\partial z} = D_h \left(\frac{\partial^2 T}{\partial x^2} + \frac{\partial^2 T}{\partial y^2} \right) + D_v \frac{\partial^2 T}{\partial z^2} \quad (5)$$

Equation of state:

$$\rho = \rho(T) \quad (6)$$

In the above formulation, u , v and w are the local velocity components in the x , y and z directions respectively; p is the pressure; T is the local water temperature; ρ is the local density of water; ρ_0 is a fixed reference density to be defined later; and ν_h , ν_v , D_h and D_v are the horizontal and vertical components of eddy viscosity and eddy diffusivity of heat, respectively. The above differential equations are solved simultaneously, subject to the boundary conditions described in the following section. To solve the above mathematical formulation, the following steps will be taken:

1) Non-dimensionalization of the governing equations and derivation of time-derivative equations for velocity components u , v , w and temperature T .

2) Derivation of the final equations including the pressure equation.

In this study, the thermal discharge parameters are used as reference values for the non-dimensionalization of the governing equations. Let U_0 , ρ_0 , T_0 and d_0 be the velocity, density, temperature and half width of thermal discharge flow respectively. The dimensionless quantities are defined by

$$x^* = \frac{x}{d_0} \quad y^* = \frac{y}{d_0} \quad z^* = \frac{z}{d_0} \quad (7)$$

$$u^* = \frac{u}{U_0} \quad v^* = \frac{v}{U_0} \quad w^* = \frac{w}{U_0} \quad (8)$$

$$t^* = \frac{tU_0}{d_0} \quad T^* = \frac{T}{T_0} \quad p^* = \frac{p}{\rho_0 g d_0} \quad (9)$$

$$\rho^* = \frac{\rho}{\rho_0} \quad (10)$$

Employing the above dimensionless variables in the continuity, momentum and energy equations, one obtains the time derivatives for velocity components u , v , w and temperature T in parabolic differential form as follows:

$$\begin{aligned} \frac{\partial u^*}{\partial t^*} = & -u^* \frac{\partial u^*}{\partial x^*} - v^* \frac{\partial u^*}{\partial y^*} - w^* \frac{\partial u^*}{\partial z^*} - \frac{1}{F_0^2} \frac{\partial p^*}{\partial x^*} \\ & + \frac{v_h}{v_0} \frac{1}{R_0} \left(\frac{\partial^2 u^*}{\partial x^{*2}} + \frac{\partial^2 u^*}{\partial y^{*2}} \right) + \frac{v_v}{v_0} \frac{1}{R_0} \frac{\partial^2 u^*}{\partial z^2} \end{aligned} \quad (11)$$

$$\frac{\partial v^*}{\partial t^*} = -u^* \frac{\partial v^*}{\partial x^*} - v^* \frac{\partial v^*}{\partial y^*} - w^* \frac{\partial v^*}{\partial z^*} - \frac{1}{F_0^2} \frac{\partial p^*}{\partial y^*}$$

$$+ \frac{v_h}{v_0} \frac{1}{R_0} \left(\frac{\partial^2 v^*}{\partial x^{*2}} + \frac{\partial^2 v^*}{\partial y^{*2}} \right) + \frac{v_v}{v_0} \frac{1}{R_0} \frac{\partial^2 v^*}{\partial z^{*2}} \quad (12)$$

$$\frac{\partial w^*}{\partial t^*} = -u^* \frac{\partial w^*}{\partial x^*} - v^* \frac{\partial w^*}{\partial y^*} - w^* \frac{\partial w^*}{\partial z^*} - \frac{1}{F_0^2} \frac{\partial p^*}{\partial z^*} - \frac{\rho^*}{F_0^2}$$

$$+ \frac{v_h}{v_0} \frac{1}{R_0} \left(\frac{\partial^2 w^*}{\partial x^{*2}} + \frac{\partial^2 w^*}{\partial y^{*2}} \right) + \frac{v_v}{v_0} \frac{1}{R_0} \frac{\partial^2 w^*}{\partial z^{*2}} \quad (13)$$

$$\frac{\partial T^*}{\partial t^*} = -u^* \frac{\partial T^*}{\partial x^*} - v^* \frac{\partial T^*}{\partial y^*} - w^* \frac{\partial T^*}{\partial z^*}$$

$$+ \frac{D_h}{v_0} \frac{1}{R_0} \left(\frac{\partial^2 T^*}{\partial x^{*2}} + \frac{\partial^2 T^*}{\partial y^{*2}} \right) + \frac{D_v}{v_0} \frac{\partial^2 T^*}{\partial z^{*2}} \quad (14)$$

with continuity and state equations given by

$$\frac{\partial u^*}{\partial x^*} + \frac{\partial v^*}{\partial y^*} + \frac{\partial w^*}{\partial z^*} = 0 \quad (15)$$

$$\rho^* = \rho^*(T^*) \quad (16)$$

In the above equations, the following dimensionless numbers are used

$$1) \text{ Froude number } F_0 = U_0 / (gd_0)^{1/2} \quad (17)$$

$$2) \text{ Reynolds number } R_0 = U_0 d_0 / \nu_0 \quad (18)$$

Equations (11) to (16) are a set of six equations that will be used for the determination of six unknowns, u^* , v^* , w^* , p^* , ρ^* , and T^* , as functions of time and space.

The final equations, including the pressure equation, are obtained from the above equations as follows. First, the continuity equation is differentiated with respect to time to give

$$\frac{\partial}{\partial t} \left(\frac{\partial u^*}{\partial x} + \frac{\partial v^*}{\partial y} + \frac{\partial w^*}{\partial z} \right) = 0 \quad (19)$$

The order of differentiation in equation (19) is then interchanged

$$\frac{\partial}{\partial x} \left(\frac{\partial u^*}{\partial t} \right) + \frac{\partial}{\partial y} \left(\frac{\partial v^*}{\partial t} \right) + \frac{\partial}{\partial z} \left(\frac{\partial w^*}{\partial t} \right) = 0 \quad (20)$$

To facilitate spatial differentiations indicated by equation (20), equations (11) through (13) are first rewritten in the following form

$$\frac{\partial u^*}{\partial t} = Q_x^* - \frac{1}{F_0^2} \frac{\partial p^*}{\partial x} \quad (21)$$

$$\frac{\partial v^*}{\partial t} = Q_y^* - \frac{1}{F_0^2} \frac{\partial p^*}{\partial y} \quad (22)$$

$$\frac{\partial w^*}{\partial t} = Q_z^* - \frac{1}{F_0^2} \frac{\partial p^*}{\partial z} \quad (23)$$

where

$$Q_x^* = -u^* \frac{\partial u^*}{\partial x} - v^* \frac{\partial u^*}{\partial y} - w^* \frac{\partial u^*}{\partial z}$$

$$+ \frac{v_h}{v_0} \frac{1}{R_0} \left(\frac{\partial^2 u^*}{\partial x^{*2}} + \frac{\partial^2 u^*}{\partial y^{*2}} \right) + \frac{v_v}{v_0} \frac{1}{R_0} \frac{\partial^2 u^*}{\partial z^{*2}} \quad (24)$$

$$Q_y^* = -u^* \frac{\partial v^*}{\partial x^*} - v^* \frac{\partial v^*}{\partial y^*} - w^* \frac{\partial v^*}{\partial z^*} \\ + \frac{v_h}{v_0} \frac{1}{R_0} \left(\frac{\partial^2 v^*}{\partial x^{*2}} + \frac{\partial^2 v^*}{\partial y^{*2}} \right) + \frac{v_v}{v_0} \frac{1}{R_0} \frac{\partial^2 v^*}{\partial z^{*2}} \quad (25)$$

$$Q_z^* = -u^* \frac{\partial w^*}{\partial x^*} - v^* \frac{\partial w^*}{\partial y^*} - w^* \frac{\partial w^*}{\partial z^*} \\ + \frac{v_h}{v_0} \frac{1}{R_0} \left(\frac{\partial^2 w^*}{\partial x^{*2}} + \frac{\partial^2 w^*}{\partial y^{*2}} \right) + \frac{v_v}{v_0} \frac{1}{R_0} \frac{\partial^2 w^*}{\partial z^{*2}} - \frac{\rho^*}{F_0} \quad (26)$$

with

$$\frac{\partial w^*}{\partial z^*} = - \left(\frac{\partial u^*}{\partial x^*} + \frac{\partial v^*}{\partial z^*} \right) \quad (27)$$

substituting equations (21) through (23) into (20), yields

$$\nabla^2 p^* = F_0^2 \left(\frac{\partial Q_x^*}{\partial x^*} + \frac{\partial Q_y^*}{\partial y^*} + \frac{\partial Q_z^*}{\partial z^*} \right) \quad (28)$$

This elliptic differential equation will provide a means for the calculation of the pressure distribution in the water body. Equations (27), (24), (25), (26), (28), (21), (22), (14), and (16) constitute our final equations which will be solved for updated values of w^* , Q_x^* ,

Q_y^* , Q_z^* , p^* , u^* , v^* , T^* , and ρ^* , respectively. It should be noted that the vertical momentum equation (23) is used indirectly in this computation through its usage in the derivation of the pressure equation (28). Since the water-level variation provides a vertical freedom for the water body, it is imperative to calculate the vertical velocity component w^* from the continuity equation and not by the application of the momentum equation in the vertical direction equation (23). However, the momentum balance in the vertical direction is fully satisfied since equation (23) is explicitly used in the derivation of the pressure equation (28). In fact, equation (23) may be easily derived from the final equations stated above.

The mathematical formulation for the water-level flow region is derived by the application of the fundamental equations of mass, momentum, and energy as follows. The water-level in cells located at the air-water interface varies with time and space. For this reason, the continuity equation, as given by equation (27), is not valid for these cells. Application of a mass balance to such a cell with all sides fixed in space except the top which moves with the water-level, yields

$$\begin{aligned} \frac{\partial}{\partial t} (\rho \Delta x \Delta y H) &= (\rho u \Delta y H)_x - (\rho u \Delta y H)_{x + \Delta x} \\ &+ (\rho v \Delta x H)_y - (\rho v \Delta x H)_{y + \Delta y} \\ &+ (\rho w \Delta x \Delta y)_z - (\rho w \Delta x \Delta y)_{z + H} \end{aligned} \quad (29)$$

The last term represents the mass flux leaving the water-level surface. Since the evaporation losses are considered insignificant, this term is equal to zero. Dividing both sides of the above equation by $\rho\Delta x\Delta y$, considering ρ to be constant and equal to ρ_0 consistent with Boussinesq approximation used earlier, and letting Δx , Δy approach zero, yields

$$\frac{\partial H}{\partial t} = - \frac{\partial}{\partial x} (uH) - \frac{\partial}{\partial y} (vH) + w \quad (30)$$

The above equation will be integrated in time to obtain the water-level. In nondimensional form, the above equation becomes

$$\frac{\partial H^*}{\partial t^*} = - \frac{\partial}{\partial x^*} (u^* H^*) - \frac{\partial}{\partial y^*} (v^* H^*) + w^* \quad (31)$$

where

$$H^* = \frac{H}{d_0} \quad (32)$$

Similarly, the energy equation, as given by equation (14), is not valid for the water-level elements. Application of the energy balance to these cells yields

$$\begin{aligned} \frac{\partial}{\partial t} (\rho C_p \Delta x \Delta y H T) &= (\rho C_p u \Delta y H T)_x - (\rho C_p u \Delta y H T)_{x + \Delta x} \\ &+ (\rho C_p v \Delta x H T)_y - (\rho C_p v \Delta x H T)_{y + \Delta y} \\ &+ (\rho C_p w \Delta x \Delta y T)_z - (\rho C_p w \Delta x \Delta y T)_{z + H} \end{aligned}$$

$$\begin{aligned}
& + (-\rho C_p D_x \Delta y H \frac{\partial T}{\partial x})_x - (-\rho C_p D_x \Delta y H \frac{\partial T}{\partial x})_x + \Delta x \\
& + (-\rho C_p D_y \Delta x H \frac{\partial T}{\partial y})_y - (-\rho C_p D_y \Delta x H \frac{\partial T}{\partial y})_y + \Delta y \\
& + (-\rho C_p D_z \Delta x \Delta y \frac{\partial T}{\partial z})_z - (-\rho C_p D_z \Delta x \Delta y \frac{\partial T}{\partial z})_z + H
\end{aligned} \tag{33}$$

It should be noted again that the enthalpy flux leaving the water-level surface is equal to zero. Furthermore, the turbulent thermal diffusion at the water-level surface, is (3)

$$(\rho C_p D_z \frac{\partial T}{\partial z})_z + H = -K (T - E) \tag{34}$$

Considering the above statements, dividing energy equation (33) by $\rho C_p \Delta x \Delta y$ with $\rho = \rho_0$, and letting $\Delta x, \Delta y$ approach zero, yields

$$\begin{aligned}
\frac{\partial}{\partial t} (HT) &= \frac{\partial}{\partial x} (uHT) - \frac{\partial}{\partial y} (vHT) + wT \\
&+ D_x \frac{\partial}{\partial x} (H \frac{\partial T}{\partial x}) + D_y \frac{\partial}{\partial y} (H \frac{\partial T}{\partial y}) \\
&- D_z \frac{\partial T}{\partial z} - \frac{K (T - E)}{\rho_0 C_p}
\end{aligned} \tag{35}$$

Performing the differentiations indicated by the first three terms of the above, and employing equation (30) gives

$$\frac{\partial T}{\partial t} = -u \frac{\partial T}{\partial x} - v \frac{\partial T}{\partial y} + \frac{D_x}{H} \frac{\partial}{\partial x} (H \frac{\partial T}{\partial x})$$

$$+ \frac{D_y}{H} \frac{\partial}{\partial y} \left(H \frac{\partial T}{\partial y} \right) - \frac{D_z}{H} \frac{\partial T}{\partial z} - \frac{K(T - E)}{\rho_0 C_p H} \quad (36)$$

By a similar reasoning, after employing the surface boundary conditions, to be derived later, the velocity components in the water-level element is found to be

$$\begin{aligned} \frac{\partial u}{\partial t} = & -u \frac{\partial u}{\partial x} - v \frac{\partial u}{\partial y} - \frac{1}{\rho_0 H} \frac{\partial H p}{\partial x} + \frac{v_h}{H} \frac{\partial}{\partial x} \left(H \frac{\partial u}{\partial x} \right) \\ & + \frac{v_h}{H} \frac{\partial}{\partial y} \left(H \frac{\partial u}{\partial y} \right) - \frac{v_v}{H} \frac{\partial u}{\partial z} + \frac{1}{2} \frac{C_{fx} \rho_a W_x^2}{\rho_0 H} \end{aligned} \quad (37)$$

$$\begin{aligned} \frac{\partial v}{\partial t} = & -u \frac{\partial v}{\partial x} - v \frac{\partial v}{\partial y} - \frac{1}{\rho_0 H} \frac{\partial H p}{\partial y} + \frac{v_h}{H} \frac{\partial}{\partial x} \left(H \frac{\partial v}{\partial x} \right) \\ & + \frac{v_h}{H} \frac{\partial}{\partial y} \left(H \frac{\partial v}{\partial y} \right) - \frac{v_v}{H} \frac{\partial v}{\partial z} + \frac{1}{2} \frac{C_{fy} \rho_a W_y^2}{\rho_0 H} \end{aligned} \quad (38)$$

After setting $D_x = D_y = D_h$, and $D_z = D_v$, equations (36), (37), and (38) in non-dimensional form become

$$\begin{aligned} \frac{\partial T^*}{\partial t^*} = & -u^* \frac{\partial T^*}{\partial x^*} - v^* \frac{\partial T^*}{\partial y^*} + \frac{D_h}{v_0} \frac{1}{R_0 H^*} \left(\frac{\partial}{\partial x^*} \left(H^* \frac{\partial T^*}{\partial x^*} \right) \right. \\ & \left. + \frac{\partial}{\partial y^*} \left(H^* \frac{\partial T^*}{\partial y^*} \right) \right) - \frac{v_v}{v_0} \frac{1}{R_0 H^*} \frac{\partial T^*}{\partial z^*} - \frac{S_0}{H^*} (T^* - E^*) \end{aligned} \quad (39)$$

$$\frac{\partial u^*}{\partial t^*} = -u^* \frac{\partial u^*}{\partial x^*} - v^* \frac{\partial u^*}{\partial y^*} - \frac{1}{F_0^2 H^*} \frac{\partial H^* p^*}{\partial x^*}$$

$$\begin{aligned}
& + \frac{v_h}{v_0} \frac{1}{R_0 H^*} \left(\frac{\partial}{\partial x^*} \left(H^* \frac{\partial u^*}{\partial x^*} \right) + \frac{\partial}{\partial y^*} \left(H^* \frac{\partial u^*}{\partial y^*} \right) \right) \\
& - \frac{v_v}{v_0} \frac{1}{R_0 H^*} \frac{\partial u^*}{\partial z^*} + \frac{1}{2} \frac{C_{fx} \rho_a W_x^{*2}}{H^*}
\end{aligned} \tag{40}$$

$$\begin{aligned}
\frac{\partial v^*}{\partial t^*} &= - u^* \frac{\partial v^*}{\partial x^*} - v^* \frac{\partial v^*}{\partial y^*} - \frac{1}{F_0^2 H^*} \frac{\partial H^* p^*}{\partial y^*} \\
& + \frac{v_h}{v_0} \frac{1}{R_0 H^*} \left(\frac{\partial}{\partial x^*} \left(H^* \frac{\partial v^*}{\partial x^*} \right) + \frac{\partial}{\partial y^*} \left(H^* \frac{\partial v^*}{\partial y^*} \right) \right) \\
& - \frac{v_v}{v_0} \frac{1}{R_0 H^*} \frac{\partial v^*}{\partial z^*} + \frac{1}{2} \frac{C_{fy} \rho_a W_y^{*2}}{H^*}
\end{aligned} \tag{41}$$

where S_0 , the Stanton number, is defined by

$$S_0 = \frac{K}{\rho_0 C_p U_0} \tag{42}$$

and

$$\rho_a^* = \frac{\rho_a}{\rho_0} \tag{43}$$

$$W_x^* = \frac{W_x}{U_0} \quad W_y^* = \frac{W_y}{U_0} \tag{44}$$

The pressure at the air-water interface is always atmospheric and can be considered equal to zero no matter where the water-level may be at any instant of time. For this reason, the pressure at the center of the water-level element can be considered to be hydrostatic,

i.e., for the water-level element

$$P^* = \rho^* H^* \quad (45)$$

This equation may also be derived by applying a momentum balance, in the vertical direction, on the water-level flow region and noting that the water vertical velocity in this region is relatively small and, therefore, negligible. If the pressure distribution given by equation (45) is substituted into the horizontal momentum equations, the terms involving pressure gradient would simplify as follows

$$\frac{1}{F_0^2 H^*} \frac{\partial H^* P^*}{\partial x^*} = \frac{1}{F_0^2} \left(H^* \frac{\partial \rho^*}{\partial x^*} + 2\rho^* \frac{\partial H^*}{\partial x^*} \right) \quad (46)$$

$$\frac{1}{F_0^2 H^*} \frac{\partial H^* P^*}{\partial y^*} = \frac{1}{F_0^2} \left(H^* \frac{\partial \rho^*}{\partial y^*} + 2\rho^* \frac{\partial H^*}{\partial y^*} \right) \quad (47)$$

It is important to realize the difference between the flow patterns in the sub-water-level and the water-level elements. In the sub-water-level elements, an increase in the fluid boundary velocity will induce horizontal velocity components. However, the time constant associated with the momentum equations will not permit these equations to respond instantaneously to the boundary velocity disturbance at the river or thermal discharge inlets. Because of the flow incompressibility and since the fluid vertical inertia is considerably smaller than the horizontal inertias, the mass imbalance will result in a positive vertical fluid motion designated as "upwelling". This

vertical motion will continue upward until it reaches the water-level element where by virtue of equation (30) will cause the water-level to rise. This rise is large for elements close to the boundary disturbance and small for elements farther away. This spatial change affects the horizontal momentum equations (40) and (41) and induce a horizontal flow in the water-level elements which in turn tends to reduce the water-level through equation (30). A reverse situation takes place when the boundary velocity is decreased. The mass imbalance results in a negative vertical fluid motion, designated as "downwelling". This will cause the lowering of the water-level with an attendant horizontal flow into the element to increase the water-level.

2.3 Boundary Conditions

The water body boundary surfaces may be classified as follows:

Type (1): The surface boundary located at the water-level free surface.

Type (2): The lateral solid-fluid boundary located at the interface between the impoundment wall and the water body.

Type (3): The bottom boundary located at the bottom of the impoundment.

Type (4): The fluid-fluid boundary located at the interface between the water body in the region of interest and surrounding waters.

Implementation of the boundary conditions for the above boundary

surfaces involves calculation of the first spatial derivative of the system variables at the boundary. Two methods have been in usage for the calculation of the derivative:

- 1) Application of the reflection principle and usage of fictitious fluid nodes behind the solid boundary;
- 2) Calculation of the derivative at the boundary on the basis of three internal nodes adjacent to the boundary node.

The latter technique requires less computer storage space and is employed in the present study as detailed in the Appendix.

Pressure boundary conditions. For the surface boundary, the pressure is considered atmospheric at the air-water interface and is set equal to zero.

For the lateral solid-fluid and bottom boundaries, application of the momentum equation in the direction normal to the boundaries gives the following equations respectively:

$$\frac{\partial p^*}{\partial y^*} = F_0^2 \frac{v_h}{v_0} \frac{1}{R_0} \frac{\partial^2 v^*}{\partial y^{*2}} \quad (48)$$

$$\frac{\partial p^*}{\partial z^*} = -\rho^* + F_0^2 \frac{v_v}{v_0} \frac{1}{R_0} \frac{\partial^2 w^*}{\partial z^{*2}} \quad (49)$$

For the fluid-fluid boundaries, the pressure distribution at the inflow interfaces is considered to be static. The pressure distribution at any outflow interface node is set equal to that of the adjacent upstream node in the flow field. The surrounding waters

entering a water body is characterized by a uniform channel flow. The pressure variation at any cross section of a uniform channel flow may be proved to be hydrostatic by applying the Bernoulli's equation. This equation is applied to two streamlines, one at the channel top and the other at an arbitrary depth. Since, in steady, frictionless, one-dimensional, uniform channel flow, the constants in the Bernoulli's equation are the same for all streamlines, it can be easily shown that the pressure distribution across the channel, as well as the interface between the channel and the impoundment is hydrostatic.

Velocity boundary conditions. For the surface boundary, the wind effect produces shear at the water surface which is equal to the shear stress in the fluid near the surface.

$$\rho_0 \nu_v \left(\frac{\partial u}{\partial z} \right)_{z = z_s} = \frac{1}{2} \rho_a C_{fx} W_x |W_x| \quad (50)$$

$$\rho_0 \nu_v \left(\frac{\partial v}{\partial z} \right)_{z = z_s} = \frac{1}{2} \rho_a C_{fy} W_y |W_y| \quad (51)$$

where C_{fx} and C_{fy} are skin coefficients along x and y axis respectively; ρ_a and ρ_0 are the air and water densities; W_x and W_y are x and y components of the wind velocity; u and v are the fluid velocity components along x and y; and ν_v is the vertical eddy viscosity. Equations (50) and (51) are based on the assumption that the wind velocity components are steady and moderate with zero vertical component. Under these conditions, the water surface will remain

smooth and the skin coefficients may be taken from experimental measurements (22, 16). In dimensionless form, the equations (50) and (51) become

$$\frac{v_v}{v_0} \frac{1}{R_0} \left(\frac{\partial u^*}{\partial z^*} \right)_{z^* = z_s^*} = \frac{1}{2} C_{fx} \rho_a^* W_x^* \left| W_x^* \right| \quad (52)$$

$$\frac{v_v}{v_0} \frac{1}{R_0} \left(\frac{\partial v^*}{\partial z^*} \right)_{z^* = z_s^*} = \frac{1}{2} C_{fy} \rho_a^* W_y^* \left| W_y^* \right| \quad (53)$$

For the lateral solid-fluid and bottom boundaries, the velocity components normal to the boundaries is always zero. The tangential velocity components is calculated by the application of no-slip condition. With this condition, both tangential velocity components at the solid boundary are set equal to zero. The no-slip boundary condition is ideal for very fine grid spacing. However, computational economy requires a coarse grid spacing for most dynamic three-dimensional hydrothermal analysis. Since, a coarse grid does not afford sufficient resolution at the boundary, a limited amount of numerical error will be induced in the solution.

For the fluid-fluid boundaries, the velocity distribution at the inflow interfaces is set equal to the channel inflow distribution which is considered to be a known function of time. The velocity distribution at any outflow interface node is set equal to that of the adjacent upstream node in the flow field.

Temperature boundary condition. For the surface boundary, the rate of heat dispersion through the water (involving both conduction and turbulent thermal diffusion) is equal to the heat transfer from water to air.

$$\rho_0 C_p D_v \left(\frac{\partial T}{\partial z} \right)_{z = z_s} = -K (T_s - E) \quad (54)$$

where ρ_0 , C_p , and D_v are the density, specific heat, and eddy diffusivity of heat respectively; T_s is the water surface temperature; K is the heat exchange coefficient; and E is the equilibrium water temperature. Value of K and E may be estimated in terms of meteorological conditions (3). In dimensionless form, equation (54) becomes

$$\frac{D_v}{\nu_0} \frac{1}{R_0} \left(\frac{\partial T^*}{\partial z^*} \right)_{z^* = z_s^*} = -S_0 (T_s^* - E^*) \quad (55)$$

For the lateral solid-fluid and bottom boundaries, the heat transfer is considered to be negligible leading to zero temperature gradient at the boundary, i.e.,

$$\frac{\partial T^*}{\partial n} = 0 \quad (56)$$

The temperature distribution at the fluid-fluid boundaries is treated similar to the velocity distribution. The temperature at the inflow interface is set equal to the channel inflow temperature

which is considered to be a known function of time. The temperature at any outflow interface node is set equal to that of the adjacent upstream node in the flow field.

Water-level boundary condition. The water-level at the fluid-fluid boundaries is treated similar to the temperature distribution. The water-level at the inflow interface is set equal to the channel inflow which is considered to be known. The water-level at outflow interface nodes is set equal to that of the adjacent upstream nodes in the flow field. For the solid-fluid, the water level is calculated by simplifying the continuity equation for the water-level-region considering no-slip boundary conditions.

Boundary conditions on Q_x^* , Q_y^* and Q_z^* . In the development of pressure equation, values of $\partial Q_x^*/\partial x^*$, $\partial Q_y^*/\partial y^*$, and $\partial Q_z^*/\partial z^*$ should be evaluated in the flow field. It was shown earlier that variables Q_x^* , Q_y^* , and Q_z^* , are related to the velocity field by equations (24), (25), and (26). It should be realized that the only spatial derivatives of Q_x^* needed in the present analysis is $\partial Q_x^*/\partial x^*$. Spatial derivatives $\partial Q_x^*/\partial y^*$ and $\partial Q_x^*/\partial z^*$ are not required. This situation simplifies the analysis. Calculation of $\partial Q_x^*/\partial x^*$ in the fluid cell next to the boundary requires the calculation of Q_x^* at the boundaries normal to the x axis. Similarly, calculation of $\partial Q_y^*/\partial y^*$ and $\partial Q_z^*/\partial z^*$ require the calculation of Q_y^* and Q_z^* at the boundaries normal to the the y and z axes respectively.

For the surface boundary, only Q_z^* is needed. As stated earlier,

since the vertical velocity component in the water-level element is small, equation (26) gives

$$(Q_z^*)_{z = z_s^*} = -\frac{\rho^*}{F_0^*} z^* \quad (57)$$

For the lateral solid-fluid boundaries, only the component of Q^* normal to the solid boundary is required. Since the velocity component normal to the solid boundary is zero, the component Q^* normal to the solid boundary can be obtained after simplifying equation (25) as follows:

$$(Q_y^*)_{y^* = 0} = (Q_y^*)_{y = y_{\max}^*} = \frac{v_h}{v_0} \frac{1}{R_0} \frac{\partial^2 v^*}{\partial y^{*2}} \quad (58)$$

For the bottom boundary only Q_z^* is needed. Simplification of equation (26) at the bottom boundary gives

$$(Q_z^*)_{z^* = 0} = -\frac{\rho^*}{F_0^*} z^* + \frac{v_v}{v_0 R_0} \left(\frac{\partial^2 w^*}{\partial z^{*2}} \right) \quad (59)$$

For the fluid-fluid boundary, the surrounding waters entering the water body are characterized by a uniform channel flow, with a uniform velocity at any cross section and zero axial pressure gradient. As an example, application of the momentum equation along the channel flowing in the x-direction gives

$$(Q_x^*)_{x^* = 0} = 0 \quad (60)$$

Furthermore, at the outflow interface nodes, the values of Q_x^* are set equal to that of the adjacent upstream nodes in the flow field.

3. NUMERICAL SOLUTION

The numerical scheme, used in this study, is mainly based on the application of spatial and temporal finite difference technique to the governing equation described in the previous section. This scheme is divided into the following steps:

- 1) Initialization and setting of boundary conditions;
- 2) Calculation of vertical velocity components;
- 3) Calculation of pressure distribution;
- 4) Calculation of the time derivatives for the horizontal velocity components and temperature in the elements of the sub-water-level region;
- 5) Calculation of the time derivative for the water-level, horizontal velocity components and temperature in the elements of the water-level region;
- 6) Time integration and updating of the horizontal velocity components and temperature in the sub-water-level region;
- 7) Calculation of the water-level, velocity components, and temperature in the water-level region;
- 8) Calculation of density distribution.

After the completion of the last step, the problem time is incremented, the computational procedure is repeated starting at the second step, and the integration cycle is continued until the final problem time is reached.

The initial conditions of the problem consist of the initial

values of temperature, velocity components, and pressure at time $t = 0$. These quantities are most easily obtained by the application of the present dynamic analysis. To demonstrate this application, let us consider that it is desired to determine the dynamic response of a water body when it is subjected to a prescribed thermal discharge. The solution to the problem can be obtained in two steps. In the first step, the dynamic response is obtained by setting the thermal discharge temperature equal to the river inflow temperature. Maintaining the atmospheric and the inflow conditions unchanged, the water body will reach equilibrium conditions and the pressure, velocity and temperature distribution will be obtained. In the second step, these equilibrium conditions are entered as the initial conditions and the thermal discharge temperature is set equal to the prescribed value. Again, if the atmospheric and inflow conditions are held unchanged, the water body will reach new equilibrium conditions. An examination of this dynamic response will enable the program user to predict the effects of the thermal discharge on the state of the water body. The results of this study could then be used to determine if the state's allowable temperature standards have been violated. These standards generally vary from state to state. Many states impose a maximum allowable water temperature, a temperature rise, and a rate of temperature rise on the use of water bodies.

From the temporal viewpoint, the boundary conditions stated earlier are of two types -- time-independent and time-dependent. The

setting of the time-independent boundary conditions are undertaken prior to the beginning of the dynamic solution. The time-dependent boundary conditions are satisfied during the execution of the integration cycle. For example, the time-dependent boundary conditions on vertical velocity and pressure are satisfied during the execution of steps 2 and 3 stated above respectively.

3.1 Calculation of the Vertical Velocity Components

As discussed earlier, the vertical component of velocity w^* is calculated from the continuity equation (27). This equation can be used to determine the vertical velocity component w^* , once the horizontal velocity components u^* and v^* are known throughout the flow field. For the first step of numerical integration, the latter quantities as initialized will be used to calculate the velocity component w^* . For the numerical integration steps higher than the first, the updated values of u^* and v^* will be available by the time integration of the momentum equations in the x and y directions to be discussed later.

To facilitate the writing of finite difference equations, to be presented in this section, it is convenient to introduce the following conventions:

1) Variables defining any quantity within an element i,j,k will be written without subscripts, for example, $u_{i,j,k}^*$ will be written as u^* .

2) Variables defining any quantity within an element adjacent

to element i,j,k will be written with the subscript that is different from i,j,k , for example $u_{i+1,j,k}^*$ or $u_{i,j-1,k}^*$ will be written as u_{i+1}^* or u_{j-1}^* respectively.

The finite difference form of $\partial u^*/\partial x^*$ and $\partial v^*/\partial y^*$, using a parabolic representation (three points with unequal spacing), are given in the Appendix 3.

Hence, expressing the right hand side equation (27) in finite difference form, yields:

$$\begin{aligned} \frac{\partial w^*}{\partial z^*} = & a_x \frac{u_{i+1}^* - u_{i-1}^*}{x_{i+1}^* - x_{i-1}^*} + b_x \frac{u_{i+1}^* - u_{i-1}^*}{x_{i+1}^* - x_{i-1}^*} \\ & + a_y \frac{v_{j+1}^* - v_{j-1}^*}{y_{j+1}^* - y_{j-1}^*} + b_y \frac{v_{j+1}^* - v_{j-1}^*}{y_{j+1}^* - y_{j-1}^*} \end{aligned} \quad (61)$$

where

$$a_x = \frac{x_{i+1}^* - x_{i-1}^*}{x_{i+1}^* - x_{i-1}^*} \quad (62)$$

$$b_x = \frac{x_{i+1}^* - x_{i-1}^*}{x_{i+1}^* - x_{i-1}^*} \quad (63)$$

$$a_y = \frac{y_{j+1}^* - y_{j-1}^*}{y_{j+1}^* - y_{j-1}^*} \quad (64)$$

and

$$b_y = \frac{y_{j+1}^* - y^*}{y_{j+1}^* - y_{j-1}^*} \quad (65)$$

Starting from the bottom boundary, equation (59) is then integrated along z to obtain the vertical velocity component in each element.

3.2 Calculation of Pressure Distribution

The updated values of pressure in the flow field are found from equations (24), (25), (26), and (28) by applying a centered three-point finite difference for the first spatial derivatives of u^* , v^* , w^* , Q_x^* , Q_y^* , and Q_z^* as well as for the second spatial derivatives of p^* , u^* , v^* , and w^* . A variable mesh size finite difference scheme is used as detailed in Appendix 3. Equation (28) becomes

$$\begin{aligned} c_x \left(\frac{p_{j+1}^* - p^*}{x_{i+1}^* - x^*} - \frac{p^* - p_{i-1}^*}{x^* - x_{i-1}^*} \right) + c_y \left(\frac{p_{j+1}^* - p^*}{y_{j+1}^* - y^*} - \frac{p^* - p_{j-1}^*}{y^* - y_{j-1}^*} \right) \\ + c_z \left(\frac{p_{k+1}^* - p^*}{z_{k+1}^* - z^*} - \frac{p^* - p_{k-1}^*}{z^* - z_{k-1}^*} \right) = \xi \end{aligned} \quad (66a)$$

where

$$\begin{aligned} \xi = F_0^2 \left(a_x \frac{(Q_x^*)_{i+1} - Q_x^*}{x_{i+1}^* - x^*} + b_x \frac{Q_x^* - (Q_x^*)_{i-1}}{x^* - x_{i-1}^*} \right. \\ \left. + a_y \frac{(Q_y^*)_{j+1} - Q_y^*}{y_{j+1}^* - y^*} + b_y \frac{Q_y^* - (Q_y^*)_{j-1}}{y^* - y_{j-1}^*} \right) \end{aligned} \quad (66b)$$

$$+ a_z \frac{(Q_z^*)_{k+1} - Q_z^*}{z_{k+1}^* - z^*} + b_z \frac{Q_z^* - (Q_z^*)_{k-1}}{x^* - z_{k-1}^*}$$

$$a_x = (x^* - x_{i-1}^*) / (x_{i+1}^* - x_{i-1}^*)$$

$$b_x = (x_{i+1}^* - x^*) / (x_{i+1}^* - x_{i-1}^*)$$

$$c_x = 2 / (x_{i+1}^* - x_{i-1}^*)$$

$$a_y = (y^* - y_{j-1}^*) / (y_{j+1}^* - y_{j-1}^*)$$

$$b_y = (y_{j+1}^* - y^*) / (y_{j+1}^* - y_{j-1}^*)$$

$$c_y = 2 / (y_{j+1}^* - y_{j-1}^*)$$

$$a_z = (z^* - z_{k-1}^*) / (z_{k+1}^* - z_{k-1}^*)$$

$$b_z = (z_{k+1}^* - z^*) / (z_{k+1}^* - z_{k-1}^*)$$

$$c_z = 2 / (z_{k+1}^* - z_{k-1}^*)$$

(67)

$$Q_x^* = -u^* \left(a_x \frac{u_{i+1}^* - u^*}{x_{i+1}^* - x^*} + b_x \frac{u^* - u_{i-1}^*}{x^* - x_{i-1}^*} \right)$$

$$\begin{aligned}
& - v^* \left(a_y \frac{u_{j+1}^* - u^*}{y_{j+1}^* - y^*} + b_y \frac{u^* - u_{j-1}^*}{y^* - y_{j-1}^*} \right) \\
& - w^* \left(a_z \frac{u_{k+1}^* - u^*}{z_{k+1}^* - z^*} + b_z \frac{u^* - u_{k-1}^*}{z^* - z_{k-1}^*} \right) \\
& + \frac{v_h}{v_0} \frac{1}{R_0} \left(c_x \left(\frac{u_{i+1}^* - u^*}{x_{i+1}^* - x^*} - \frac{u^* - u_{i-1}^*}{x^* - x_{i-1}^*} \right) \right. \\
& \left. + c_y \left(\frac{u_{j+1}^* - u^*}{y_{j+1}^* - y^*} - \frac{u^* - u_{j-1}^*}{y^* - y_{j-1}^*} \right) \right) \\
& + \frac{v_v}{v_0 R_0} c_z \left(\frac{u_{k+1}^* - u^*}{z_{k+1}^* - z^*} - \frac{u^* - u_{k-1}^*}{z^* - z_{k-1}^*} \right)
\end{aligned} \tag{68}$$

$$\begin{aligned}
Q_y^* &= - u^* \left(a_x \frac{v_{i+1}^* - v^*}{x_{i+1}^* - x^*} + b_x \frac{v^* - v_{i-1}^*}{x^* - x_{i-1}^*} \right) \\
& - v^* \left(a_y \frac{v_{j+1}^* - v^*}{y_{j+1}^* - y^*} + b_y \frac{v^* - v_{j-1}^*}{y^* - y_{j-1}^*} \right) \\
& - w^* \left(a_z \frac{v_{k+1}^* - v^*}{z_{k+1}^* - z^*} + b_z \frac{v^* - v_{k-1}^*}{z^* - z_{k-1}^*} \right) \\
& + \frac{v_h}{v_0} \frac{1}{R_0} \left(c_x \left(\frac{v_{i+1}^* - v^*}{x_{i+1}^* - x^*} - \frac{v^* - v_{i-1}^*}{x^* - x_{i-1}^*} \right) \right.
\end{aligned}$$

$$\begin{aligned}
& + c_y \left(\frac{v_{j+1}^* - v^*}{y_{j+1}^* - y^*} - \frac{v^* - v_{j-1}^*}{y^* - y_{j-1}^*} \right) \\
& + \frac{v_v}{v_0 R_0} c_z \left(\frac{v_{k+1}^* - v^*}{z_{k+1}^* - z^*} - \frac{v^* - v_{k-1}^*}{z^* - z_{k-1}^*} \right)
\end{aligned} \tag{69}$$

and

$$\begin{aligned}
Q_z^* &= -u^* \left(a_x \frac{w_{i+1}^* - w^*}{x_{i+1}^* - x^*} + b_x \frac{w^* - w_{i-1}^*}{x^* - x_{i-1}^*} \right) \\
& - v^* \left(a_y \frac{w_{j+1}^* - w^*}{y_{j+1}^* - y^*} + b_y \frac{w^* - w_{j-1}^*}{y^* - y_{j-1}^*} \right) \\
& - w^* \left(a_z \frac{w_{k+1}^* - w^*}{z_{k+1}^* - z^*} + b_z \frac{w^* - w_{k-1}^*}{z^* - z_{k-1}^*} \right) \\
& + \frac{v_h}{v_0 R_0} \left(c_x \left(\frac{w_{i+1}^* - w^*}{x_{i+1}^* - w^*} - \frac{w^* - w_{i-1}^*}{x^* - x_{i-1}^*} \right) \right. \\
& \left. + c_y \left(\frac{w_{j+1}^* - w^*}{y_{j+1}^* - y^*} - \frac{w^* - w_{j-1}^*}{y^* - y_{j-1}^*} \right) \right) \\
& + \frac{v_v}{v_0 R_0} c_z \left(\frac{w_{k+1}^* - w^*}{z_{k+1}^* - z^*} - \frac{w^* - w_{k-1}^*}{z^* - z_{k-1}^*} \right) - \frac{1}{F_0^2} \rho^*
\end{aligned} \tag{70}$$

Numerical values for Q_x^* , Q_y^* , and Q_z^* at any element i, j, k needed for the right hand side of equation (66b) are provided by equation (68), (69), and (70) respectively. Numerical values for $(Q_x^*)_{i+1}$, $(Q_x^*)_{i-1}$, $(Q_y^*)_{j+1}$, $(Q_y^*)_{j-1}$, $(Q_z^*)_{k+1}$, and $(Q_z^*)_{k-1}$, also needed for the right hand side of equation (66b) are provided by the same equations, after applying these equations to the entire flow field from the first cell ($i=1, j=1, k=1$) to the last ($i=i_{\max}, j=j_{\max}, k=k_{\max}$).

Defining

$$\begin{aligned}
 e_x &= \frac{c_x}{x_{i+1}^* - x_i^*} & f_x &= \frac{c_x}{x_i^* - x_{i-1}^*} \\
 e_y &= \frac{c_y}{y_{j+1}^* - y_j^*} & f_y &= \frac{c_y}{y_j^* - y_{j-1}^*} \\
 e_z &= \frac{c_z}{z_{k+1}^* - z_k^*} & f_z &= \frac{c_z}{z_k^* - z_{k-1}^*}
 \end{aligned} \tag{71}$$

Equation (66a) becomes

$$\begin{aligned}
 e_x (p_{i+1}^* - p^*) - f_x (p^* - p_{i-1}^*) + \\
 e_y (p_{j+1}^* - p^*) - f_y (p^* - p_{j-1}^*) + \\
 e_z (p_{k+1}^* - p^*) - f_z (p^* - p_{k-1}^*) = \xi
 \end{aligned} \tag{72}$$

Within one small time increment, values $e_x, e_y, e_z, f_x, f_y, f_z,$ and ξ are constants and equation (72) constitutes a system of linear algebraic equations of p^* which can be solved simultaneously. An iterative method, based on a modified Gauss-Seidel iteration technique, called successive over-relaxation is employed to solve the above system of equations. This method basically consists of solving equation (72) for p^* and designating that by p_t^* .

$$p_t^* = \frac{e_x p_{i+1}^* + f_x p_{i-1}^* + e_y p_{j+1}^* + f_y p_{j-1}^* + e_z p_{k+1}^* + f_z p_{k-1}^* - \xi}{e_x + f_x + e_y + f_y + e_z + f_z} \quad (73)$$

and then calculating the actual pressure distribution employing a weighted average of this value and the value of pressure found from the preceding iteration such that

$$p_{\ell+1}^* = \omega (p_t^*)_{\ell+1} + (1 - \omega) p_{\ell}^* \quad (74)$$

where

p_{ℓ}^* = pressure in cell i, j, k at the preceding iteration ℓ

$(p_t^*)_{\ell+1}$ = pressure in cell i, j, k calculated from equation (73) at the present iteration $\ell+1$

$p_{\ell+1}^*$ = actual pressure in cell i, j, k calculated from equation (74) at the present iteration $\ell+1$

ω = relaxation factor to be discussed later in this section

The pressure calculation, employing equation (74) is performed for every cell in the flow field and old values of p_{ℓ}^* are replaced by $p_{\ell+1}^*$ as soon as the latter value is calculated for any cell. The iterative procedure, involving the calculation of the pressure for the entire flow field, is repeated until values of pressure at every cell converge within a prescribed error. It should be noted that for $\omega = 1$, this approach becomes identical with the Gauss-Seidel iterative method. The quantity ω , designated as relaxation factor varies between 0 and 2. For values of $\omega < 1$, the iterative method is referred to as underrelaxation and is usually employed to make a nonconvergent iterative process converge. For values of $1 < \omega < 2$, the iterative method is referred to as overrelaxation and is ordinarily used to accelerate an already convergent iterative process. In this study, successive overrelaxation procedure is used and value of ω is chosen between one and two.

The optimum value of ω for finite difference method applied to the elliptic differential equation, involved in this study, is found (11, 6, 37) from

$$\omega_{\text{opt}} = \frac{2}{1 + (1 - \lambda^2)^{1/2}} \quad (75)$$

with

$$\lambda^2 = \lim \frac{||Y_{\ell+1}||}{||Y_{\ell}||} \quad (76)$$

and

$$\| Y_{\ell + 1} \| = \sum_{i=1}^{i_{\max}} \sum_{j=1}^{j_{\max}} \sum_{k=1}^{k_{\max}} \left| (p_t^*)_{\ell + 1} - (p_t^*)_{\ell} \right| \quad (77)$$

where

ω_{opt} = optimum value of overrelaxation factor

λ^2 = norms ratio

$\| Y_{\ell + 1} \|$ = norm of the pressure at the present iteration $\ell + 1$

$\| Y_{\ell} \|$ = norm of the pressure at the preceding iteration ℓ

The above procedure indicates that the calculation of the optimum value of overrelaxation factor is based on the ratio of norms at the present iteration $\ell + 1$ and the preceding iteration as given by equations (76) and (77) using Gauss-Seidel iterative scheme ($\omega = 1$). After having found the optimum value of ω , the numerical scheme employs the modified Gauss-Seidel technique with ω set equal to the optimum value. A question now arises as to how many Gauss-Seidel iterations should be performed before switching to the modified Gauss-Seidel scheme. If the number of iterations ℓ is too few, the value of λ^2 may become larger than unity which will not result in an optimum value for ω . On the other hand, if the number of iterations ℓ are too many, the value of λ^2 will approach unity ($\omega_{\text{opt}} = 2$). In this case, the converged values of pressure would be obtained by using the uneconomical Gauss-Seidel iterations. One effective procedure to solve this problem is to take enough Gauss-Seidel iterations to ensure that the estimate of λ^2 has settled down to a reasonably constant level less than one,

and then employ equation (75) to estimate ω_{opt} . This estimate of ω_{opt} is then used in equation (74) and the numerical solution iterated to convergence using this overrelaxation factor.

From the foregoing analysis, it is evident that the optimum value of ω depends on the pressure distribution which is a time-dependent function. For this reason, the optimum value of ω also becomes time-dependent and should be calculated once during each integration step. However, since values of ω will only affect the speed of convergence and not the accuracy of the solution, it is not mandatory to compute the optimum value of ω at each integration step. A considerable amount of computer time will be saved if one calculates the optimum value of ω for the first integration step and use that value for the following integration steps. Experience with this numerical approach has indicated that since the time variation of pressure in the flow field is not excessively large, the optimum value of ω found in the first integration step provides a near optimum value for the following steps and the calculation of ω_{opt} for all integration steps may be eliminated at a considerable cost saving.

3.3 Calculation of the Time Derivative for the Horizontal Velocity Components and Temperature

Time derivatives of horizontal velocity components and temperature in the flow field are found from equations (21), (22), and (14) by applying a centered three point finite difference for the first spatial derivative of p^* and T^* as well as the second spatial deriva-

tive of T^* . A variable mesh size finite difference scheme is used as detailed in Appendix 3.

$$\frac{\partial u^*}{\partial t^*} = Q_x^* - \frac{1}{F_0^2} \left(a_x \frac{p_{i+1}^* - p^*}{x_{i+1}^* - x^*} + b_x \frac{p^* - p_{i-1}^*}{x^* - x_{i-1}^*} \right) \quad (78)$$

$$\frac{\partial v^*}{\partial t^*} = Q_y^* - \frac{1}{F_0^2} \left(a_y \frac{p_{j+1}^* - p^*}{y_{j+1}^* - y^*} + b_y \frac{p^* - p_{j-1}^*}{y^* - y_{j-1}^*} \right) \quad (79)$$

$$\begin{aligned} \frac{\partial T^*}{\partial t^*} = & -u^* \left(a_x \frac{T_{i+1}^* - T^*}{x_{i+1}^* - x^*} + b_x \frac{T^* - T_{i-1}^*}{x^* - x_{i-1}^*} \right) \\ & -v^* \left(a_y \frac{T_{j+1}^* - T^*}{y_{j+1}^* - y^*} + b_y \frac{T^* - T_{j-1}^*}{y^* - y_{j-1}^*} \right) \\ & -w^* \left(a_z \frac{T_{k+1}^* - T^*}{z_{k+1}^* - z^*} + b_z \frac{T^* - T_{k-1}^*}{z^* - z_{k-1}^*} \right) \\ & + \frac{D_h}{v_0} \frac{1}{R_0} \left(c_x \left(\frac{T_{i+1}^* - T^*}{x_{i+1}^* - x^*} - \frac{T^* - T_{i-1}^*}{x^* - x_{i-1}^*} \right) \right. \\ & \left. + c_y \left(\frac{T_{j+1}^* - T^*}{y_{j+1}^* - y^*} - \frac{T^* - T_{j-1}^*}{y^* - y_{j-1}^*} \right) \right) \\ & + \frac{D_v}{v_0} \frac{1}{R_0} c_z \left(\frac{T_{k+1}^* - T^*}{z_{k+1}^* - z^*} - \frac{T^* - T_{k-1}^*}{z^* - z_{k-1}^*} \right) \end{aligned} \quad (80)$$

where Q_x^* , Q_y^* , a_x , b_x , a_y , b_y , a_z , b_z , c_x , c_y , and c_z , are previously

defined by equations (68), (69), and (67).

3.4 Calculation of the Time Derivative for the Water -Level, Horizontal Velocity Components and Temperature in the Water-Level Elements

Time-derivatives of the water-level horizontal velocity components and temperature in the water-level region are found from equations (31), (41), (40), and (39), respectively, by applying a centered three-point finite difference for the first and second spatial derivatives of H^* , u^* , v^* , and T^* . A variable mesh size finite difference scheme is used as detailed in Appendix 3.

$$\begin{aligned} \frac{\partial u^*}{\partial t} = & - u^* \left(a_x \frac{u_{i+1}^* - u^*}{x_{i+1}^* - x^*} + b_x \frac{u^* - u_{i-1}^*}{x^* - x_{i-1}^*} \right) \\ & - v^* \left(a_y \frac{u_{j+1}^* - u^*}{y_{j+1}^* - y^*} + b_y \frac{u^* - u_{j-1}^*}{y^* - y_{j-1}^*} \right) \\ & - \frac{H^*}{F_0^2} \left(a_x \frac{\rho_{i+1}^* - \rho^*}{x_{i+1}^* - x^*} + b_x \frac{\rho^* - \rho_{i-1}^*}{x^* - x_{i-1}^*} \right) \\ & + \left(a_x \frac{H_{i+1}^* - H^*}{x_{i+1}^* - x^*} + b_x \frac{H^* - H_{i-1}^*}{x^* - x_{i-1}^*} \right) , \\ & \left(\frac{v_h}{v_0} \frac{1}{R_0 H^*} \left(a_x \frac{u_{i+1}^* - u^*}{x_{i+1}^* - x^*} + b_x \frac{u^* - u_{i-1}^*}{x^* - x_{i-1}^*} \right) - \frac{2\rho^*}{F_0^2} \right) \end{aligned}$$

$$\begin{aligned}
& + \frac{v_h}{v_0} \frac{1}{R_0} c_x \left(\frac{u_{i+1}^* - u^*}{x_{i+1}^* - x^*} - \frac{u^* - u_{i-1}^*}{x^* - x_{i-1}^*} \right) \\
& + \frac{v_h}{v_0} \frac{1}{R_0 H^*} \left(a_y \frac{H_{j+1}^* - H^*}{y_{j+1}^* - y^*} + b_y \frac{H^* - H_{j-1}^*}{y^* - y_{j-1}^*} \right) \cdot \\
& \quad \left(a_y \frac{u_{j+1}^* - u^*}{y_{j+1}^* - y^*} + b_y \frac{u^* - u_{j-1}^*}{y^* - y_{j-1}^*} \right) \\
& + \frac{v_h}{v_0} \frac{1}{R_0} c_y \left(\frac{u_{j+1}^* - u^*}{y_{j+1}^* - y^*} - \frac{u^* - u_{j-1}^*}{y^* - y_{j-1}^*} \right) \\
& - \frac{v_v}{v_0} \frac{1}{R_0 H^*} \left(\frac{u^* - u_{k-1}^*}{z^* - z_{k-1}^*} \right) + \frac{1}{2H^*} c_{fx} \rho_a^* W_x^{*2} \tag{81}
\end{aligned}$$

$$\begin{aligned}
\frac{\partial v^*}{\partial t^*} & = -u^* \left(a_x \frac{v_{i+1}^* - v^*}{x_{i+1}^* - x^*} + b_x \frac{v^* - v_{i-1}^*}{x^* - x_{i-1}^*} \right) \\
& - v^* \left(a_y \frac{v_{j+1}^* - v^*}{y_{j+1}^* - y^*} + b_y \frac{v^* - v_{j-1}^*}{y^* - y_{j-1}^*} \right) \\
& - \frac{H^*}{F_0^2} \left(a_y \frac{\rho_{j+1}^* - \rho^*}{y_{j+1}^* - y^*} + b_y \frac{\rho^* - \rho_{j-1}^*}{y^* - y_{j-1}^*} \right) \\
& + \left(a_y \frac{H_{j+1}^* - H^*}{y_{j+1}^* - y^*} + b_y \frac{H^* - H_{j-1}^*}{y^* - y_{j-1}^*} \right) \cdot
\end{aligned}$$

$$\left(\frac{v_h}{v_0} \frac{1}{R_0 H^*} \left(a_y \frac{v_{j+1}^* - v^*}{y_{j+1}^* - y^*} + b_y \frac{v^* - v_{j-1}^*}{y^* - y_{j-1}^*} \right) - \frac{2\rho^*}{F_0} \right)$$

$$+ \frac{v_h}{v_0} \frac{1}{R_0} c_x \left(\frac{v_{i+1}^* - v^*}{x_{i+1}^* - x^*} - \frac{v^* - v_{i-1}^*}{x^* - x_{i-1}^*} \right)$$

$$+ \frac{v_h}{v_0} \frac{1}{R_0 H^*} \left(a_x \frac{H_{i+1}^* - H^*}{x_{i+1}^* - x^*} + b_x \frac{H^* - H_{i-1}^*}{x^* - x_{i-1}^*} \right) \cdot$$

$$\left(a_x \frac{v_{i+1}^* - v^*}{x_{i+1}^* - x^*} + b_x \frac{v^* - v_{i-1}^*}{x^* - x_{i-1}^*} \right)$$

$$+ \frac{v_h}{v_0} \frac{1}{R_0} c_y \left(\frac{v_{j+1}^* - v^*}{y_{j+1}^* - y^*} - \frac{v^* - v_{j-1}^*}{y^* - y_{j-1}^*} \right)$$

$$- \frac{v_v}{v_0} \frac{1}{R_0 H^*} \left(\frac{v^* - v_{k-1}^*}{z^* - z_{k-1}^*} \right) + \frac{1}{2H^*} c_{fy} \rho_a^* W_y^{*2}$$

(82)

$$\frac{\partial T^*}{\partial t^*} = -u^* \left(a_x \frac{T_{i+1}^* - T^*}{x_{i+1}^* - x^*} + b_x \frac{T^* - T_{i-1}^*}{x^* - x_{i-1}^*} \right)$$

$$- v^* \left(a_y \frac{T_{j+1}^* - T^*}{y_{j+1}^* - y^*} + b_y \frac{T^* - T_{j-1}^*}{y^* - y_{j-1}^*} \right)$$

$$\begin{aligned}
& + \frac{D_h}{v_0} \frac{1}{R_0 H^*} \left(a_x \frac{H_{i+1}^* - H^*}{x_{i+1}^* - x^*} + b_y \frac{H^* - H_{i-1}^*}{x^* - x_{i-1}^*} \right) \cdot \\
& \left(a_x \frac{T_{i+1}^* - T^*}{x_{i+1}^* - x^*} + b_x \frac{T^* - T_{i-1}^*}{x^* - x_{i-1}^*} \right) \\
& + \frac{D_h}{v_0} \frac{1}{R_0} c_x \left(\frac{T_{i+1}^* - T^*}{x_{i+1}^* - x^*} - \frac{T^* - T_{i-1}^*}{x^* - x_{i-1}^*} \right) \\
& + \frac{D_h}{v_0} \frac{1}{R_0 H^*} \left(a_y \frac{H_{j+1}^* - H^*}{y_{j+1}^* - y^*} + b_y \frac{H^* - H_{j-1}^*}{y^* - y_{j-1}^*} \right) \cdot \\
& \left(a_y \frac{T_{j+1}^* - T^*}{y_{j+1}^* - y^*} + b_y \frac{T^* - T_{j-1}^*}{y^* - y_{j-1}^*} \right) \\
& + \frac{D_h}{v_0} \frac{1}{R_0} c_y \left(\frac{T_{j+1}^* - T^*}{y_{j+1}^* - y^*} - \frac{T^* - T_{j-1}^*}{y^* - y_{j-1}^*} \right) \\
& - \frac{D_v}{v_0} \frac{1}{R_0 H^*} \left(\frac{T^* - T_{k-1}^*}{z^* - z_{k-1}^*} \right) - \frac{S_0}{H^*} (T^* - E^*) \tag{83}
\end{aligned}$$

$$\begin{aligned}
\frac{\partial H^*}{\partial t} & = -u^* \left(\frac{H^* - H_{i-1}^*}{x^* - x_{i-1}^*} \right) - v^* \left(\frac{H^* - H_{j-1}^*}{y^* - y_{j-1}^*} \right) \\
& - H^* \left(\frac{u^* - u_{i-1}^*}{x^* - x_{i-1}^*} + \frac{v^* - v_{j-1}^*}{y^* - y_{j-1}^*} \right) + W_{k-1}^* \tag{84}
\end{aligned}$$

It should be noted that a backward spatial differencing method is employed in equation (84) to ensure a conditional stability of this equation.

3.5 Calculations of the Horizontal Velocity Components and Temperature in the Sub-Water-Level-Region

The updated values of horizontal velocity components and temperature in the sub-water-level region are found by applying a forward two-point finite difference for the first temporal derivatives.

$$u^{*n} = u^* + \Delta t^* \left(\frac{\partial u^*}{\partial t^*} \right) \quad (85)$$

$$v^{*n} = v^* + \Delta t^* \left(\frac{\partial v^*}{\partial t^*} \right) \quad (86)$$

$$T^{*n} = T^* + \Delta t^* \left(\frac{\partial T^*}{\partial t^*} \right) \quad (87)$$

where superscript n represents the updated values, Δt^* is time integration step, and values $(\partial u^*/\partial t^*)$, $(\partial v^*/\partial t^*)$, and $(\partial T^*/\partial t^*)$ are calculated from equations (78), (79), and (80).

3.6 Calculation of the Water-Level, Velocity Components and Temperature in the Water-Level Element

The updated values of the water-level, velocity components and temperature in the water-level region are found by applying a forward two-point finite difference for the first temporal derivatives.

$$u^{*n} = u^* + \Delta t^* \left(\frac{\partial u^*}{\partial t^*} \right) \quad (88)$$

$$v^{*n} = v^* + \Delta t^* \left(\frac{\partial v^*}{\partial t^*} \right) \quad (89)$$

$$T^{*n} = T^* + \Delta t^* \left(\frac{\partial T^*}{\partial t^*} \right) \quad (90)$$

$$H^{*n} = H^* + \Delta t^* \left(\frac{\partial H^*}{\partial t^*} \right) \quad (91)$$

The values of $(\partial u^*/\partial t^*)$, $(\partial v^*/\partial t^*)$, $(\partial T^*/\partial t^*)$, and $(\partial H^*/\partial t^*)$ are calculated from equations (81), (82), (83), and (84).

3.7 Calculation of Density Distribution

The fluid density is considered to be a function of temperature only. The dependence of density on pressure, within the range of the variation of pressure in this study, is extremely weak and is, therefore, neglected. The functional relationship between the density and temperature is defined in tabular form as shown in columns 1 and 2 of Table 1 and is designated by

$$\rho = \rho (T) \quad (92)$$

in which T is the fluid temperature in degrees Fahrenheit, and ρ is the density in lbm/ft^3 . In non-dimensional form, equation (92) becomes

$$\rho^* = \rho^* (T^*) \quad (93)$$

as exemplified in columns 3 and 4 of Table 1 with temperatures normalized to 90°F .

TABLE 1. DENSITY VERSUS TEMPERATURE

No.	T OF	$\rho \left(\frac{\text{lbm}}{\text{ft}^3} \right)$	Dimensionless	
			T	ρ
1	70	62.30296	.7777777	1.0030153
2	72	62.29876	.8000000	1.0028028
3	74	62.27426	.8222222	1.0025533
4	76	62.25487	.8444444	1.0022411
5	78	62.23938	.8666666	1.0019917
6	80	62.22002	.8888888	1.0016800
7	82	62.20065	.9111111	1.0013682
8	84	62.18132	.9333333	1.0010570
9	86	62.16199	.9555555	1.0007458
10	88	62.13882	.9777777	1.0003728
11	90	62.11566	1.0000000	1.0000000
12	92	62.09251	1.0222222	.99962730
13	94	62.06940	1.0444444	.99925526
14	96	62.04628	1.0666666	.99888305
15	98	62.02319	1.0888888	.998511325
16	100	61.99628	1.1111111	.99807810
17	102	61.96939	1.1333333	.997645199
18	104	61.94252	1.1555555	.99721262
19	106	61.91566	1.1777777	.9967802
20	108	61.88885	1.2000000	.9963485
21	110	61.86205	1.2222222	.9959171
22	112	61.83145	1.2444444	.9954245
23	114	61.80470	1.2666666	.99499385
24	116	61.77415	1.2888888	.9945020
25	118	61.74364	1.3111111	.9940108
26	120	61.71315	1.3333333	.9935199

4. NUMERICAL STABILITY AND CONVERGENCE

The numerical procedure, presented in the previous section, involves spatial integrations for the calculation of the vertical velocity component and pressure, as well as time integrations for the calculation of the horizontal velocity components, temperature and water-level. These spatial and time integration procedures are plagued by numerical stability problems caused by round-off errors in the numerical solution and convergence problems caused by the truncation error due to the finite magnitude of the space and time increments. The numerical solution is said to be stable if round-off errors remain either constant or decrease as the solution to the difference equations is carried out. The numerical solution is termed convergent if, as the space or time increment approaches zero, the numerical solution approaches the exact solution to the differential equations. Using these definitions, stability is a necessary condition for convergence and in this sense the problems of stability and convergence are interdependent.

To achieve numerical stability in nonlinear numerical solutions, the governing system of differential equations must first be linearized and the absolute value of the largest eigenvalue of the amplification matrix for the system of differential equations should be made less or equal to unity (1, 26). This condition will provide an upper bound on the integration step size that ensures the numerical stability of the solution. The computation of the eigenvalues of a large

system of differential equations is very laborious and time consuming. It constitutes a problem of the same order of difficulty as the problem to be solved. Fortunately, an exact calculation of the upper bound of integration step size is not necessary. A conservative estimate of the upper bound of integration step size is all that is required. The calculation of a conservative estimate for spatial and time steps will be based on the following arguments:

- 1) In general, numerical instabilities develop locally in one variable, grow with increasing time and adversely affect other variables through coupling terms in the governing differential equations. If the differential equation for every variable is numerically stable in any small region of the domain, without feedback from the others, instability will not have its origin within the domain and, therefore, will not occur. Employing this hypothesis, each of the differential equations involved in the analysis will be examined independently and their stability criterion will be determined.

- 2) Experience with the computational properties of various numerical schemes for the solution of the heat and mass transport phenomena has led to a simple rule (3) based on the fact that water temperature is generally a non-negative quantity. This rule, which applies to an explicit computational scheme, is that the coefficients multiplying each of the temperature values should be positive quantities. This rule, used herein for the numerical stability analysis of the differential equations for the horizontal velocity components and temperature, is substantiated in Appendix 2.

4.1 Stability Analysis of Horizontal Momentum and Energy Equations

As discussed earlier, the horizontal momentum and energy equations are solved explicitly.

Using the above-mentioned rule on these explicit solutions, one obtains a set of stability conditions which must be satisfied, otherwise the solutions will not converge. The stability analysis is demonstrated for energy equation only. In view of similarity between the momentum and energy equations, the same results are applicable to the horizontal momentum equations. After linearizing and expressing the energy equation in an explicit finite difference form in terms of a small variation in temperature, as exemplified in Appendix 2, one obtains:

$$\begin{aligned}
 \dot{T}^n &= \dot{T}_{i+1} \left(-a_x \frac{u\Delta t}{x_{i+1}-x} + c_x \frac{D_h \Delta t}{x_{i+1}-x} \right) \\
 &+ \dot{T} \left(1 + a_x \frac{u\Delta t}{x_{i+1}-x} - b_x \frac{u\Delta t}{x-x_{i-1}} + a_y \frac{v\Delta t}{y_{j+1}-y} - b_y \frac{v\Delta t}{y-y_{j-1}} \right. \\
 &+ a_z \frac{w\Delta t}{z_{k+1}-z} - b_z \frac{w\Delta t}{z-z_{k-1}} - c_x \frac{D_h \Delta t}{x_{i+1}-x} - c_x \frac{D_h \Delta t}{x-x_{i-1}} - c_y \frac{D_h \Delta t}{y_{j+1}-y} \\
 &\left. - c_y \frac{D_h \Delta t}{y-y_{j-1}} - c_z \frac{D_v \Delta t}{z_{k+1}-z} - c_z \frac{D_v \Delta t}{z-z_{k-1}} \right) \\
 &+ \dot{T}_{i-1} \left(b_x \frac{u\Delta t}{x-x_{i-1}} + c_x \frac{D_h \Delta t}{x-x_{i-1}} \right)
 \end{aligned}$$

$$\begin{aligned}
& + \dot{T}_{j+1} \left(-a_y \frac{v\Delta t}{y_{j+1}-y} - c_y \frac{D_h\Delta t}{y_{j+1}-y} \right) \\
& + \dot{T}_{j-1} \left(b_y \frac{v\Delta t}{y-y_{j-1}} + c_y \frac{D_h\Delta t}{y-y_{j-1}} \right) \\
& + \dot{T}_{k+1} \left(-a_z \frac{w\Delta t}{z_{k+1}-z} + c_z \frac{D_v\Delta t}{z_{k+1}-z} \right) \\
& + \dot{T}_{k-1} \left(b_z \frac{w\Delta t}{z-z_{k-1}} + c_z \frac{D_v\Delta t}{z-z_{k-1}} \right)
\end{aligned} \tag{94}$$

where superscript (\cdot) indicates a small variation in temperature.

Setting the temperature coefficients equal to positive value and substituting for a_x , b_x , c_x , a_y , b_y , c_y , a_z , b_z , and c_z from equations (67) yields

$$x - x_{i-1} < \frac{2D_h}{|u|}$$

$$x_{i+1} - x < \frac{2D_h}{|u|}$$

$$y - y_{j-1} < \frac{2D_h}{|v|}$$

$$y_{j+1} - y < \frac{2D_h}{|v|}$$

$$\begin{aligned}
 z - z_{k-1} &\leq \frac{2D_v}{|w|} \\
 z_{k+1} - z &\leq \frac{2D_v}{|w|}
 \end{aligned} \tag{95}$$

and

$$\Delta t \leq 1 / \left(\frac{2D_h}{(x - x_{i-1})^2} + \frac{2D_h}{(y - y_{j-1})^2} + \frac{2D_v}{(z - z_{k-1})^2} \right) \tag{96}$$

The increments of the nodal coordinates may be expressed in terms of the elements mesh size Δx , Δy , and Δz by

$$\begin{aligned}
 x - x_{i-1} &= \frac{1}{2} (\Delta x + \Delta x_{i-1}) \\
 y - y_{j-1} &= \frac{1}{2} (\Delta y + \Delta y_{j-1}) \\
 z - z_{k-1} &= \frac{1}{2} (\Delta z + \Delta z_{k-1})
 \end{aligned} \tag{97}$$

Substituting equations (97) into equations (95) and (96) results in

$$\begin{aligned}
 \Delta x + \Delta x_{i-1} &\leq \frac{4D_h}{|u|} \\
 \Delta y + \Delta y_{j-1} &\leq \frac{4D_h}{|v|} \\
 \Delta z + \Delta z_{k-1} &\leq \frac{4D_v}{|w|}
 \end{aligned}$$

$$\Delta t \leq 1 / \left(\frac{8D_h}{(\Delta x + \Delta x_{i-1})^2} + \frac{8D_h}{(\Delta y + \Delta y_{j-1})^2} + \frac{8D_v}{(\Delta z + \Delta z_{k-1})^2} \right) \quad (98)$$

Since the energy equation has the same form as the momentum equations, equations (97) and (98) constitute the numerical stability criteria for the energy and horizontal momentum equations.

4.2 Convergence of Pressure Equation

As discussed in Section 3.2, the pressure equation provides a system of linear equations, which is solved by the modified Gauss-Seidel iterative method designated as successive over-relaxation. A sufficient condition for the convergence of this iterative process is given by Isaacson and Keller (12) and Williams (43) as follows:

$$\Delta = \frac{\sum_{m=1}^{m'} |a_{rm}|}{|a_{rr}|} \leq 1 \quad (99)$$

where a_{rr} is the value of the diagonal terms of matrix and a_{rm} is the value of the off-diagonal terms, with m' notation signifying that the value of a_{rr} is omitted from the summation. The convergence of the numerical solution is ensured if $\Delta < 1$ for at least one equation in the system of linear pressure equations, given by equations (72). An examination of the later equations shows that for nodal points within the fluid field $\Delta = 1$ except for nodal points just below the water-level region or near other boundaries for which pressure is known. The known value of pressure causes a transfer of the corre-

sponding term to the right hand side of equation (72) leading to $\Delta < 1$ which is a sufficient condition for convergence.

4.3 Stability Analysis of Water-Level Equation

The stability criterion for the water-level equation is obtained by linearizing equation (31), as exemplified in Appendix 2 for the energy equation, as follows:

$$\frac{\dot{\partial H}}{\partial t} = -u \frac{\dot{\partial H}}{\partial x} - v \frac{\dot{\partial H}}{\partial y} \quad (100)$$

where superscript (.) indicates small variation about steady-state. In this study, equation (100) is expressed in a finite difference form, by applying a forward two point finite difference approximation to the temporal derivative and backward finite difference approximation to first spatial derivatives respectively. Applying Von-Neumann stability analysis, shown in Appendix 2, to the discretized equation accounting for variable mesh size, one obtains:

$$e^{\delta \Delta t} = 1 - \frac{u \Delta t}{x - x_{i-1}} (1 - e^{-i \alpha \Delta x_{i-1}}) - \frac{v \Delta t}{y - y_{j-1}} (1 - e^{-i \beta \Delta y_{j-1}}) \quad (101)$$

The necessary and sufficient condition for equation (101) to be stable is that the absolute value of $e^{\delta \Delta t}$ must be less than or equal to one.

After expressing the exponential terms in trigonometric forms and calculating the absolute value of $e^{\delta\Delta t}$, equation (101) reduces to

$$\begin{aligned}
 |e^{\delta\Delta t}| &= (1 - 4 (1 - \frac{u\Delta t}{x-x_{i-1}} - \frac{v\Delta t}{y-y_{j-1}}) \frac{u\Delta t}{x-x_{i-1}} \sin^2 \frac{\alpha\Delta x_{i-1}}{2} \\
 &\quad - 4 (1 - \frac{u\Delta t}{x-x_{i-1}} - \frac{v\Delta t}{y-y_{j-1}}) \frac{v\Delta t}{y-y_{j-1}} \sin^2 \frac{\beta\Delta y_{j-1}}{2} \\
 &\quad - 4 (\frac{u\Delta t}{x-x_{i-1}}) (\frac{v\Delta t}{y-y_{j-1}}) \sin^2 \frac{\alpha\Delta x_{i-1} - \beta\Delta y_{j-1}}{2}) \leq 1 \quad (102)
 \end{aligned}$$

In the above inequality, the maximum values of the trigonometric terms are equal to one. Therefore, further examination shows that this inequality will be satisfied if

$$1 - \frac{u\Delta t}{x-x_{i-1}} - \frac{v\Delta t}{y-y_{j-1}} \geq 0 \quad (103)$$

To satisfy the above stability conditions for negative values of the velocity components, one must have

$$\Delta t \leq 1/(|u|/(x-x_{i-1}) + |v|/(y-y_{j-1})) \quad (104)$$

Expressing equation (104) in terms of Δx and Δy as shown in Section (4.1), one obtains

$$\Delta t \leq 1/(2|u|/(\Delta x + \Delta x_{i-1}) + 2|v|/(\Delta y + \Delta y_{j-1})) \quad (105)$$

It should be noted that estimates of the largest values of the velocity components would suffice for the calculation of conservative values of space and time increments.

5. PRESENTATION OF RESULTS

The mathematical formulation and the computer program, developed in this study, is verified by applying the dynamic analysis to a number of cases where the final steady-state solution is either known experimentally or the final steady-state flow pattern has been determined by other investigators. Employing the present program, the fluid flow problem considered is brought to steady-state condition through a dynamic analysis. The initial values of thermal and hydraulic variables in the flow field, needed for this purpose, were selected in such a way to facilitate the entry of input data. Starting from these rough initial values and maintaining the appropriate boundary conditions, the present program is employed to bring the flow problem to steady-state. These steady-state solutions are then compared either with the experimental work or the known flow patterns. These verification studies are considered in the next section.

5.1 Verification Studies

The following problems were selected for verification purposes:

- 1) Experimental studies on the laminar flow development in a square duct (7)
- 2) Natural circulation in a pond partially heated from the bottom or the side.

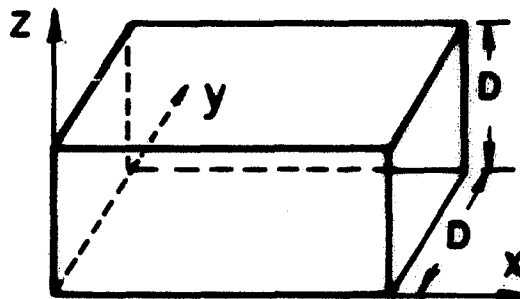
The first study consists of a duct with a square cross section with a spatially and temporally uniform input velocity at the duct

entrance. Since the geometric dimensions of the laboratory model used in the experimental studies were small and our interest here is in water bodies of appreciable size, the similarity between the present and the experimental study is based on two dimensionless parameters: 1) the quotient of aspect ratio (x/D) to Reynolds number; and 2) the ratio of the fluid velocity to the average inlet velocity. The geometric and hydraulic input data for this case is given in Table 2 and Fig. 2a. The computational grid used in this dynamic analysis consists of $12 \times 5 \times 5$ mesh points with non-uniform spacing in the longitudinal direction and uniform spacing in the square cross section as shown in Table 3. The initial values of the velocity components in the flow field were selected uniform in the longitudinal direction equal to the inlet velocity with zero transverse and vertical velocities. The dynamic solution was continued until the flow pattern in the duct reached steady-state. Typical time histories for a number of upstream and a downstream points along the duct axial central plane are obtained and used to plot the final steady-state centerline and vertical velocity profiles in the square duct. These velocity profiles, in non-dimensional form, are compared in Figs. 2b and 2c with the experimental results of Goldstein and Kreid (37) who measured the laminar flow development in a square duct using laser-doppler flowmeter. This comparison shows that the results of this study are in good agreement with the experimental data. This confirms that the present formulation can predict the three dimensional flow behavior of water bodies.

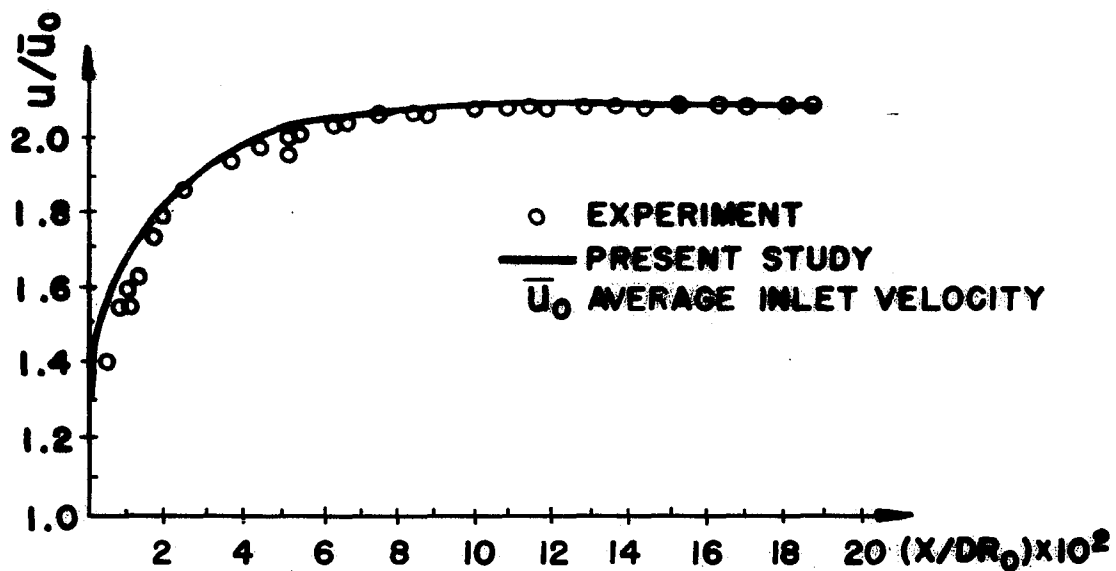
The second study consists of natural circulation in a pond par-

TABLE 2. THE GEOMETRIC AND HYDRAULIC INPUT DATA
FOR LAMINAR FLOW IN A SQUARE DUCT

<u>Specifications</u>	<u>Dimensions</u>	
	<u>British Unit</u>	<u>SI Unit</u>
Water Body Length	280 ft.	85.34 m
Water Body Width	72 ft.	21.946 m
Water Body Depth	72 ft.	21.946 m
Inlet Water Velocity	1 ft/sec	0.3048 m/sec
Water Temperature	75°F	23.89 °C
Reynolds Number, R_0	20.83	20.83
Hydraulic Diameter of Square Duct, D	72 ft.	21.946 m



(a) THE SQUARE DUCT CONFIGURATION



(b) CENTER-LINE VELOCITY DEVELOPMENT

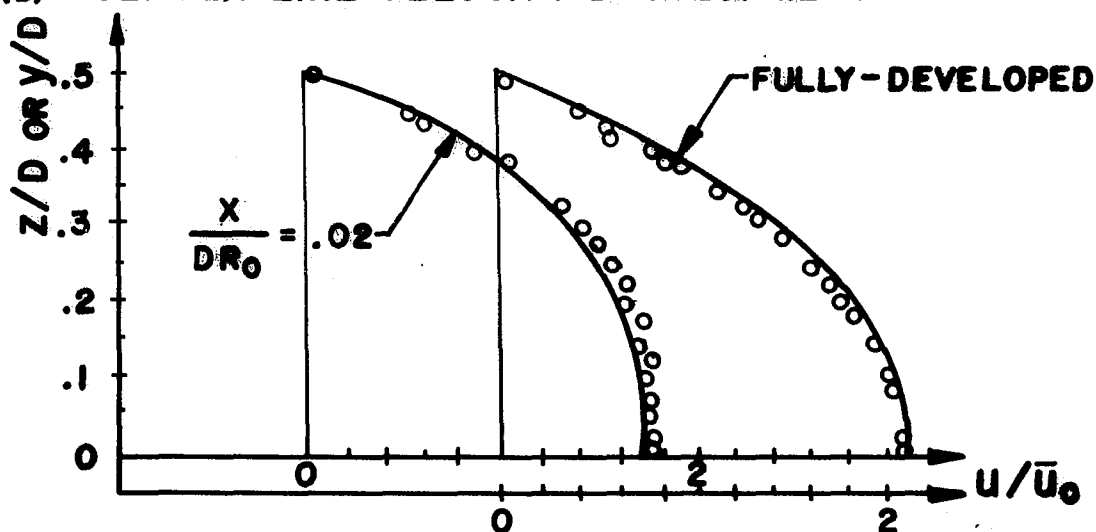
(c) DEVELOPMENT OF VELOCITY PROFILE
CENTRAL PLANE

FIG 2. VELOCITY DEVELOPMENT IN A SQUARE DUCT

TABLE 3. GRID DIMENSIONS FOR THE LAMINAR FLOW
IN A SQUARE DUCT

Element No.	<u>x</u>		<u>y</u>		<u>z</u>	
	<u>ft</u>	<u>m</u>	<u>ft</u>	<u>m</u>	<u>ft</u>	<u>m</u>
1	10	3.048	18	5.486	18	5.486
2	10	3.048	18	5.486	18	5.486
3	10	3.048	18	5.486	18	5.486
4	10	3.048	18	5.486	18	5.486
5	20	6.096	18	5.486	18	5.486
6	30	9.144				
7	30	9.144				
8	30	9.144				
9	30	9.144				
10	30	9.144				
11	30	9.144				
12	30	9.144				
13	30	9.144				

tially heated from the bottom or the side with uniform initial temperature, zero initial velocities and zero inlet and outlet mass flux. The geometric and hydraulic input data for this case is given in Table 4. The computational grid used in the dynamic analysis consists of $10 \times 6 \times 6$ mesh points with uniform spacing in the longitudinal and transversal directions and non-uniform spacing in the vertical direction as shown in Table 5. Two cases were examined. In these cases, a step temperature is applied to a portion of side wall or the bottom surface of the pond. The dynamic response of the pond is obtained for each case and the natural circulation flows developed are compared with expected flow patterns. Typical natural circulation flow patterns for partially heated side wall and bottom surface are shown in Figs. 3, 4, 5, and 6 at times $t = 40$ and $t = 25$ seconds, respectively, into the transient. An examination of these flow patterns demonstrates that for the case of partially heated side wall one natural circulation vortex in transversal direction and two symmetric natural circulation vortices in the longitudinal direction are developed which lose their strength as the distance from the heated wall increases (see Figs. 3 and 4). For the case of partially heated bottom surface two symmetric natural vortices in vertical longitudinal and vertical diagonal planes are developed which lose their strength with increasing distance from the heated surface (see Figs. 5 and 6). In both cases, described above, the natural circulation vortices formed and the resultant mixing are caused by density differences in the water body. These results establish that the present formulation can predict the three dimensional flow and thermal

TABLE 4. GEOMETRIC AND HYDRAULIC INPUT
DATA FOR PARTIALLY HEATED POND

<u>Specifications</u>	<u>Dimensions</u>	
	<u>British Unit</u>	<u>SI Unit</u>
Water Body Length	540 ft	164.592 m
Water Body Width	250 ft	76.2 m
Water Body Depth	20.5 ft	6.248 m
Water Body Temperature	75 °F	23.89 °C
Water Body Equilibrium Temperature	74 °F	23.33 °C
Temperature of Heated Area	100 °F	37.77 °C
Reference Velocity, U_0	1 ft/sec	0.3048 m/sec
Reference Length, d_0	50 ft	15.24 m
Reference Temperature, T_0	100 °F	37.77 °C
Reference Time, t_0	50 sec	50 sec
Reference Density, ρ_0	61.9963 lbm/ft ⁻³	960.590 kg m ⁻³
Reference Pressure, p_0	21.52 lbf in ⁻²	14639.4 kg m ⁻²
Reynolds Number, R_0	0.6061 x 10 ⁷	0.6061 x 10 ⁷
Heat Exchange Coefficient, K	100 Btu/ft ² day °F	23.64 W/m ² °C

TABLE 5. GRID DIMENSIONS FOR PARTIALLY HEATED POND

Element No.	<u>x</u>		<u>y</u>		<u>z</u>	
	<u>ft.</u>	<u>m</u>	<u>ft.</u>	<u>m</u>	<u>ft.</u>	<u>m</u>
1	60	18.29	50	15.24	3.5	1.07
2	60	18.29	50	15.24	4.	1.22
3	60	18.29	50	15.24	5	1.52
4	60	18.29	50	15.24	4	1.22
5	60	18.29	50	15.24	4	1.22
6	60	18.29	50	15.24	3.5	1.07
7	60	18.29				
8	60	18.29				
9	60	18.29				
10	60	18.29				

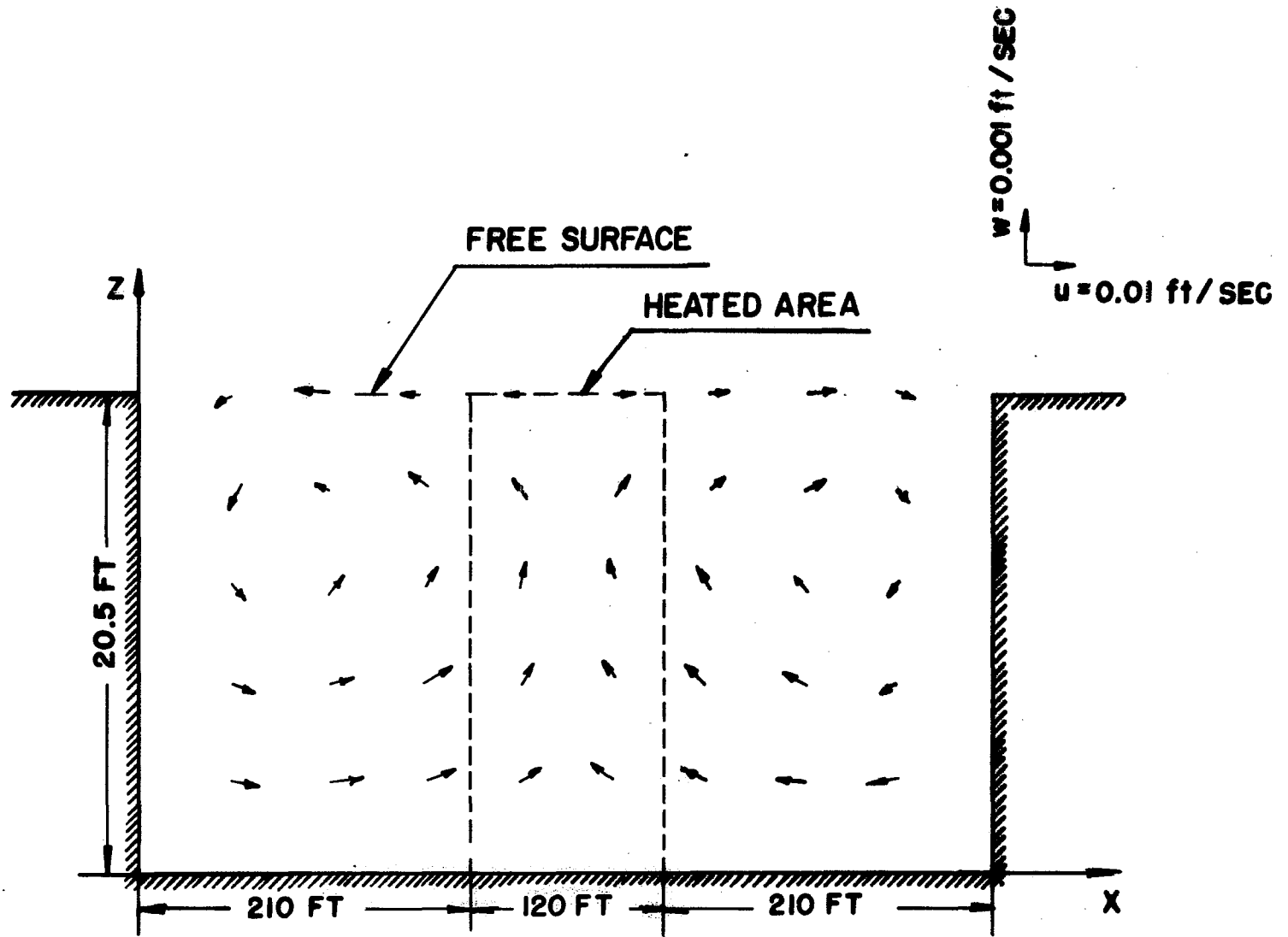
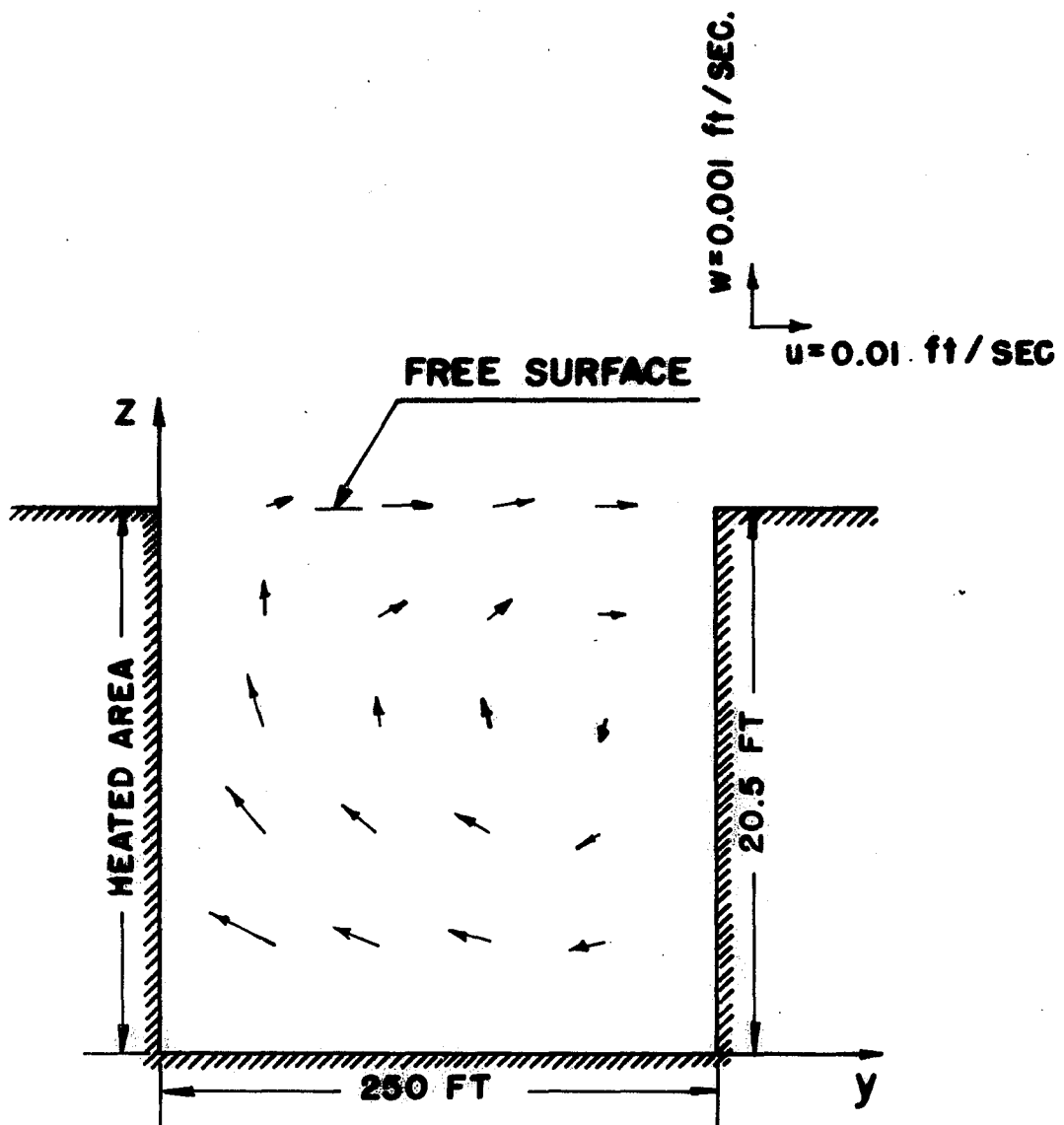


FIG 3. NATURAL CIRCULATION AT $y=100 \text{ FT.}$ AND $t=40 \text{ SEC}$ IN A POND PARTIALLY HEATED FROM SIDE.



**FIG 4. NATURAL CIRCULATION AT $x = 240 \text{ FT.}$
AND $t = 40 \text{ SEC.}$ IN A POND PARTIALLY
HEATED FROM SIDE.**

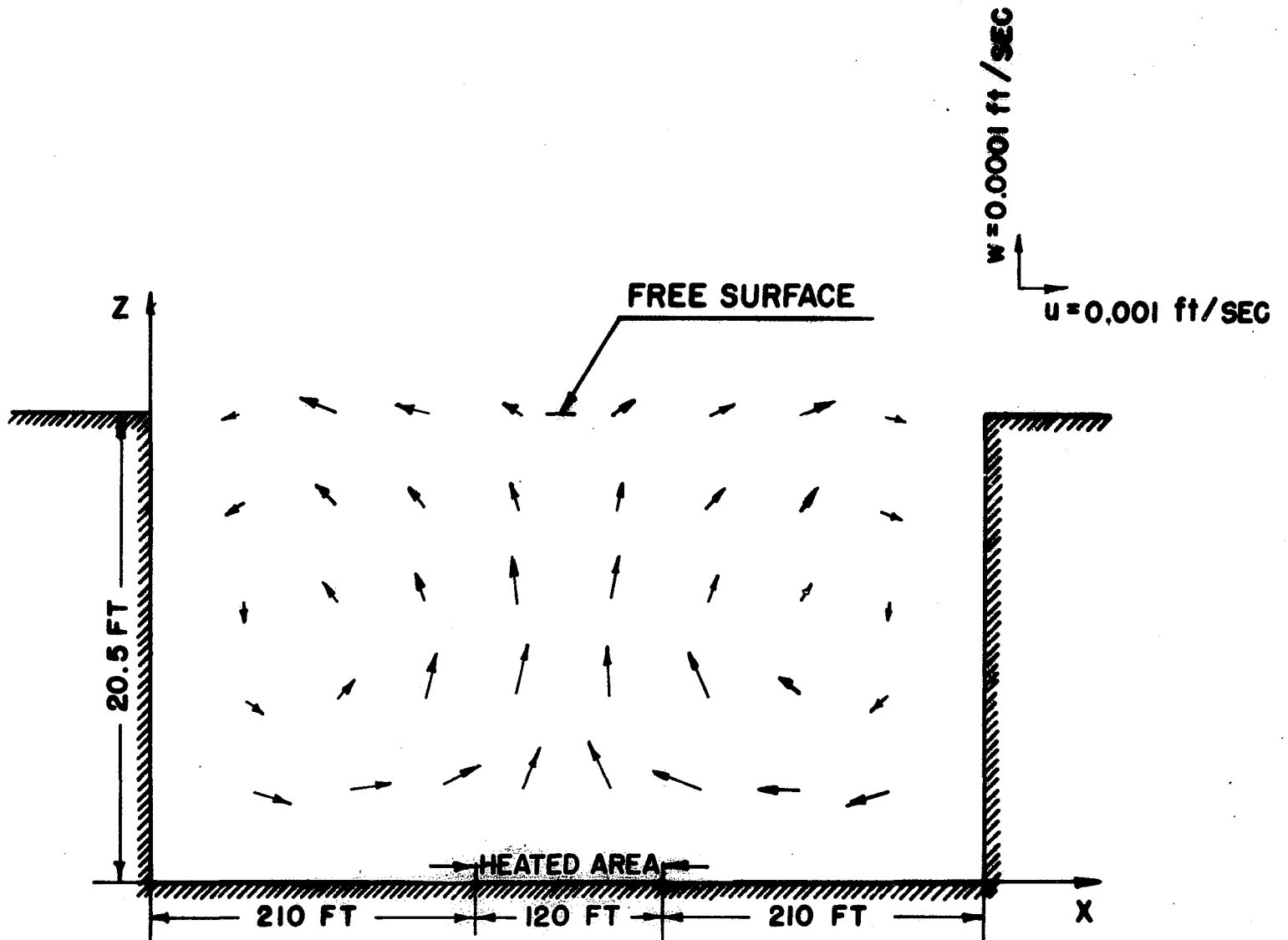


FIG 5. NATURAL CIRCULATION AT $y=100 \text{ FT.}$ AND $t=25 \text{ SEC.}$
 IN A POND PARTIALLY HEATED FROM BOTTOM.

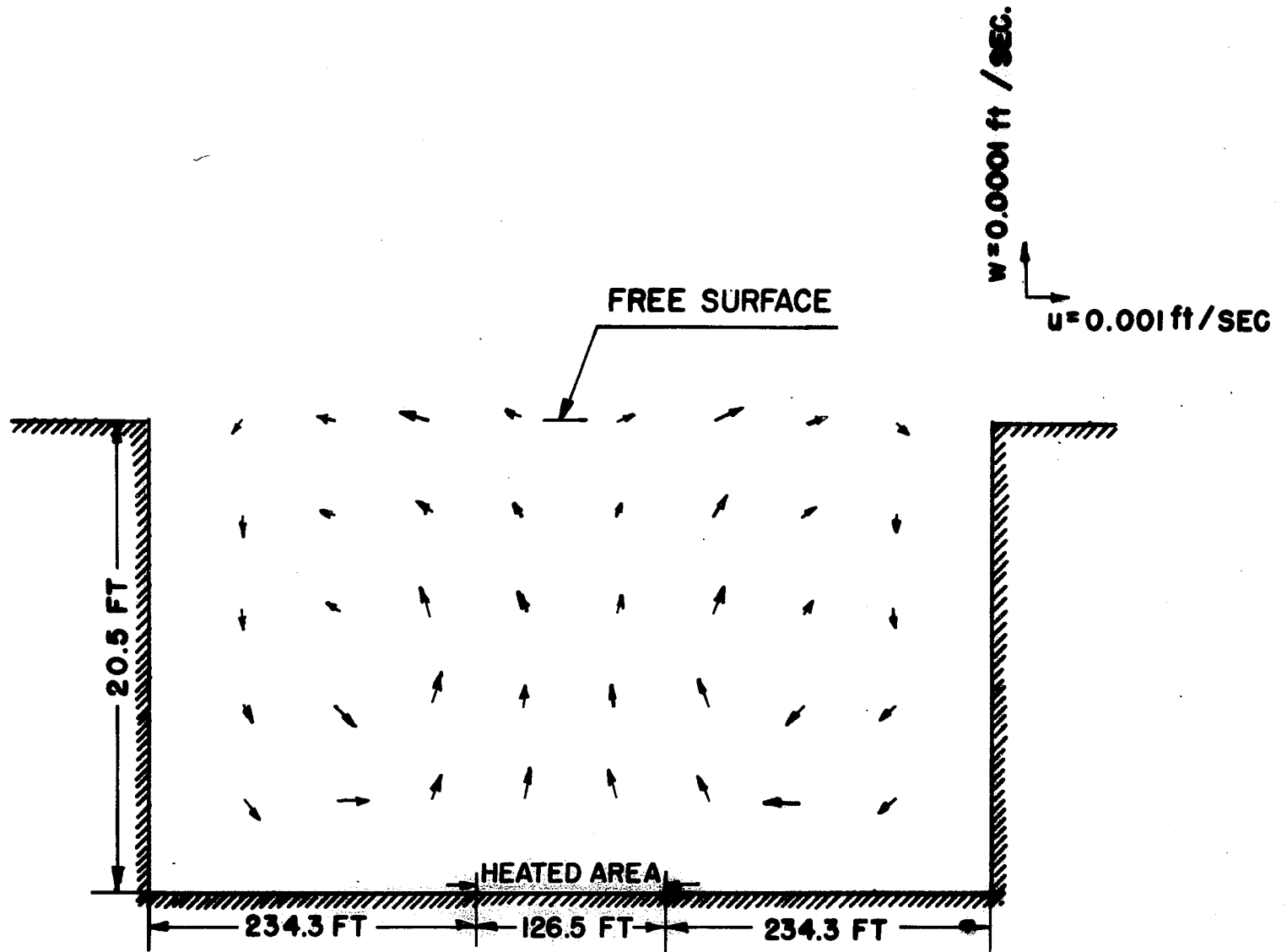


FIG 6. NATURAL CIRCULATION AT $t = 25 \text{ SEC}$ IN VERTICAL DIAGONAL PLANE IN A POND PARTIALLY HEATED FROM BOTTOM

aspects of water bodies.

5.2 Circulation and Stratification in Water Bodies

The mathematical formulation, presented in this study, is applied to the study of circulation and stratification of large water bodies. The water body is considered to be flowing initially at a uniform velocity and temperature and suddenly exposed to a 90° angle jet with higher velocity but the same temperature. This problem is referred to as three-dimensional non-buoyant jet in a cross current. When the dynamic problem reaches steady-state, the water jet temperature is suddenly raised to simulate a thermal discharge. This problem is referred to as three-dimensional buoyant jet in a cross current. To facilitate the understanding of the results, the two problems indicated above (i.e., non-buoyant and buoyant jets) are discussed separately in the following sections but the time histories of the results are shown on the same plots for easy reference.

5.3 Three Dimensional Non-Buoyant Jet in a Cross Current

The problem of a three-dimensional non-buoyant jet in a cross current, as modeled in Fig. 1, was first analyzed. The geometric and hydraulic input data for the case studied are given in Table 6.

The water body is assumed to be flowing initially at a uniform velocity of 0.40 ft/sec when suddenly exposed to a jet velocity of 2.00 ft/sec at a 90 degree angle. The surface wind velocity are considered to be zero. Furthermore, the temperature of the thermal

TABLE 6. GEOMETRIC AND HYDRAULIC INPUT DATA FOR
THREE-DIMENSIONAL, NON-BUOYANT AND BUOYANT JETS
IN A CROSS CURRENT

<u>Specifications</u>	<u>Dimensions</u>	
	<u>British Unit</u>	<u>SI Unit</u>
Water Body Length	600 ft	182.88 m
Water Body Width	360 ft	109.73 m
Water Body Depth	16.5 ft	5.03 m
Jet Width, $2 d_0$	100 ft	30.48 m
Jet Depth	5.75 ft	1.75 m
Jet Velocity, U_0	2 ft/sec	.61 m/sec
River Velocity	.4 ft/sec	.12 m/sec
Water Body Temperature	75 °F	23.89 °C
Water Body Equilibrium Temperature	74 °F	23.33 °C
Thermal Discharge Temperature, T_0	90 °F	32.22 °C
Reference Time, t_0	25 sec	25 sec
Reference Density, ρ_0	62.1156 lbm ft ⁻³	962.730 kg m ⁻³
Reference Pressure, P_0	21.56 lbm in ⁻²	14672.0 kg m ⁻²
Reynolds Number, R_0	0.1212×10^8	0.1212×10^8
Heat Exchange, Coefficient, k	100 Btu/ft ² day °F	23.64 W/m ² °C

discharge is made equal to the water body temperature to simulate a non-buoyant jet. A three-dimensional grid was superimposed on the flow field as detailed in Table 7. The boundaries of this grid coincide with the physical river boundaries. The nodal points, where hydrothermal variables are defined, are located at the point i,j,k of each grid and are shown in Fig. 7.

The time histories of the dynamic variables are shown, for points A, B, C, D, E, F, G, and H marked on Fig. 7, as follows:

- 1) Velocity components and water-level for two nodes A and B in front of the incoming jet at the free surface shown in Figs. 8 and 9.
- 2) Velocity components for two nodes C and D in front of the incoming jet at 12.75 feet below the free surface shown in Figs. 10 and 11.
- 3) Velocity components and water-level for two nodes E and F located upstream and downstream respectively at the free surface shown in Figs. 12 and 13.
- 4) Velocity components of two nodes G and H located upstream and downstream respectively at 12.75 feet below the free surface shown in Figs. 14 and 15.

Examination of the above plots shows the following features:

Considering the flow at the water surface, the transversal velocities at points A and B, in front of the incoming jet, rise initially with a subsequent rise in the vertical velocity leading to a rise in water-level in the neighborhood of the incoming jet which

TABLE 7. GRID DIMENSIONS FOR THREE-DIMENSIONAL
NON-BUOYANT AND BUOYANT JETS IN A CROSS CURRENT

Element <u>No.</u>	<u>x</u>		<u>y</u>		<u>z</u>	
	<u>ft</u>	<u>m</u>	<u>ft</u>	<u>m</u>	<u>ft</u>	<u>m</u>
1	65	19.81	50	15.24	3.5	1.07
2	60	18.29	50	15.24	4.0	1.22
3	55	16.76	55	16.76	5.0	1.52
4	50	15.24	60	18.29	4.0	1.22
5	50	15.24	65	19.81	3.5	1.07
6	55	16.76	70	21.34		
7	60	16.76	70	21.34		
8	65	19.81				
9	70	21.34				
10	70	21.34				

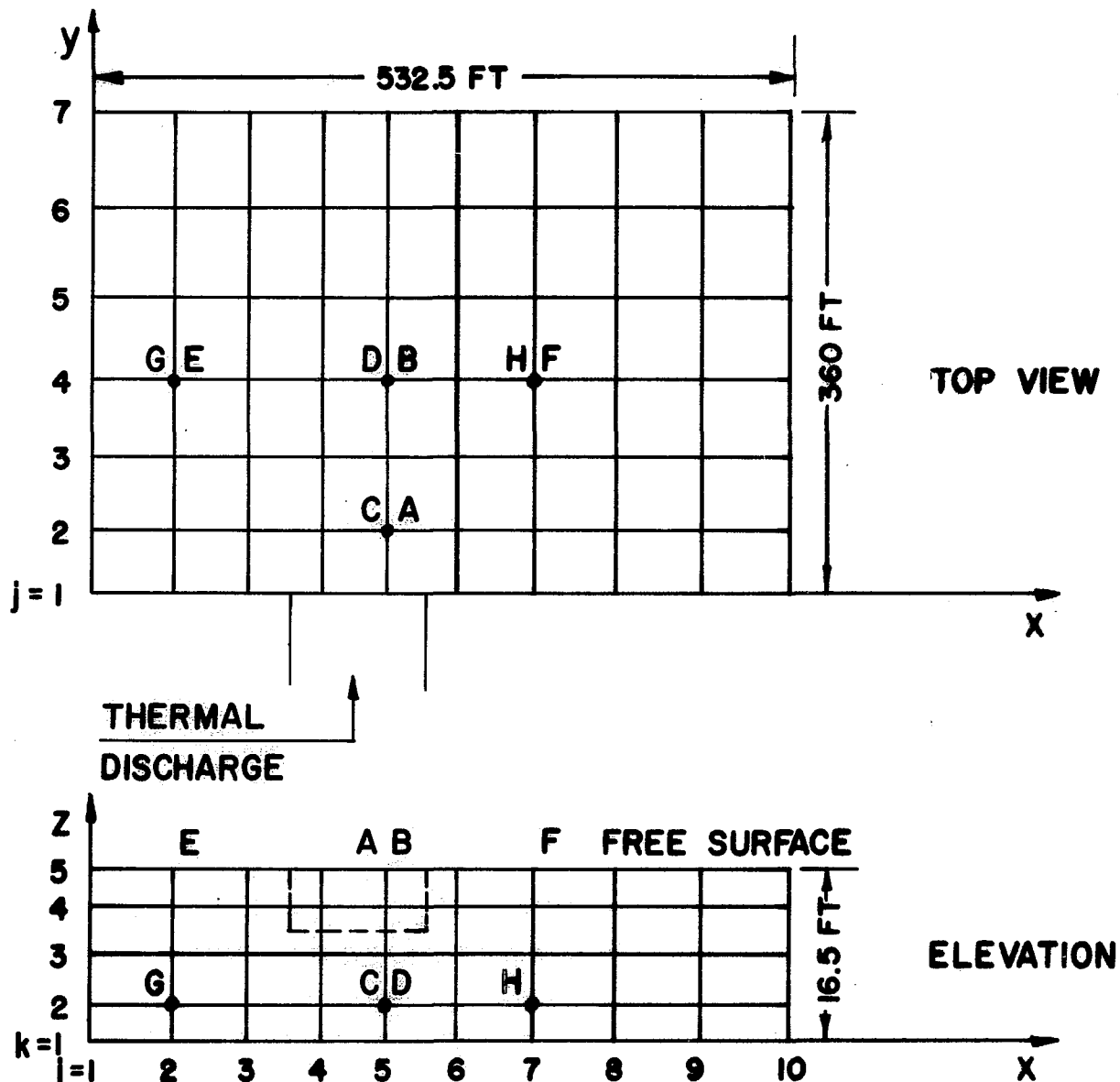


FIG 7. GRID WORK WITH VARIABLE MESH SIZE SUPERIMPOSED ON THE WATER BODY

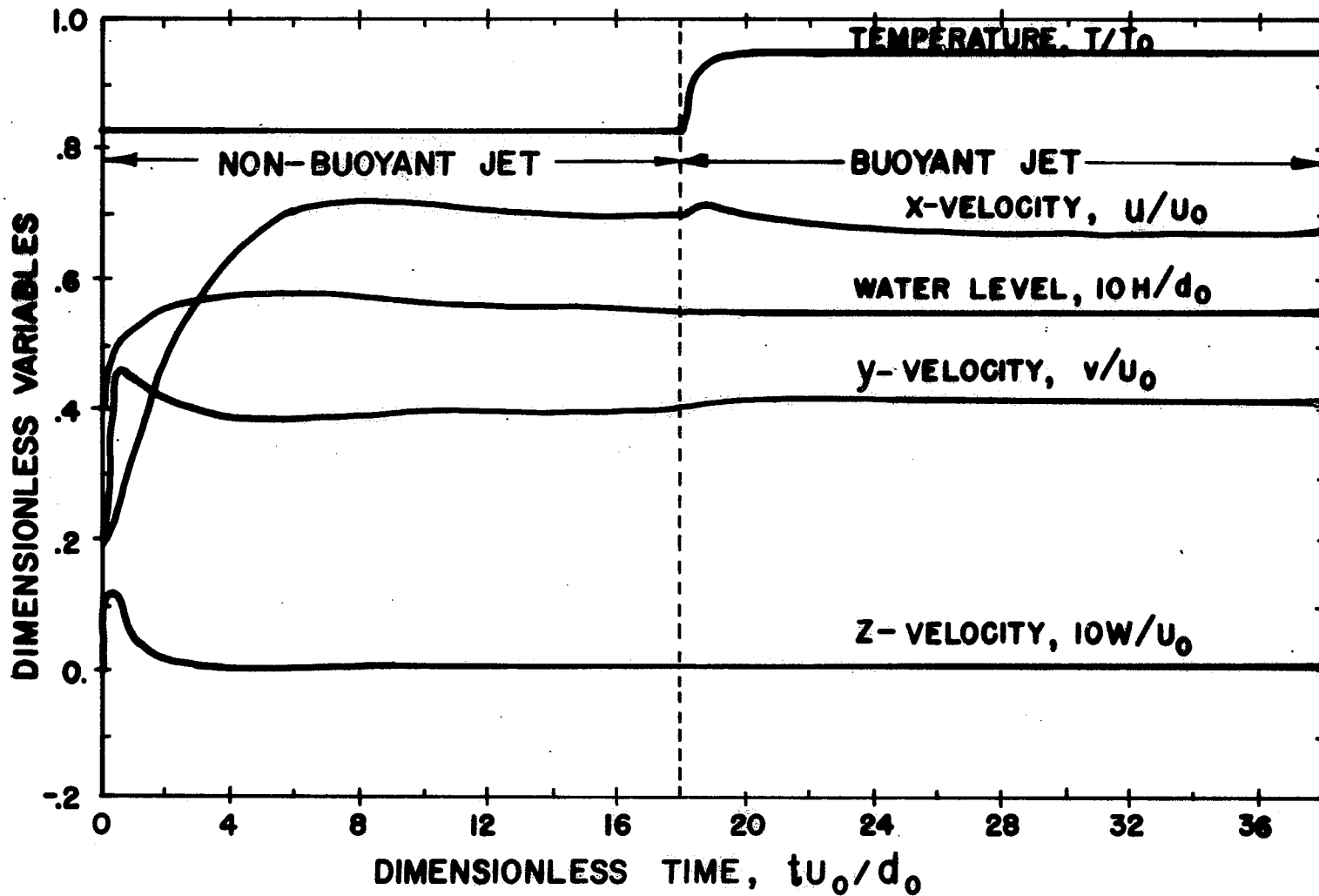


FIG 8. TIME HISTORIES OF VARIABLES AT POINT "A" IN THE WATER BODY SUBJECTED TO NON-BUOYANT AND BUOYANT JETS

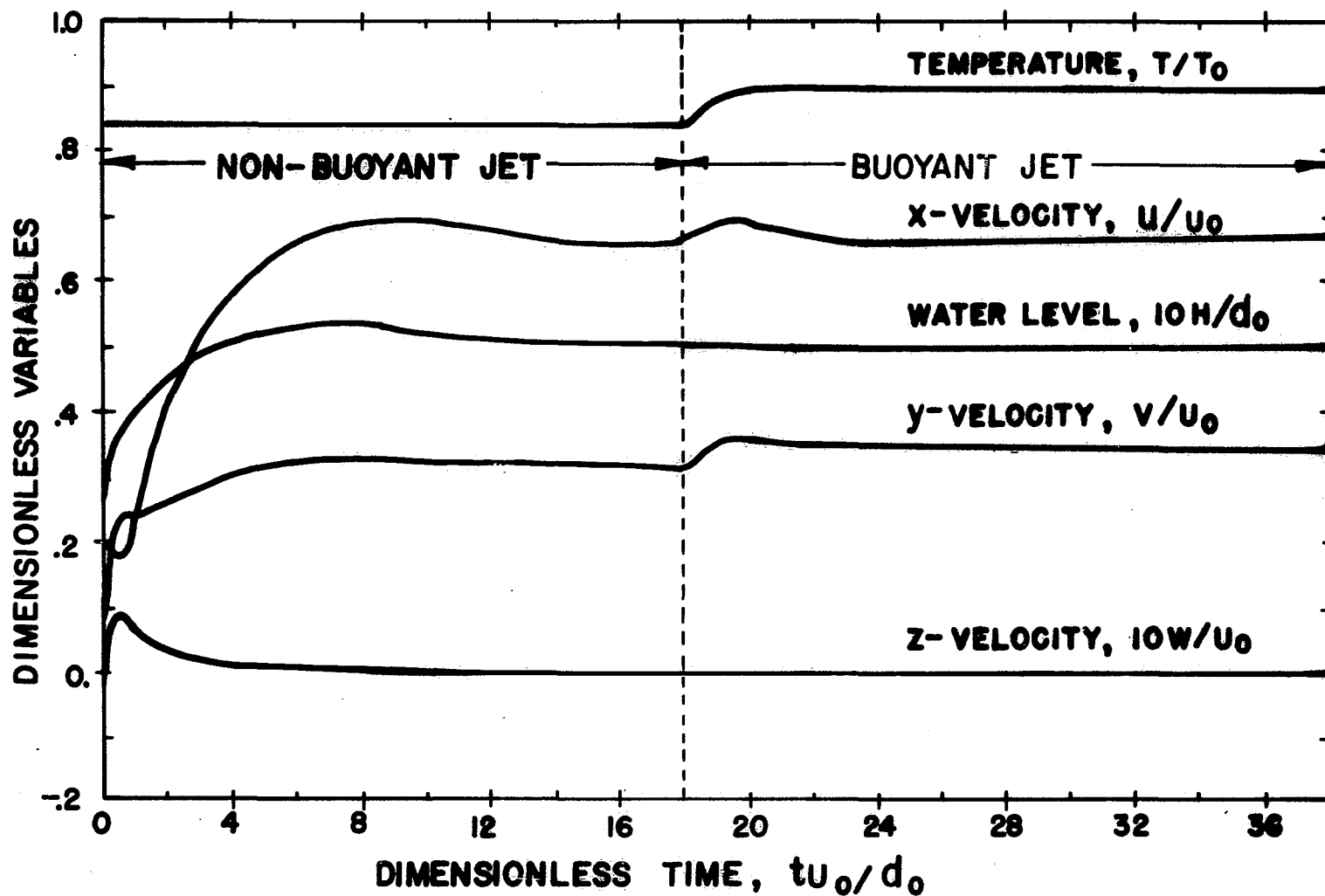


FIG 9. TIME HISTORIES OF VARIABLES AT POINT "B" IN THE WATER BODY SUBJECTED TO NON-BUOYANT AND BUOYANT JETS

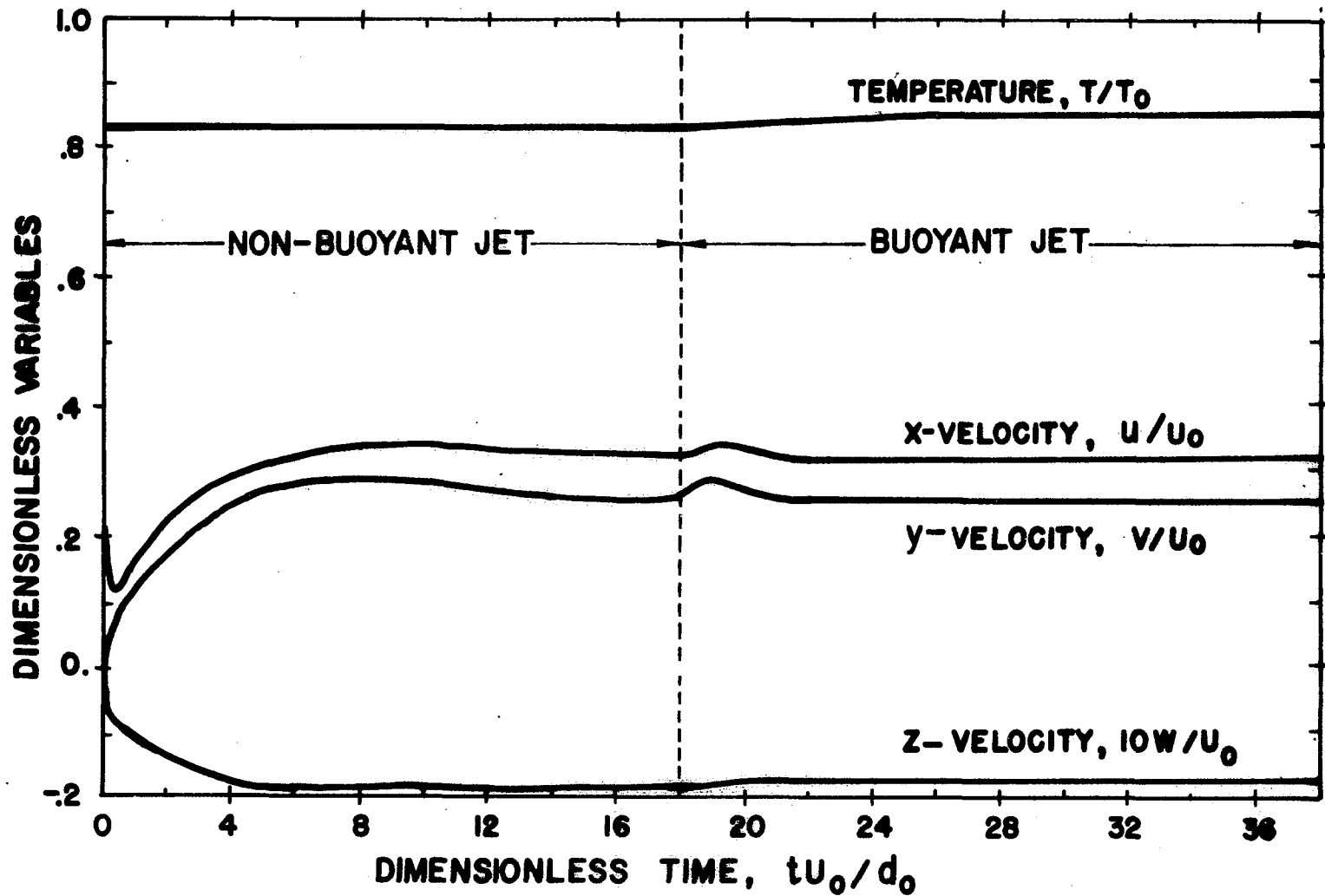


FIG 10. TIME HISTORIES OF VARIABLES AT POINT "C" IN THE WATER BODY SUBJECTED TO NON-BUOYANT AND BUOYANT JETS

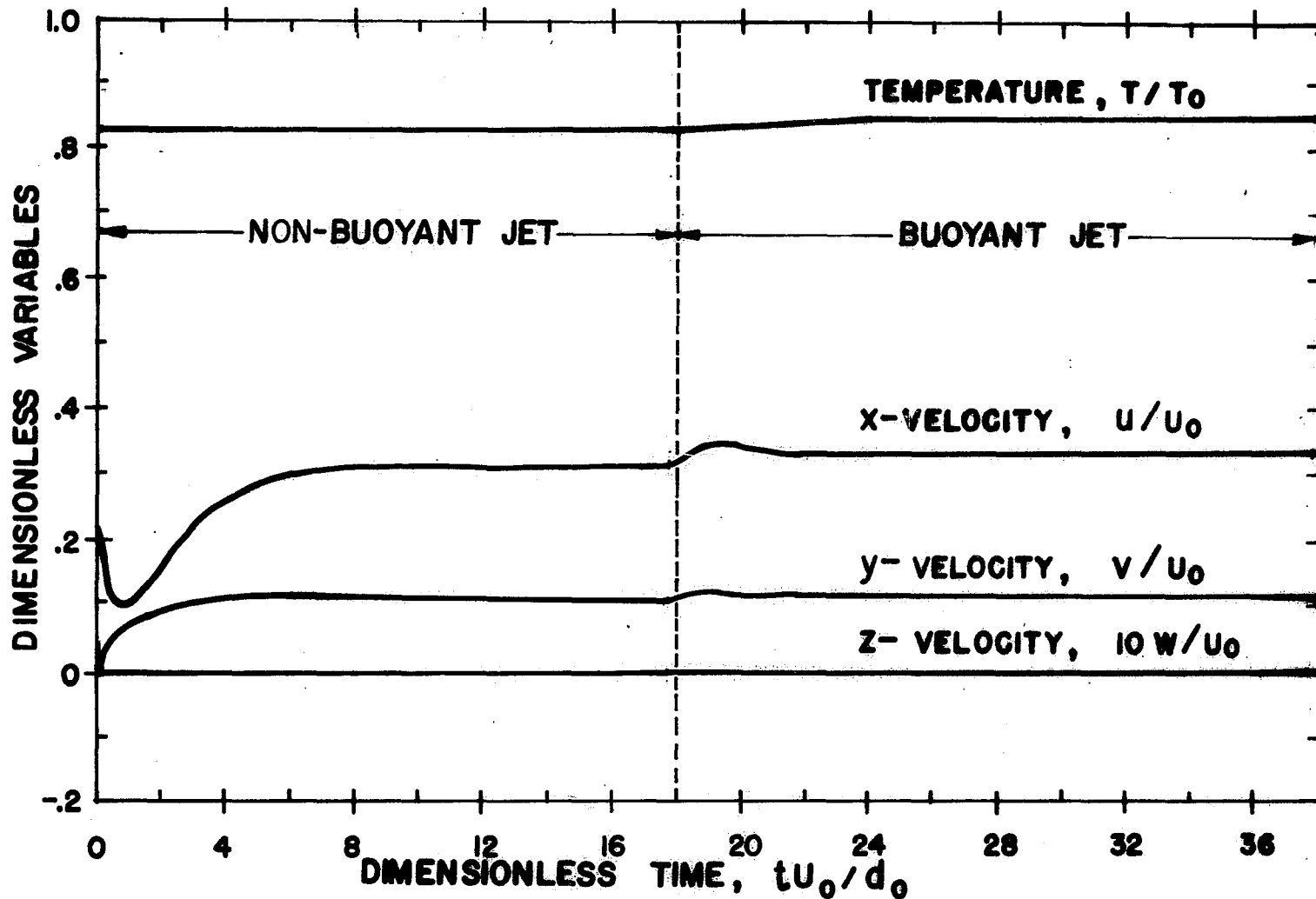


FIG II. TIME HISTORIES OF VARIABLES AT POINT "D" IN THE WATER BODY
SUBJECTED TO NON-BUOYANT AND BUOYANT JETS

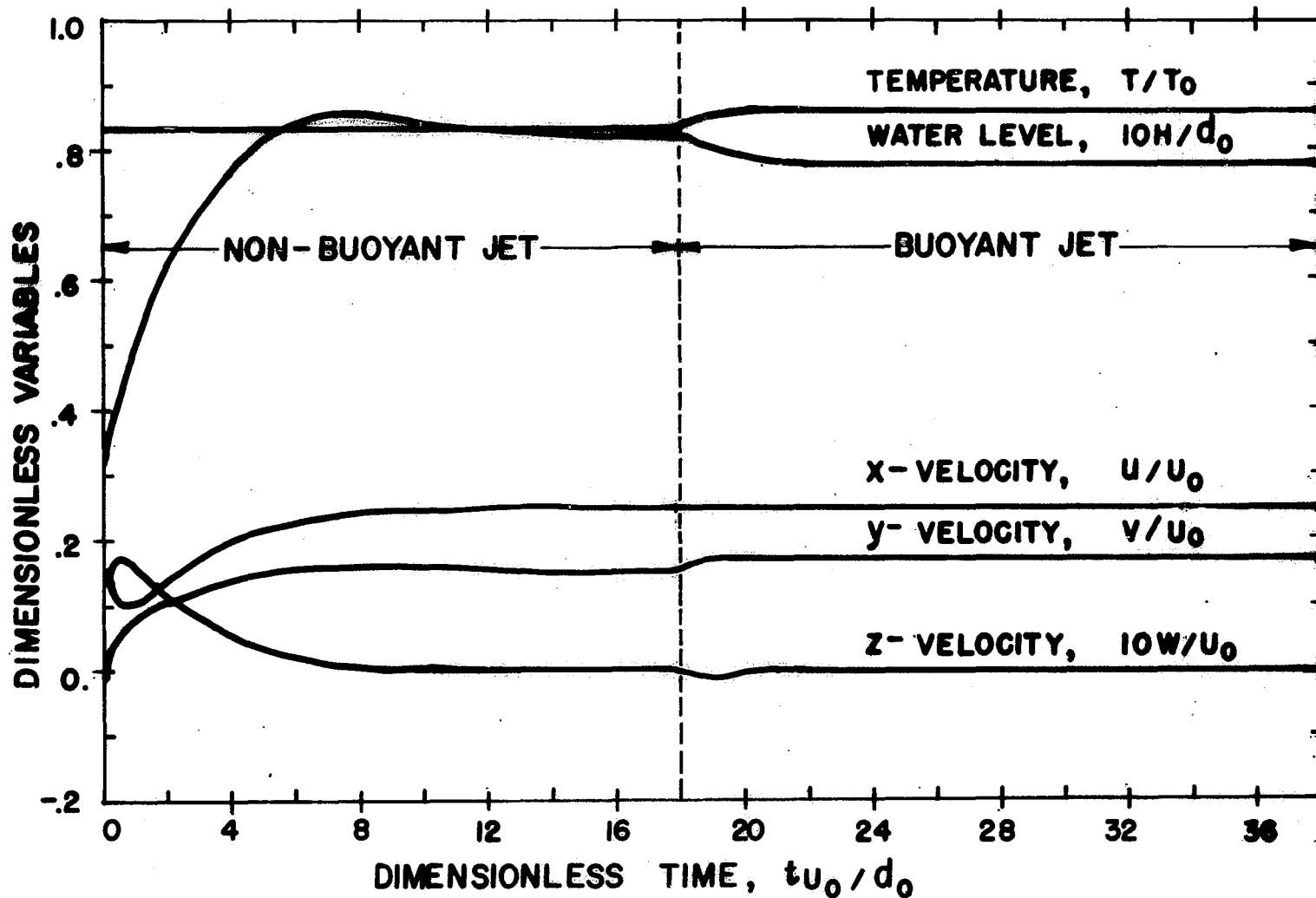


FIG 12. TIME HISTORIES OF VARIABLES AT POINT "E" IN THE WATER BODY
SUBJECTED TO NON-BUOYANT AND BUOYANT JETS

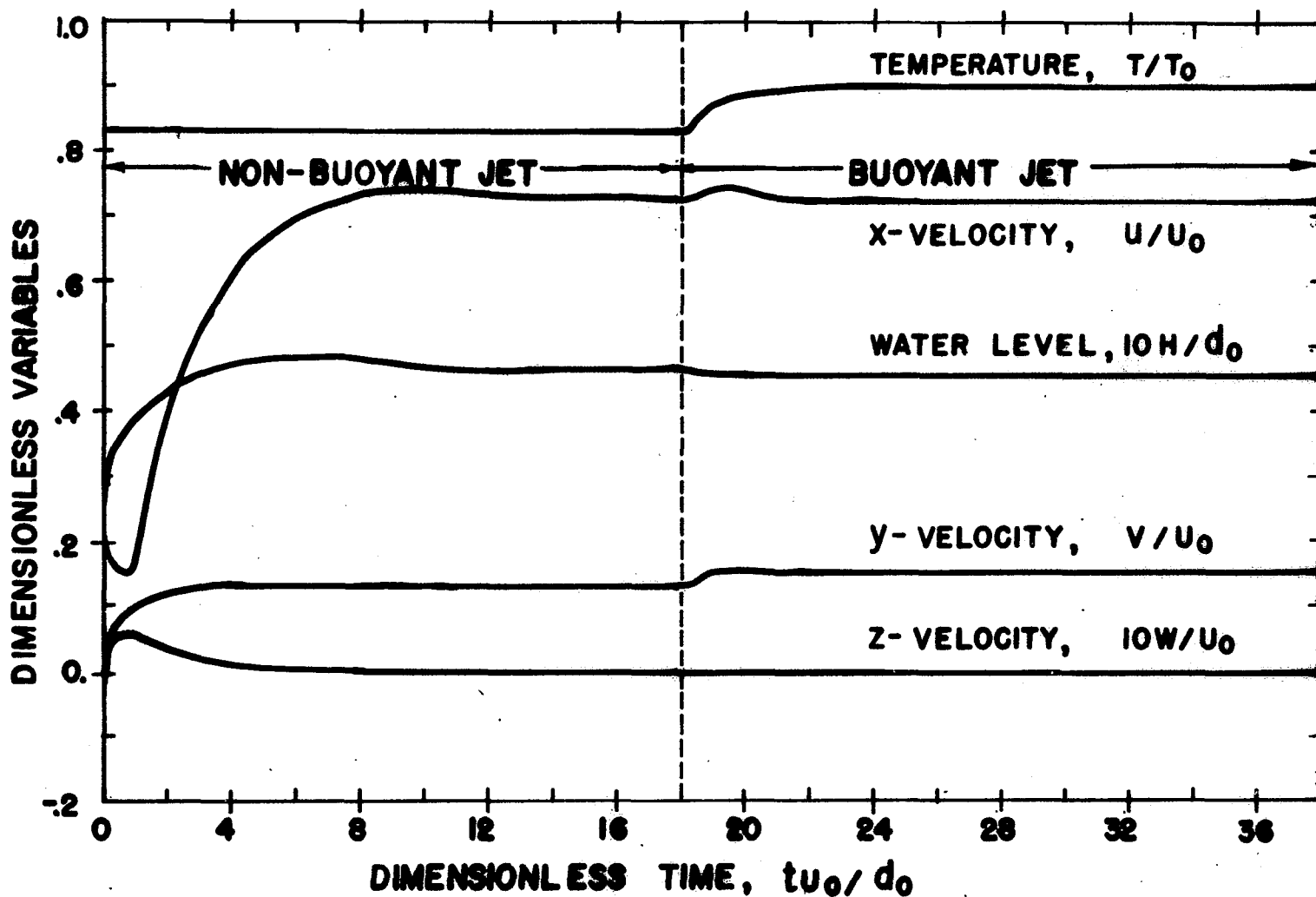


FIG 13. TIME HISTORIES OF VARIABLES AT POINT "F" IN THE WATER BODY
SUBJECTED TO NON-BUOYANT AND BUOYANT JETS

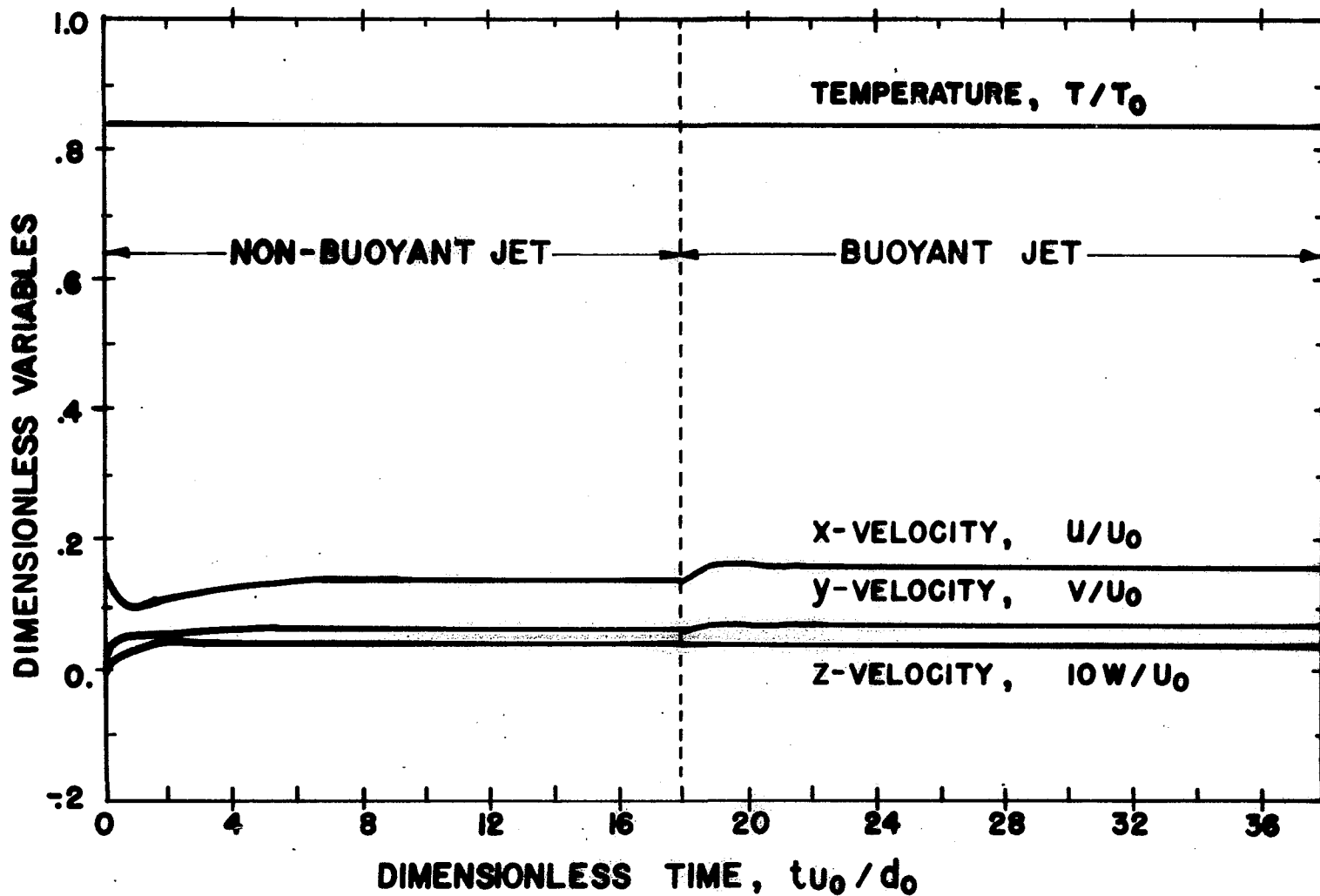


FIG 14. TIME HISTORIES OF VARIABLES AT POINT "G" IN THE WATER BODY
SUBJECTED TO NON-BUOYANT AND BUOYANT JETS

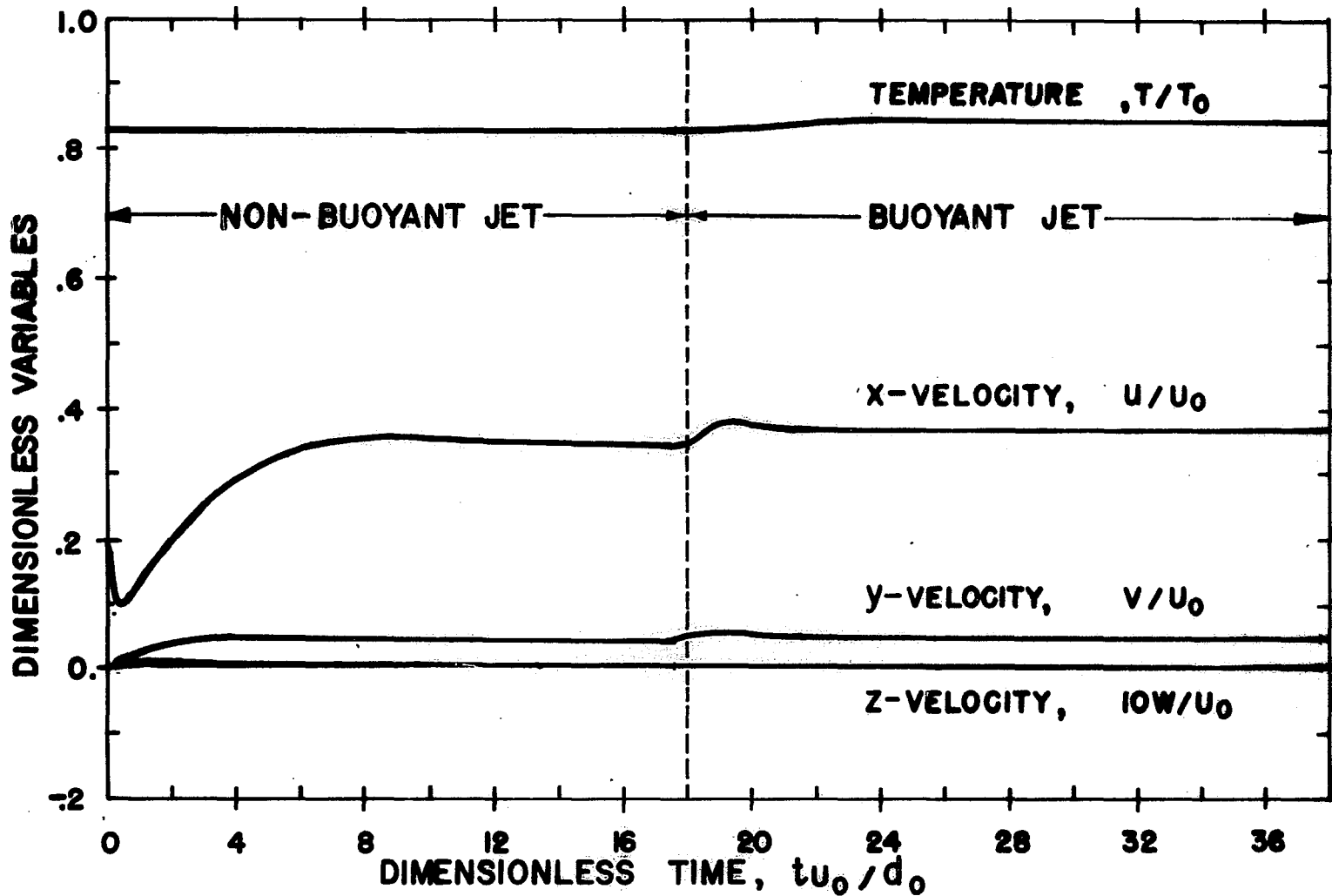


FIG 15. TIME HISTORIES OF VARIABLES AT POINT "H" IN THE WATER BODY SUBJECTED TO NON-BUOYANT AND BUOYANT JETS

in turn affects the field flow conditions as time progresses. Since the total head associated with the jet and river flow entries are constant on a short term basis, an increase in the water-level must be accomplished with a corresponding decrease in the field velocity. The disturbance caused by the incoming jet travel from the discharge point both upstream and downstream of the river. This demonstrates the propagation of the surface gravity waves (surge waves). As time progresses, these disturbances reach the river exit and are reflected back upstream. This demonstrates the reflection of the surface gravity waves from the river exit. However, on a long term basis, a further increase in the water-level in the neighborhood of the incoming jet causes additional horizontal velocity components for further downstream points which will gradually affect the downstream flow. These flow patterns are observed both in axial and transversal direction as discussed hereunder:

a) The increase in the water-level initially decelerates the axial flow on a short term basis. With a further increase in the water-level, the axial flow at various points accelerates to their final steady state values which are larger for the downstream points and smaller for the upstream points from the incoming jet position as expected. These effects can be clearly observed in u/U_0 curves for points E, B, and F, in Figs. 12, 9, and 13, where the transient start with a small dip in the flow rate followed by a steep rise settling to its final steady state value.

b) Similarly, the transversal flow generated by the incoming jet, increases the water-level, which in turn, decelerates the trans-

versal flow. However, since, unlike the axial flow patterns, the boundary condition in the transversal direction (solid wall) does not accommodate flow, the transversal flow reaches its final steady state value without undergoing large swings observed in the axial flow curves. These effects can be clearly observed in v/U_0 curves for points A and B in Figs. 8 and 9, where the transversal flow after the initial rise undergoes a reduction in magnitude followed by a mild rise settling to its final steady state values.

At the end of transients discussed above, the increase in transversal flow caused by the incoming jet is accommodated by an increase in the axial flow. For this reason, the water-level reaches a steady state value which in turn results in zero vertical velocity and constant axial and transversal flow, observed in Figs. 8 and 9, as the end of non-buoyant analysis is approached.

Examining the flow at 12.75 feet below the water surface, the axial flow exhibit a pattern similar to that of the water surface except that the results are further accentuated due to the bottom boundary condition. Figs. 11, 14 and 15 show that the flow transients u/U_0 start with a relatively large dip followed by relatively smaller rise as it approaches its final steady state value. The effect of the non-slip bottom boundary condition is to reduce the final steady state value to a quantity below that of the surface and to enlarge the initial dip. Furthermore, the shear initiated by the incoming jet creates transversal flow components v/U_0 at this level

as observed in Figs. 10 and 11. The transversal flow reduces from the top to the bottom. At the end of transients discussed above, the increase in transversal flow caused by the incoming jet cannot be fully accommodated by an increase in the axial flow in cells near the solids boundaries. For this reason, a downward velocity component develops which induces a vertical downward velocity in the upper cells, observed in w/U_0 curve in Fig. 10. This downward flow extends in a few cells from the jet entrance and is changed to an upwards flow in cells in the center of the flow field.

5.4 Three Dimensional Buoyant Jet in a Cross Current

The problem of three-dimensional buoyant jet in a cross current is next analyzed. The geometric and hydraulic input data for this case is exactly similar to the non-buoyant jet case detailed in Fig. 7 and Table 6. The water body is assumed to be initially under the steady state conditions reached in the non-buoyant case discussed earlier when the thermal discharge temperature is suddenly increased from 75° to 90° F.

The dynamic buoyant jet problem is analyzed in a manner similar to the non-buoyant jet case. However, in view of the introduction of thermal effects, the water body becomes stratified as discussed hereunder. The time histories of the dynamic variables are shown, for points A, B, C, D, E, F, G, and H marked in Fig. 7, as a continuation of the non-buoyant time history plots in Figs. 8 to 15. Examination

of these plots shows the following features:

From the circulation viewpoint, the effect of the heated thermal discharge entering the flow field can be decomposed into two components:

1) unheated discharge entering the flow field studied in the non-buoyant case problem; and 2) thermal effect of the discharge considered as a heated wall studied earlier under the topic of ponds partially heated from the side. According to the above decomposition, the velocity time history plots in Figs. 8 to 15 for buoyant case should be the sum of the non-buoyant case and the corresponding heated wall problem.

Examining the flow at the water surface at points A and B, in front of the incoming jet, the transversal velocity for the buoyant case shows a small increase as compared with the non-buoyant results due to the natural circulation caused by the heated wall. This additional transversal velocity induces an axial velocity transient similar in nature to the non-buoyant case. However, since the magnitude of the transversal velocity transient is small, the resultant axial velocity transient would also be small. This behavior can be clearly observed in u/U_0 and v/U_0 curves in Figs. 8 and 9 which also show the temperature transients T/T_0 for both points A and B. A similar pattern can also be observed at the surface points E and F in Figs. 12 and 13 as well as the points C, G, and H located at 12.75 feet below the water surface in Figs. 10, 14 and 15.

At the end of the transients, the coldest and the hottest regions are at the river and the thermal discharge entrances respectively.

For this reason, the strongest natural circulation patterns would develop mainly between these coldest and hottest regions. A comparison of the surface velocity vectors at points A and E and the velocity vectors at 12.75 feet depth for buoyant and non-buoyant cases confirm the existence of this natural circulation pattern. The surface velocity field at the end of the transient is shown in Fig. 16. The slowing down of the inlet flow near the thermal discharge entrance and the turning of the thermal discharge and incoming flows as a result of their interaction can be clearly seen in this figure.

The thermal plume effect is shown in Figs. 17 to 22. Isotherms are plotted for the water surface as well as the vertical planes in the longitudinal and transversal direction at times 60 and 500 seconds after the initiation of the heated discharge. An examination of these plots indicate the following features:

- 1) The stratification pattern is three dimensional in nature and shows the expansion of the thermal plume in all directions. In view of the prevailing advective currents this expansion is much more intense toward the downstream as compared to the other directions (vertical and upstream) as expected.

- 2) The three distinctive regions which constitute the characteristic behavior of a stratified water body (epilimnion, thermocline and hypolimnion) can be clearly observed in Fig. 22.

- 3) These results provide the water temperature rise and the rate of temperature rise needed for the assessment of the extent of thermal pollution in the water body.

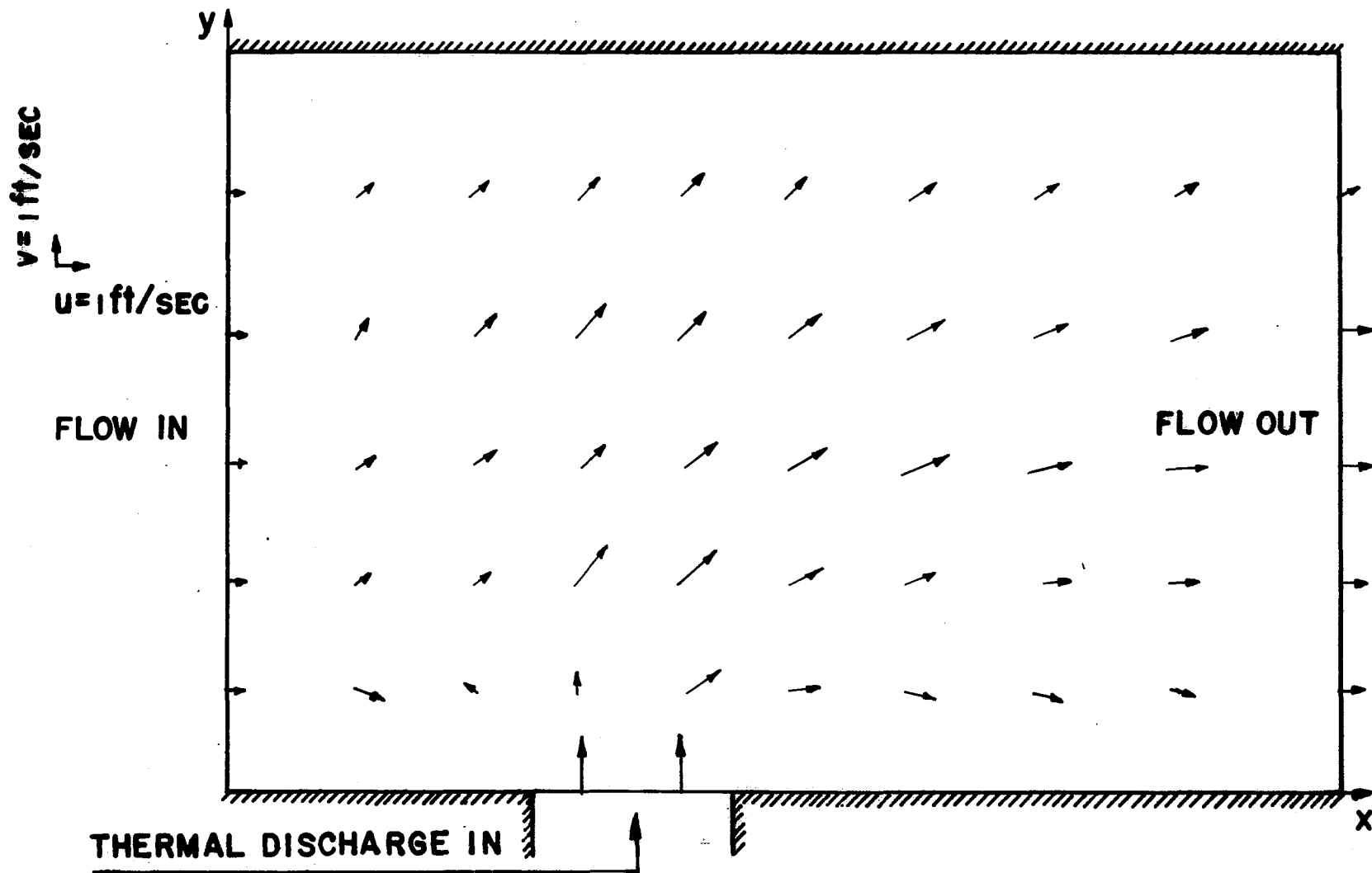
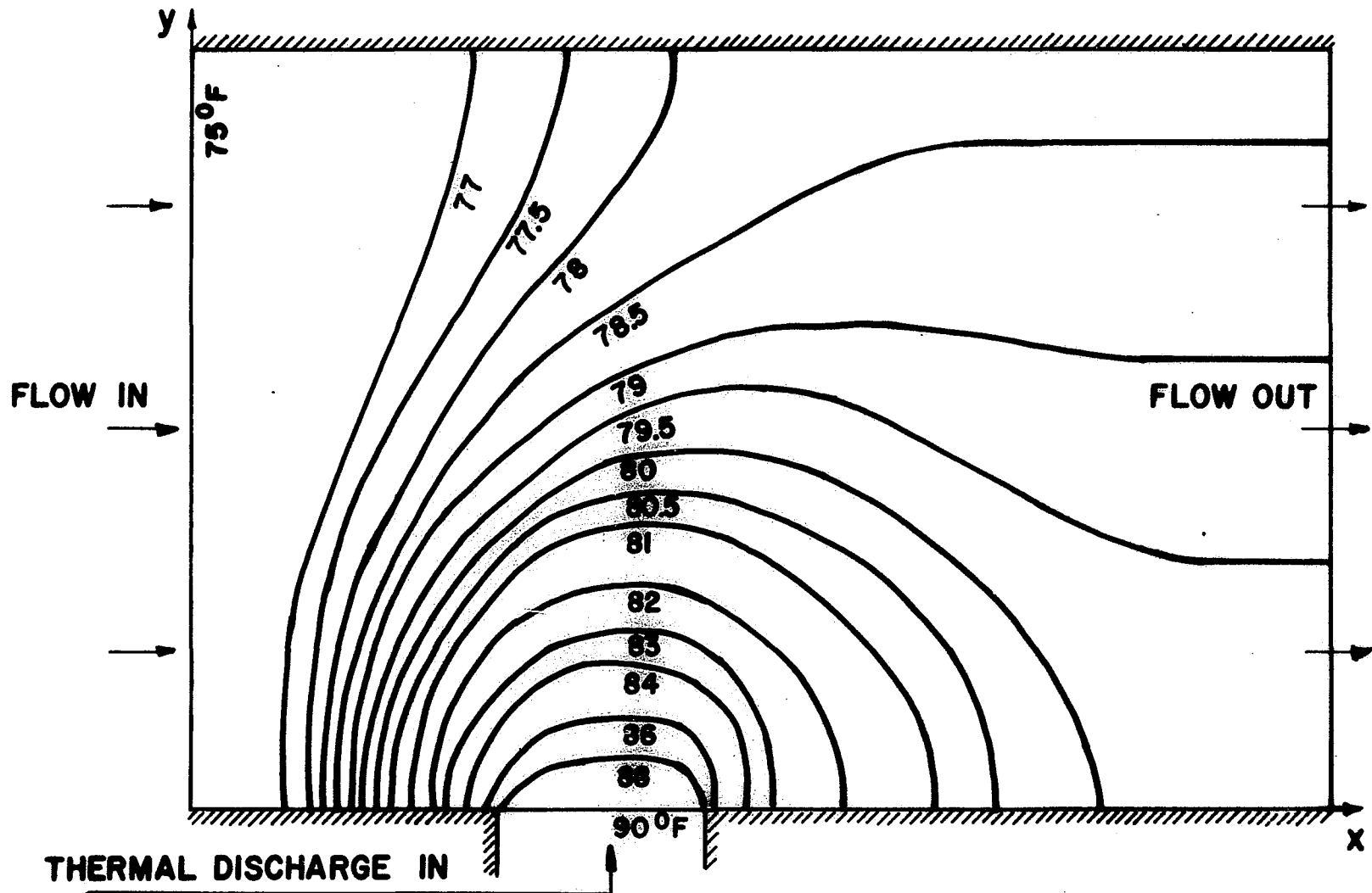


FIG 16. SURFACE VELOCITY FIELD AT $t = 500 \text{ SEC.}$ IN THE WATER
 SUBJECTED TO A BUOYANT JET



**FIG 17. SURFACE ISOTHERMS AT $t = 60$ SEC. IN THE WATER BODY
SUBJECTED TO A BUOYANT JET**

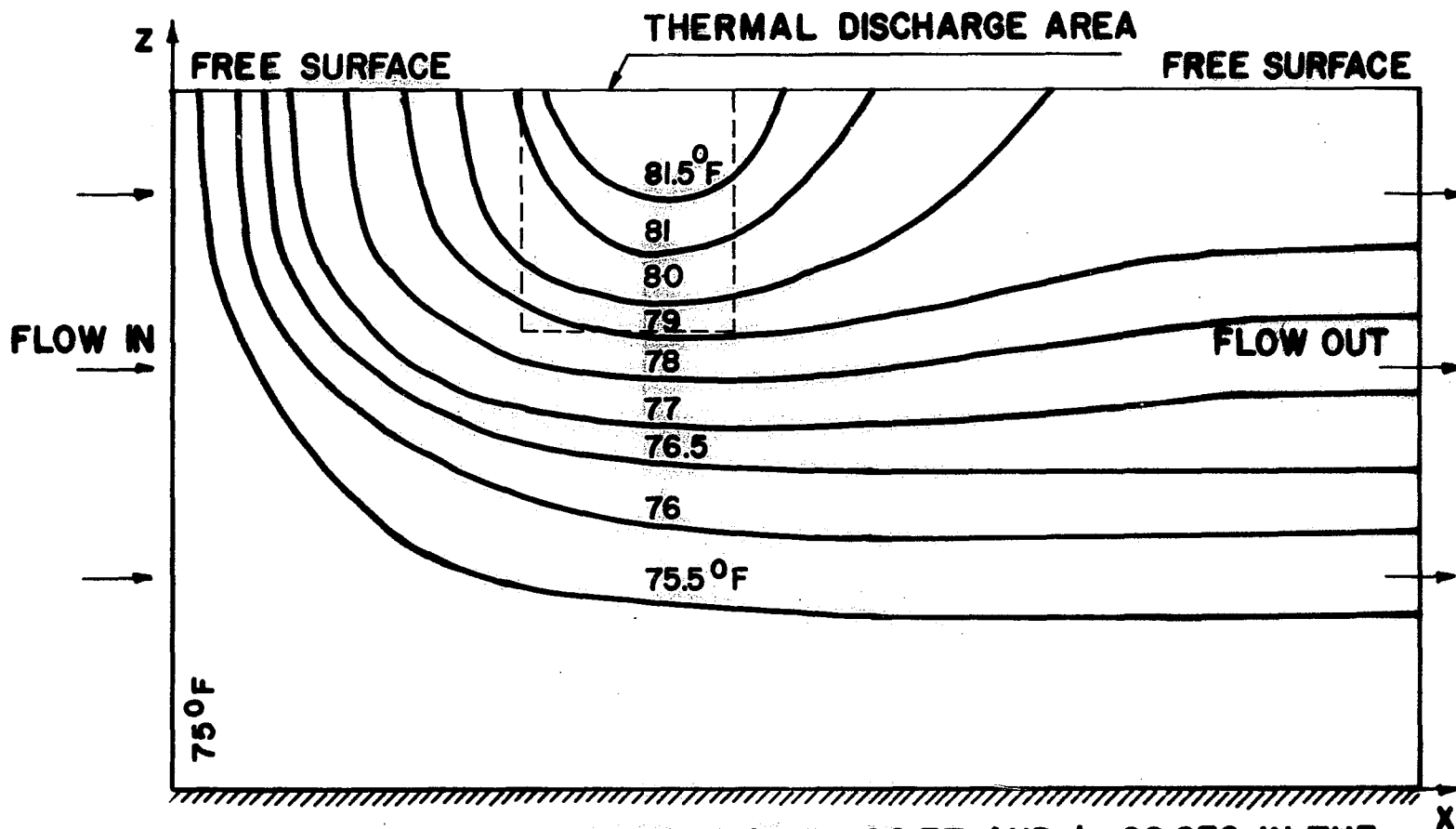


FIG 18. VERTICAL ISOTHERMS AT $y = 100$ FT. AND $t = 60$ SEC. IN THE WATER BODY SUBJECTED TO A BUOYANT JET

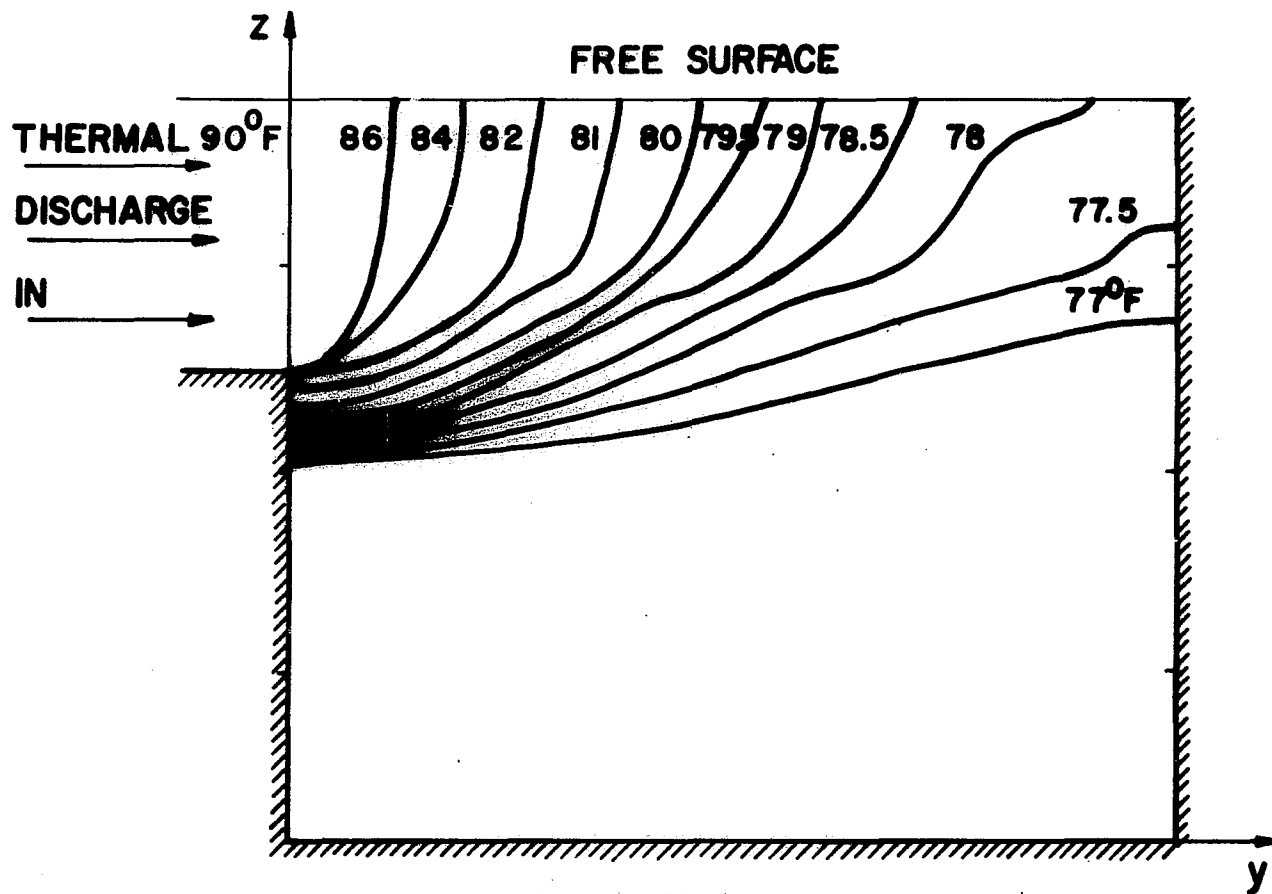


FIG 19. VERTICAL ISOTHERMS AT $X = 222.5$ FT. AND $t = 60$ SEC. IN THE WATER BODY SUBJECTED TO A BUOYANT JET

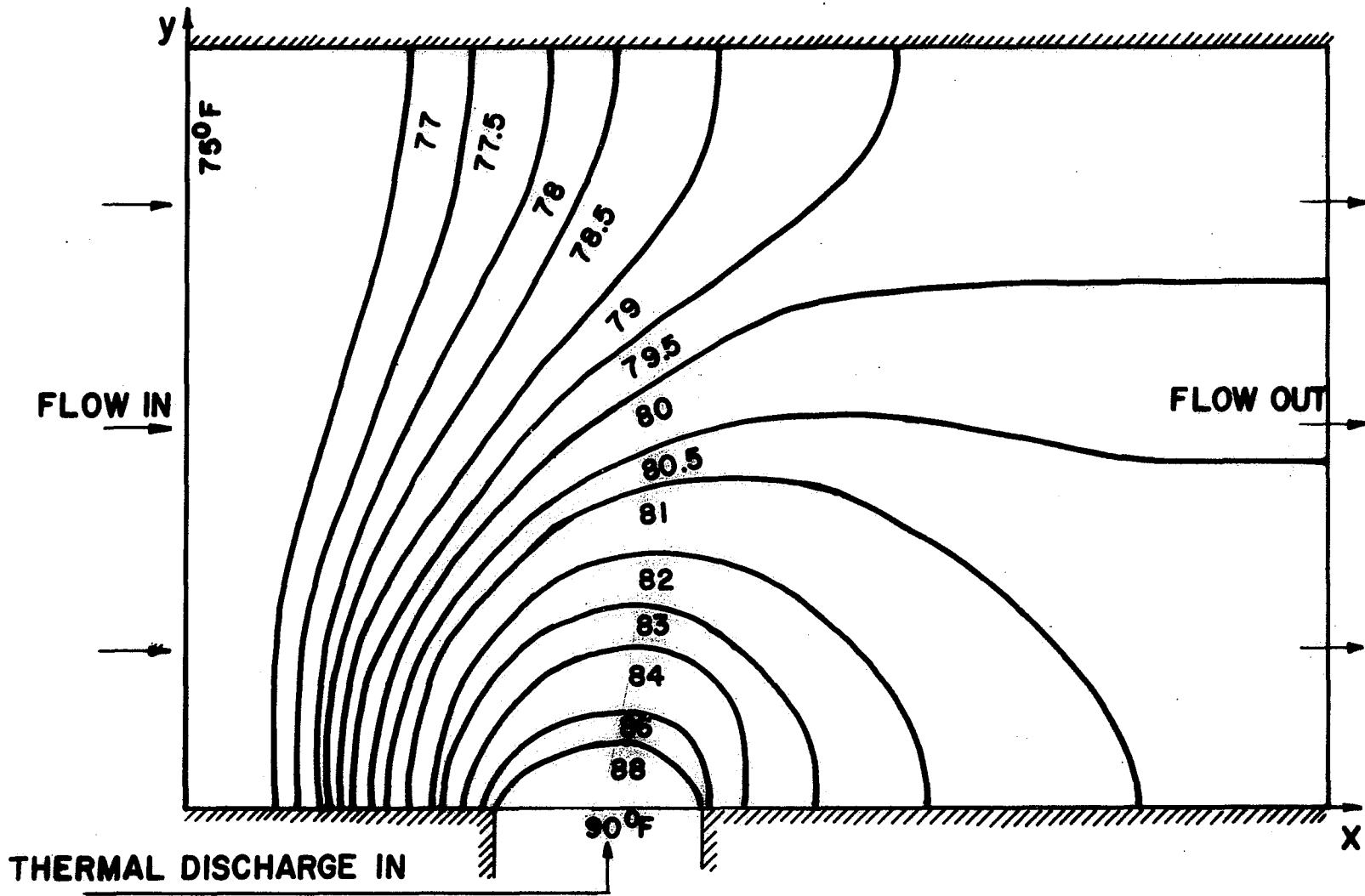


FIG 20. SURFACE ISOTHERMS AT $t = 500$ SEC. IN THE WATER BODY
SUBJECTED TO A BUOYANT JET

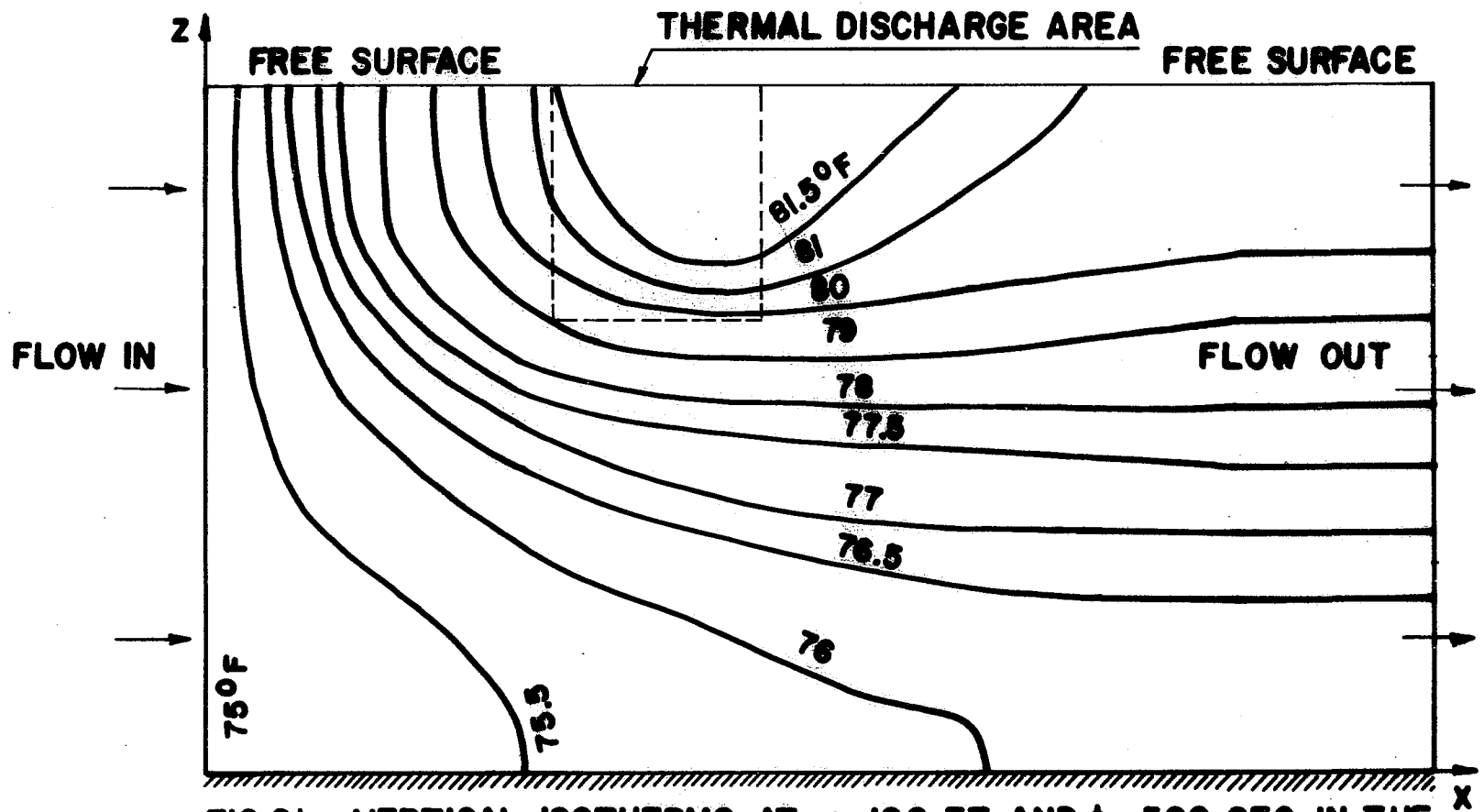


FIG 21. VERTICAL ISOTHERMS AT $y = 100$ FT. AND $t = 500$ SEC. IN THE WATER BODY SUBJECTED TO A BUOYANT JET

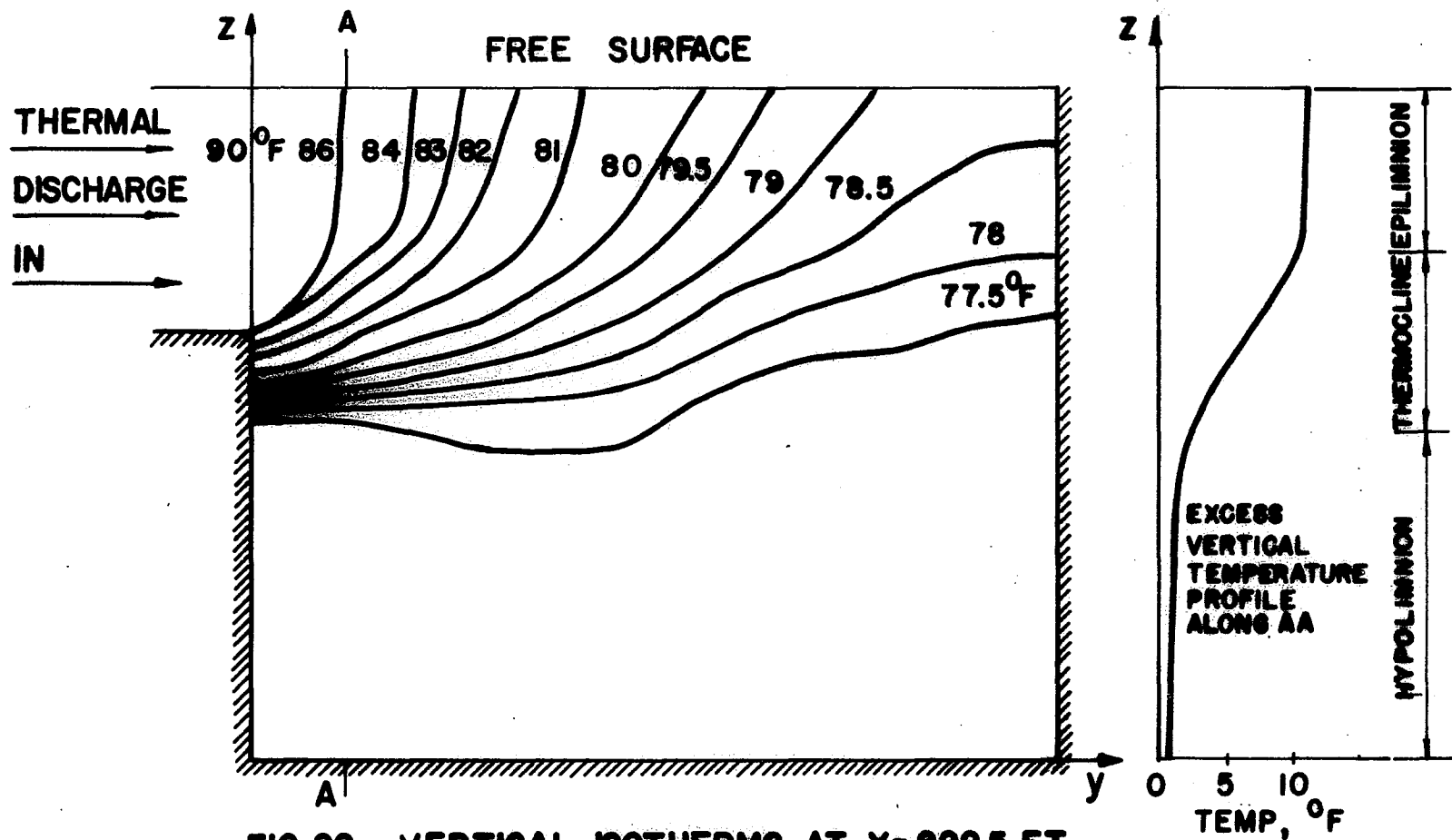


FIG 22. VERTICAL ISOTHERMS AT $x=222.5$ FT. AND $t=500$ SEC. IN THE WATER BODY SUBJECTED TO A BUOYANT JET

As a further verification of the validity of the results, a mass and energy balance was performed on the water body as shown in Table 8. This table shows that conservation equations are satisfied with a good accuracy. The small magnitude of the surface heat transfer verifies the often used assumption that, at high values of equilibrium temperature used herein for the purpose of near field studies, the contribution of heat exchange to the atmosphere is insignificant and may be neglected in simplified analysis.

TABLE 8. MASS AND ENERGY BALANCE

<u>Mass and Energy Balance</u>	<u>Incoming Flow</u>	<u>Thermal Discharge</u>	<u>Outgoing Flow</u>	<u>Surface Flow</u>	<u>% Error</u>
Mass Balance, in 10^6 lbm/sec	0.12252	0.08464	0.20748	0.0	0.15
Energy Balance, in 10^8 Btu/sec	0.09189	0.07454	0.16204	0.15096 $\times 10^{-4}$	0.18

6. CONCLUSIONS AND RECOMMENDATIONS

An improved three-dimensional analytical model for the mathematical description of a large rectangular water body subjected to a thermal discharge has been developed and presented herein. The improvement results from the facts that:

- A) Compliance with the conservation of mass and momentum is obtained by the introduction of two flow regions in the entire flow field: a) the water-level region containing a portion of water near the free surface in which the water-level rises or falls, as the case may be, during the dynamic solution; and b) the sub-water-level region located under the water-level region which remains totally filled with fluid at all times during the transient.
- B) A different set of differential equations for the conservation of mass, momentum, and energy is applied to the cells located in the water-level and sub-water-level flow regions.
- C) The pressure distribution is obtained accurately by combining the momentum and continuity equations and not by the hydrostatic approximation.
- D) A detailed numerical stability analysis is performed which provides accurate criteria for the selection of the space and time increments.

The analysis, programmed on a digital computer, is generally capable of predicting velocity, pressure and temperature distribution

in water bodies subjected to thermal discharge. Particularly, the present study can be employed to determine the temperature rise and rate of temperature rise needed for the assessment of the extent of thermal pollution in water bodies.

Further studies should be undertaken with the following objectives:

- a) To develop a three-dimensional analytical model for the mathematical description of actual rivers with arbitrary bottom and side configurations.
- b) To develop accurate correlations for momentum and thermal eddy diffusivities and to incorporate these correlations in the digital program.
- c) To provide an experimental verification for the thermal discharge problem.

7. NOMENCLATURE

a	Terms of matrix involved in the solution of the pressure equations
$a_x, a_y, a_z, b_x, b_y, b_z, c_x, c_y, c_z$	Weighting factors for derivatives
C_p	Specific heat of water
C_{fx}	Skin coefficient along x-axis
C_{fy}	Skin coefficient along y-axis
d_0	Half width of thermal discharge
d_x, d_y, d_z	Thermal eddy diffusivities
D_x, D_y, D_z	Sum of thermal conductivity plus thermal eddy diffusivity as defined by equations (A-19)
D_h	Horizontal eddy diffusivity of heat
D_v	Vertical eddy diffusivity of heat
$e_x, e_y, e_z, f_x, f_y, f_z$	Weighting factors for pressure
E	Equilibrium temperature of water
F_0	Froude number
g	Gravitational acceleration
H	Water-level height
K	Heat exchange coefficient for water-air interface
n	Normal to the solid boundary
p	Pressure
R_0	Reynolds number
S_0	Stanton number
T_0	Thermal discharge temperature
T	Local water temperature
t	Time

NOMENCLATURE (Cont'd)

U_0	Thermal discharge velocity
u	Horizontal velocity components in x-direction
v	Horizontal velocity components in y-direction
w	Vertical velocity components in z-direction
W_x	Wind velocity components in x-direction
W_y	Wind velocity components in y-direction
x, y, z	Cartesian coordinate system
Δx_i	Dimension of element i in x-direction
Δy_j	Dimension of element j in y-direction
Δz_k	Dimension of element k in z-direction

Greek Symbols

$\alpha_x, \alpha_y, \alpha_z$	Momentum eddy diffusivities
$\alpha, \beta, \gamma, \delta$	Coefficients in Von-Neumann stability analysis
Δ	A quantity defined by equation (99)
ξ	A quantity defined by equation (66b)
λ^2	Norm ratio
ν	Viscosity
ν_0	Kinematic viscosity of thermal discharge
$\nu_x^2, \nu_y^2, \nu_z^2$	Sum of viscosity and momentum eddy diffusivity as defined by equation (A-19)
ν_h	Horizontal eddy viscosity
ν_v	Vertical eddy viscosity
ρ	Local density of water

NOMENCLATURE (Cont'd)Greek Symbols (Cont'd)

ρ_a	Air density
ρ_0	Thermal discharge density
ω	Relaxation factor
ω_{opt}	Optimum value of relaxation factor
∇^2	Laplacian operator $\frac{\partial^2}{\partial x^2} + \frac{\partial^2}{\partial y^2} + \frac{\partial^2}{\partial z^2}$

Superscripts

*	Refers to non-dimensional quantities
n	Refers to updated value
.	Refers to a small variation
-	Refers to time average
'	Refers to fluctuating components used in Appendix 1 and to steady-state components used in Appendix 2

Subscripts

0	Refers to thermal discharge
i,j,k	Refers to element numbers
max	Refers to maximum
s	Refers to surface
rm	Refers to components of matrix "a"

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APPENDIX 1. Derivation of Governing Equation (1) through (15)

To derive the governing equations for the sub-water-level region, the Navier-Stokes equations for incompressible flow (31) are used in conjunction with the Boussinesq's approximation which neglects the fluid density variation in all terms involved in the momentum equations except in the body force term as follows:

Momentum Equations

$$\frac{\partial u}{\partial t} + u \frac{\partial u}{\partial x} + v \frac{\partial u}{\partial y} + w \frac{\partial u}{\partial z} = - \frac{1}{\rho_0} \frac{\partial p}{\partial x} + \nu \nabla^2 u \quad (\text{A-1})$$

$$\frac{\partial v}{\partial t} + u \frac{\partial v}{\partial x} + v \frac{\partial v}{\partial y} + w \frac{\partial v}{\partial z} = - \frac{1}{\rho_0} \frac{\partial p}{\partial y} + \nu \nabla^2 v \quad (\text{A-2})$$

$$\frac{\partial w}{\partial t} + u \frac{\partial w}{\partial x} + v \frac{\partial w}{\partial y} + w \frac{\partial w}{\partial z} = - \frac{1}{\rho_0} \frac{\partial p}{\partial z} + \nu \nabla^2 w - \frac{\rho g}{\rho_0} \quad (\text{A-3})$$

Energy Equation

$$\frac{\partial T}{\partial t} + u \frac{\partial T}{\partial x} + v \frac{\partial T}{\partial y} + w \frac{\partial T}{\partial z} = k \nabla^2 T \quad (\text{A-4})$$

Continuity Equation

$$\frac{\partial u}{\partial x} + \frac{\partial v}{\partial y} + \frac{\partial w}{\partial z} = 0 \quad (\text{A-5})$$

In describing a turbulent flow in mathematical terms, it is convenient to separate the flow into a mean motion and a fluctuating, or eddying motion. Denoting the time-average of the u-component of

velocity by \bar{u} and its velocity fluctuation by u' and using a similar notation for the other variables, one can express the following relations for the velocity components, temperature and pressure:

$$u = \bar{u} + u'; \quad v = \bar{v} + v'; \quad w = \bar{w} + w'; \quad T = \bar{T} + T'; \quad p = \bar{p} + p' \quad (\text{A-6})$$

By definition, the time-average of all fluctuating quantities is equal to zero. For example,

$$\bar{u}' = 0, \quad \bar{v}' = 0, \quad \bar{w}' = 0, \quad \bar{T}' = 0, \quad \bar{p}' = 0; \quad \frac{\partial \bar{u}'}{\partial x} = 0, \quad (\text{A-7})$$

Upon forming the time-average of equations (A-1) through (A-5), one obtains

$$\frac{\partial \bar{u}}{\partial t} + \bar{u} \frac{\partial \bar{u}}{\partial x} + \bar{v} \frac{\partial \bar{u}}{\partial y} + \bar{w} \frac{\partial \bar{u}}{\partial z} = -\frac{1}{\rho_0} \frac{\partial \bar{p}}{\partial x} + \nu \nabla^2 \bar{u} - \left(\frac{\partial \bar{u}'^2}{\partial x} + \frac{\partial \bar{u}'\bar{v}'}{\partial y} + \frac{\partial \bar{u}'\bar{w}'}{\partial z} \right) \quad (\text{A-8})$$

$$\frac{\partial \bar{v}}{\partial t} + \bar{u} \frac{\partial \bar{v}}{\partial x} + \bar{v} \frac{\partial \bar{v}}{\partial y} + \bar{w} \frac{\partial \bar{v}}{\partial z} = -\frac{1}{\rho_0} \frac{\partial \bar{p}}{\partial y} + \nu \nabla^2 \bar{v} - \left(\frac{\partial \bar{u}'\bar{v}'}{\partial x} + \frac{\partial \bar{v}'^2}{\partial y} + \frac{\partial \bar{v}'\bar{w}'}{\partial z} \right) \quad (\text{A-9})$$

$$\frac{\partial \bar{w}}{\partial t} + \bar{u} \frac{\partial \bar{w}}{\partial x} + \bar{v} \frac{\partial \bar{w}}{\partial y} + \bar{w} \frac{\partial \bar{w}}{\partial z} = -\frac{1}{\rho_0} \frac{\partial \bar{p}}{\partial z} + \nu \nabla^2 \bar{w} - \left(\frac{\partial \bar{u}'\bar{w}'}{\partial x} + \frac{\partial \bar{v}'\bar{w}'}{\partial y} + \frac{\partial \bar{w}'^2}{\partial z} \right) - \frac{\rho g}{\rho_0} \quad (\text{A-10})$$

$$\frac{\partial \bar{T}}{\partial t} + \bar{u} \frac{\partial \bar{T}}{\partial x} + \bar{v} \frac{\partial \bar{T}}{\partial y} + \bar{w} \frac{\partial \bar{T}}{\partial z} = k \nabla^2 \bar{T} - \left(\frac{\partial \bar{u}'\bar{T}'}{\partial x} + \frac{\partial \bar{v}'\bar{T}'}{\partial y} + \frac{\partial \bar{w}'\bar{T}'}{\partial z} \right) \quad (\text{A-11})$$

$$\frac{\partial \bar{u}}{\partial x} + \frac{\partial \bar{v}}{\partial y} + \frac{\partial \bar{w}}{\partial z} = 0 \quad (\text{A-12})$$

Employing the eddy diffusivity concept,

$$-\bar{u}'^2 = \alpha_x \frac{\partial \bar{u}}{\partial x} + \alpha_x \frac{\partial \bar{u}}{\partial x}$$

$$-\bar{u}'\bar{v}' = \alpha_y \frac{\partial \bar{u}}{\partial y} + \alpha_x \frac{\partial \bar{v}}{\partial x}$$

$$-\bar{u}'\bar{w}' = \alpha_z \frac{\partial \bar{u}}{\partial z} + \alpha_x \frac{\partial \bar{w}}{\partial x}$$

$$-\bar{v}'^2 = \alpha_y \frac{\partial \bar{v}}{\partial y} + \alpha_y \frac{\partial \bar{v}}{\partial y}$$

$$-\bar{v}'\bar{u}' = \alpha_x \frac{\partial \bar{v}}{\partial x} + \alpha_y \frac{\partial \bar{u}}{\partial y}$$

$$-\bar{v}'\bar{w}' = \alpha_z \frac{\partial \bar{v}}{\partial z} + \alpha_y \frac{\partial \bar{w}}{\partial y}$$

$$-\bar{w}'\bar{u}' = \alpha_x \frac{\partial \bar{w}}{\partial x} + \alpha_z \frac{\partial \bar{u}}{\partial z}$$

$$-\bar{w}'\bar{v}' = \alpha_y \frac{\partial \bar{w}}{\partial y} + \alpha_z \frac{\partial \bar{v}}{\partial z}$$

$$-\bar{w}'^2 = \alpha_z \frac{\partial \bar{w}}{\partial z} + \alpha_z \frac{\partial \bar{w}}{\partial z}$$

$$-\bar{u}'\bar{T}' = d_x \frac{\partial \bar{T}}{\partial x}$$

$$-\bar{v}'\bar{T}' = d_y \frac{\partial \bar{T}}{\partial y}$$

$$-\bar{w}'\bar{T}' = d_z \frac{\partial \bar{T}}{\partial z}$$

(A-13)

after substituting equations (A-13) into equations (A-8) through (A-11), one obtains the following equations for incompressible turbulent flow:

$$\frac{\partial \bar{u}}{\partial t} + \bar{u} \frac{\partial \bar{u}}{\partial x} + \bar{v} \frac{\partial \bar{u}}{\partial y} + \bar{w} \frac{\partial \bar{u}}{\partial z} = -\frac{1}{\rho_0} \frac{\partial \bar{p}}{\partial x} + v_x \frac{\partial^2 \bar{u}}{\partial x^2} + v_y \frac{\partial^2 \bar{u}}{\partial y^2} + v_z \frac{\partial^2 \bar{u}}{\partial z^2} \quad (\text{A-14})$$

$$\frac{\partial \bar{v}}{\partial t} + \bar{u} \frac{\partial \bar{v}}{\partial x} + \bar{v} \frac{\partial \bar{v}}{\partial y} + \bar{w} \frac{\partial \bar{v}}{\partial z} = -\frac{1}{\rho_0} \frac{\partial \bar{p}}{\partial y} + v_x \frac{\partial^2 \bar{v}}{\partial x^2} + v_y \frac{\partial^2 \bar{v}}{\partial y^2} + v_z \frac{\partial^2 \bar{v}}{\partial z^2} \quad (\text{A-15})$$

$$\frac{\partial \bar{w}}{\partial t} + \bar{u} \frac{\partial \bar{w}}{\partial x} + \bar{v} \frac{\partial \bar{w}}{\partial y} + \bar{w} \frac{\partial \bar{w}}{\partial z} = -\frac{1}{\rho_0} \frac{\partial \bar{p}}{\partial z} + v_x \frac{\partial^2 \bar{w}}{\partial x^2} + v_y \frac{\partial^2 \bar{w}}{\partial y^2} + v_z \frac{\partial^2 \bar{w}}{\partial z^2} - \frac{\rho g}{\rho_0} \quad (\text{A-16})$$

$$\frac{\partial \bar{T}}{\partial t} + \bar{u} \frac{\partial \bar{T}}{\partial x} + \bar{v} \frac{\partial \bar{T}}{\partial y} + \bar{w} \frac{\partial \bar{T}}{\partial z} = D_x \frac{\partial^2 \bar{T}}{\partial x^2} + D_y \frac{\partial^2 \bar{T}}{\partial y^2} + D_z \frac{\partial^2 \bar{T}}{\partial z^2} \quad (\text{A-17})$$

$$\frac{\partial \bar{u}}{\partial x} + \frac{\partial \bar{v}}{\partial y} + \frac{\partial \bar{w}}{\partial z} = 0 \quad (\text{A-18})$$

where

$$v_x = v + \alpha_x$$

$$v_y = v + \alpha_y$$

$$v_z = v + \alpha_z$$

$$D_x = k + \alpha_x$$

$$D_y = k + \alpha_y$$

$$D_z = k + \alpha_z \quad (\text{A-19})$$

In the present analysis, the horizontal components of diffusivities are considered to be equal, i.e.,

$$v_h = v_x = v_y$$

$$v_v = v_z$$

$$D_h = D_x = D_y$$

$$D_v = D_z \tag{A-20}$$

The magnitude of momentum eddy viscosities and thermal eddy diffusivities ($\alpha_x, \alpha_y, \alpha_z, d_x, d_y, d_z$) are usually so large in comparison with their molecular counterparts (ν, k) that the contributions of the latter are neglected. For practical calculations, it is satisfactory to assume that the horizontal and vertical momentum eddy viscosities are equal to the horizontal and vertical thermal eddy diffusivities respectively such that (14, 5, 13)

$$v_h = D_h$$

$$v_v = D_v \tag{A-21}$$

Liggett (19), Wada (39), and Lerman (17) are among the recent investigators of eddy diffusivity concept. Based on their studies, the following average values for the eddy viscosities in the horizontal and vertical directions, v_h and v_v , are used in this analysis

$$v_h = 1350 \text{ ft}^2/\text{sec}$$

$$v_v = .4 \text{ ft}^2/\text{sec}$$

Considering the above simplifications, equations (A-14) through (A-18) become identical with equations (1) through (5) except that, for convenience, the superscript (-) has been dropped in the latter equations.

APPENDIX 2. Verification of Stability Analysis

The purpose of this appendix is to demonstrate the validity of the rule which applies to an explicit computational scheme as discussed in Section 4. This rule is verified by developing the stability criteria for the energy equation (5) using the two following methods:

- a) Setting the coefficients multiplying each of the temperature values equal to a positive quantity (3)
- b) Von-Neumann stability analysis (1)

In order to apply stability analysis to a nonlinear equation, the equation must be first linearized. To linearize the equation, one assumes that the temperature T consists of a steady-state component T' and a small variation about steady-state \dot{T} such that

$$T = T' + \dot{T}$$

By substituting these quantities in the energy equation, and considering the velocity components constant over the integration time step, one will obtain the following linearized equation:

$$\frac{\partial \dot{T}}{\partial t} = -u \frac{\partial \dot{T}}{\partial x} - v \frac{\partial \dot{T}}{\partial y} - w \frac{\partial \dot{T}}{\partial z} + D_h \left(\frac{\partial^2 \dot{T}}{\partial x^2} + \frac{\partial^2 \dot{T}}{\partial y^2} \right) + D_v \frac{\partial^2 \dot{T}}{\partial z^2} \quad (B-1)$$

where superscript dot ($\dot{\cdot}$) indicates linearized quantities. By applying a forward two points finite difference approximation to the temperature time derivative and centered two and three points finite difference approximations to the first and second spatial differences respectively, and denoting $\dot{T}_{i,j,k}$ by \dot{T} for convenience, one obtains

$$\begin{aligned} \frac{\dot{T}^n - \dot{T}}{\Delta t} = & -u \frac{\dot{T}_{i+1} - \dot{T}_{i-1}}{2\Delta x} - v \frac{\dot{T}_{j+1} - \dot{T}_{j-1}}{2\Delta y} - w \frac{\dot{T}_{k+1} - \dot{T}_{k-1}}{2\Delta z} + \frac{\dot{T}_{i+1} - 2\dot{T} + \dot{T}_{i-1}}{\Delta x^2} \\ & + D_h \frac{\dot{T}_{j+1} - 2\dot{T} + \dot{T}_{j-1}}{\Delta x^2} + D_v \frac{\dot{T}_{k+1} - 2\dot{T} + \dot{T}_{k-1}}{\Delta x^2} \end{aligned} \quad (B-2)$$

solving for \dot{T}^n

$$\begin{aligned} \dot{T}^n = & \dot{T}_{i-1} \left(\frac{u\Delta t}{2\Delta x} + \frac{D_h\Delta t}{\Delta x^2} \right) + \dot{T} \left(1 - \frac{2\Delta t D_h}{\Delta x^2} - \frac{2\Delta t D_v}{\Delta y^2} - \frac{2\Delta t D_v}{\Delta z^2} \right) \\ & + \dot{T}_{i+1} \left(-\frac{u\Delta t}{2\Delta x} + \frac{D_h\Delta t}{\Delta x^2} \right) + \dot{T}_{j-1} \left(\frac{v\Delta t}{2\Delta y} + \frac{D_h\Delta t}{\Delta y^2} \right) \\ & + \dot{T}_{j+1} \left(-\frac{v\Delta t}{2\Delta y} + \frac{D_h\Delta t}{\Delta y^2} \right) + \dot{T}_{k-1} \left(\frac{w\Delta t}{2\Delta z} + \frac{D_v\Delta t}{\Delta z^2} \right) + \dot{T}_{k+1} \left(-\frac{w\Delta t}{2\Delta z} + \frac{D_v\Delta t}{\Delta z^2} \right) \end{aligned} \quad (B-3)$$

a) Employing the rule of positive coefficients, which applies to explicit computational scheme discussed previously, the following stability criteria will result from coefficient of \dot{T}_{i+1} , \dot{T}_{i-1}

$$\Delta x \leq \frac{2D_h}{|u|} \quad (B-4)$$

similarly, from \dot{T}_{j+1} , \dot{T}_{j-1} , \dot{T}_{k+1} , \dot{T}_{k-1} coefficients, one obtains

$$\Delta y \leq \frac{2D_h}{|v|} \quad \Delta z \leq \frac{2D_v}{|w|} \quad (B-5)$$

Finally, from the coefficient of \dot{T} , the following temporal condition results

$$\Delta t \leq 1/(2D_h/\Delta x^2 + 2D_h/\Delta y^2 + 2D_v/\Delta z^2) \quad (\text{B-6})$$

If $D_h = D_v$ and $\Delta x = \Delta y = \Delta z$, the expression for Δt simplifies to

$$\Delta t \leq \frac{\Delta x^2}{6D_h} \quad (\text{B-7})$$

b) Von-Neumann stability analysis.

According to the Von-Neumann stability analysis approach (1)

$$T^n = e^{\delta t} e^{i\alpha x} e^{i\beta y} e^{i\gamma z} \quad (\text{B-8})$$

where

$$x = \sum_{\ell=1}^{\ell} \Delta x_{\ell}$$

$$y = \sum_{m=1}^m \Delta y_m$$

$$z = \sum_{p=1}^p \Delta z_p$$

$$t = n\Delta t \quad (\text{B-9})$$

Substituting the above quantities into equation (B-3), and dividing by $e^{\delta t} e^{i\alpha x} e^{i\beta y} e^{i\gamma z}$, the following results:

$$e^{\delta \Delta t} = \left(\frac{u\Delta t}{2\Delta x} + \frac{D_h \Delta t}{\Delta x^2} \right) e^{-i\alpha \Delta t} + \left(-\frac{u\Delta t}{2\Delta x} + \frac{D_h \Delta t}{\Delta x^2} \right) e^{i\alpha \Delta t}$$

$$\begin{aligned}
& + \left(\frac{v\Delta t}{2\Delta y} + \frac{D_h\Delta t}{\Delta y^2} \right) e^{-i\beta\Delta y} + \left(-\frac{v\Delta t}{2\Delta y} + \frac{D_h\Delta t}{\Delta y^2} \right) e^{i\beta\Delta y} \\
& + \left(\frac{w\Delta t}{2\Delta z} + \frac{D_v\Delta t}{\Delta z^2} \right) e^{-i\gamma\Delta z} + \left(-\frac{w\Delta t}{2\Delta z} + \frac{D_v\Delta t}{\Delta z^2} \right) e^{i\gamma\Delta z} \\
& + \left(1 - \frac{2\Delta t D_h}{\Delta x^2} - \frac{2\Delta t D_h}{\Delta y^2} - \frac{2\Delta t D_v}{\Delta z^2} \right)
\end{aligned} \tag{B-10}$$

For the above equation to be numerically stable, the absolute value of $e^{\delta\Delta t}$ must be set less than or equal to one. This leads to

$$\left(1 - 4 \frac{\Delta t D_h}{\Delta x^2} \sin^2 \frac{\alpha\Delta x}{2} - 4 \frac{\Delta t D_h}{\Delta y^2} \sin^2 \frac{\beta\Delta y}{2} - 4 \frac{\Delta t D_v}{\Delta z^2} \sin^2 \frac{\alpha\Delta z}{2} \right)^2 + \tag{B-11}$$

$$\left(\frac{u\Delta t}{\Delta x} \sin \alpha\Delta x + \frac{v\Delta t}{\Delta y} \sin \alpha\Delta y + \frac{w\Delta t}{\Delta z} \sin \alpha\Delta z \right)^2 \leq 1 \tag{B-12}$$

Expanding and assuming that $\Delta x = \Delta y = \Delta z$, and $D_h = D_v$, the following will be obtained:

$$1 + \left(144 \frac{\Delta t^2 D_h^2}{\Delta x^4} - 36 \frac{u^2 \Delta t^2}{\Delta x^2} \right) \sin^4 \frac{\alpha\Delta x}{2} + \left(36 \frac{u^2 \Delta t^2}{\Delta x^2} - 24 \frac{\Delta t D_h}{\Delta x^2} \right) \sin^2 \frac{\alpha\Delta x}{2} \leq 1 \tag{B-13}$$

In order for the above inequality to be less or equal one, the following inequalities must hold.

$$144 \frac{\Delta t^2 D_h^2}{\Delta x^4} - 36 \frac{u^2 \Delta t^2}{\Delta x^2} < \epsilon^2 \tag{B-14}$$

$$36 \frac{u^2 \Delta t^2}{\Delta x^2} - 24 \frac{\Delta t D_h}{\Delta x^2} = -\epsilon^2 \quad (\text{B-15})$$

where ϵ^2 is very small number. From equation (B-14), the spatial stability criterion may be written as

$$144 \frac{\Delta t^2 D_h^2}{\Delta x^4} \geq 36 \frac{u^2 \Delta t^2}{\Delta x^2}$$

or

$$\Delta x \leq \frac{2D_h}{|u|} \quad (\text{B-16})$$

Adding equations (B-14) and (B-15), the upper limit for Δt becomes:

$$\Delta t \leq \frac{\Delta x^2}{6D_h} \quad (\text{B-17})$$

These results are in agreement with those obtained in part (a) of this appendix.

APPENDIX 3. Calculation of the First and
Second Derivatives at the Boundary and in the Field¹

The calculation of the first derivative in the flow field is based on a quadratic approximation. Let $q(x)$ be defined by a parabolic equation

$$q(x) = ax^2 + bx + c \quad (C-1)$$

shown in Fig. 23. Values of the function at nodal points x_1 , x_2 , and x_3 , are q_1 , q_2 , and q_3 , respectively. Therefore

$$\begin{Bmatrix} q_1 \\ q_2 \\ q_3 \end{Bmatrix} = \begin{bmatrix} x_1^2 & x_1 & 1 \\ x_2^2 & x_2 & 1 \\ x_3^2 & x_3 & 1 \end{bmatrix} \begin{Bmatrix} a \\ b \\ c \end{Bmatrix} \quad (C-2)$$

combining equations (C-1) and (C-2) yields

$$q(x) = \{x^2 \quad x \quad 1\} \begin{bmatrix} x_1^2 & x_1 & 1 \\ x_2^2 & x_2 & 1 \\ x_3^2 & x_3 & 1 \end{bmatrix}^{-1} \begin{Bmatrix} q_1 \\ q_2 \\ q_3 \end{Bmatrix} \quad (C-3)$$

¹Nomenclature used in this Appendix is defined within the text and not in Section 7

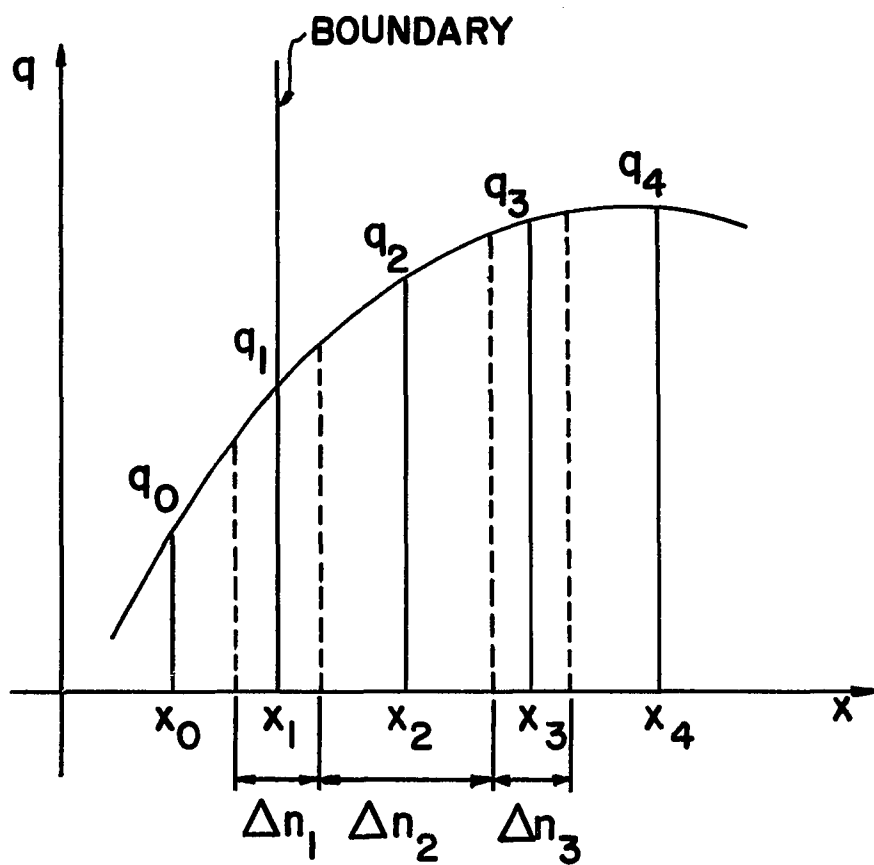


FIG.23 CALCULATION OF FIRST AND SECOND DERIVATIVES

First Derivative at x_1

The derivative at x_1 is given by

$$\left. \frac{dq(x)}{dx} \right|_{x=x_1} = \{2x_1 \quad 1 \quad 0\} \begin{bmatrix} x_1^2 & x_1 & 1 \\ x_2^2 & x_2 & 1 \\ x_3^2 & x_3 & 1 \end{bmatrix}^{-1} \begin{Bmatrix} q_1 \\ q_2 \\ q_3 \end{Bmatrix} \quad (C-4)$$

Inverting the matrix, yields

$$\left. \frac{dq(x)}{dx} \right|_{x=x_1} = \{2x_1 \quad 1 \quad 0\} \begin{bmatrix} x_2 - x_3 & x_3 - x_1 & x_1 - x_2 \\ x_3^2 - x_2^2 & x_1^2 - x_3^2 & x_2^2 - x_1^2 \\ x_3 x_2 (x_2 - x_3) & x_3 x_1 (x_3 - x_1) & x_1 x_2 (x_1 - x_2) \end{bmatrix} \begin{Bmatrix} q_1 \\ q_2 \\ q_3 \end{Bmatrix}$$

det $\begin{vmatrix} x_1^2 & x_1 & 1 \\ x_2^2 & x_2 & 1 \\ x_3^2 & x_3 & 1 \end{vmatrix}$

(C-5)

After multiplication, one obtains

$$\left. \frac{dq(x)}{dx} \right|_{x=x_1} = \frac{(x_3 - x_2)(x_2 + x_3 - 2x_1)q_1 - (x_3 - x_1)^2 q_2 + (x_1 - x_2)^2 q_3}{(x_2 - x_1)(x_3 - x_2)(x_1 - x_3)} \quad (C-6)$$

In many instances the value of the first derivative is known at the boundary, i.e.,

$$\left. \frac{dq(x)}{dx} \right|_{x=x_1} = C \quad (C-7)$$

For this boundary condition, a relationship between q_1 , q_2 , and q_3 may be obtained by first defining the element boundaries (dotted lines on Fig. 23) and observing that each nodal point is located at the midpoint between its boundary lines. Thus,

$$x_2 - x_1 = \frac{1}{2} (\Delta n_1 + \Delta n_2) \quad (C-8)$$

$$x_3 - x_2 = \frac{1}{2} (\Delta n_2 + \Delta n_3) \quad (C-9)$$

$$x_1 - x_3 = -\frac{1}{2} (\Delta n_1 + 2\Delta n_2 + \Delta n_3) \quad (C-10)$$

Defining

$$\Delta n_{12} = \Delta n_1 + \Delta n_2 \quad (C-11)$$

and

$$\sigma = \frac{\Delta n_1 + \Delta n_2}{\Delta n_2 + \Delta n_3} \quad (C-12)$$

Combining the equations (C-8) through (C-12) gives the value of the derivative at the boundary

$$\left. \frac{dq(x)}{dx} \right|_{x=x_1} = - \frac{2}{\Delta n_{12}(1+\sigma)} \left[(1+2\sigma) q_1 - (1+\sigma)^2 q_2 + \sigma^2 q_3 \right] \quad (C-13)$$

or

$$\left. \frac{dq(x)}{dx} \right|_{x=x_1} = \frac{q_2 - q_1}{x_2 - x_1} \left(\frac{1+2\sigma}{1+\sigma} \right) + \frac{q_3 - q_2}{x_3 - x_2} \left(\frac{-\sigma}{1+\sigma} \right) \quad (C-14)$$

which means that the slope at $x=x_1$ is equal to the weighted average of the slopes in intervals $x_2 - x_1$ and $x_3 - x_2$. It should be noted that for equally spaced increments, the slope in the interval $x_2 - x_1$ is weighted by 1.5, while the slope in the interval $x_3 - x_2$ is weighted by - 0.5.

Using equations (C-7) and (C-14), yields a relationship between q_1 , q_2 , and q_3

$$- \frac{2}{\Delta n_{12}(1+\sigma)} \left[(1+2\sigma) q_1 - (1+\sigma)^2 q_2 + \sigma^2 q_3 \right] = C \quad (C-15)$$

from which the value of the variable at the boundary can be determined in terms of the values within the flow region.

$$q_1 = \frac{1}{1+2\sigma} \left[(1+\sigma)^2 q_2 - \sigma^2 q_3 - \frac{1}{2} \Delta n_{12} (1+\sigma) C \right] \quad (C-16)$$

First Derivative at x_2

The value of the first derivative in the field is calculated by differentiating equation (C-3) and setting $x=x_2$

$$\frac{dq(x)}{dx} \Big|_{x=x_2} = \{2x_2 \quad 1 \quad 0\} \begin{bmatrix} x_1^2 & x_1 & 1 \\ x_2^2 & x_2 & 1 \\ x_3^2 & x_3 & 1 \end{bmatrix}^{-1} \begin{Bmatrix} q_1 \\ q_2 \\ q_3 \end{Bmatrix} \quad (\text{C-17})$$

Inverting the matrix yields,

$$\frac{dq(x)}{dx} \Big|_{x=x_2} = \{2x_2 \quad 1 \quad 0\} \det \begin{bmatrix} x_1^2 & x_1 & 1 \\ x_2^2 & x_2 & 1 \\ x_3^2 & x_3 & 1 \end{bmatrix}^{-1} \begin{bmatrix} x_2 - x_3 & x_3 - x_1 & x_1 - x_2 \\ x_3^2 - x_2^2 & x_1^2 - x_3^2 & x_2^2 - x_1^2 \\ x_3x_2(x_2-x_3) & x_3x_1(x_3-x_1) & x_1x_2(x_1-x_2) \end{bmatrix} \begin{Bmatrix} q_1 \\ q_2 \\ q_3 \end{Bmatrix} \quad (\text{C-18})$$

After multiplication, one obtains

$$\frac{dq(x)}{dx} \Big|_{x=x_2} = \frac{(x_3-x_2)^2 q_1 + (x_3-x_1)(2x_2-x_3-x_1) q_2 - (x_2-x_1)^2 q_3}{(x_2-x_1)(x_3-x_2)(x_1-x_3)} \quad (\text{C-19})$$

Using relationships (C-8), (C-9), and (C-10), the above equation becomes

$$\left. \frac{dq(x)}{dx} \right|_{x=x_2} = - \frac{2}{\Delta n_{12}(1+\sigma)} \left[q_1 + (\sigma^2 - 1) q_2 - \sigma^2 q_3 \right] \quad (C-20)$$

or

$$\left. \frac{dq(x)}{dx} \right|_{x=x_2} = \frac{q_2 - q_1}{x_2 - x_1} \left(\frac{1}{1+\sigma} \right) + \frac{q_3 - q_2}{x_3 - x_2} \left(\frac{\sigma}{1+\sigma} \right) \quad (C-21)$$

which means that the slope at $x=x_2$ is equal to the weighted average of the slopes in intervals x_2-x_1 and x_3-x_2 . It should be noted that for equally spaced increments, the slope at x_2 becomes equal to the mean value of the slopes in intervals x_2-x_1 and x_3-x_2 . Furthermore, using equations (C-12), (C-10), (C-9), and (C-8), it can easily be shown that the weighting function can be expressed in terms of spatial differences as follows:

$$\frac{1}{1+\sigma} = \frac{x_3 - x_2}{x_3 - x_1} \quad (C-22)$$

$$\frac{\sigma}{1+\sigma} = \frac{x_2 - x_1}{x_3 - x_1} \quad (C-23)$$

First Derivative at x_3

The derivative at x_3 is given by

$$\left. \frac{dq(x)}{dx} \right|_{x=x_3} = \{2x_3 \quad 1 \quad 0\} \det \begin{bmatrix} x_1^2 & x_1 & 1 \\ x_2^2 & x_2 & 1 \\ x_3^2 & x_3 & 1 \end{bmatrix} \begin{bmatrix} x_2 - x_3 & x_3 - x_1 & x_1 - x_2 \\ x_3^2 - x_2^2 & x_1^2 - x_3^2 & x_2^2 - x_1^2 \\ x_3 x_2 (x_2 - x_3) & x_3 x_1 (x_3 - x_1) & x_1 x_2 (x_1 - x_2) \end{bmatrix} \begin{Bmatrix} q_1 \\ q_2 \\ q_3 \end{Bmatrix} \quad (\text{C-24})$$

After multiplication, one obtains

$$\left. \frac{dq(x)}{dx} \right|_{x=x_3} = \frac{-(x_3 - x_2)^2 q_1 + (x_3 - x_1)^2 q_2 + (x_1 - x_2) (x_3 - x_2 + x_3 - x_1) q_3}{(x_2 - x_3) (x_1 - x_3) (x_1 - x_2)} \quad (\text{C-25})$$

Using relationships (C-8), (C-9), and (C-10) and redefining

$$\Delta n_{23} = \Delta n_2 + \Delta n_3 \quad (\text{C-26})$$

and

$$\sigma' = \frac{\Delta n_3 + \Delta n_2}{\Delta n_1 + \Delta n_2} = \frac{1}{\sigma} \quad (\text{C-27})$$

equation (C-25) becomes

$$\left. \frac{dq(x)}{dx} \right|_{x=x_3} = \frac{2}{\Delta n_{23} (1 + \sigma')} \left[(1 + 2\sigma') q_3 - (1 + \sigma')^2 q_2 + \sigma'^2 q_1 \right] \quad (\text{C-28})$$

or

$$\left. \frac{dq(x)}{dx} \right|_{x=x_3} = \frac{q_2 - q_1}{x_2 - x_1} \left(\frac{-\sigma'}{1 + \sigma'} \right) + \frac{q_3 - q_2}{x_3 - x_2} \left(\frac{1 + 2\sigma'}{1 + \sigma'} \right) \quad (C-29)$$

Second Derivative

In a similar manner, the second derivative can be calculated by differentiating equation (C-1) twice with respect to x to obtain

$$\frac{d^2q(x)}{dx^2} \Big|_2 = \frac{(x_2 - x_3)q_1 + (x_3 - x_1)q_2 + (x_1 - x_2)q_3}{(x_2 - x_1)(x_3 - x_2)(x_1 - x_3)} \quad (C-30)$$

or

$$\frac{d^2q(x)}{dx^2} = \frac{1}{\frac{1}{4}(\Delta n_{12} + \Delta n_{23})} \left(\frac{q_3 - q_2}{x_3 - x_2} - \frac{q_2 - q_1}{x_2 - x_1} \right) \quad (C-31)$$

It should be noted that:

1) The above equation is exactly identical with the central finite difference form for the second derivative. This is in contrast with the first derivative which introduced weighting factors.

2) The value of the second derivative is equal at positions x_1 , x_2 , or x_3 .

Furthermore, the coefficient of the finite difference quotients on the right hand side of equation (C-31) in terms of spatial differences becomes

$$\frac{1}{\frac{1}{4} (\Delta n_{12} + \Delta n_{23})} = \frac{2}{x_3 - x_1}$$

(C-32)

PART TWO

USER'S MANUAL

1. DESCRIPTION OF THERMA DIGITAL COMPUTER PROGRAM

The program THERMA, developed for the analysis of three dimensional thermal stratification and circulation in water bodies subjected to thermal discharge, consists of a MAIN program and several subroutines as shown in Figs. 24 to 28. A tabular outline of the MAIN program and its subroutines together with their functions are presented below and followed by a more detailed description.

<u>Program</u>	<u>Function</u>
MAIN	The MAIN program reads the input data in, initializes the problem, calls the subroutines needed to solve the problem, and controls the termination of program.
<u>Subroutine</u>	<u>Function</u>
DERIV	This subroutine calculates spatial and time derivatives of variables as needed.
PRESS	This subroutine calculates the updated values of pressure.
SUM	This subroutine performs the time integration.

READ INPUT DATA

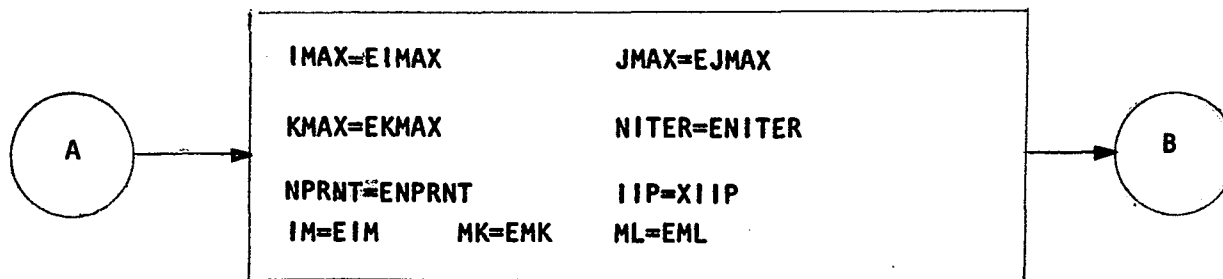
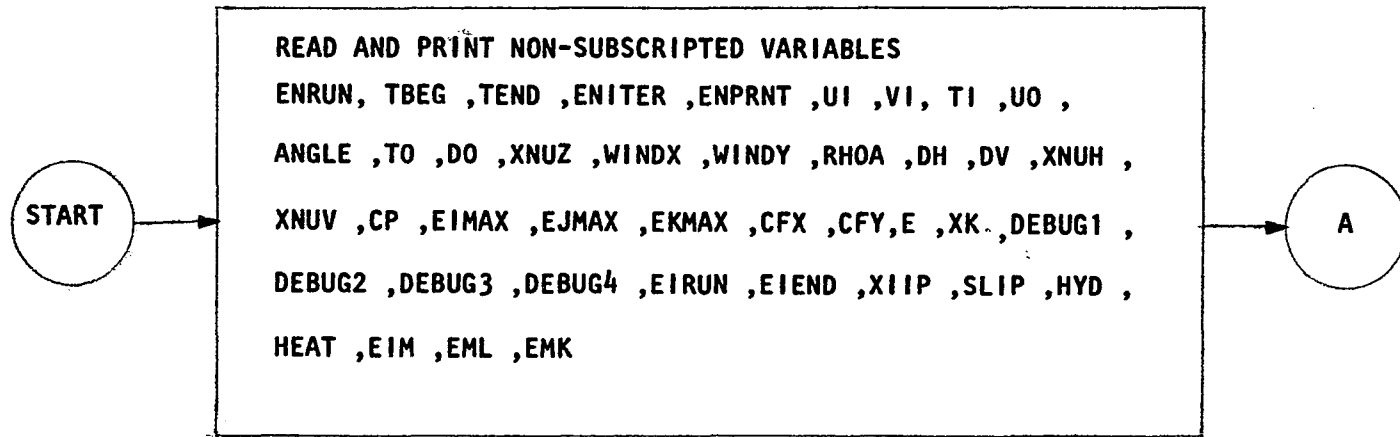


FIG 24a FLOW CHART OF MAIN PROGRAM

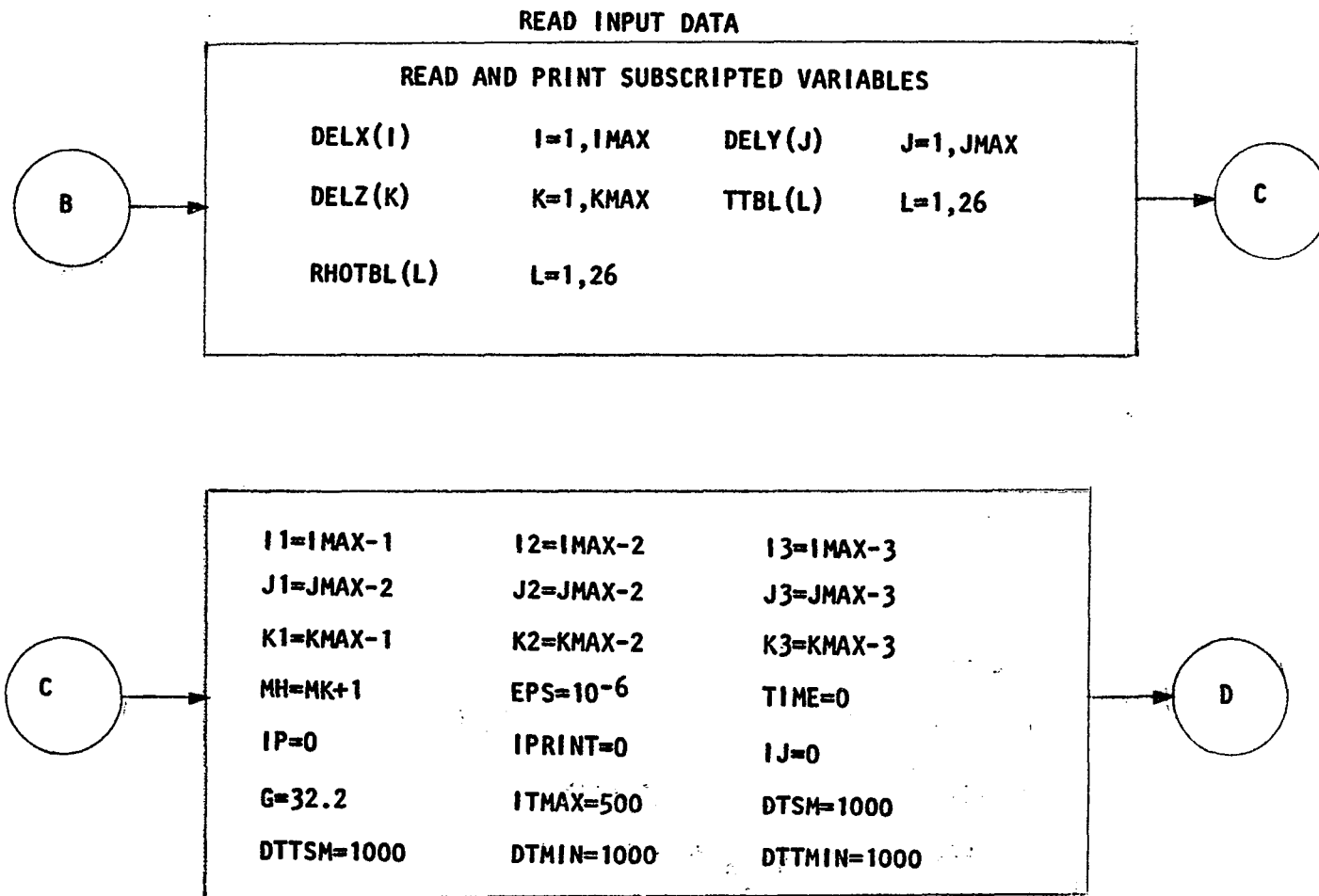


FIG 24b FLOW CHART OF MAIN PROGRAM

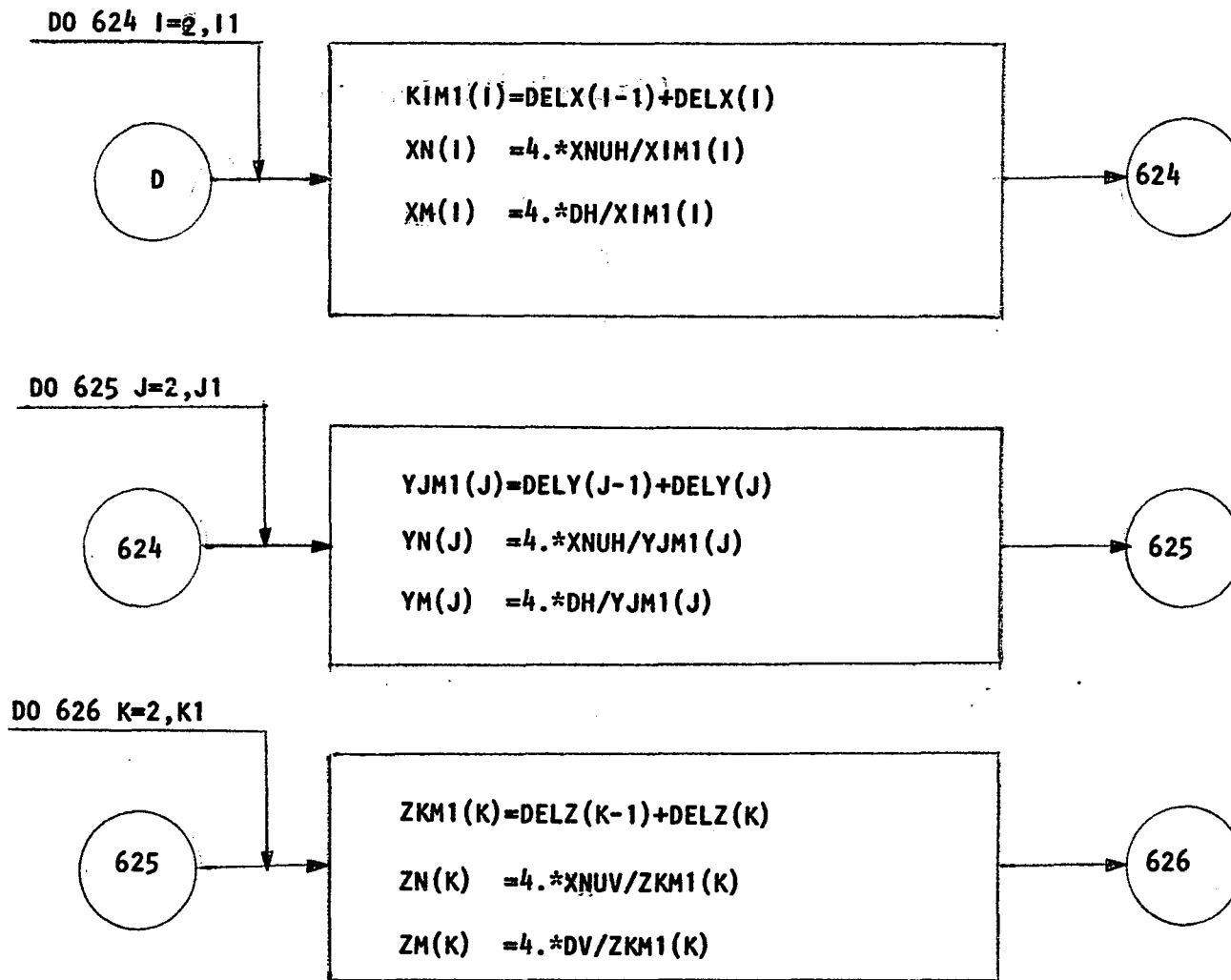


FIG 24c FLOW CHART OF MAIN PROGRAM.

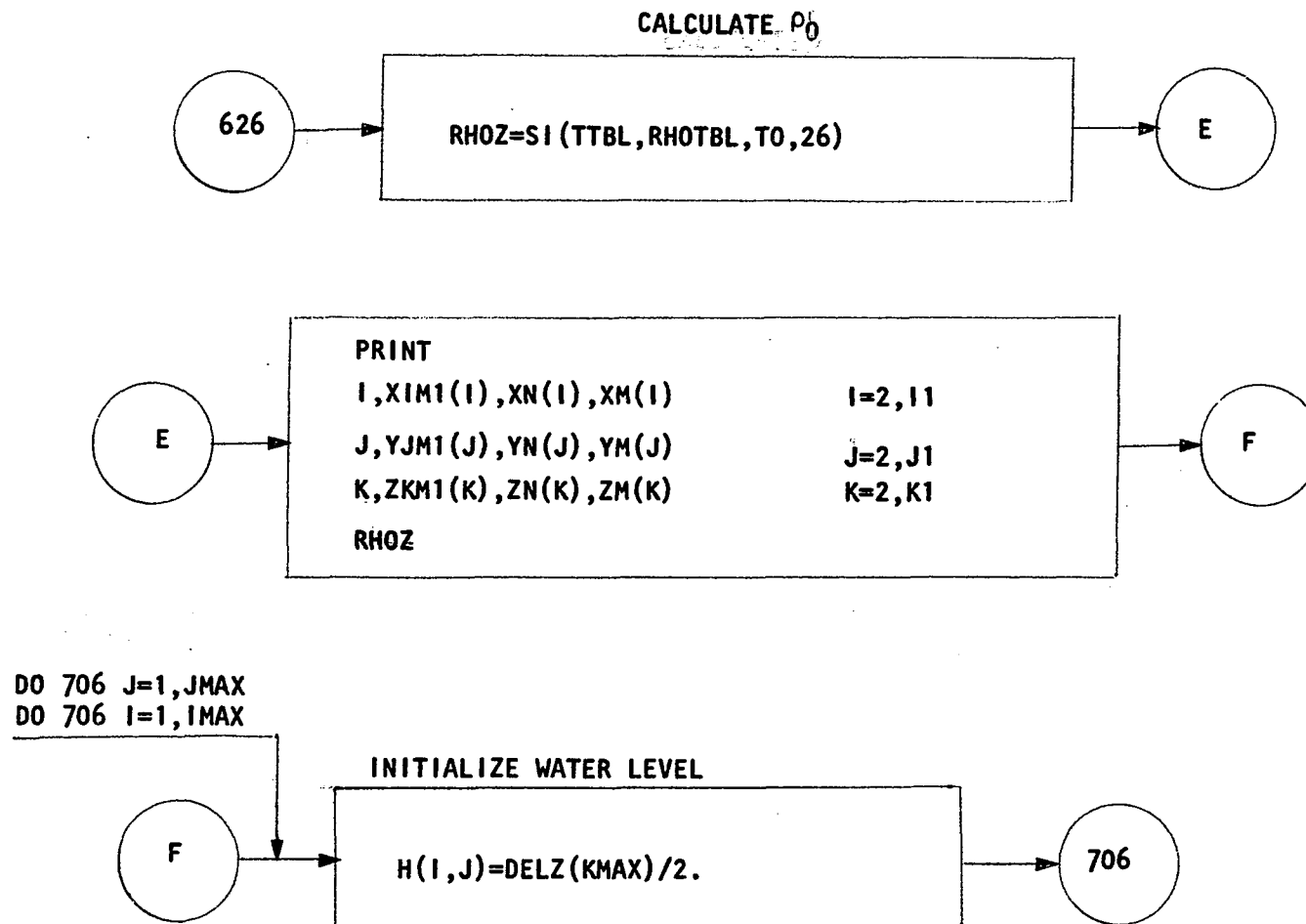
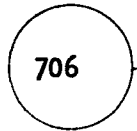


FIG 24d FLOW CHART OF MAIN PROGRAM

DO 613 K=2,K1
 DO 613 J=2,J1
 DO 613 I=2,I1



CALCULATE ALLOWABLE DT FOR FLOW REGION

$$DT=1/(2*(x_n(I)/X_{I1}(I)+Y_N(J)/Y_{J1}(J)+Z_N(K)/Z_{K1}(K)))$$

$$DTT=1/(2*(X_M(I)/X_{M1}(I)+Y_M(J)/Y_{M1}(J)+Z_M(K)/Z_{M1}(K)))$$

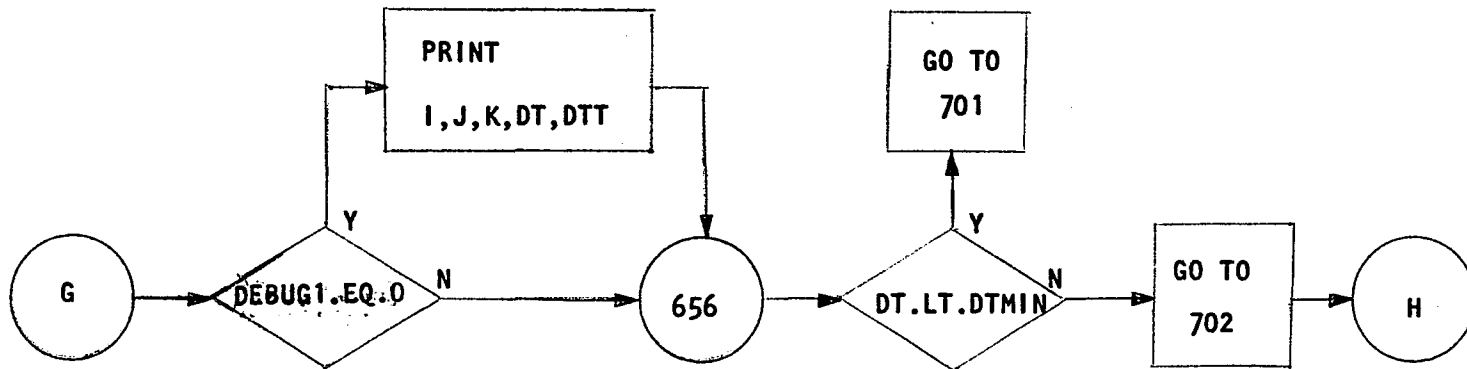
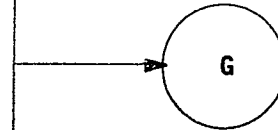


FIG 24e FLOW CHART OF MAIN PROGRAM

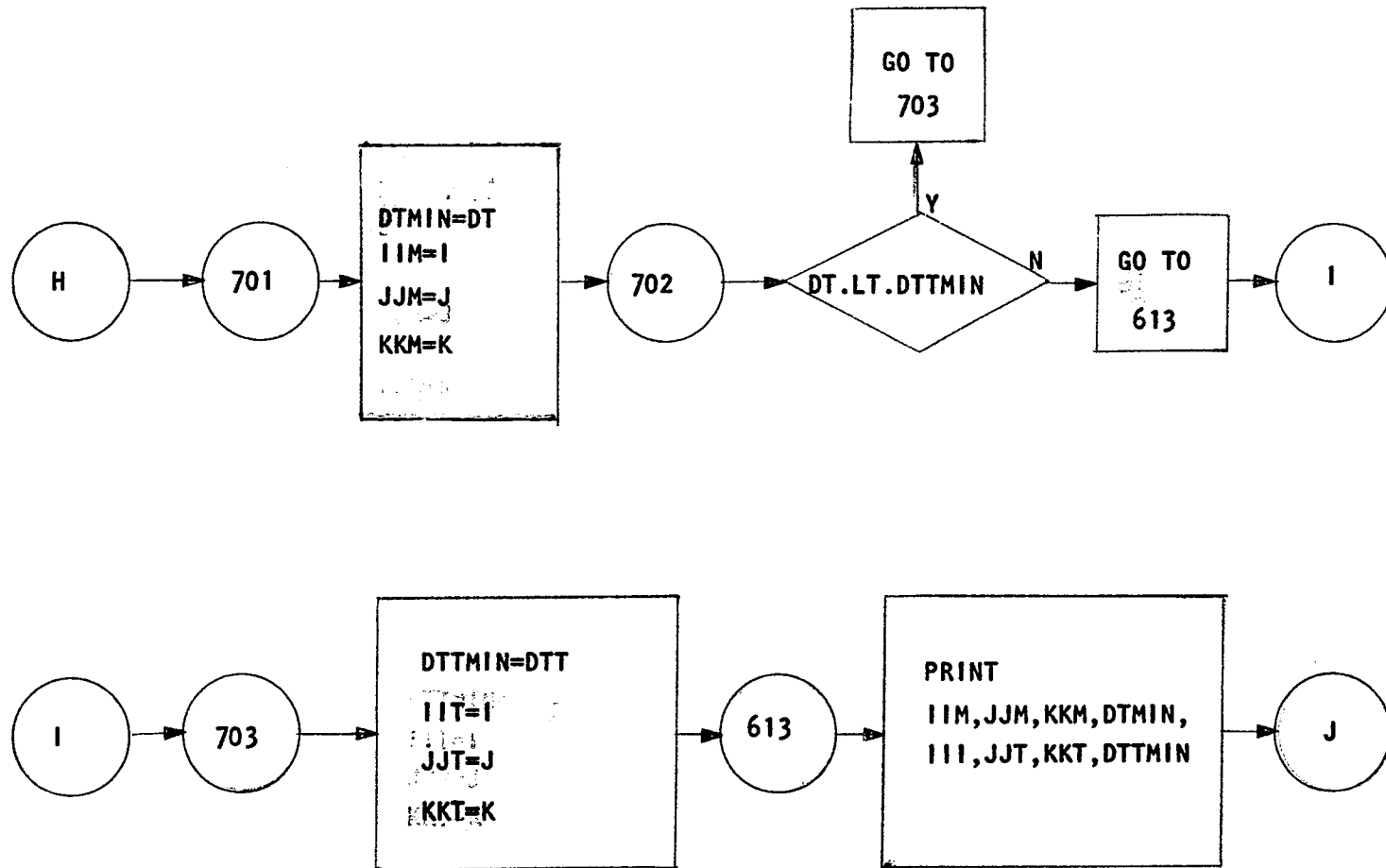


FIG 24f FLOW CHART OF MAIN PROGRAM

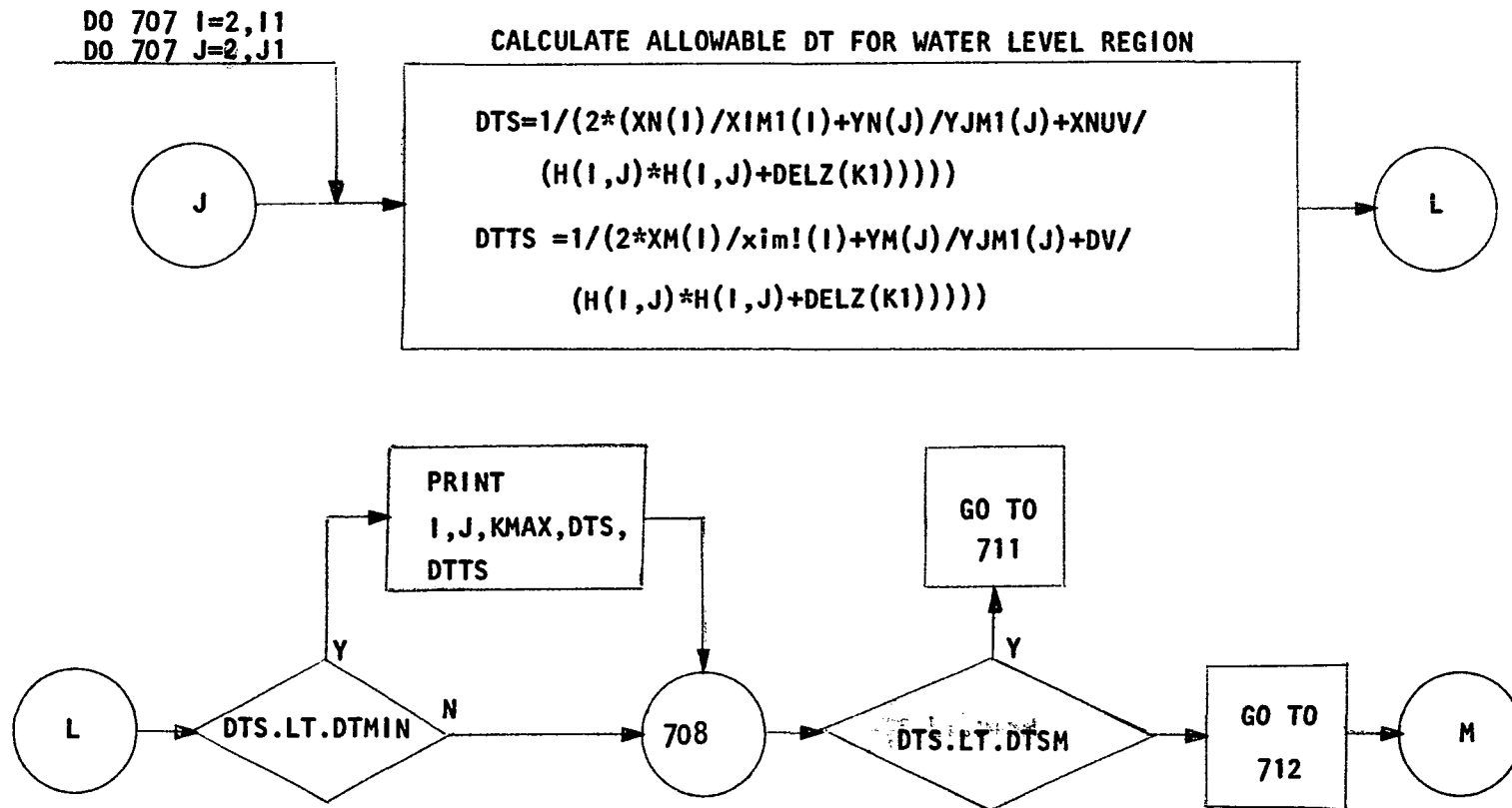


FIG 24g FLOW CHART OF MAIN PROGRAM

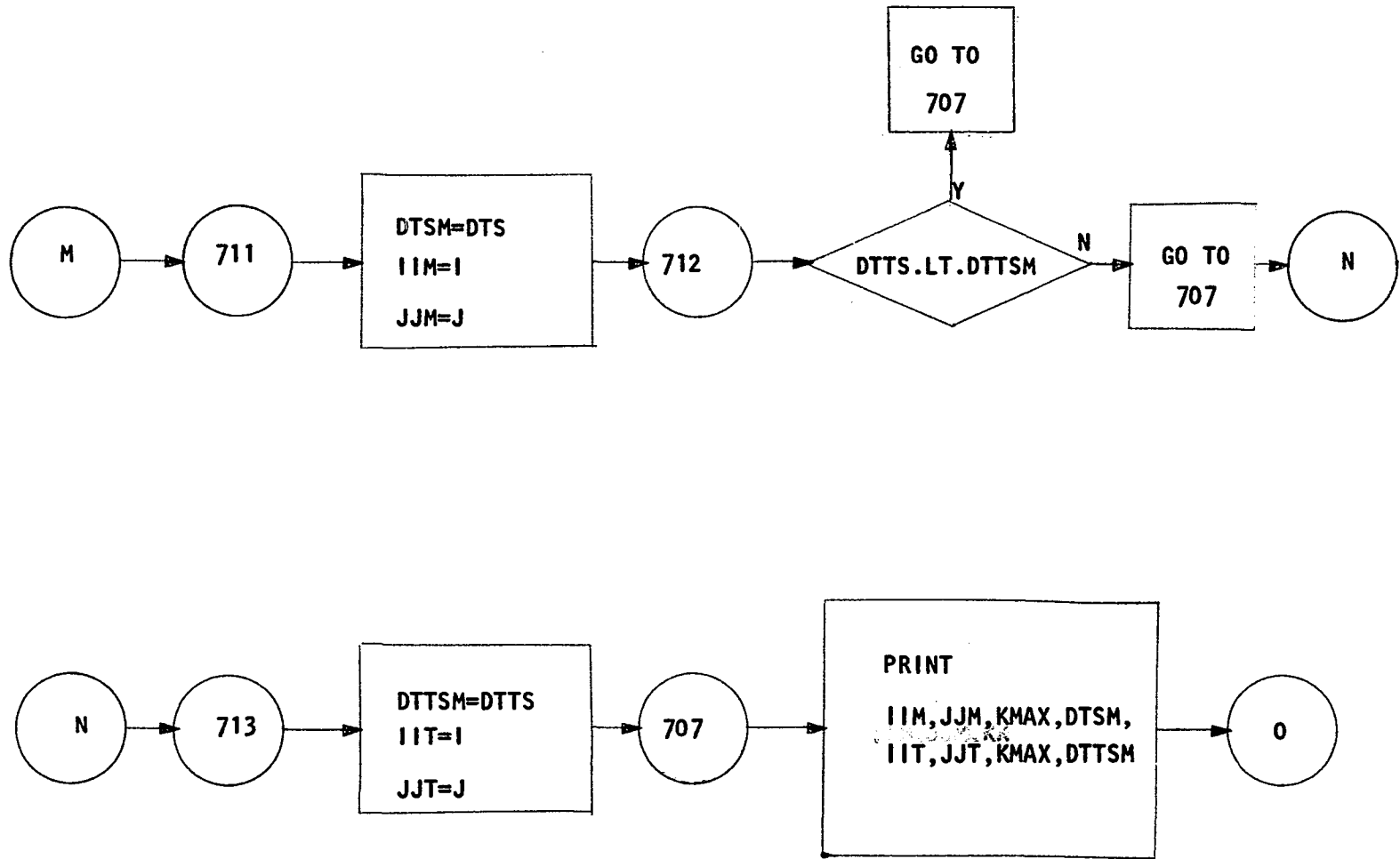


FIG 24h FLOW CHART OF MAIN PROGRAM

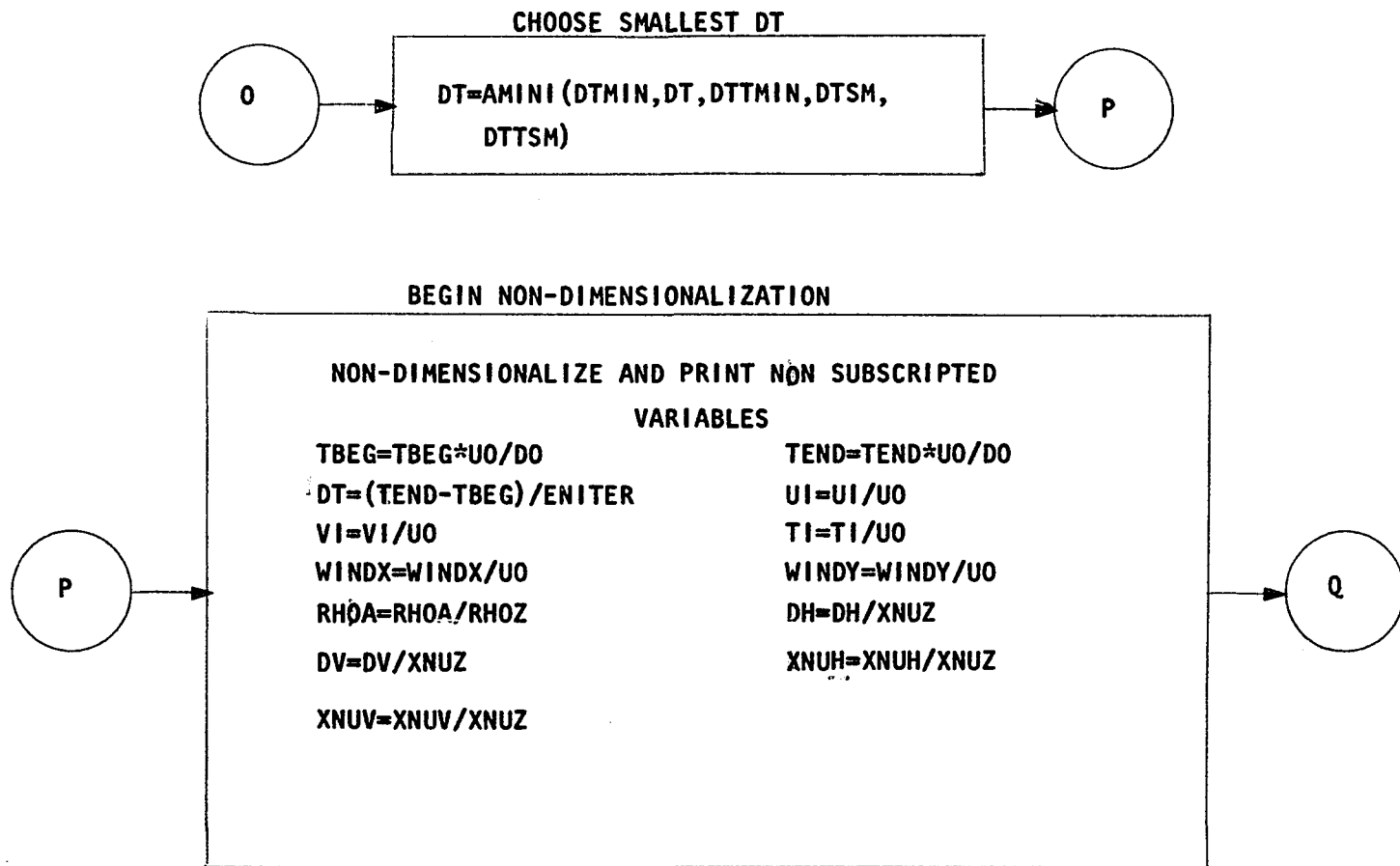
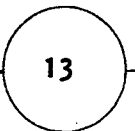
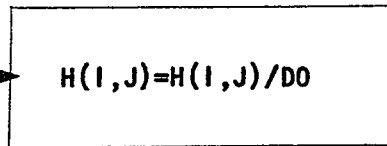
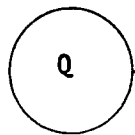
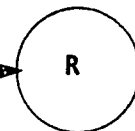
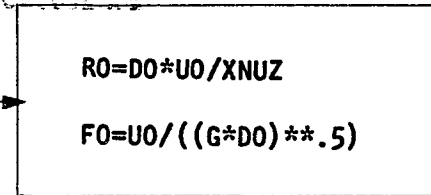


FIG 24i FLOW CHART OF MAIN PROGRAM

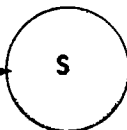
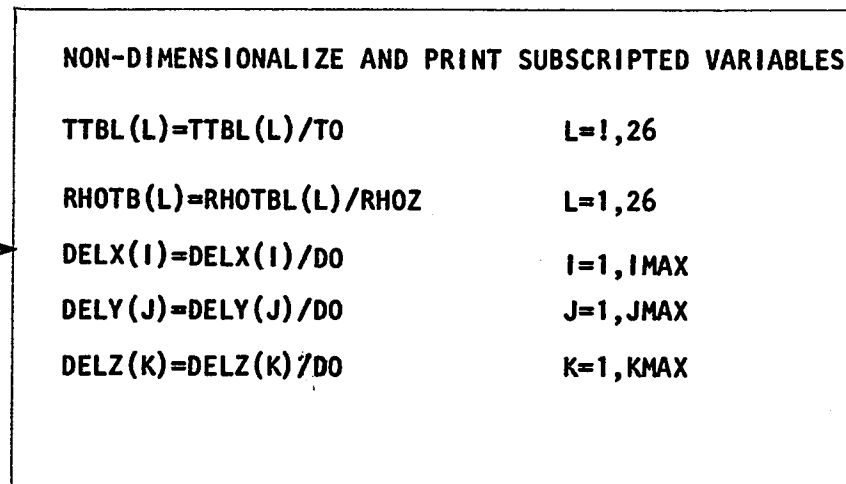
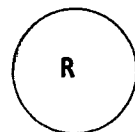
DO 13 I=1,IMAX
DO 13 J=1,JMAX



CALCULATE REYNOLDS AND FROUD NUMBER



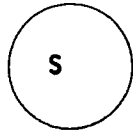
CONTINUE NON-DIMENSIONALIZATION



END NON DIMENSIONALIZATION

FIG 24j FLOW CHART OF MAIN PROGRAM

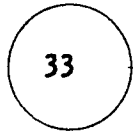
DO 33 K=2,K1



ZKP1(K)=DELZ(K+1)+delz(K)
ZKM1(K)=DELZ(K-1)+DELZ(K)
ZKPM1(K)=ZKP1(K)+ZKM1(K)
ZK123(K)=ZKP1(K)*ZKM1(K)*ZKPM1(K)
ZKPM(K)=DELZ(K-1)-DELZ(K+1)

33

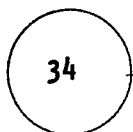
DO 34 J=2,J1



YJP1(J)=DELY(J+1)+DELY(J)
YJM1(J)=DELY(J-1)+DELY(J)
YJPM1(J)=YJP1(J)+YJM1(J)
YJPM1(J)=YJP1(J)*YJM1(J)*YJPM1(J)

34

DO 35 I=2,I1



XIP1(I)=DELX(I+1)+DELX(I)
XIM1(I)=DELX(I-1)+delx(I)
XIPM1(I)=XIP1(I)+XIM1(I)
XI123(I)=XIP1(I)*XIM1(I)*XIPM1(I)

35

FIG 24k FLOW CHART OF MAIN PROGRAM

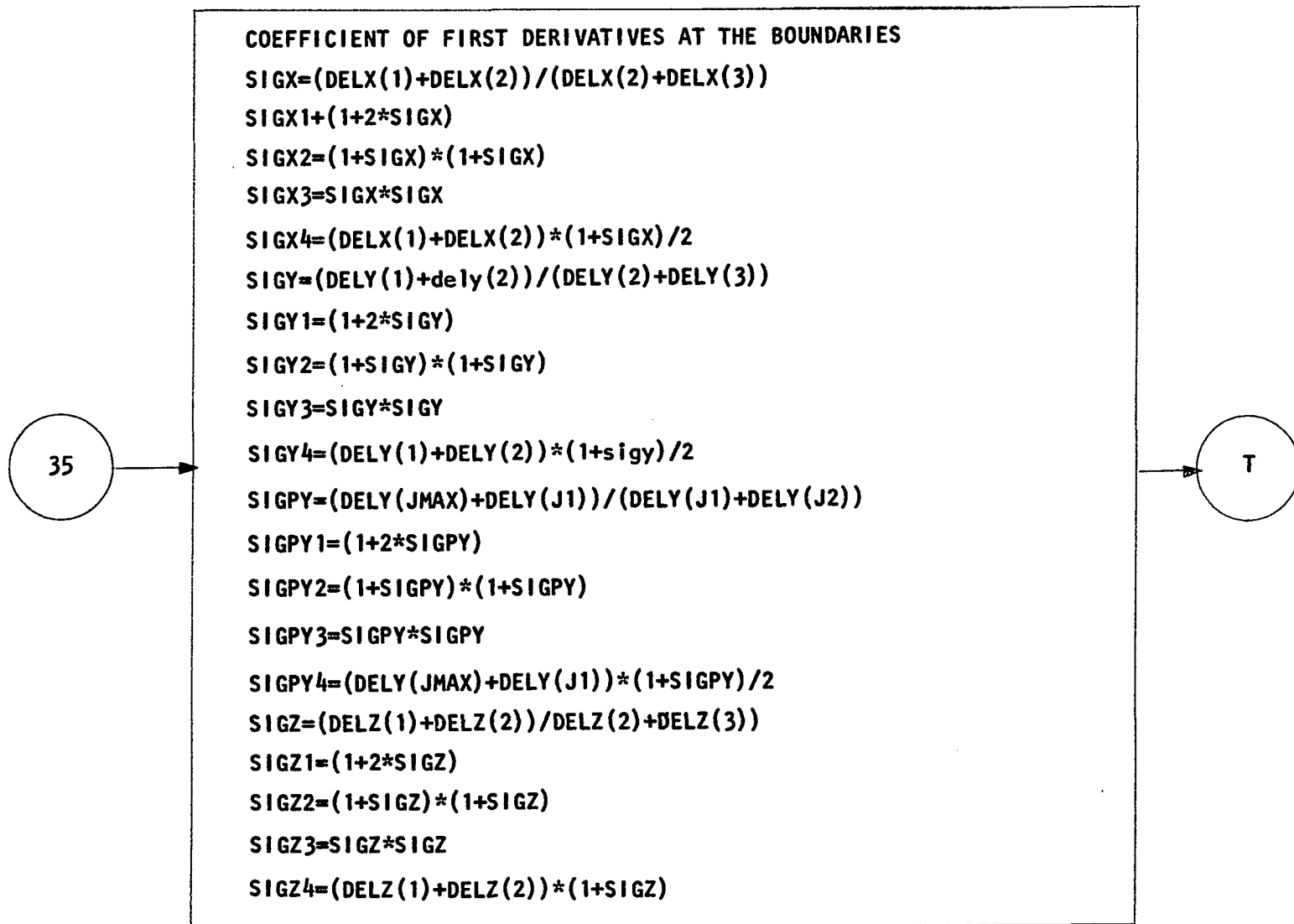


FIG 241 FLOW CHART OF MAIN PROGRAM

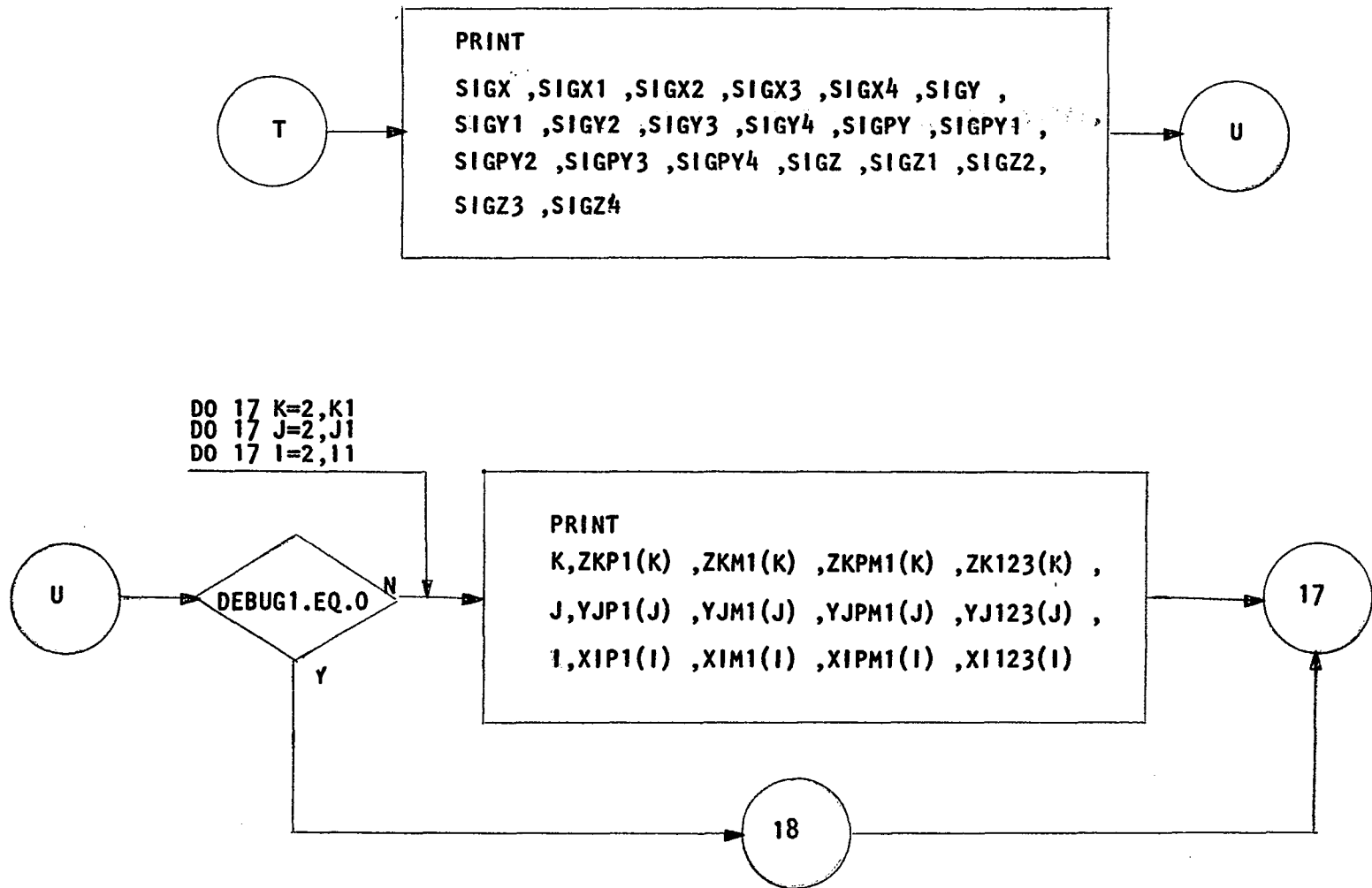


FIG 24m FLOW CHART OF MAIN PROGRAM

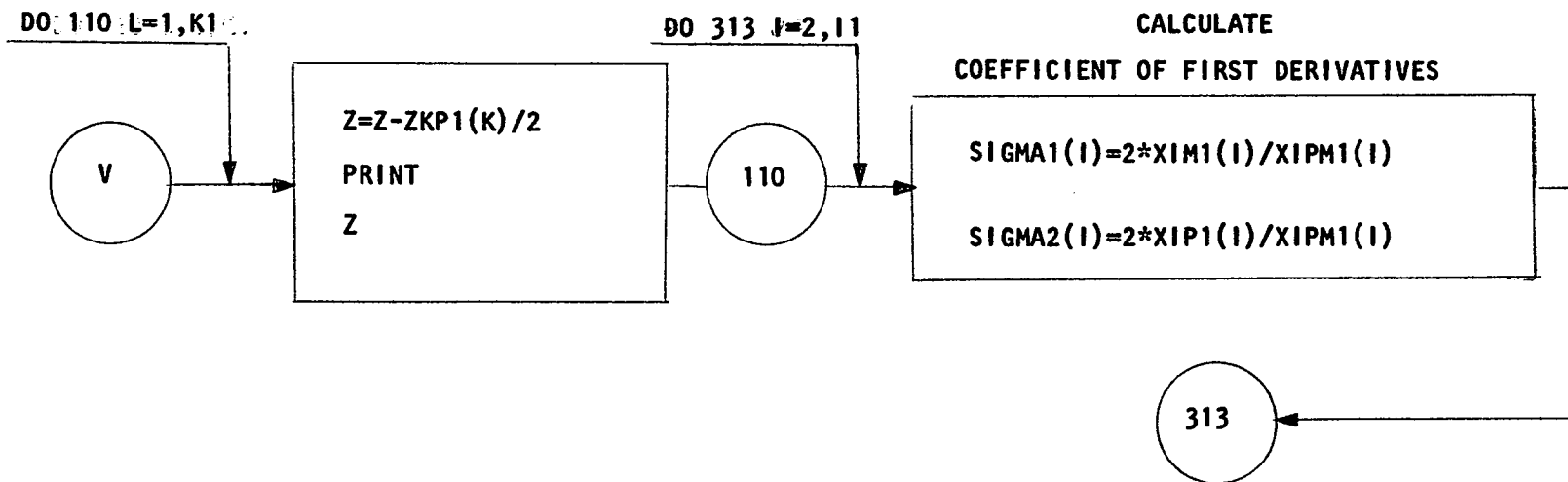
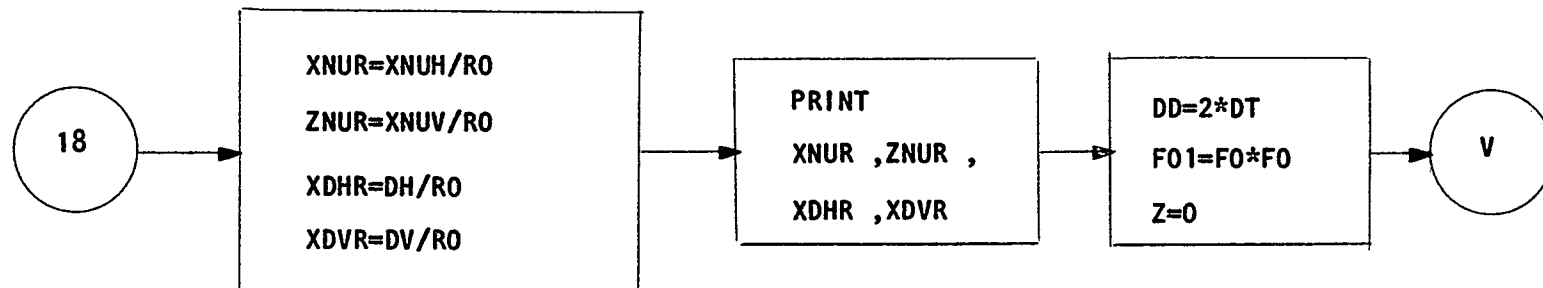


FIG 24n FLOW CHART OF MAIN PROGRAM

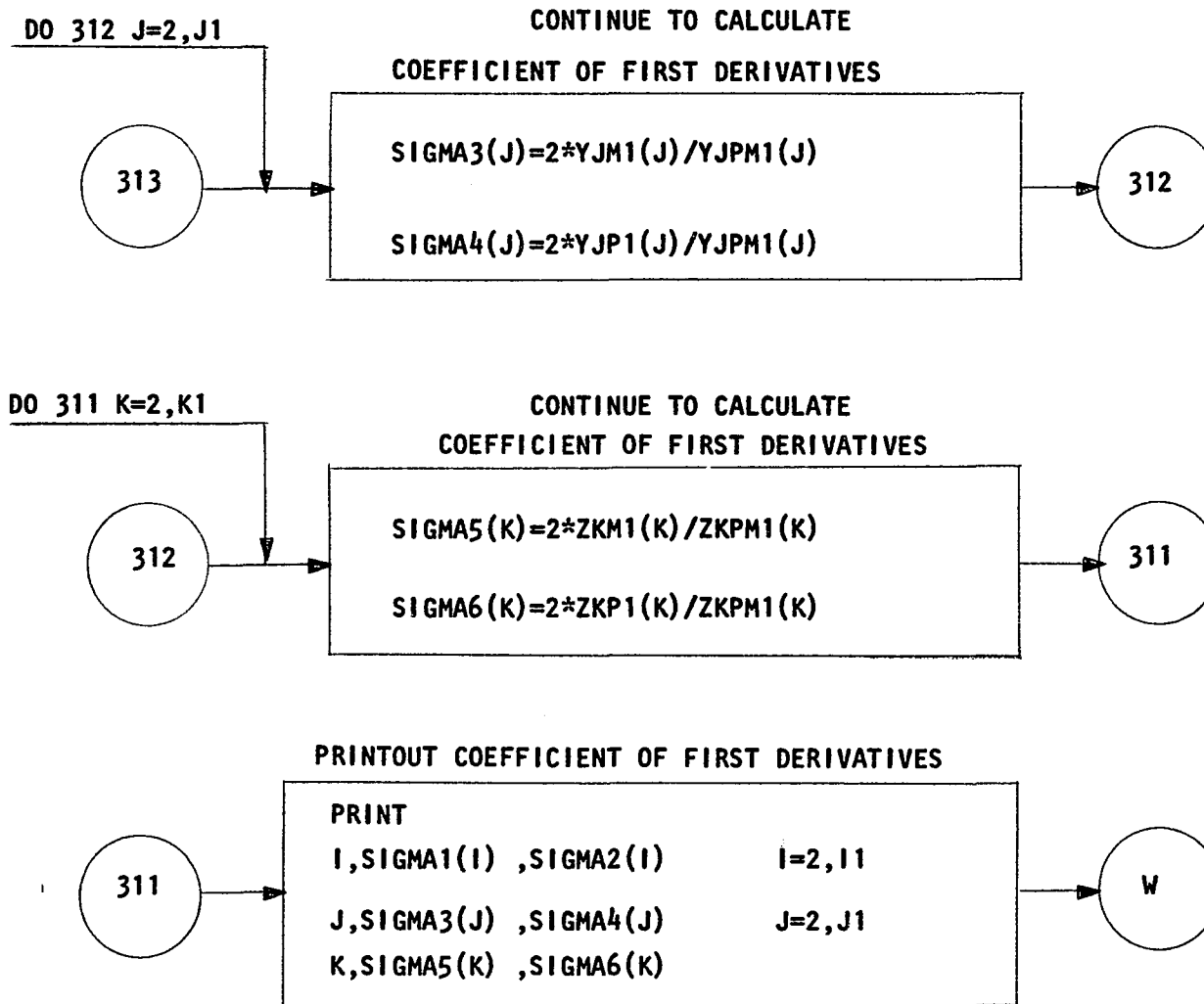


FIG 24o FLOW CHART OF MAIN PROGRAM

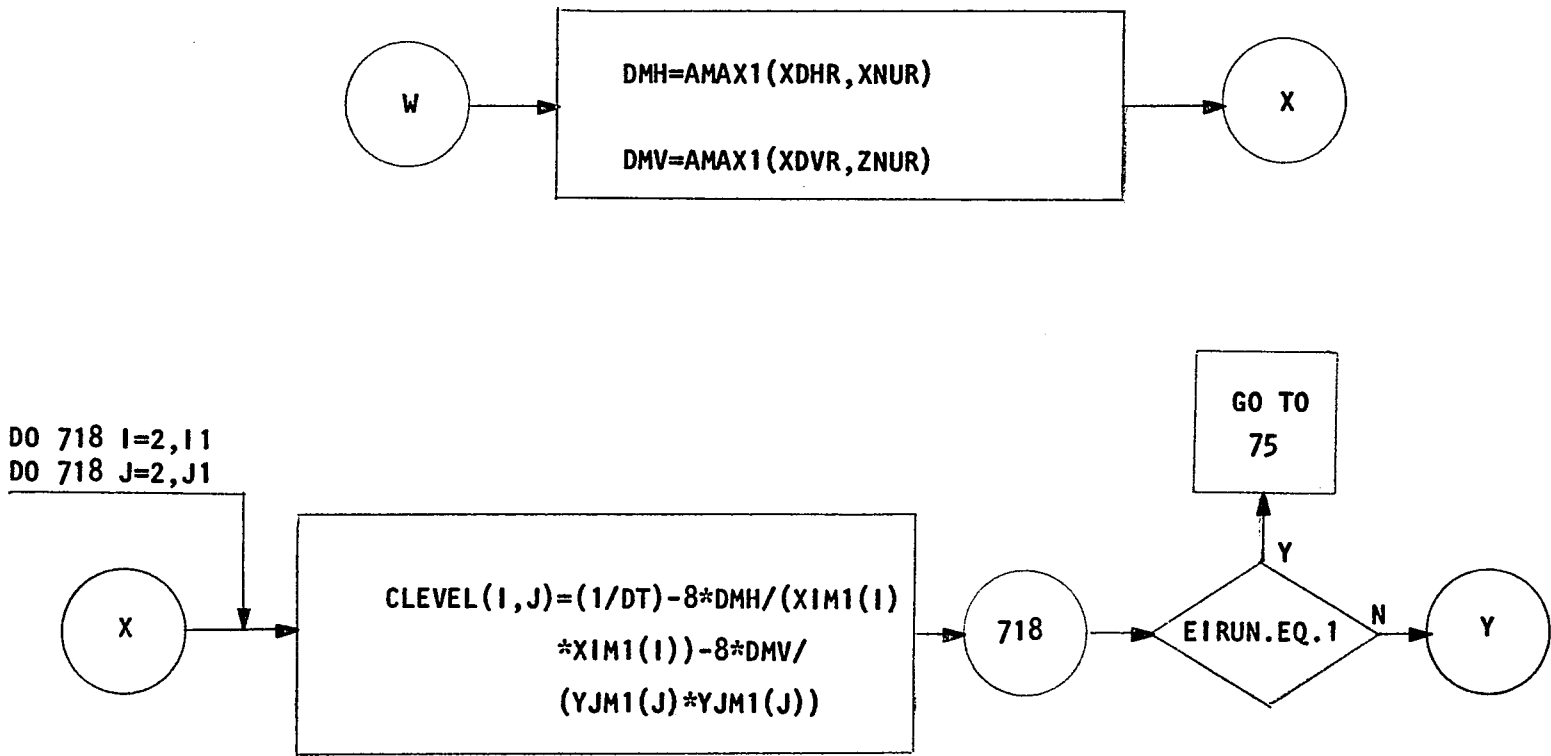
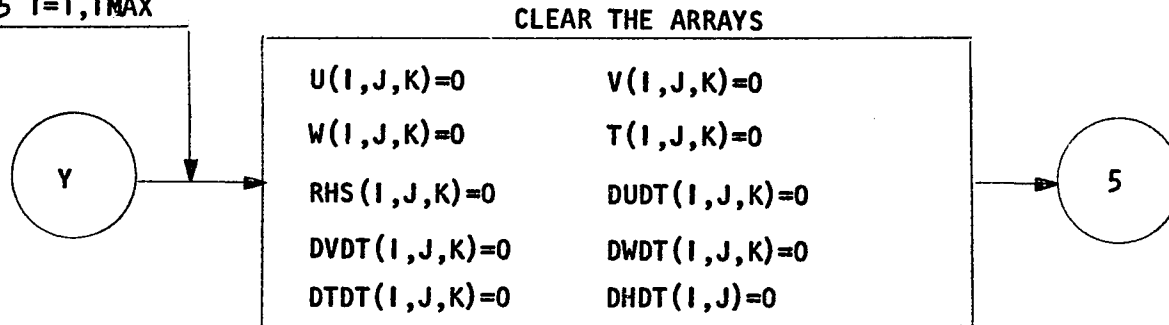


FIG 24p FLOW CHART OF MAIN PROGRAM

DO 5 K=1,KMAX
DO 5 J=1,JMAX
DO 5 I=1,IMAX



DO 25 K=1,KMAX
DO 25 J=1,JMAX
DO 25 I=1,IMAX

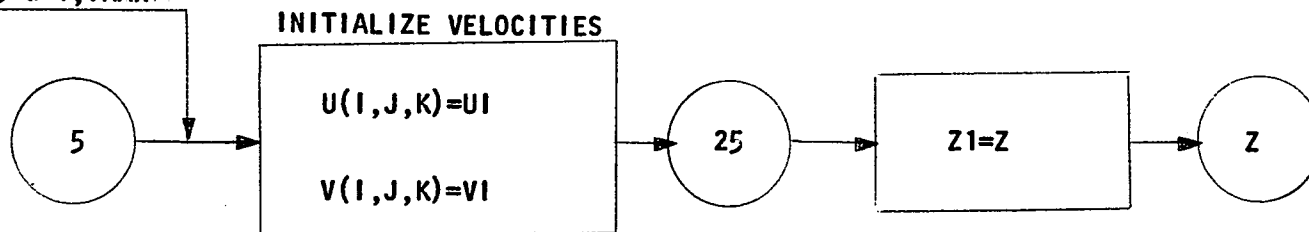


FIG 24q FLOW CHART OF MAIN PROGRAM

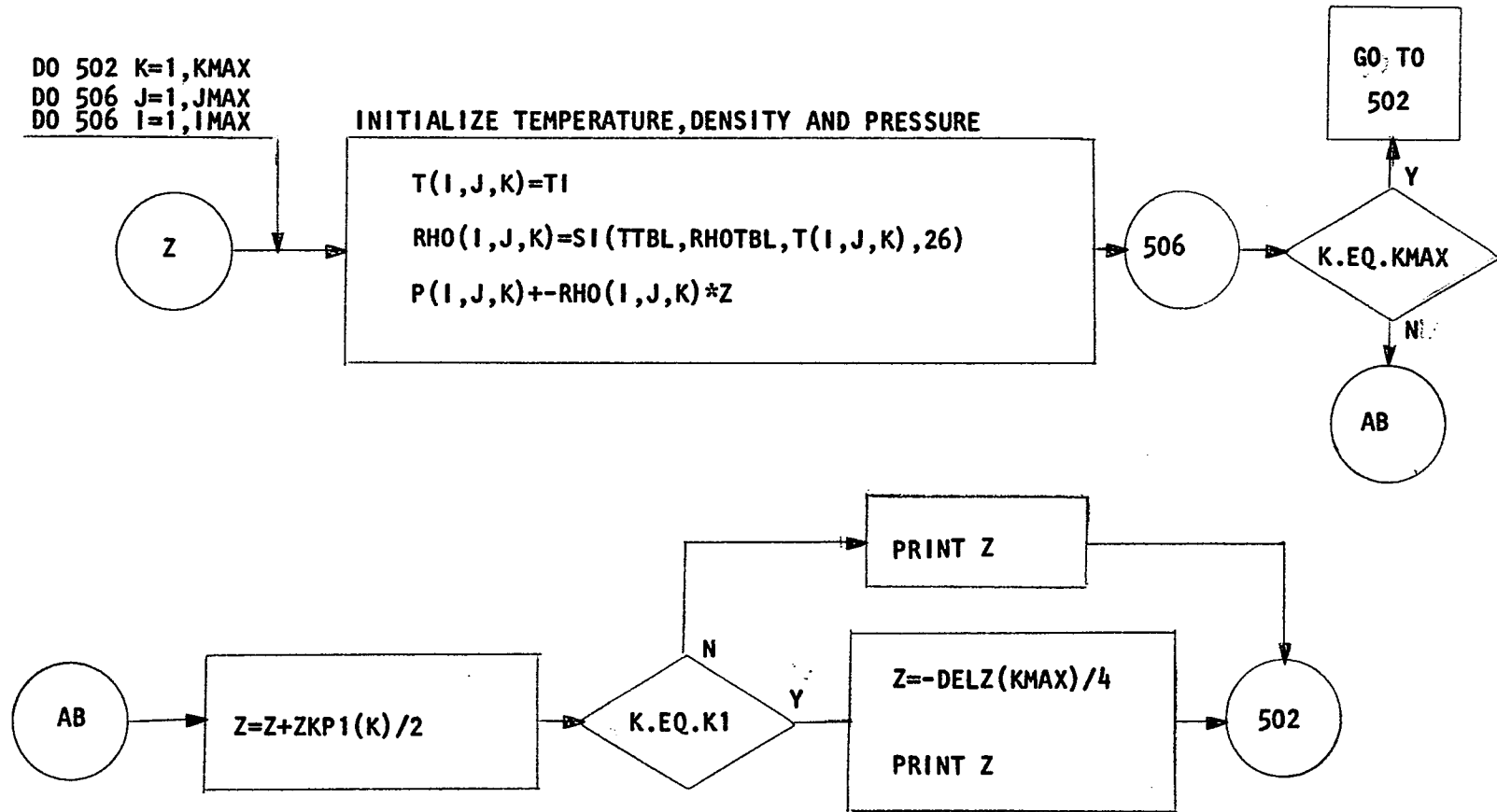


FIG 24r FLOW CHART OF MAIN PROGRAM

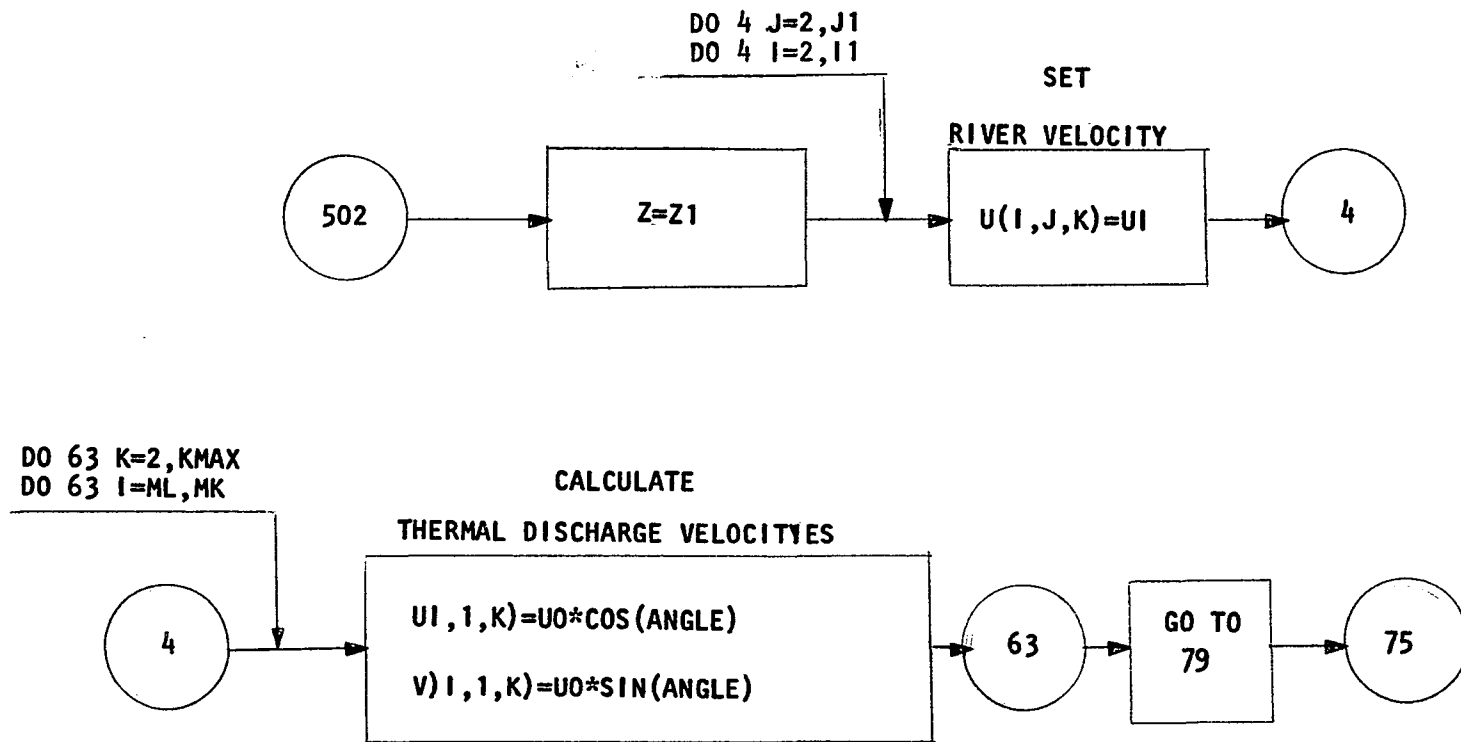


FIG 24s FLOW CHART OF MAIN PROGRAM

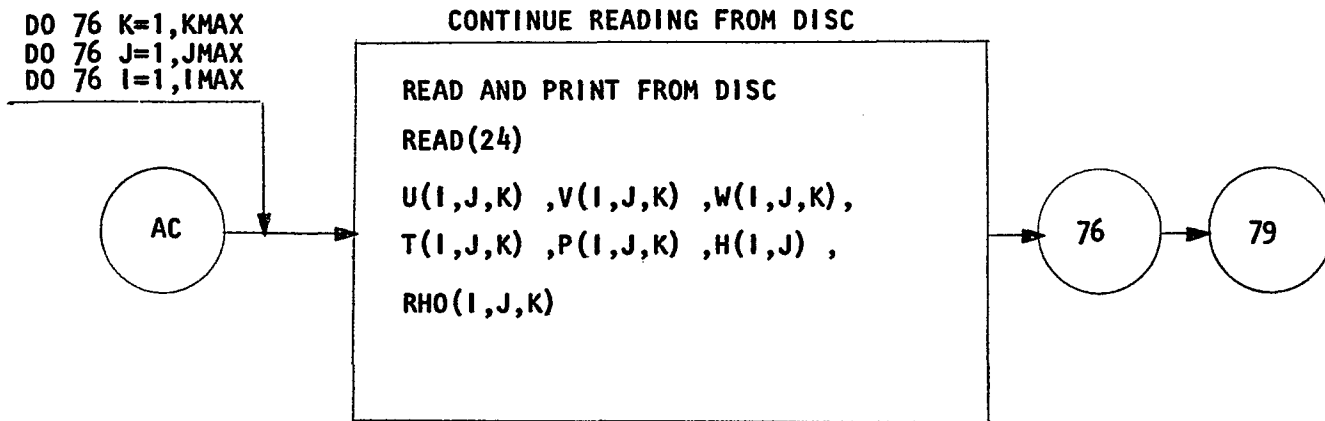
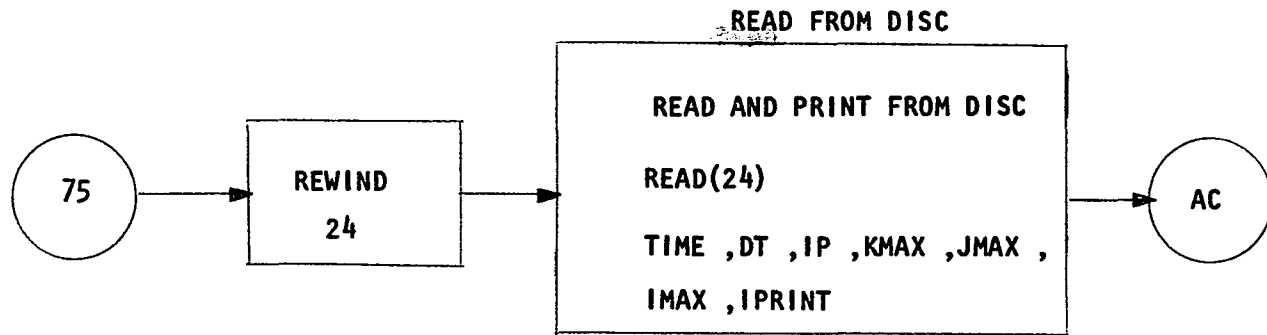


FIG 24t FLOW CHART OF MAIN PROGRAM

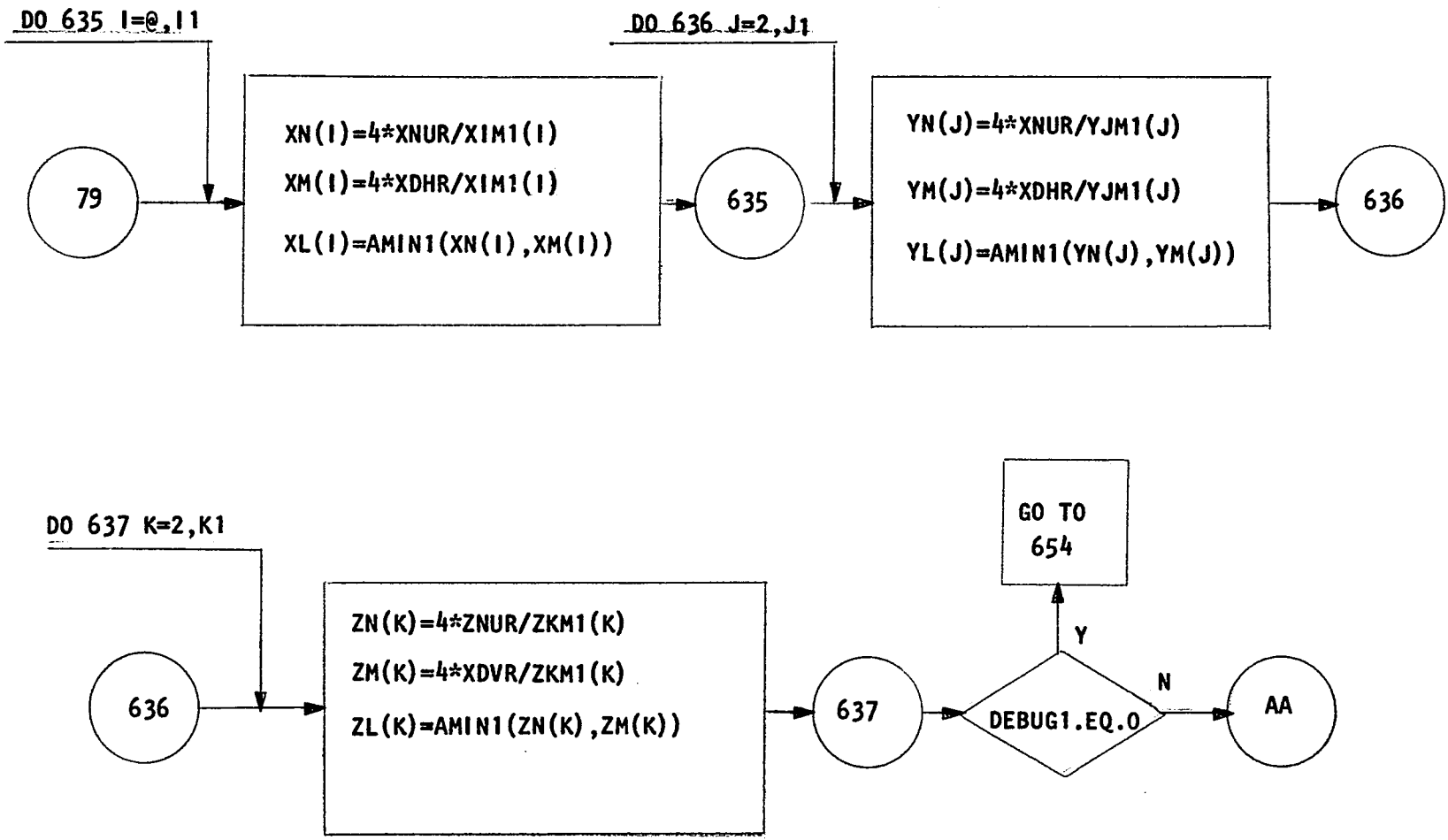


FIG 24u FLOW CHART OF MAIN PROGRAM

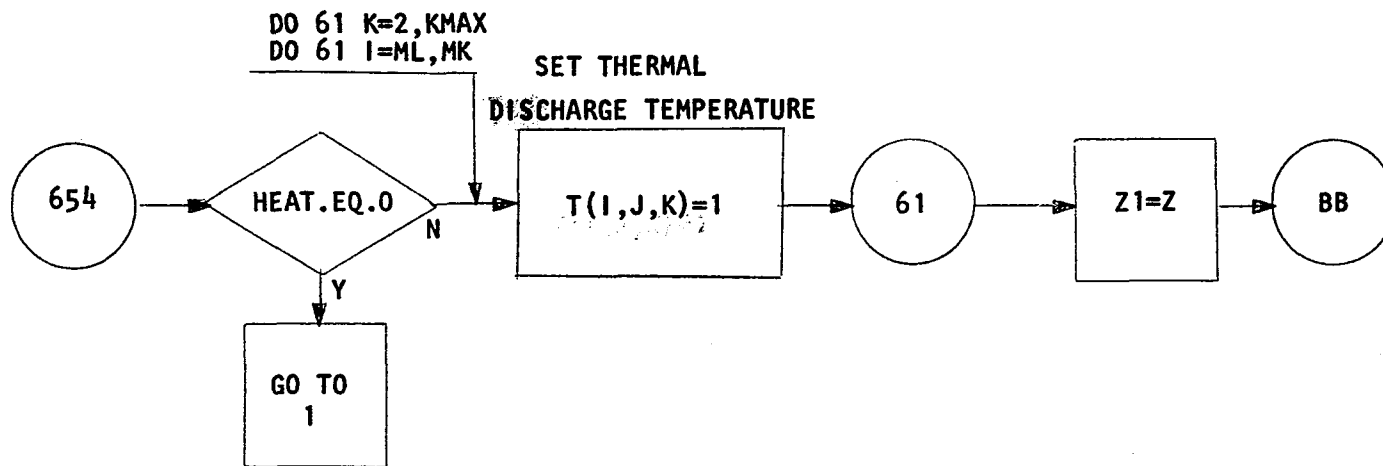
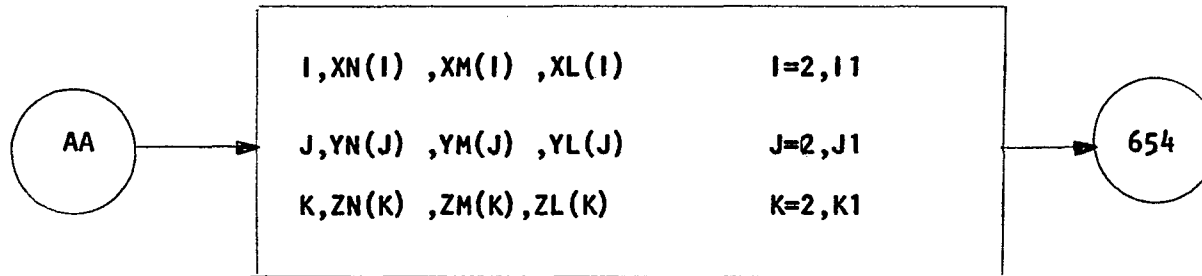


FIG 24v FLOW CHART OF MAIN PROGRAM

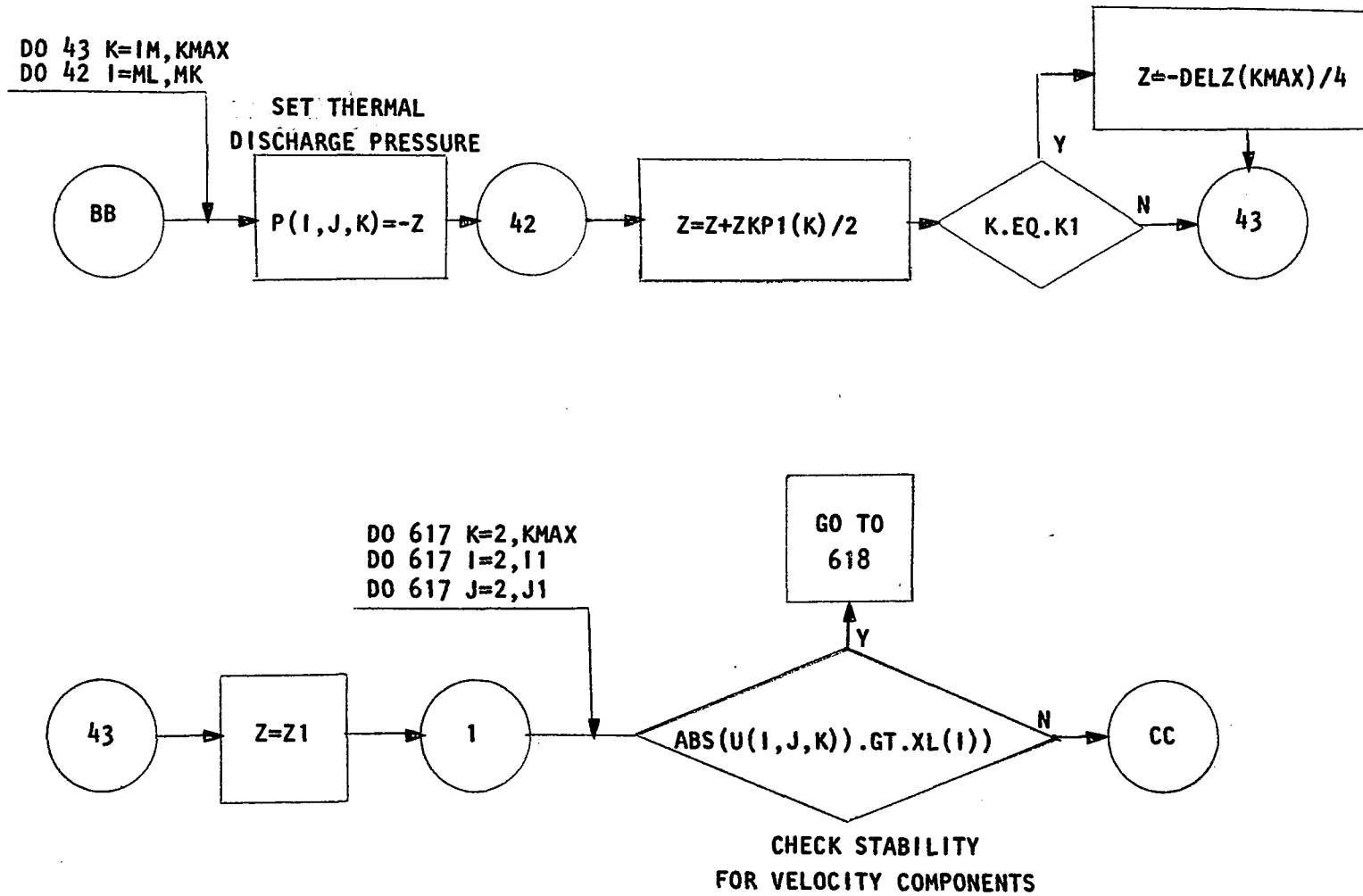


FIG 24w FLOW CHART OF MAIN PROGRAM

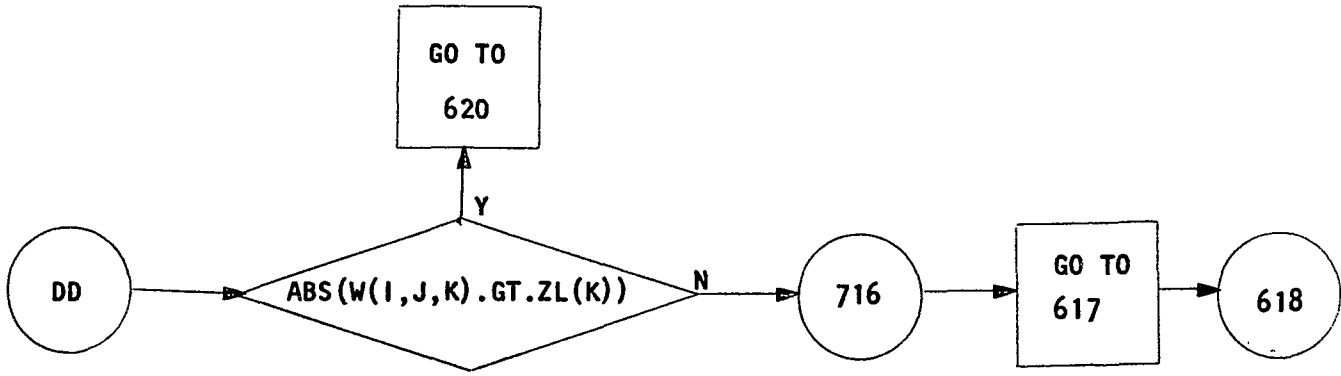
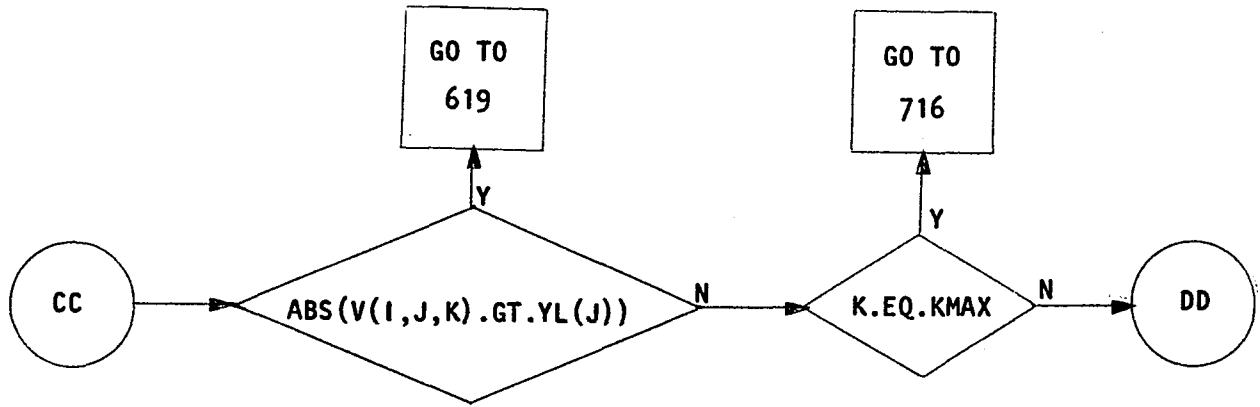


FIG 24x FLOWCHART OF MAIN PROGRAM

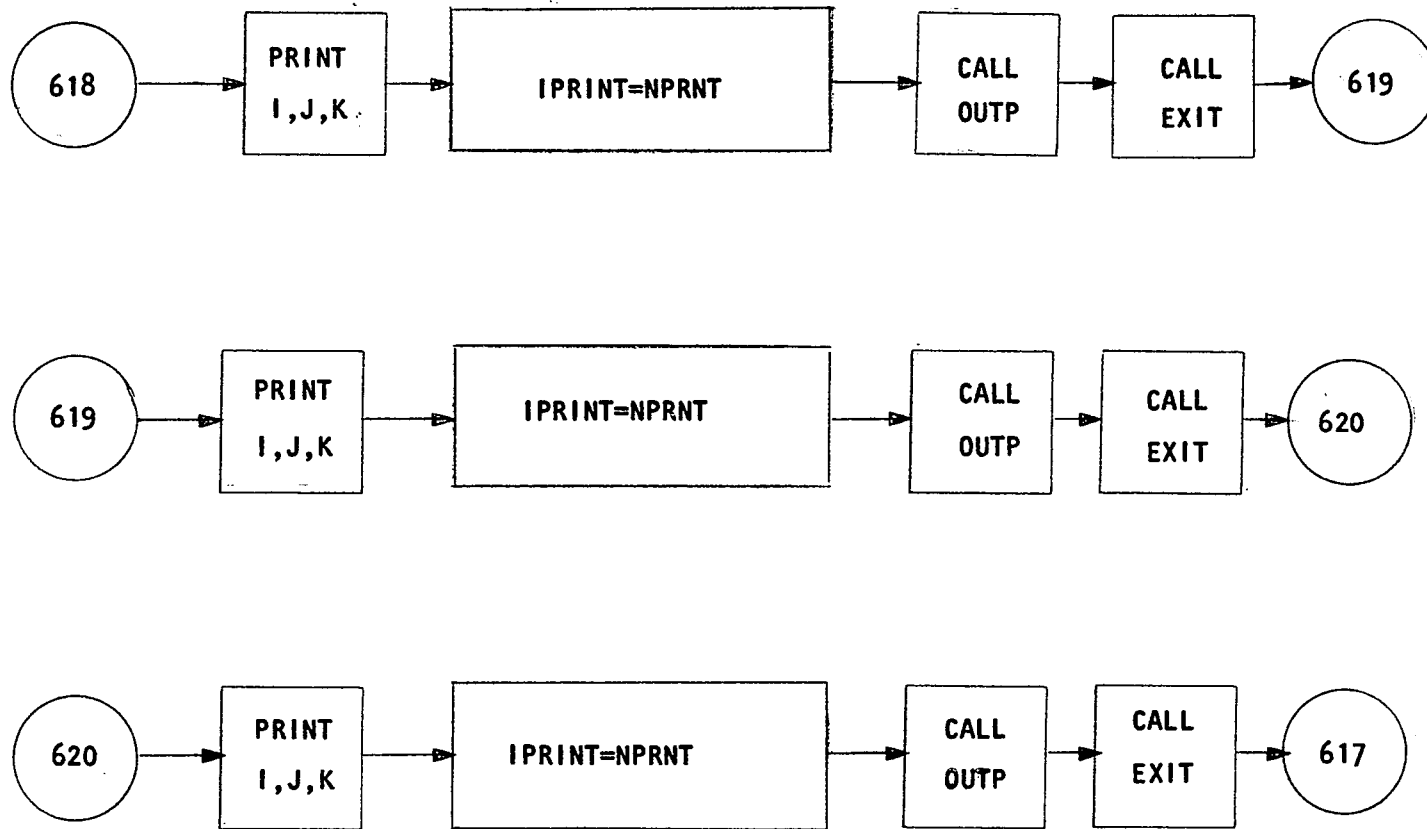


FIG 24y FLOW CHART OF MAIN PROGRAM

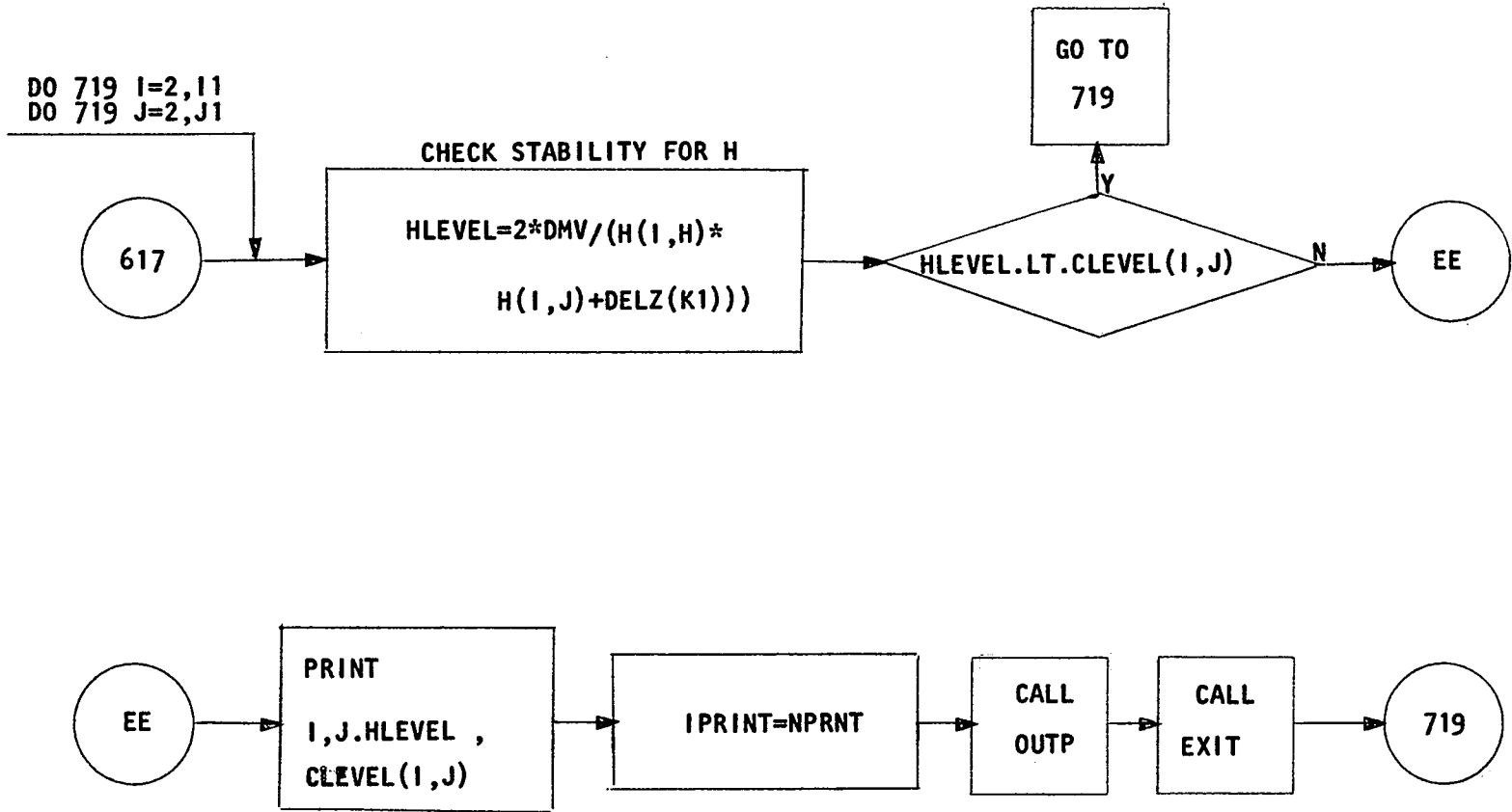


FIG 24z FLOW CHART OF MAIN PROGRAM

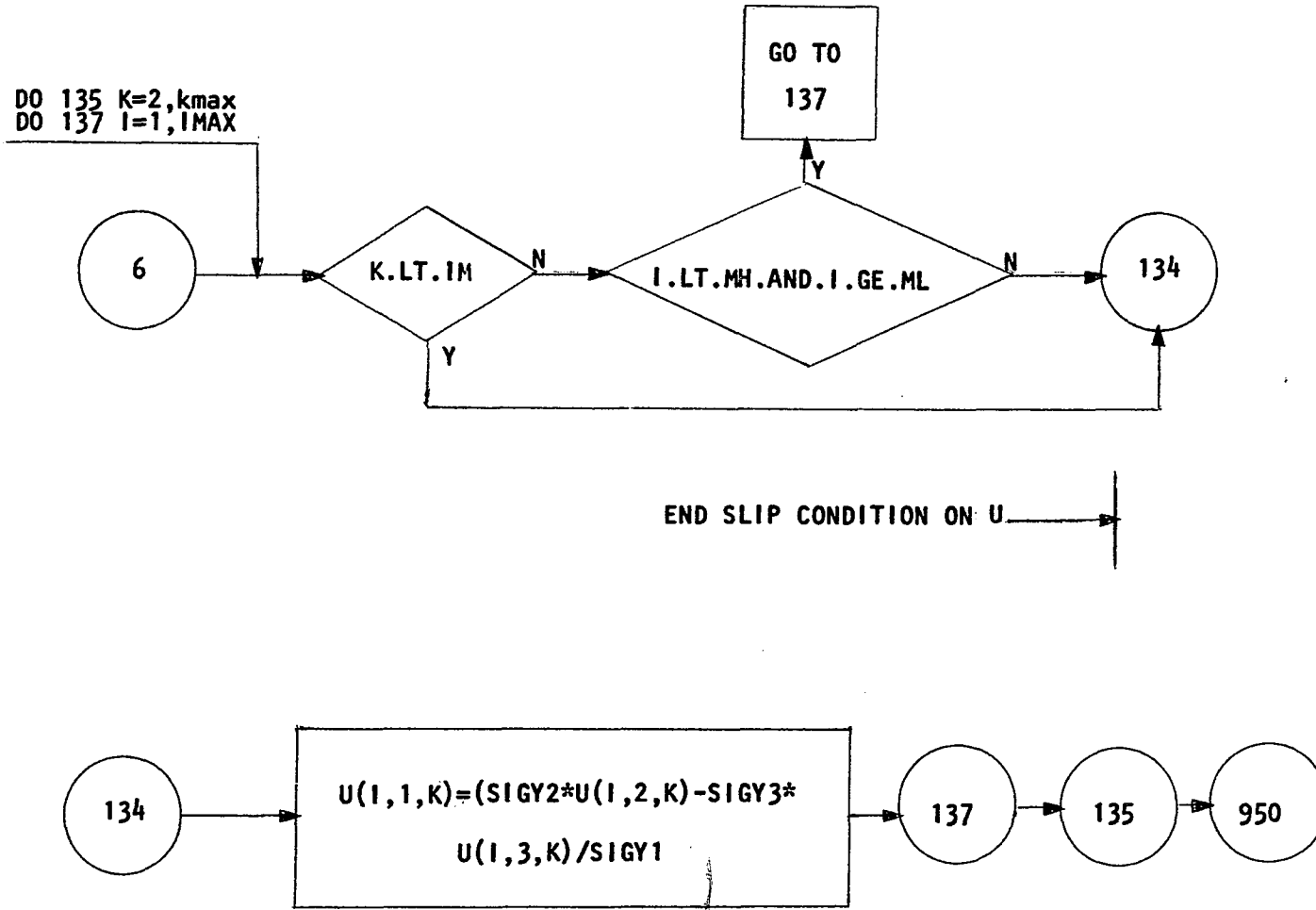


FIG 24bb FLOW CHART OF MAIN PROGRAM

← SET BOUNDARY CONDITION ON TEMPERATURE

DO 130 K=2, KMAX
DO 130 I=1, IMAX

950

$T(I, JMAX, K) = (SIGPY2 * T(I, J1, K) -$
 $SIGPY3 * T(I, J2, K)) / SIGPY1$

130

DO 125 K=2, KMAX
DO 126 I=1, IMAX

130

K.LT.IM

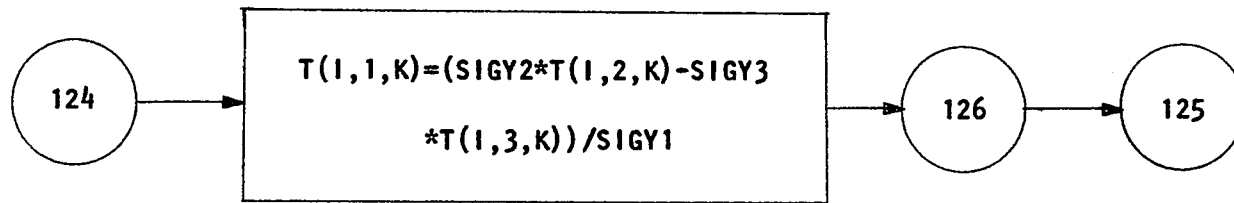
N

I.LT.MH.AND.I.GE.ML

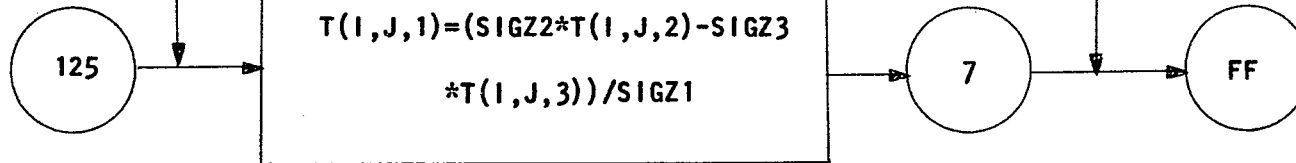
124

GO TO
126

FIG 24cc FLOW CHART OF MAIN PROGRAM



DO 7 I=1,IMAX
DO 7 J=1,JMAX



END BOUDARY CONDITION ON TEMPERATURE

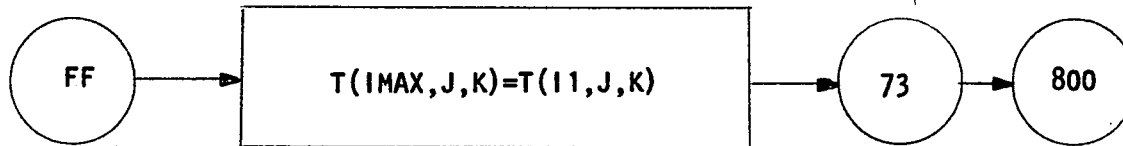


FIG 24dd FLOW CHART OF MAIN PROGRAM

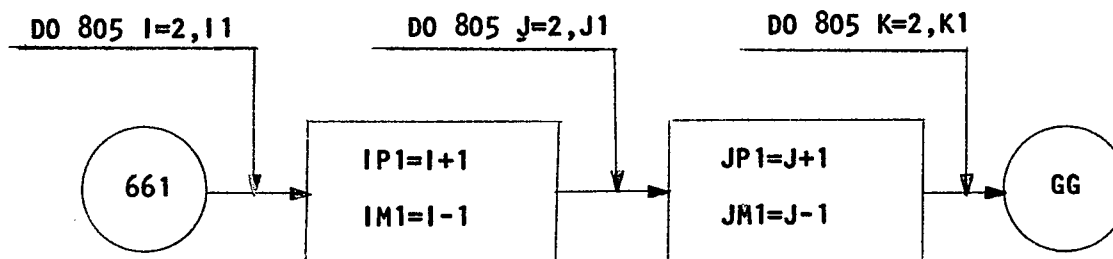
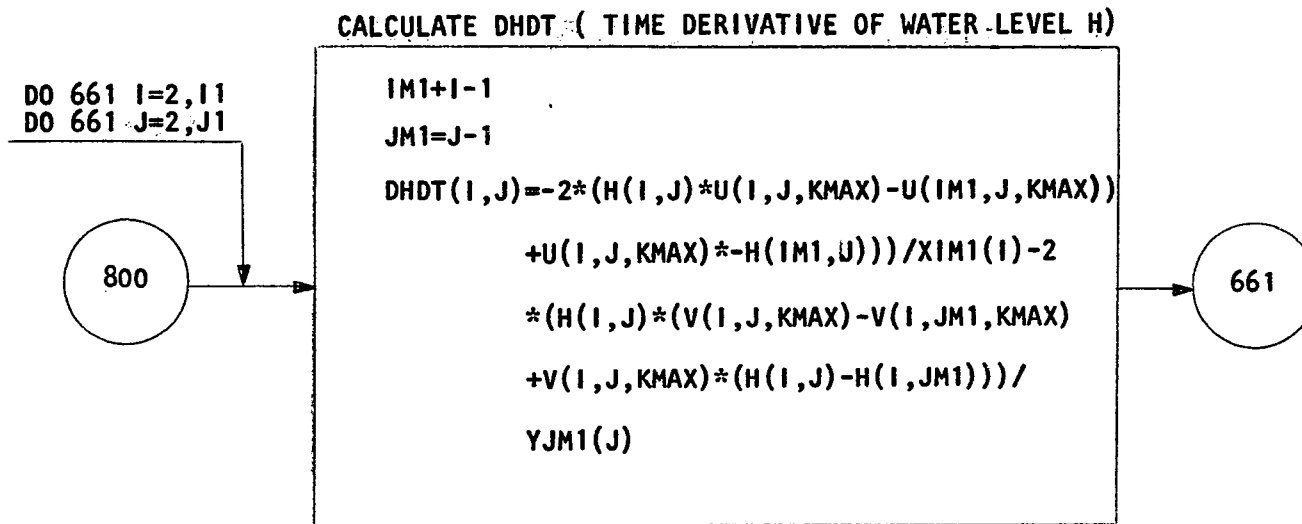
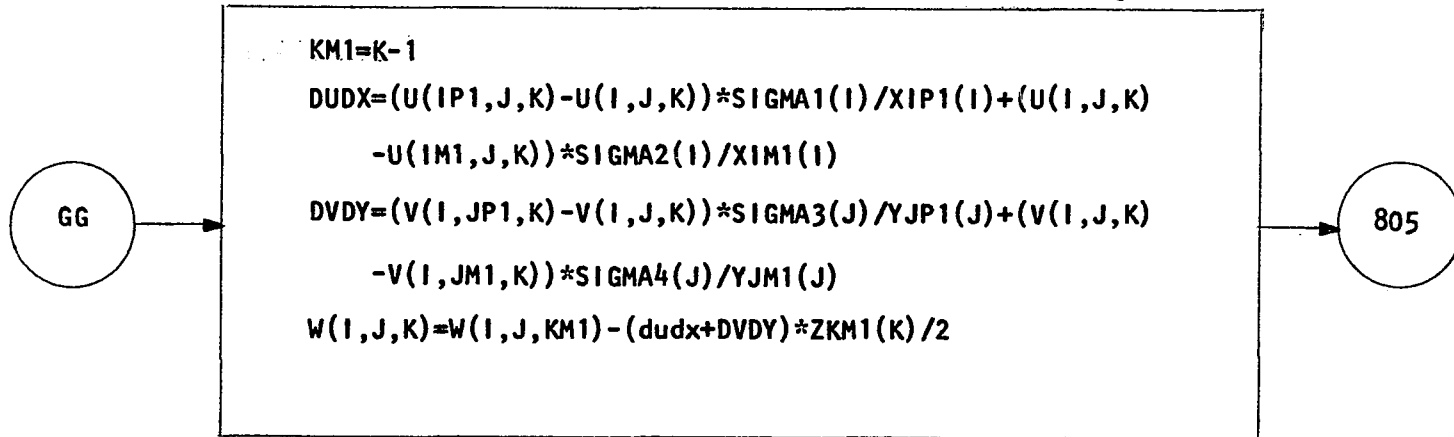


FIG 24ee FLOW CHART OF MAIN PROGRAM

CALCULATE VERTICAL VELOCITY COMPONENT FROM CONTINUITY EQUATION



DO 72 K=2,KMAX
DO 72 J=2,J1

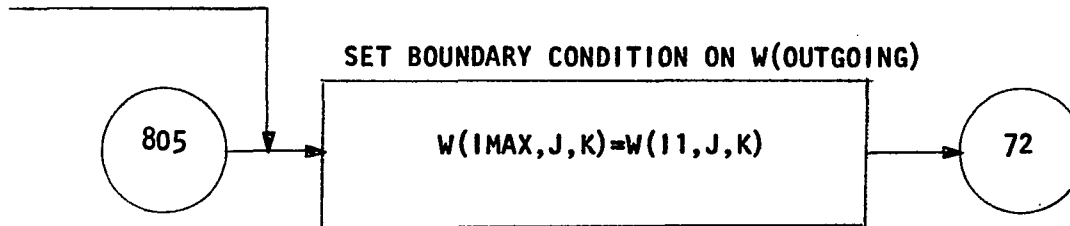


FIG 24ff FLOW CHART OF MAIN PROGRAM

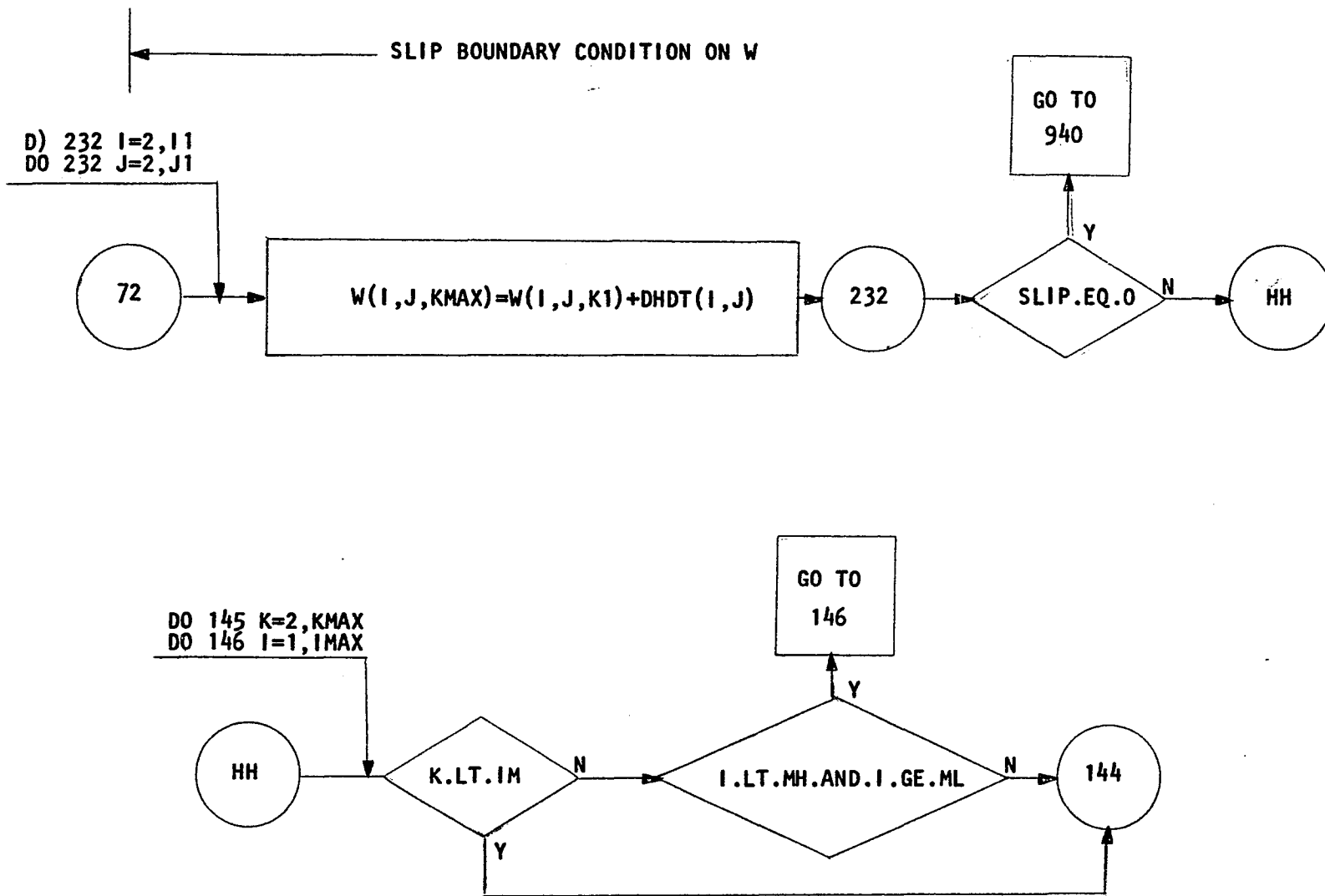


FIG 24gg FLOW CHART OF MAIN PROGRAM

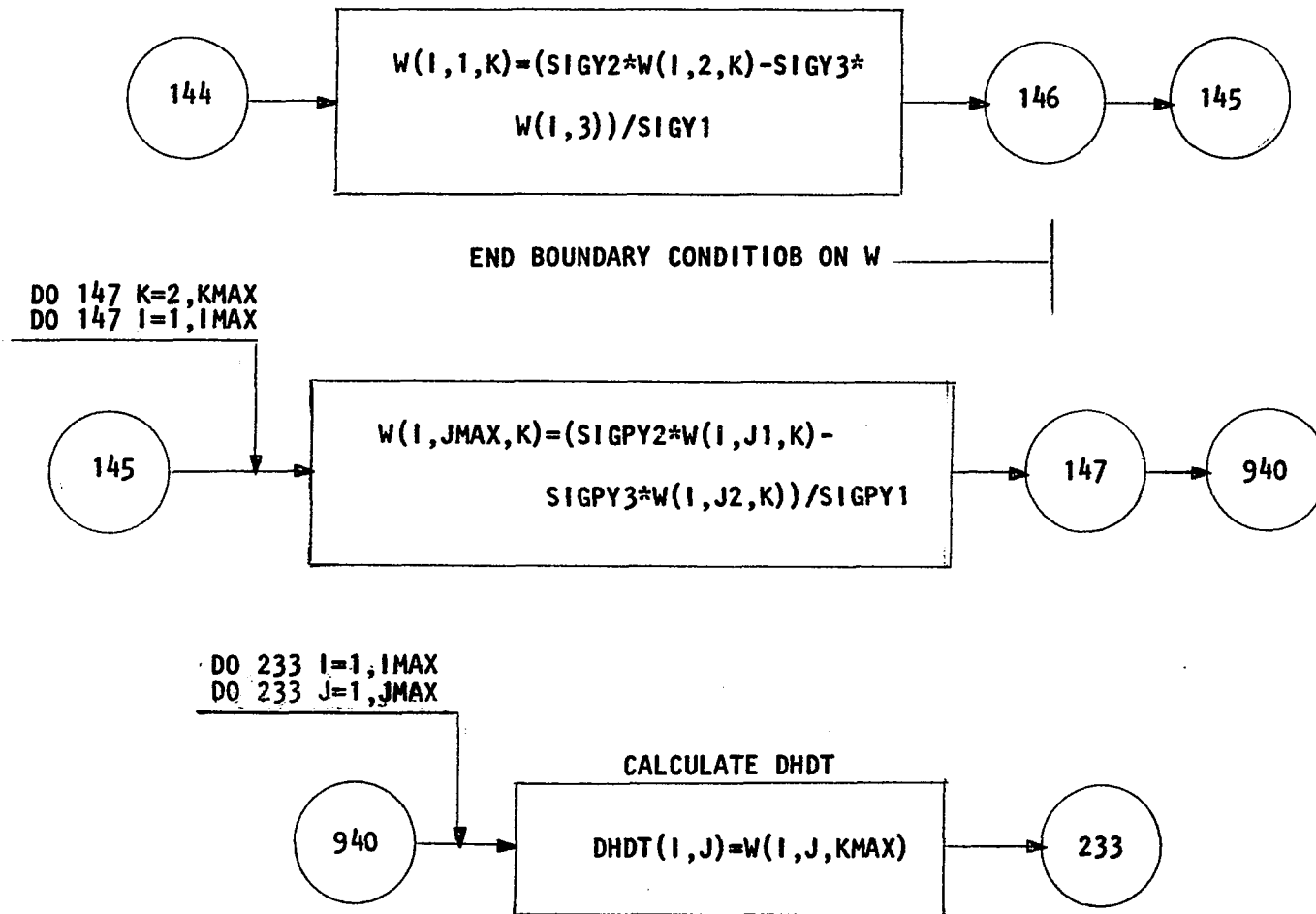


FIG 24hh FLOW CHART OF MAIN PROGRAM

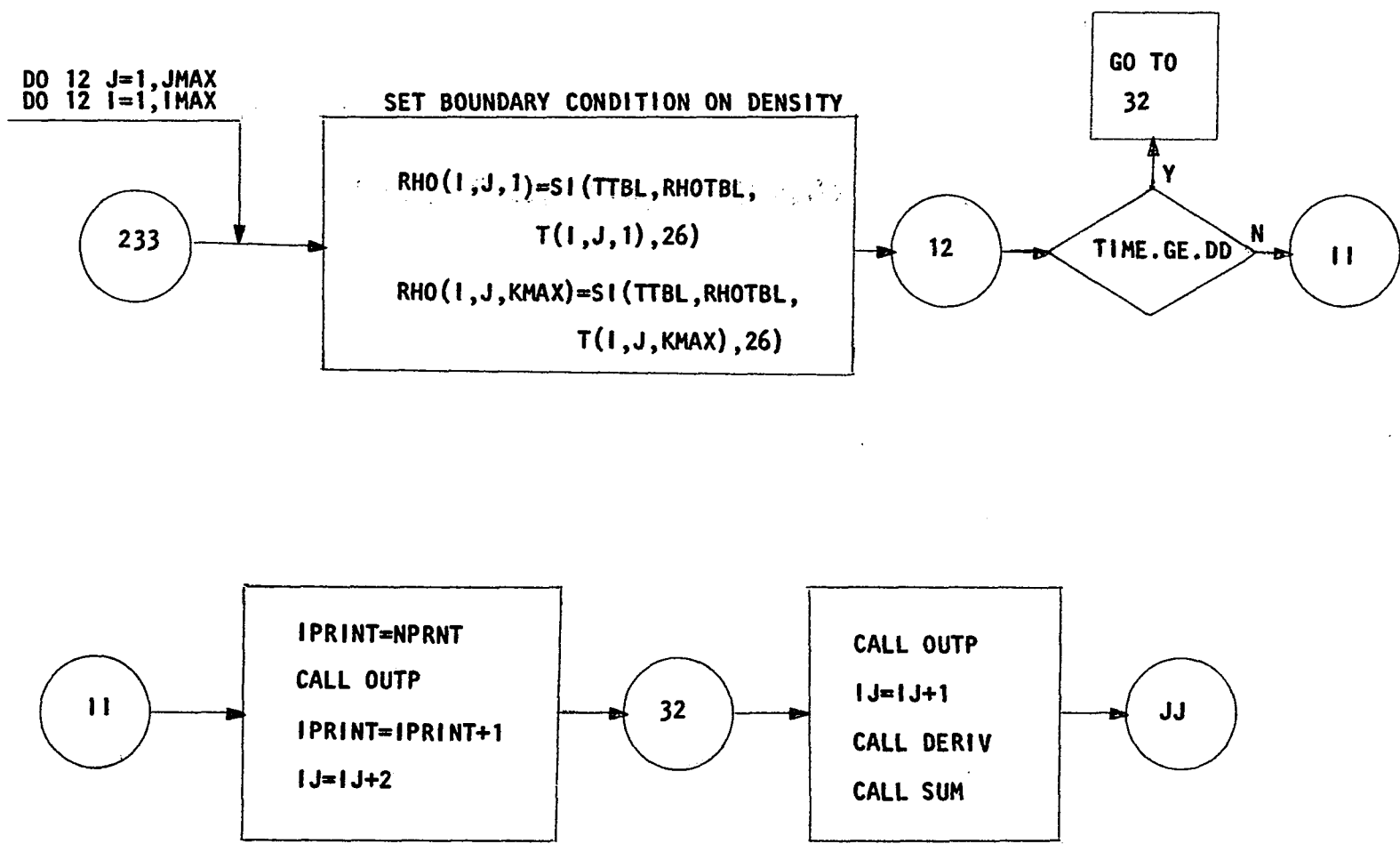


FIG 24ii FLOW CHART OF MAIN PROGRAM

```

DO 40 K=1,KMAX
DO 40 I=2,I1
DO 40 J=2,J1

```

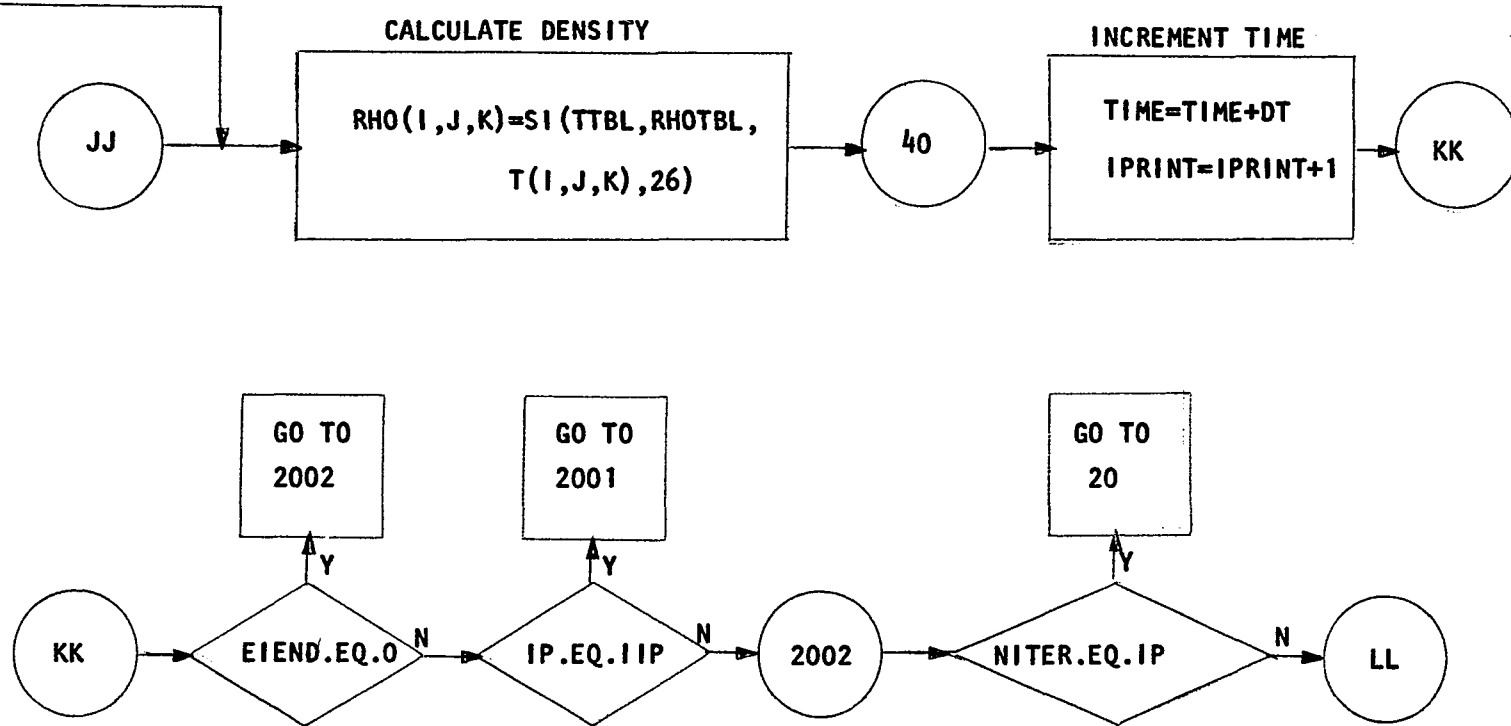


FIG 24jj FLOW CHART OF MAIN PROGRAM

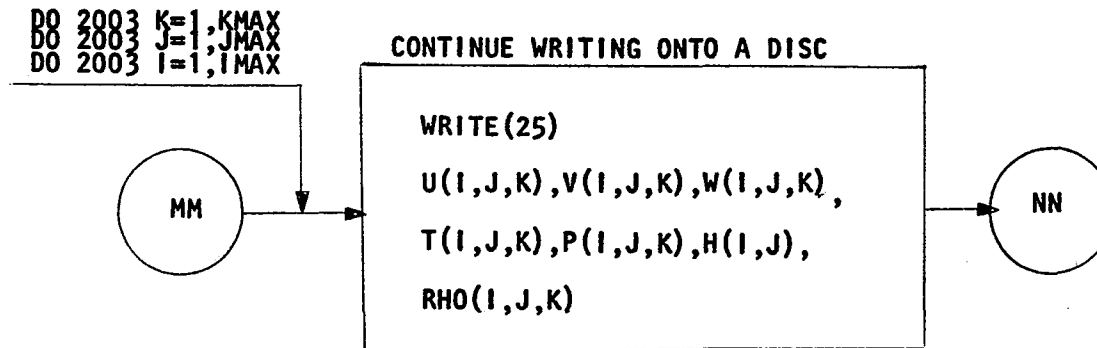
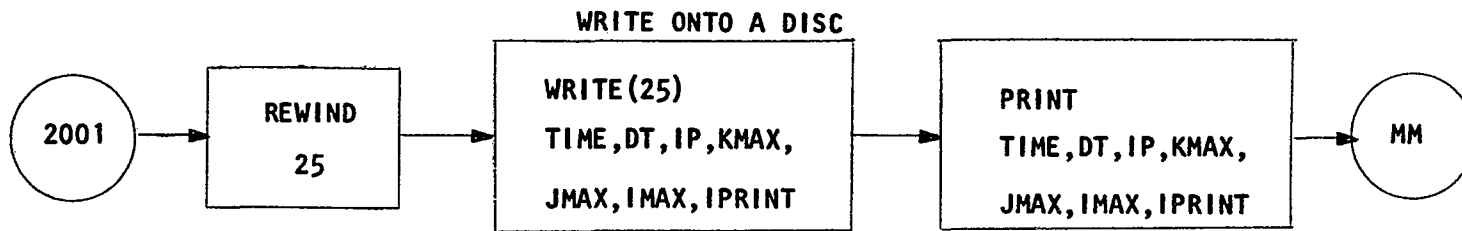
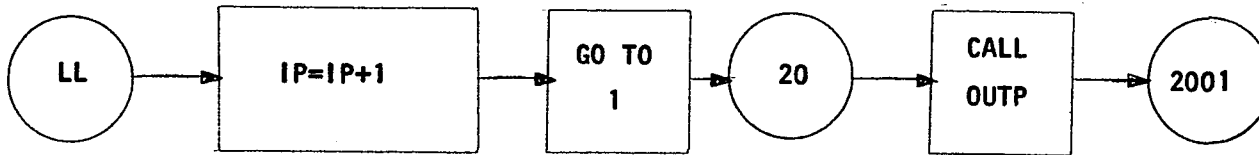


FIG 24kk FLOW CHART OF MAIN PROGRAM

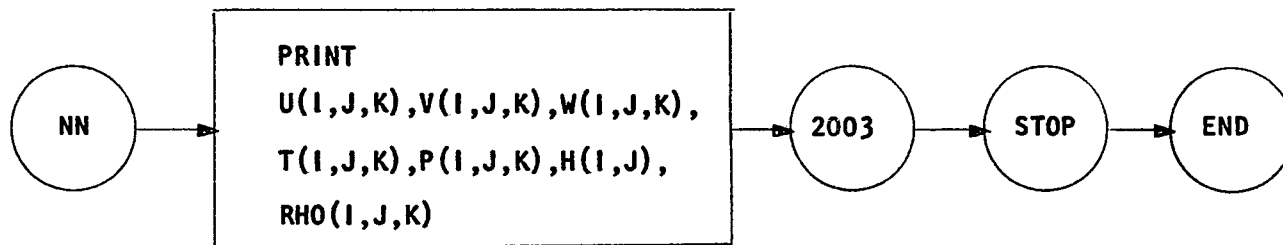


FIG2411 FLOW CHART OF MAIN PROGRAM

CALCULATE DERIVATIVES OF WATER LEVEL ELEMENTS

DO 100 I=2,I1
DO 100 J=2,J1

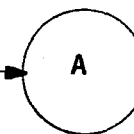
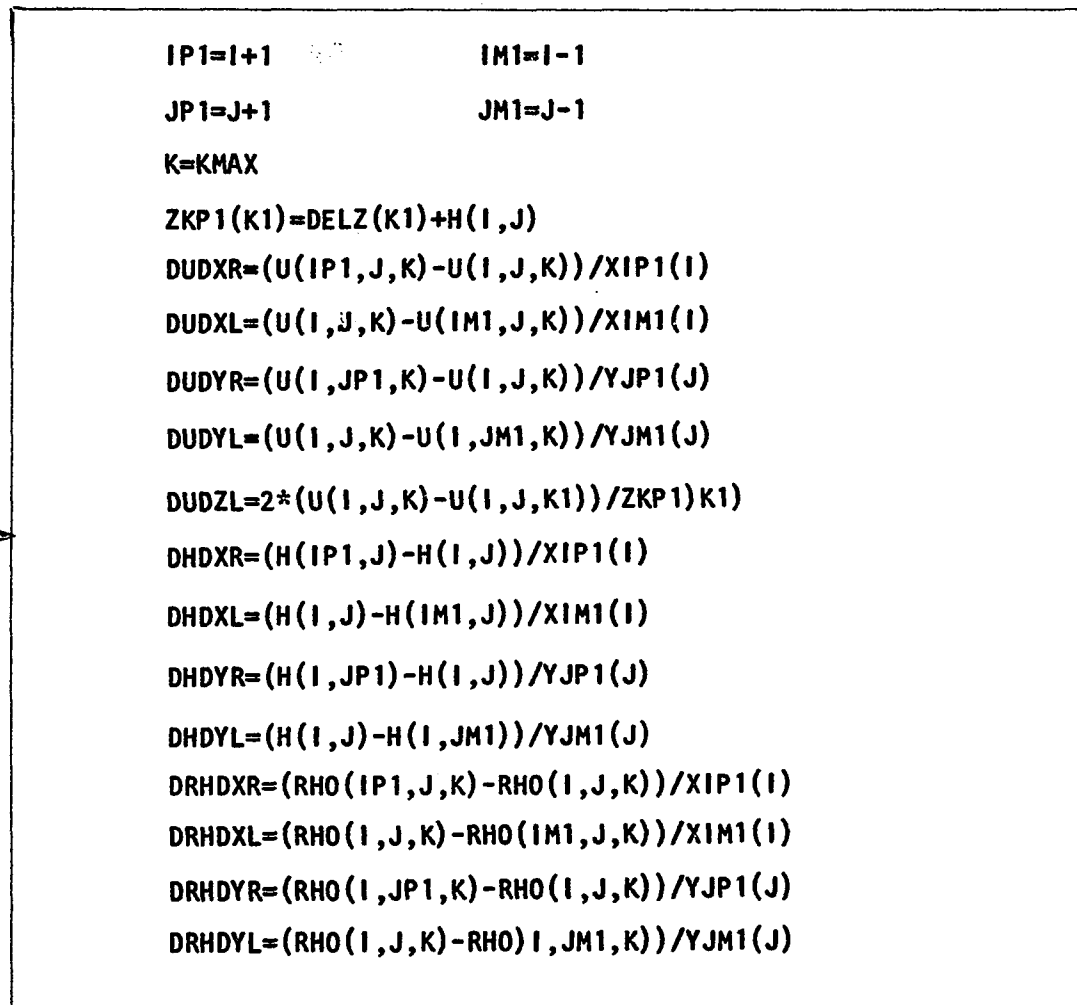


FIG 25a FLOW CHART OF SUBROUTINE DERIV

CALCULATE TIME DERIVATIVE OF U FOR WATER LEVEL ELEMENTS

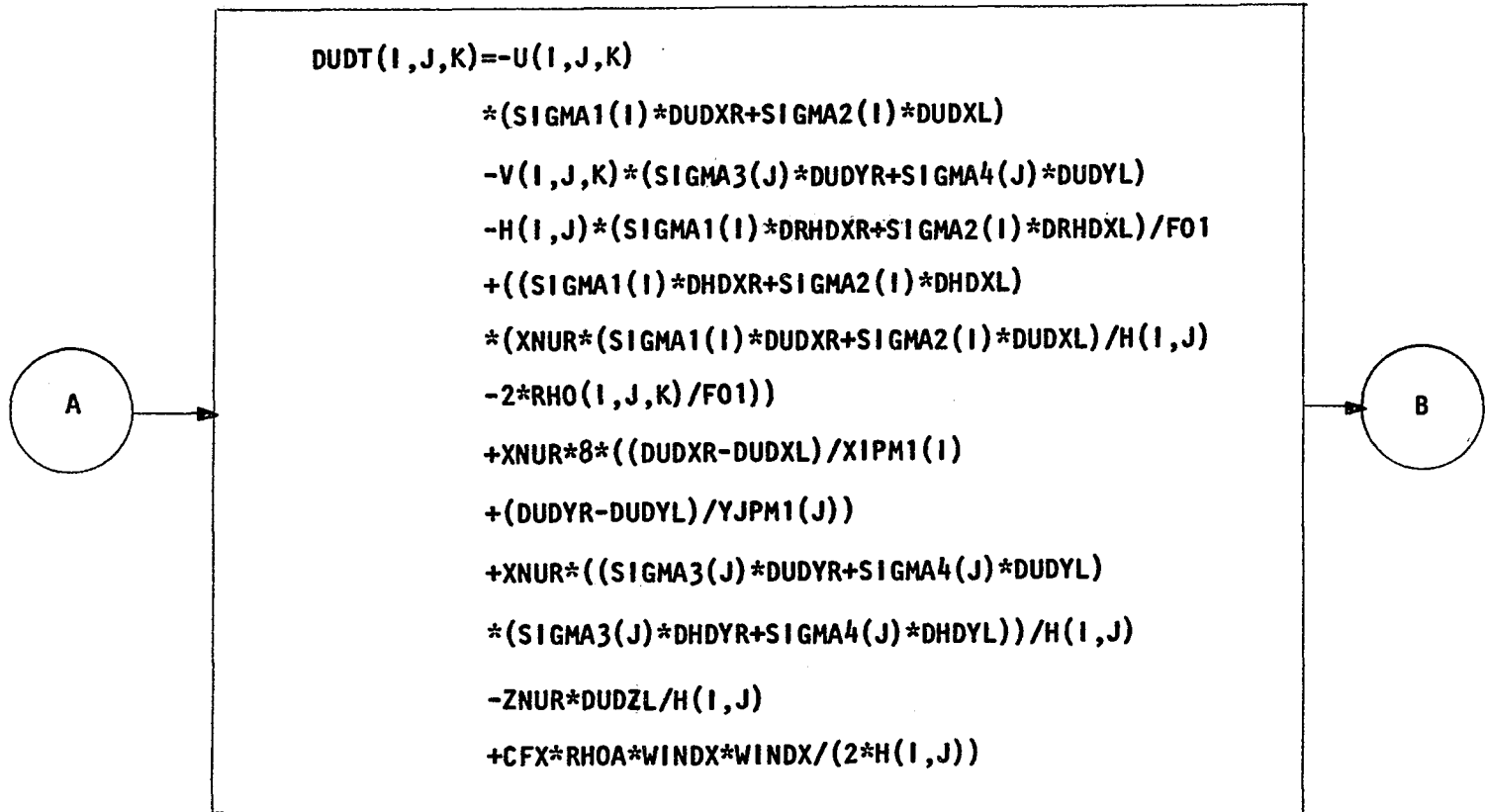


FIG 25b FLOW CHART OF SUBROUTINE DERIV

CALCULATE TIME DERIVATIVE OF V FOR WATER LEVEL ELEMENTS

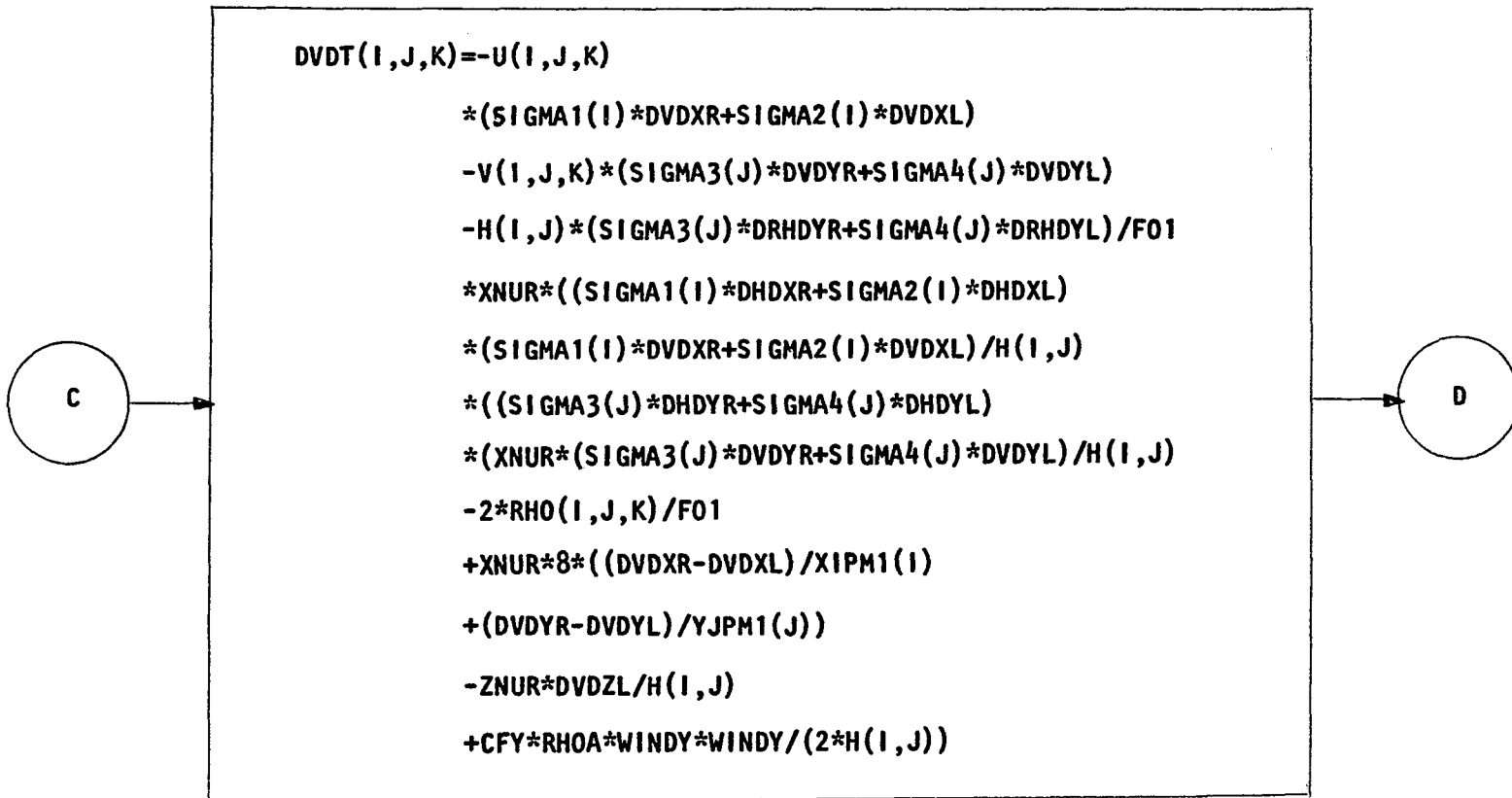


FIG 25d FLOW CHART OF SUBROUTINE DERIV

CALCULATE TIME DERIVATIVE OF T FOR WATER LEVEL ELEMENTS

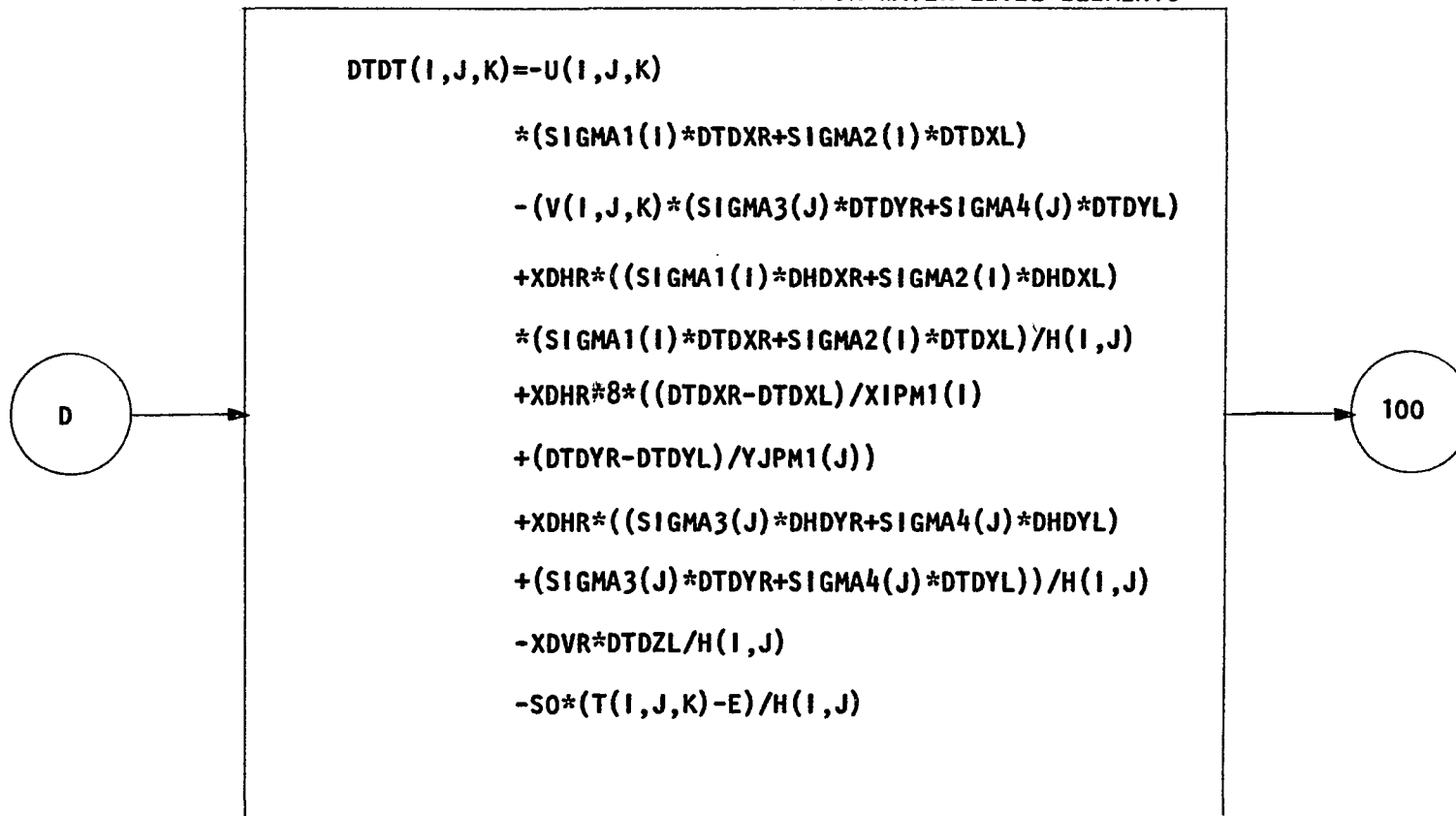


FIG 25e FLOW CHART OF SUBROUTINE DERIV

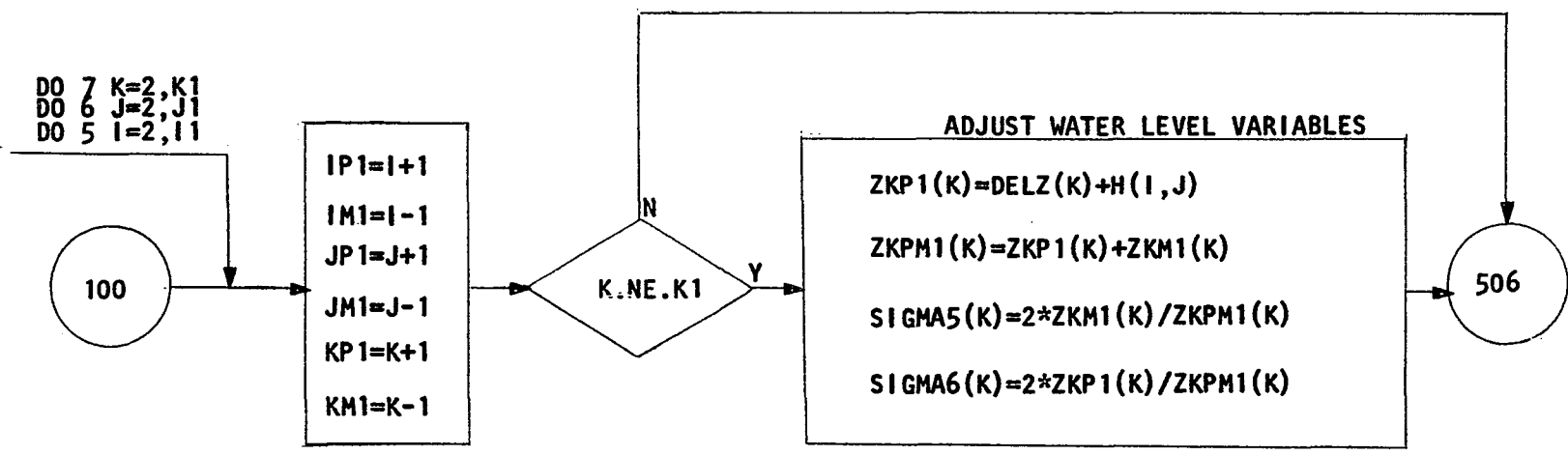


FIG 25f FLOW CHART OF SUBROUTINE DERIV

CALCULATE SPATIAL DERIVATIVE OF U AND TIME DERIVATIVE OF Q_x

506

```
DUDXR=(U(IP1,J,K)-U(I,J,K))/XIP1(I)
DUDXL=(U(I,J,K)-U(IM1,J,K))/XIM1(I)
DUDYR=(U(I,JP1,K)-U(I,J,K))/YJP1(J)
DUDYL=(U(I,J,K)-U(I,JM1,K))/YJM1(J)
DUDZR=(U(I,J,KP1)-U(I,J,K))/ZKP1(K)
DUDZL=(U(I,J,K)-U(I,J,KM1))/ZKM1(K)
DUDT(I,J,K)=-U(I,J,K)
      *(SIGMA1(I)*DUDXR+SIGMA2(I)*DUDXL)
      -V(I,J,K)*(SIGMA3(J)*DUDYR+SIGMA4(J)*DUDYL)
      -W(I,J,K)*(SIGMA5(K)*DUDZR+SIGMA6(K)*DUDZL)
      +8*(XNUR*((DUDXR-DUDXL)/XIPM1(I)
      +(DUDYR-DUDYL)/YJPM1(J)
      +ZNUR*(DUDZR-DUDZL)/ZKPM1(K)
```

E

FIG 25g FLOW CHART OF SUBROUTINE DERIV

CALCULATE SPATIAL DERIVATIVE OF V AND TIME DERIVATIVE OF Q_Y

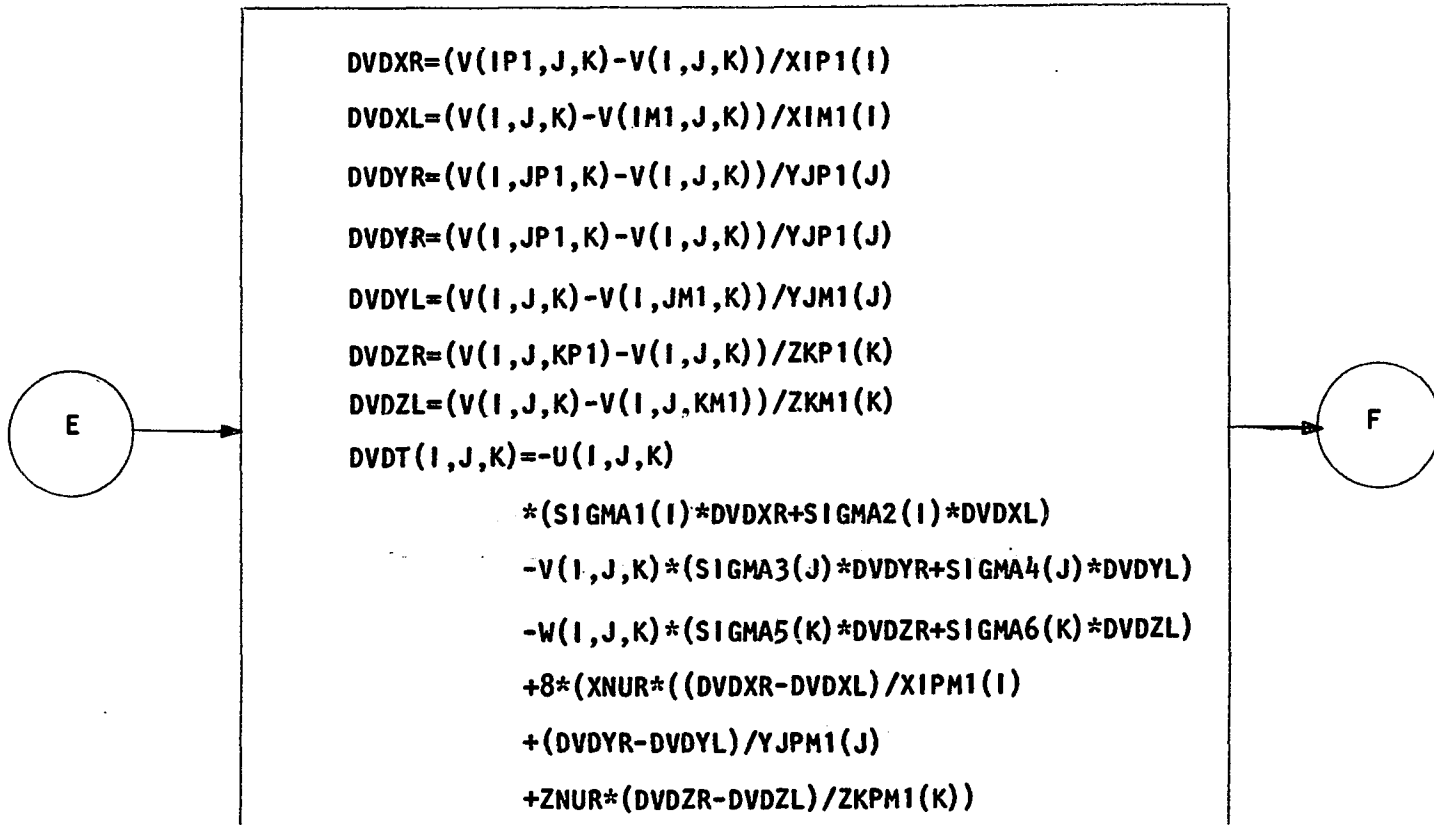


FIG 25h FLOW CHART OF SUBROUTINE DERIV

CALCULATE SPATIAL DERIVATIVE OF W AND TIME DERIVATIVE OF Q_z

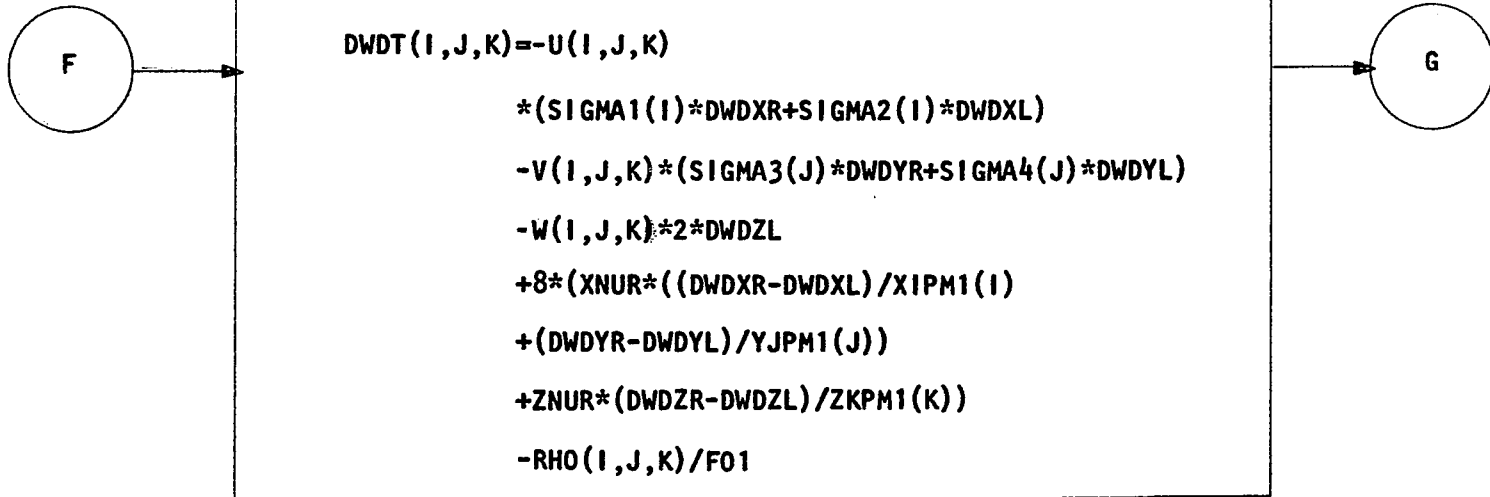


FIG 25i FLOW CHART OF SUBROUTINE DERIV

CALCULATE SPATIAL DERIVATIVE OF T AND TIME DERIVATIVE OF T

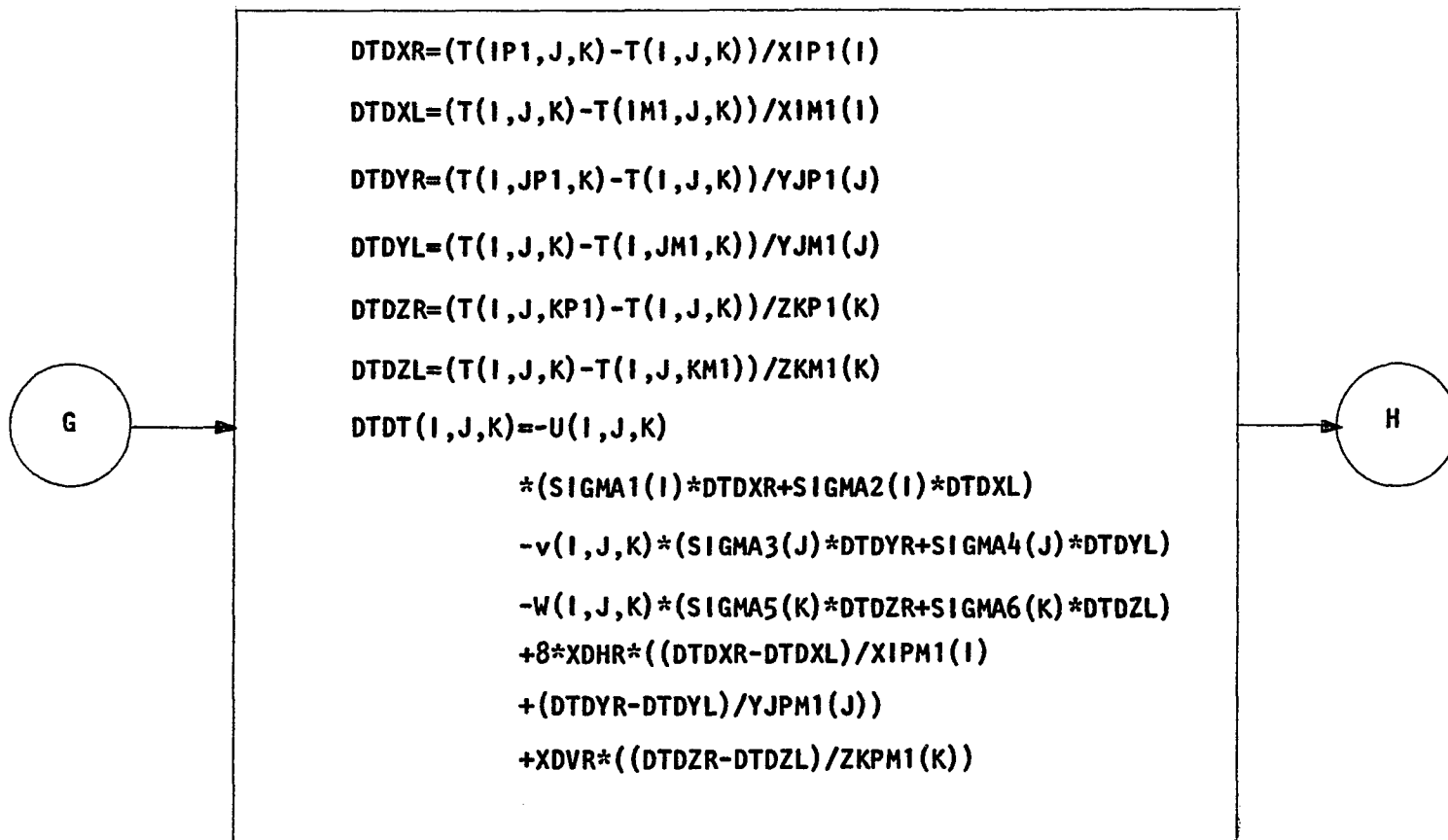


FIG 25j FLOW CHART OF SUBROUTINE DERIV

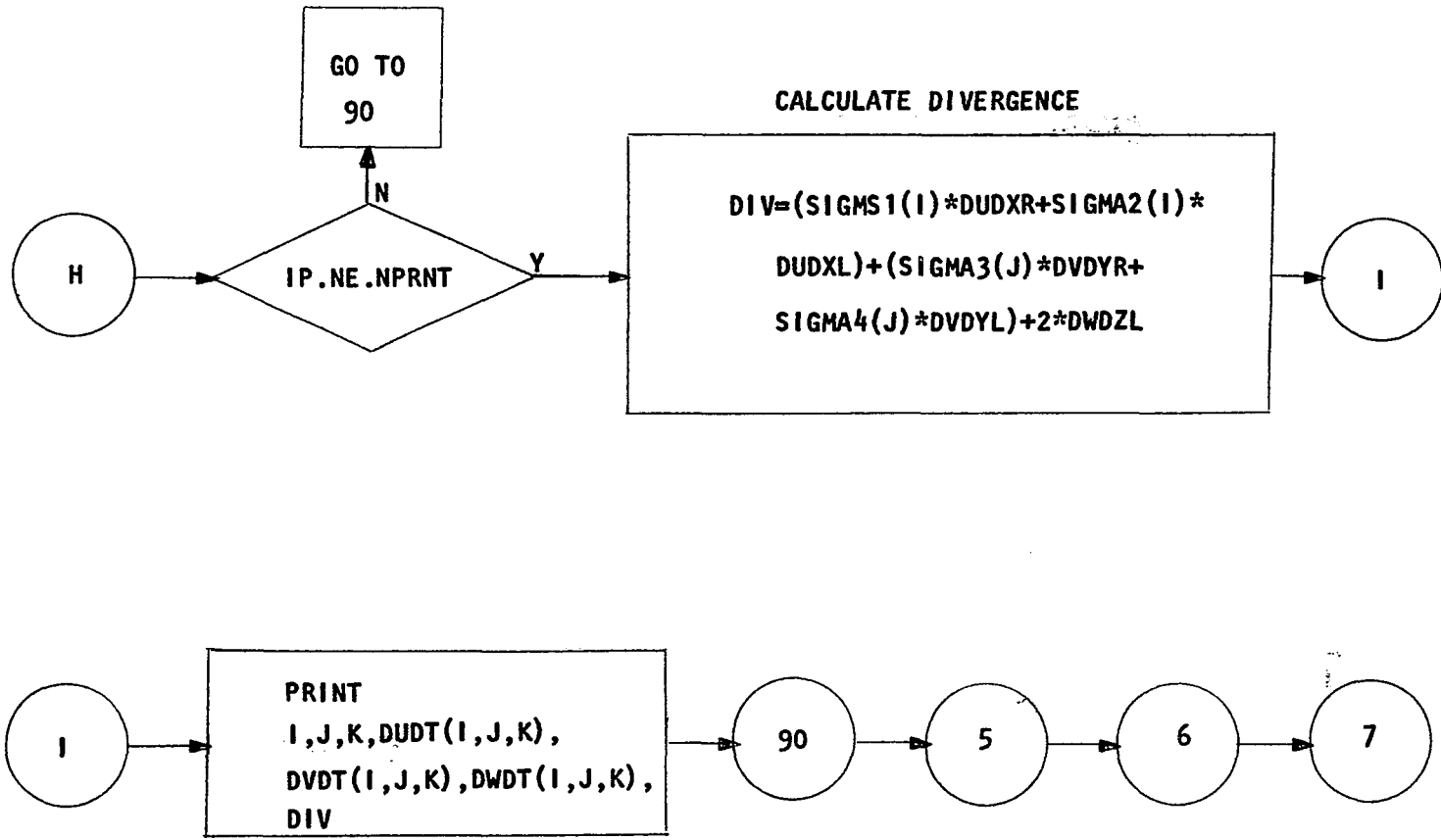


FIG 25k FLOW CHART OF SUBROUTINE DERIV

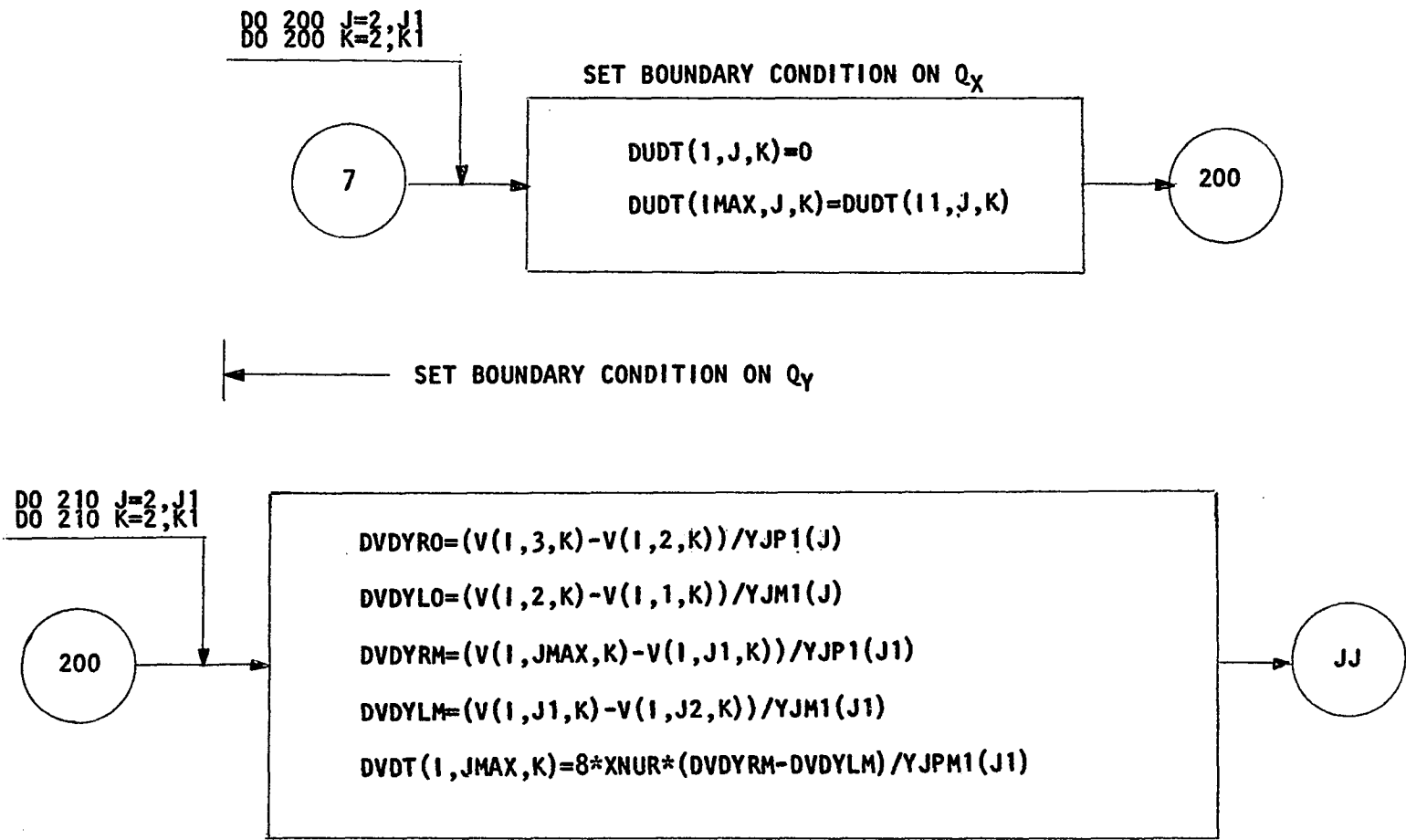


FIG 251 FLOW CHART OF SUBROUTINE DERIV

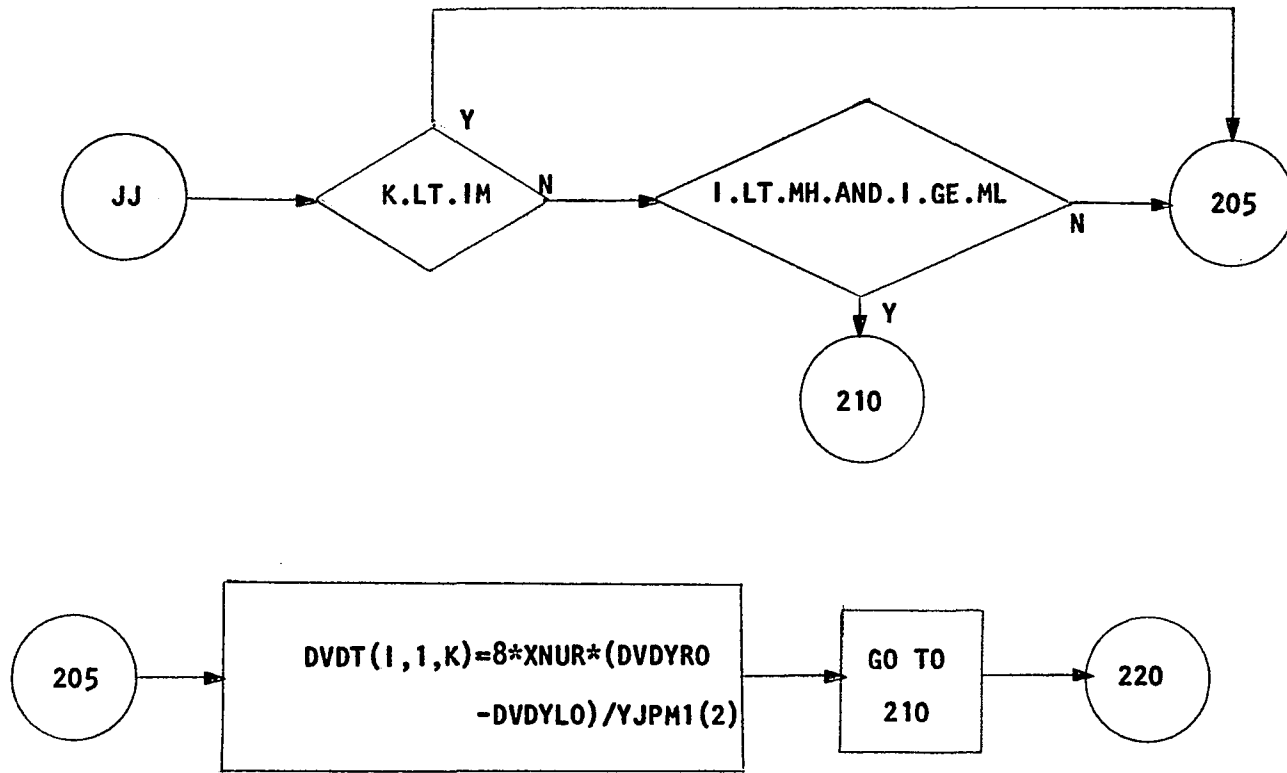
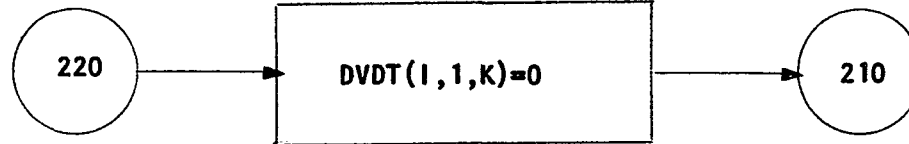


FIG 25m FLOW CHART OF SUBROUTINE DERIV

END BOUNDARY CONDITION ON Q_y



DO 230 I=2,11
DO 230 J=2,J1

SET BOUNDARY CONDITION ON Q_z

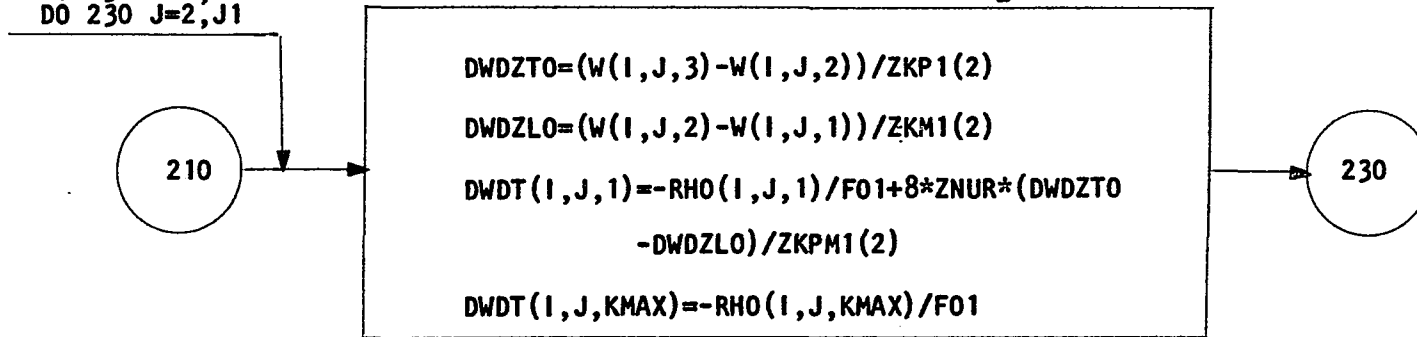


FIG 25n FLOW CHART OF SUBROUTINE DERIV

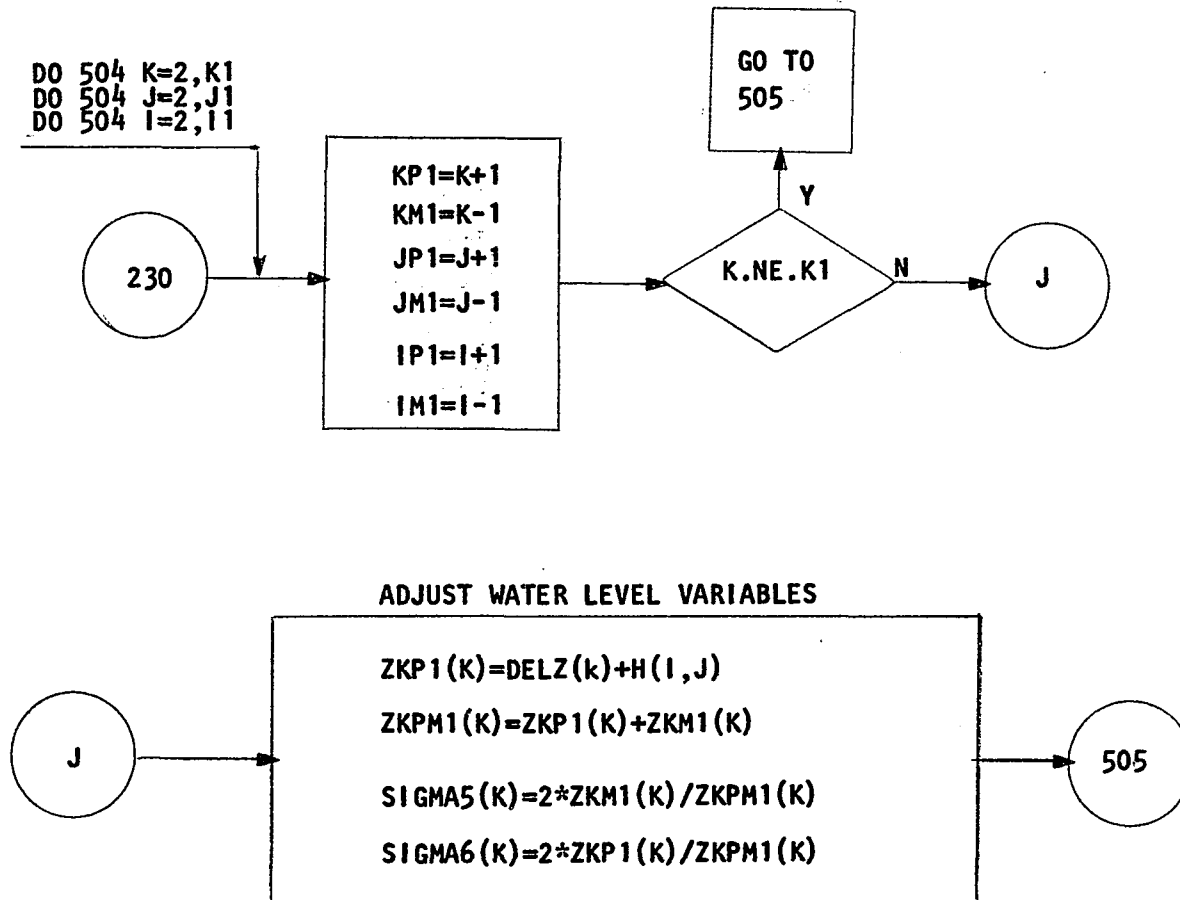


FIG 250 FLOW CHART OF SUBROUTINE DERIV

CALCULATE RHS AND SPATIAL DERIVATIVES OF $Q_x, Q_y,$ AND Q_z

505

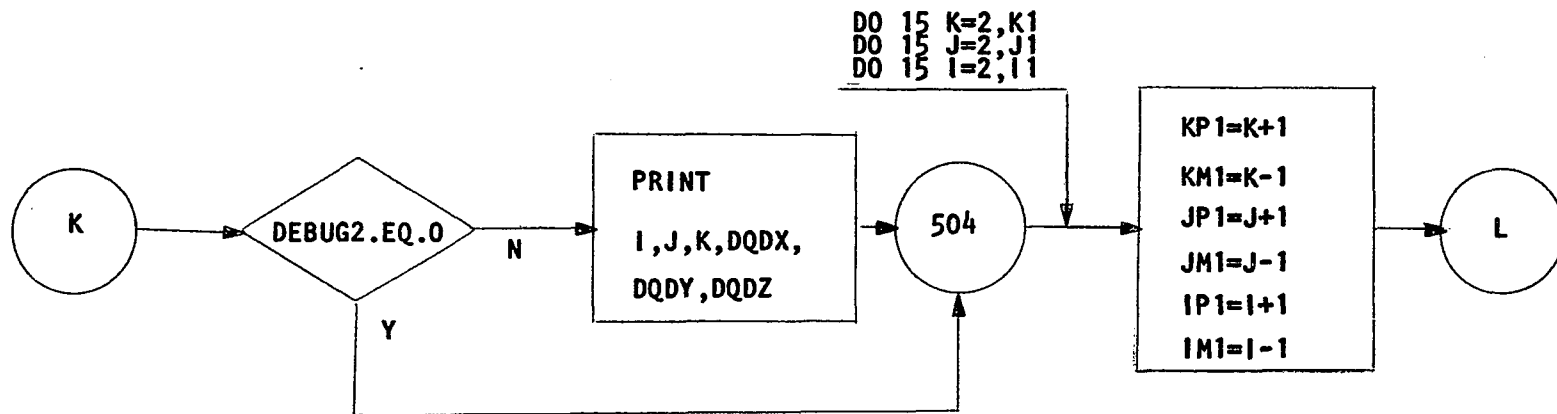
```
DQDZ=(DWDT(I,J,KP1)-DWDT(I,J,K))*SIGMA5(K)/ZKP1(K)
      +(DWDT(I,J,K)-DWDT(I,J,KM1))*SIGMA6(K)/ZKM1(K)
RHS(I,J,K)=DQDZ

DQDY=(DVDT(I,JP1,K)-DVDT(I,J,K))*SIGMA3(J)/YJP1(J)
      +(DVDT(I,J,K)-DVDT(I,JM1,K))*SIGMA4(J)/YJM1(J)
RHS(I,J,K)=RHS(I,J,K)+DQDY

DQDX=(DUDT(IP1,J,K)-DUDT(I,J,K))*SIGMA1(I)/XIP1(I)
      +(DUDT(I,J,K)-DUDT(IM1,J,K))*SIGMA2(I)/XIM1(I)
RHS(I,J,K)=(RHS(I,J,K)+DQDX)*F01
```

K

FIG 25p FLOW CHART OF SUBROUTINE DERIV



CALCULATE SPATIAL DERIVATIVE OF P AND TIME DERIVATIVES OF U AND V

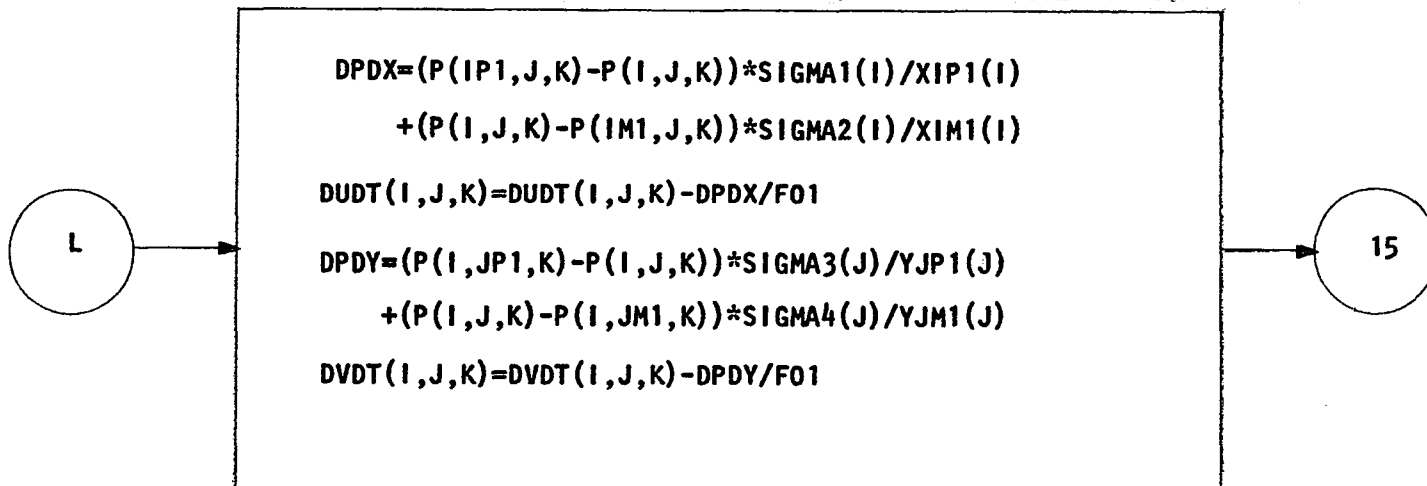


FIG 25q FLOW CHART OF SUBROUTINE DERIV

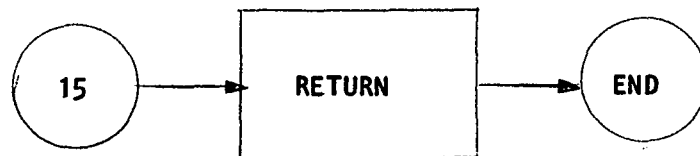


FIG 25r FLOW CHART OF SUBROUTINE DERIV

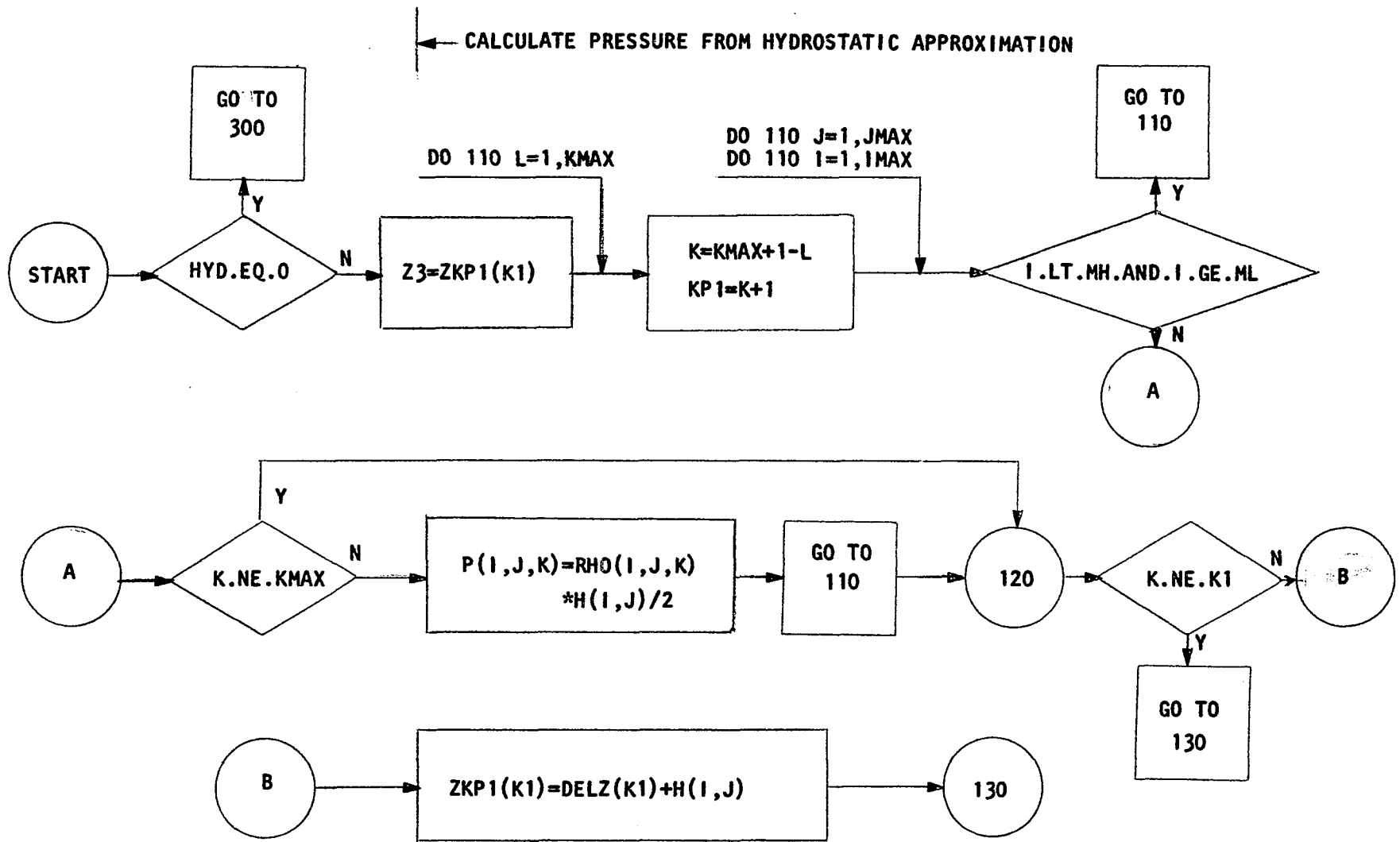


FIG 26a FLOW CHART OF SUBROUTINE PRESS

END HYDROSTATIC APPROXIMATION ON PRESSURE

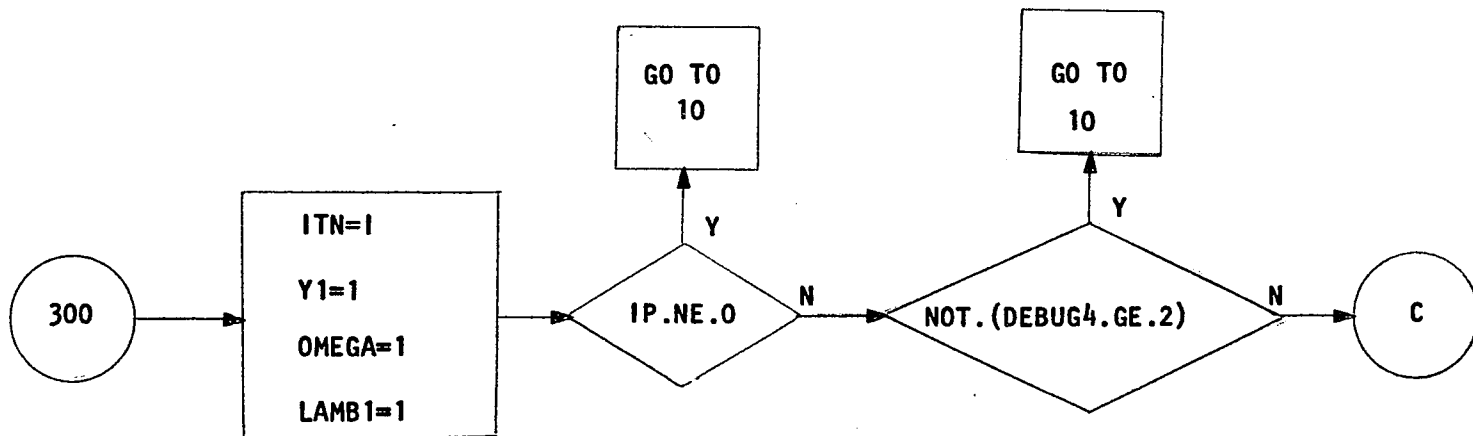
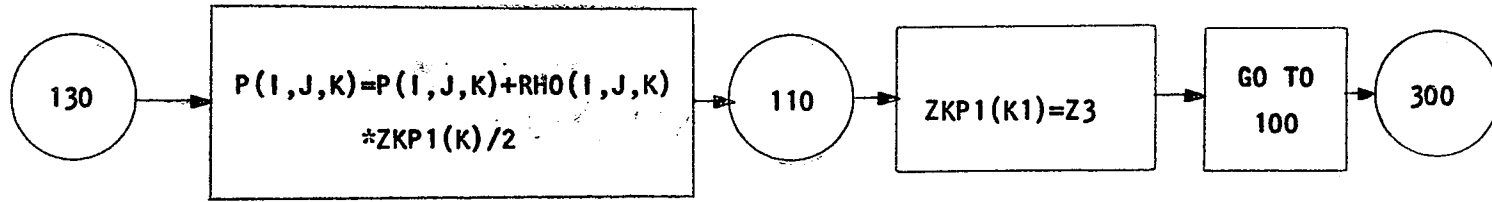


FIG 26b FLOW CHART OF SUBROUTINE PRESS

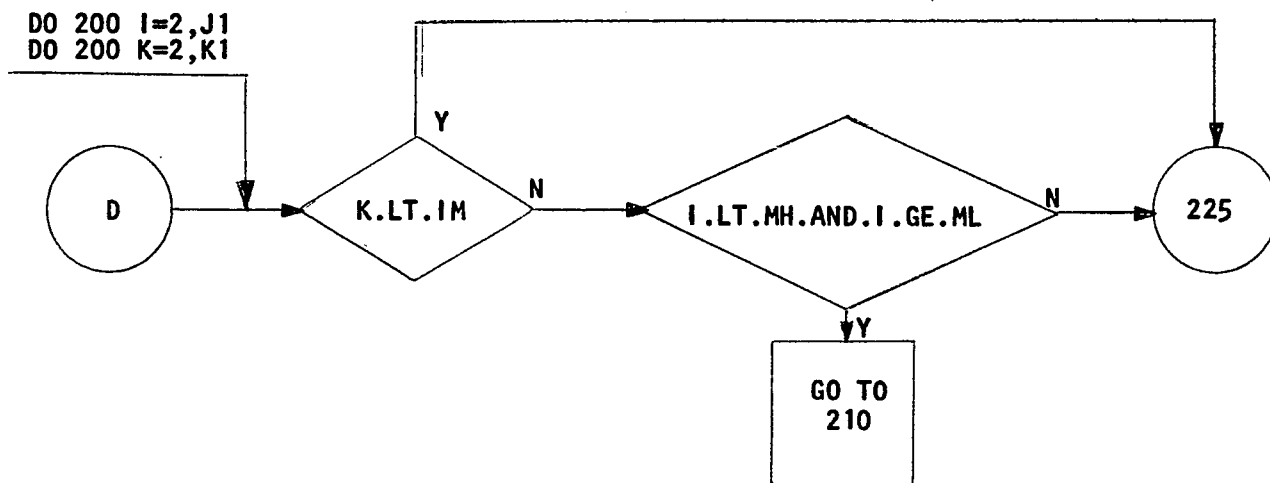
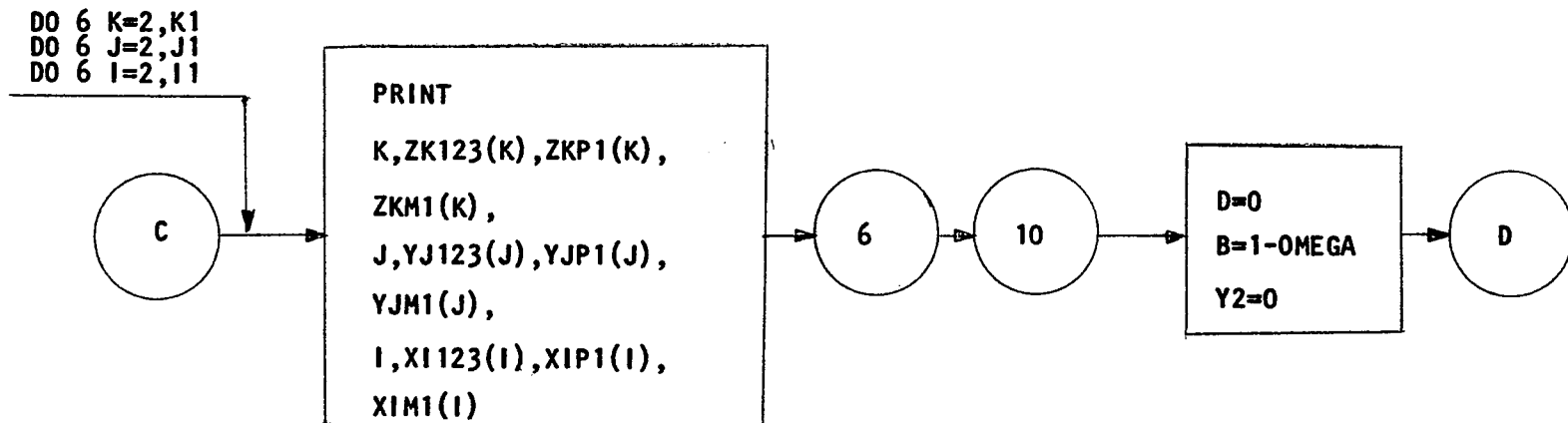


FIG 26c FLOW CHART OF SUBROUTINE PRESS

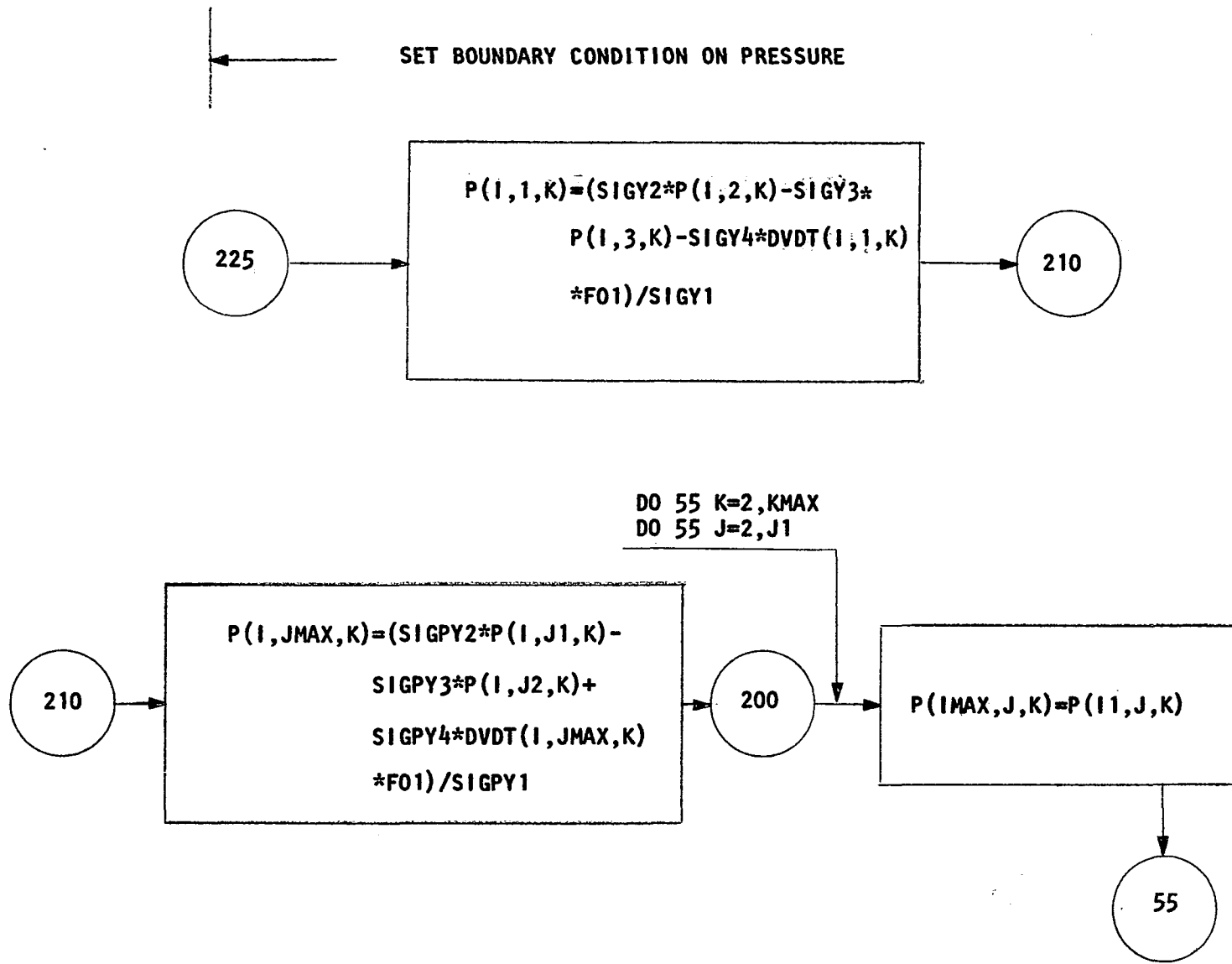
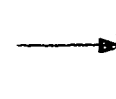
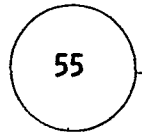


FIG 26d FLOW CHART OF SUBROUTINE PRESS

END BOUNDARY CONDITION ON PRESSURE

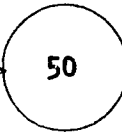


DO 50 I=2,I1
DO 50 J=2,J1

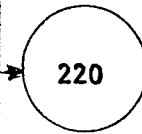


$P(I, J, KMAX) = H(I, J) * RHO(I, J, KMAX) / 2$

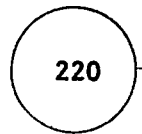
DO 220 I=2,I1
DO 220 J=2,J1



$P(I, J, 1) = (SIGZ2 * P(I, J, 2) - SIGZ3 * P(I, J, 3) - SIGZ4 * DWDT(I, J, 1) * F01) / SIGZ1$



DO 5 K=2,K1
DO 5 J=2,J1
DO 5 I=2,I1



K.NE.K1

Y

N

ADJUST WATER LEVEL VARIABLES

$ZKP1(K) = DELZ(K) + H(I, J)$

$ZKPM1(K) = ZKP1(K) + ZKM1(K)$

$ZK123(K) = ZKP1(K) * ZKM1(K) * ZKPM1(K)$

18

FIG 26e FLOW CHART OF SUBROUTINE PRESS

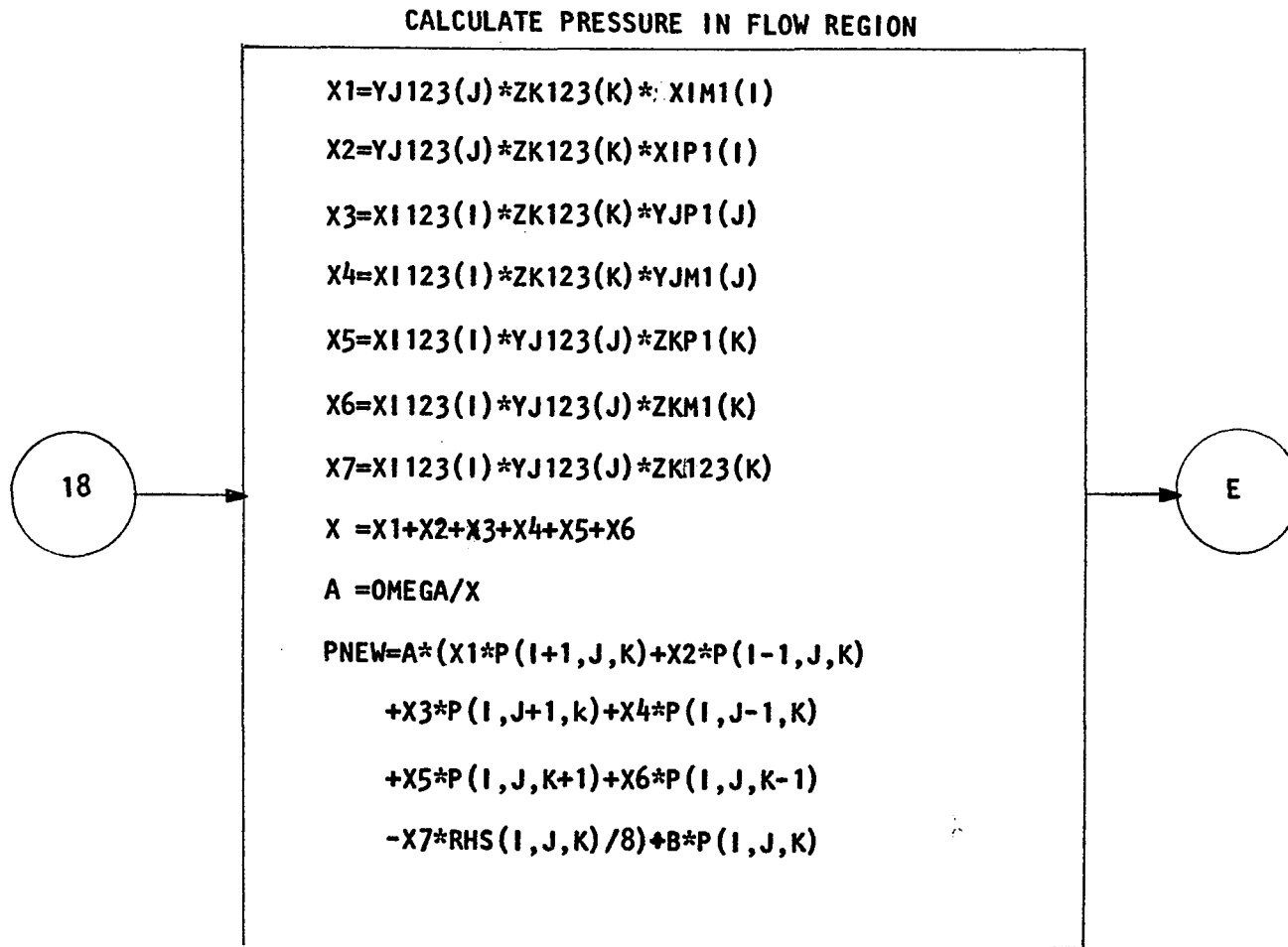


FIG 26f FLOW CHART OF SUBROUTINE PRESS

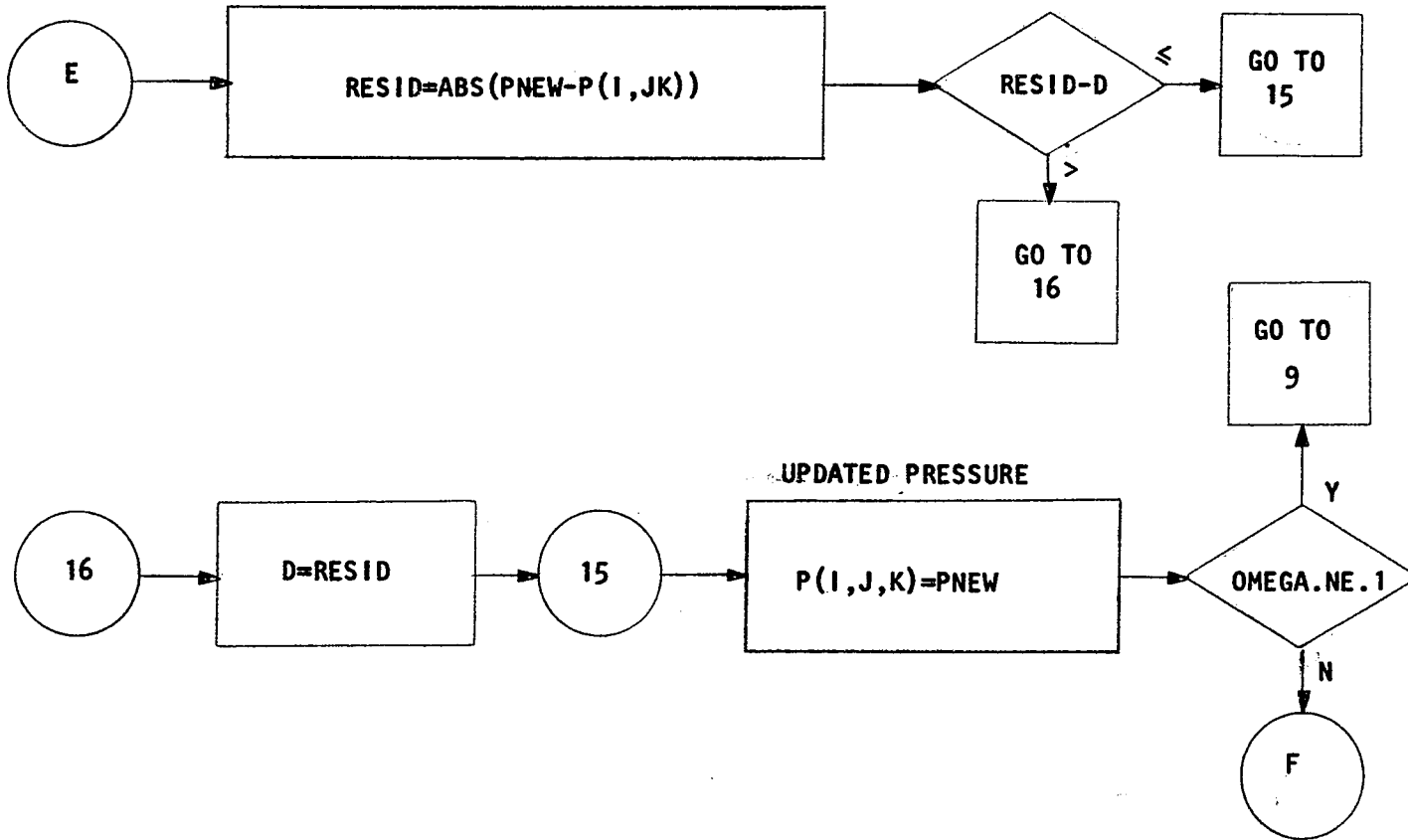


FIG 26g FLOW CHART OF SUBROUTINE PRESS

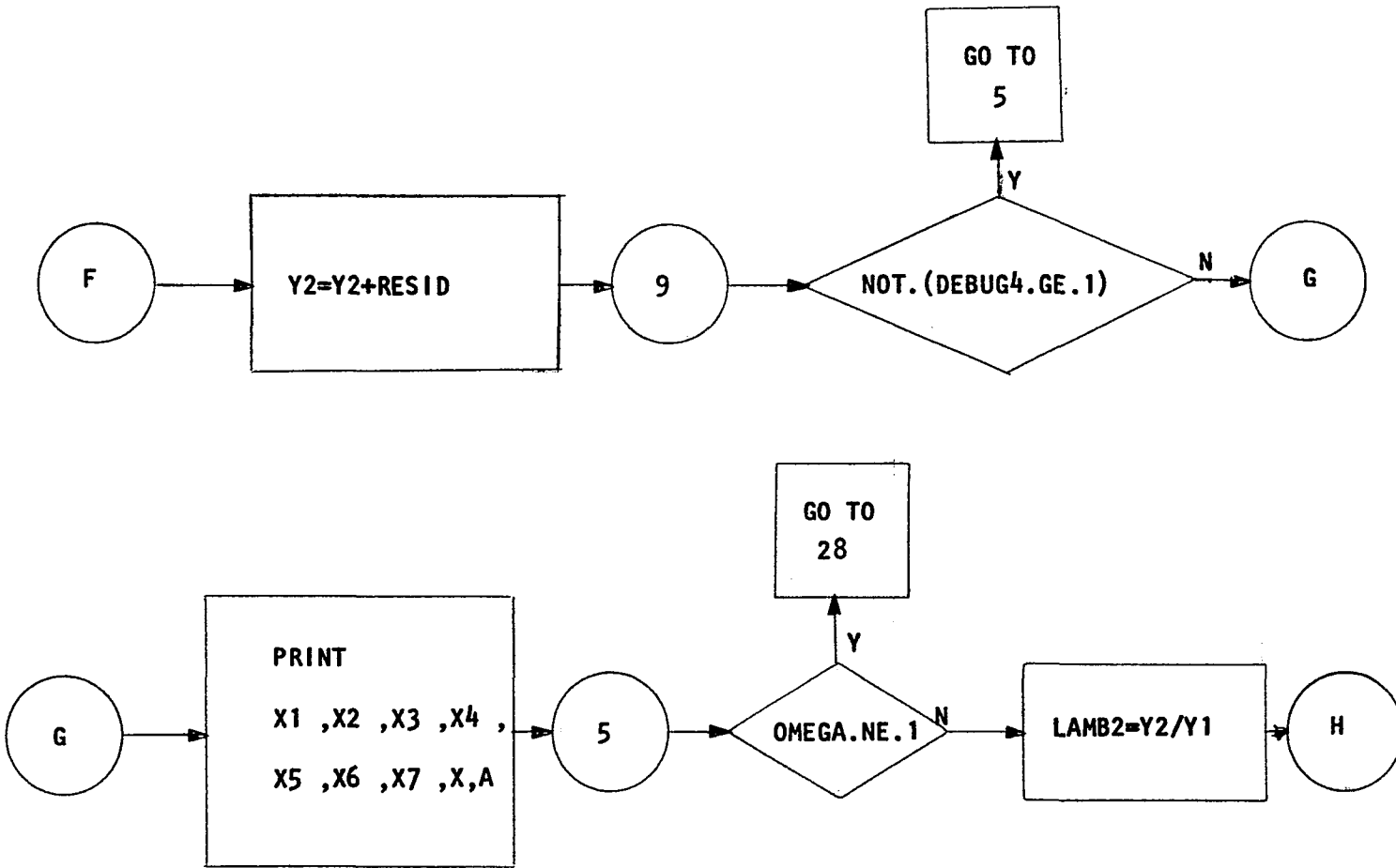


FIG 26h FLOW CHART OF SUBROUTINE PRESS

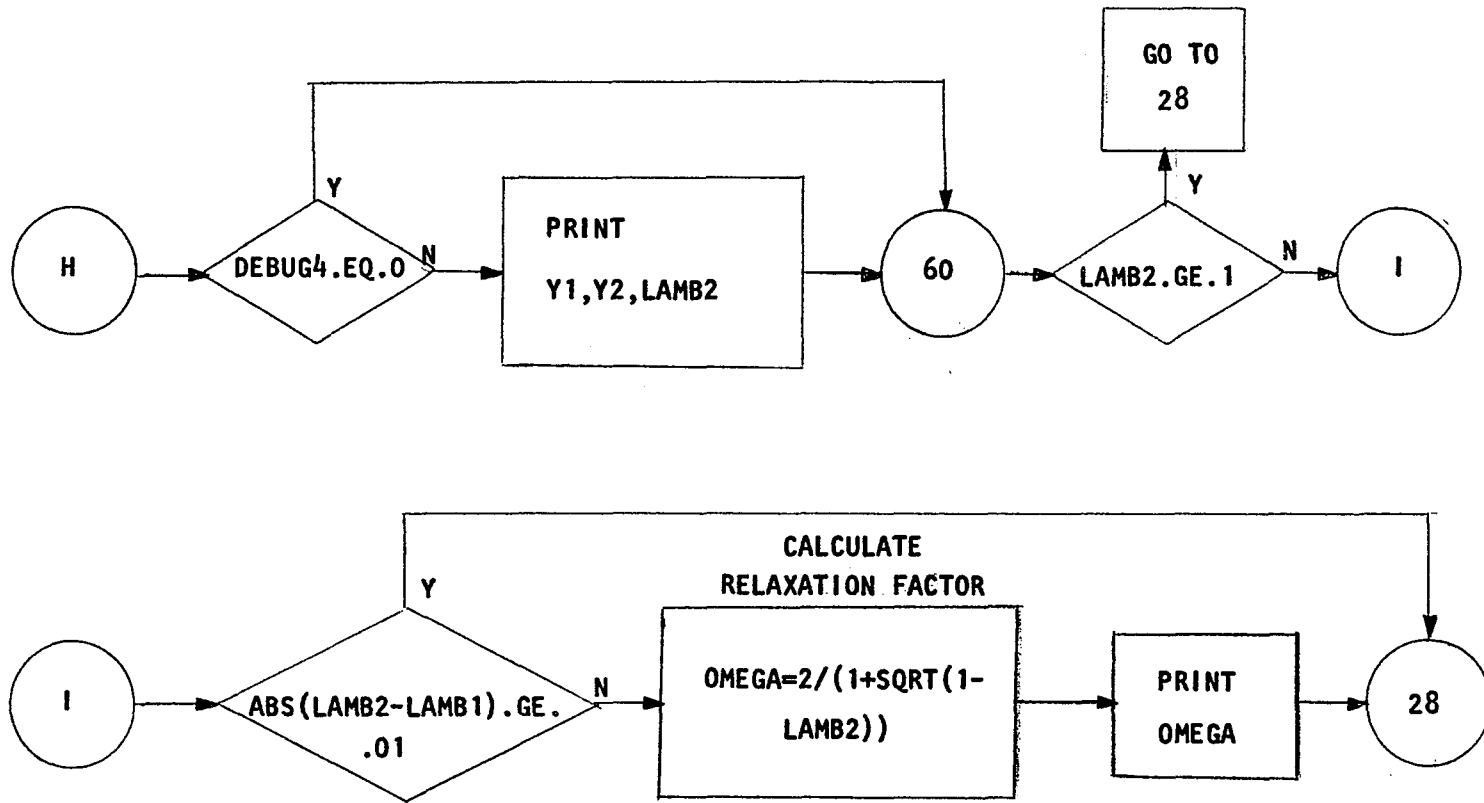


FIG 261 FLOW CHART OF SUBROUTINE PRESS

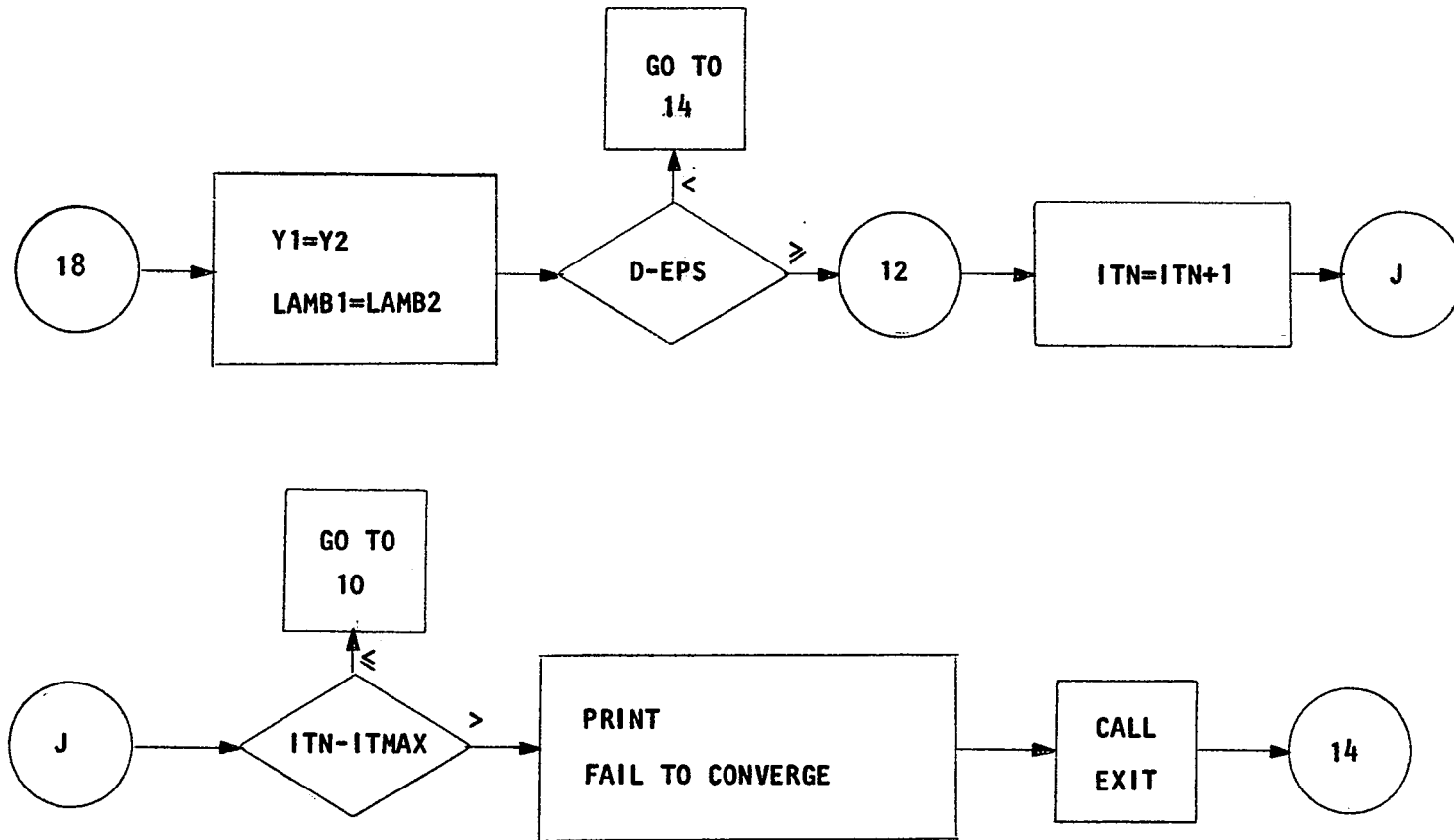


FIG 26j FLOW CHART OF SUBROUTINE PRESS

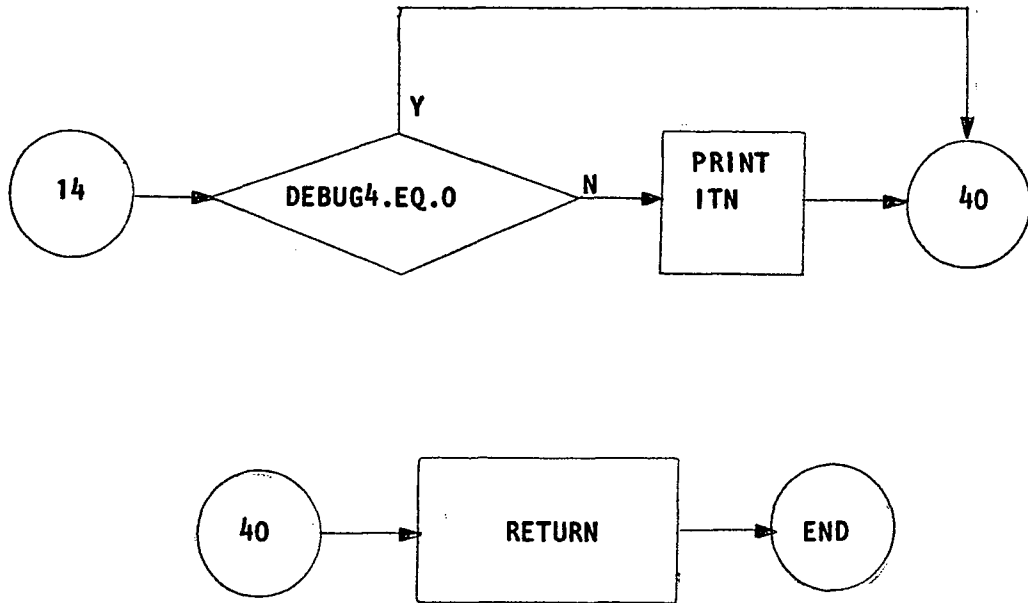
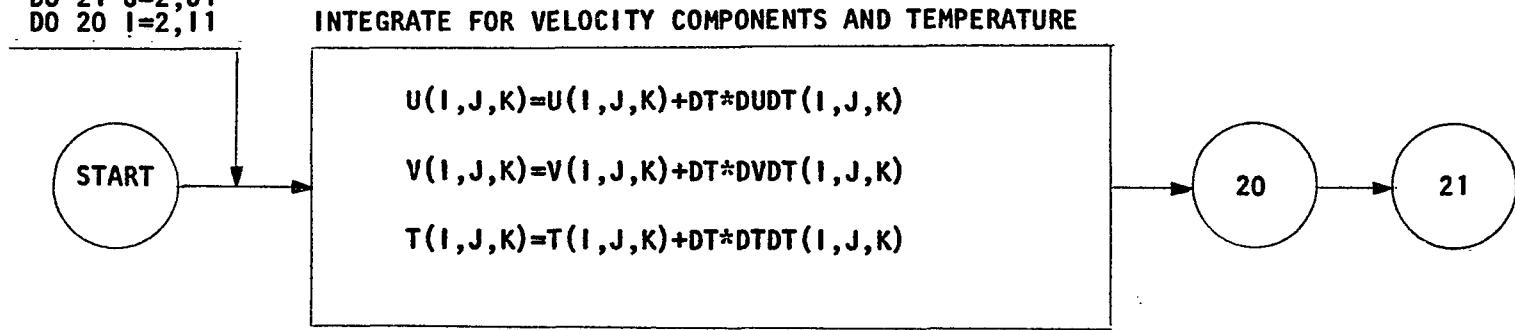


FIG 26k FLOW CHART OF SUBROUTINE PRESS

DO 22 K=2,K1
DO 21 J=2,J1
DO 20 I=2,I1



DO 23 I=1,IMAX
DO 23 J=1,JMAX

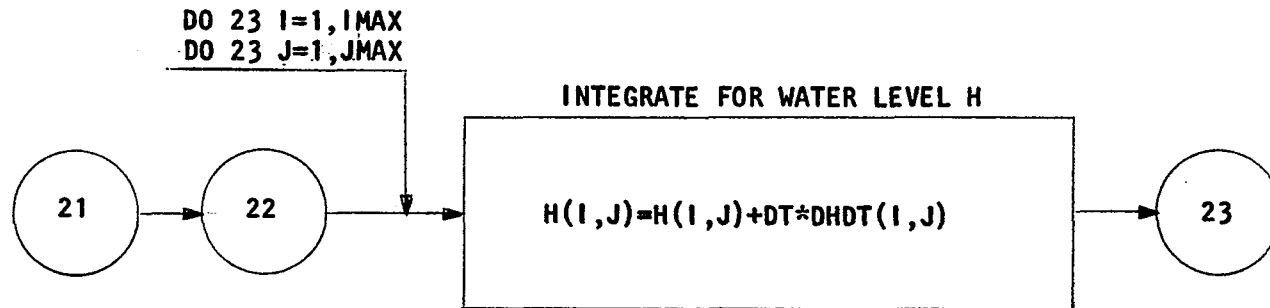


FIG 27a FLOW CHART OF SUBROUTINE SUM

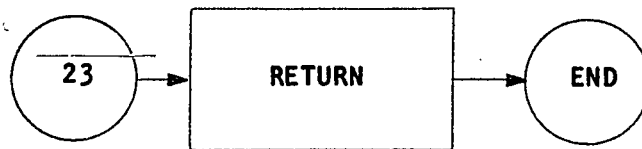


FIG 27b FLOW CHART OF SUBROUTINE SUM

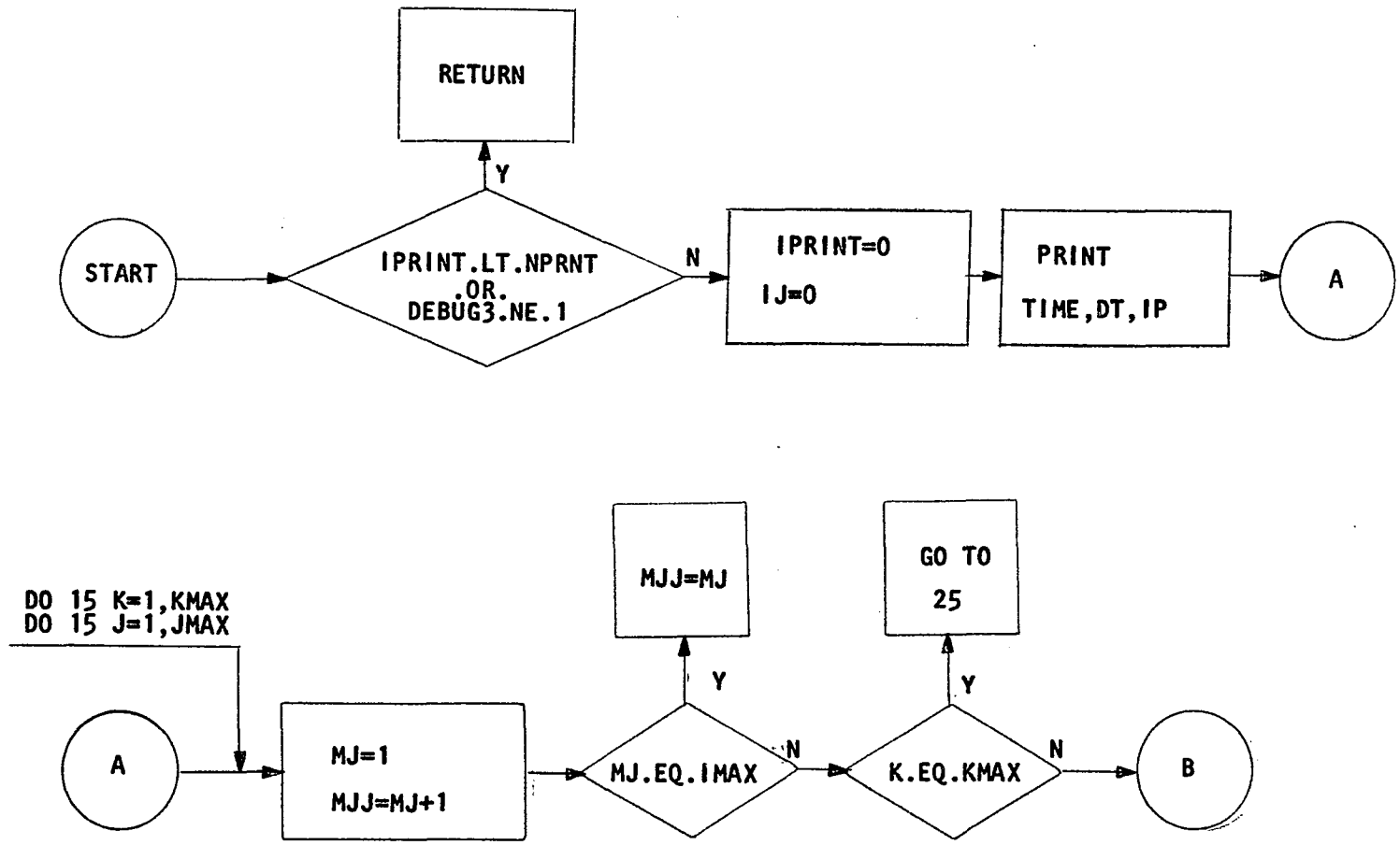


FIG 28a FLOW CHART OF SUBROUTINE OUTP

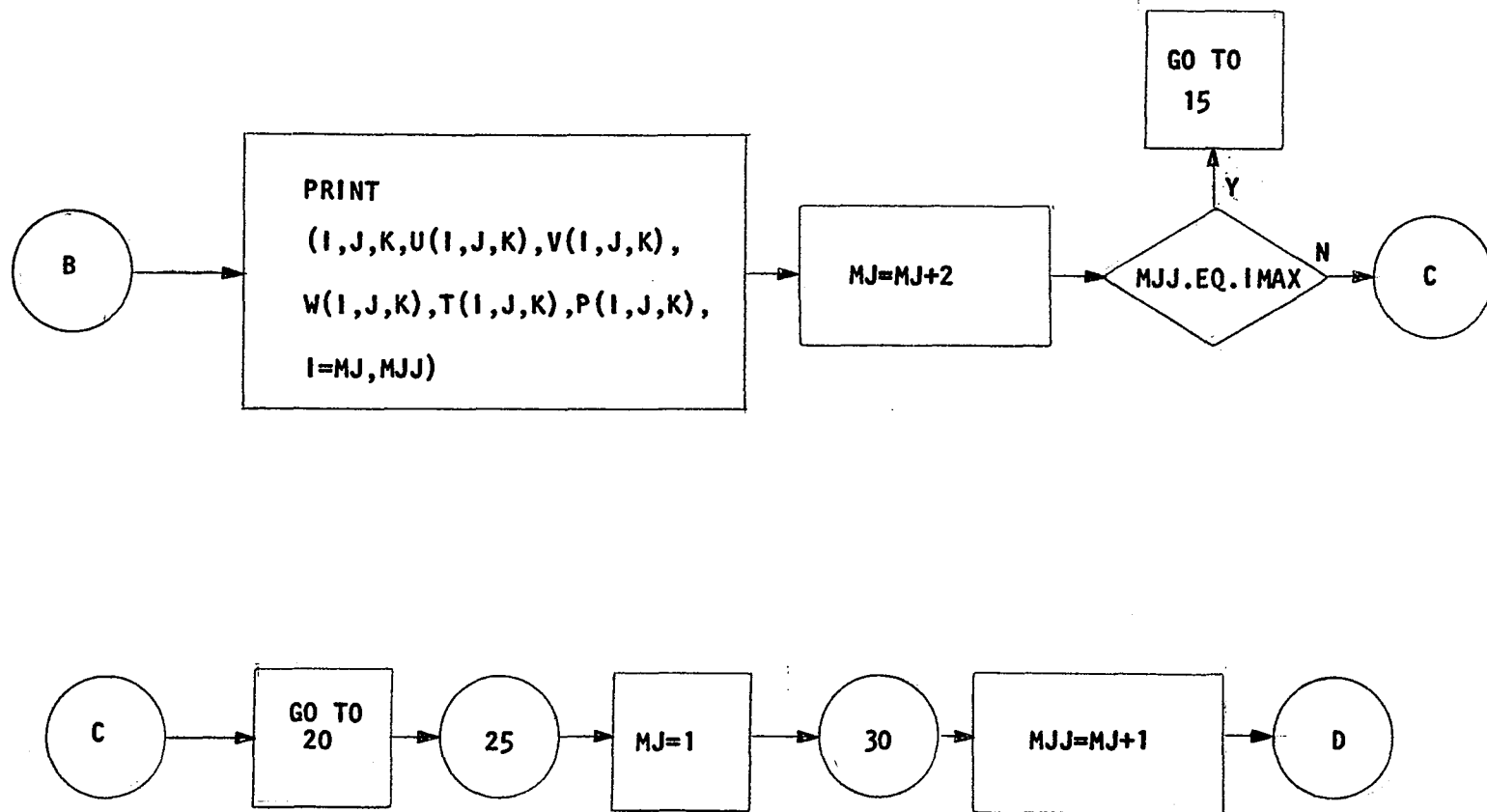


FIG 28b FLOW CHART OF SUBROUTINE OUTP

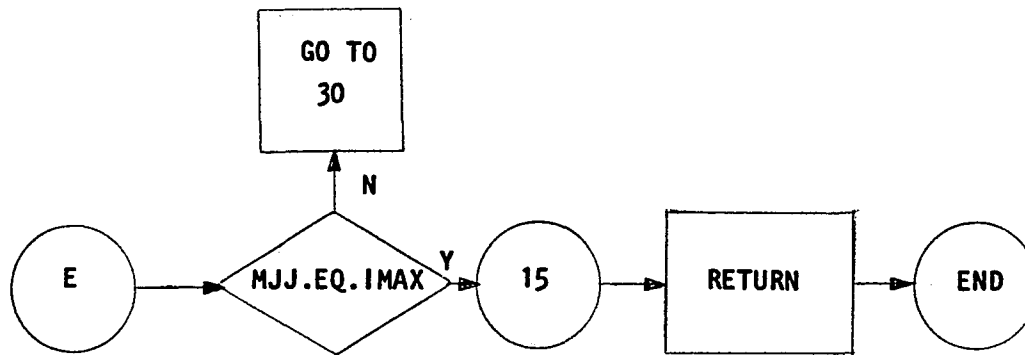
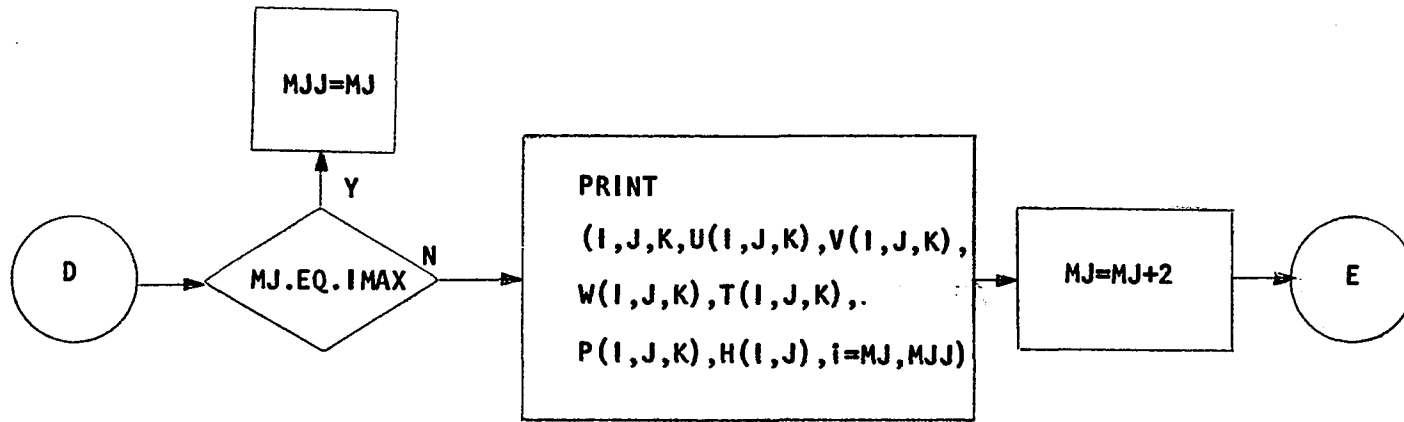


FIG 28c FLOW CHART OF SUBROUTINE OUTP

<u>Subroutine</u>	<u>Function</u>
SI Function	This subroutine performs linear interpolation.
OUTP	This subroutine prints out all the important variables.

1.1 MAIN Program

A detailed description of the MAIN program is presented in the flow charts in the Figs. 24a to 24ll. This program is employed to perform the following three main functions.

1) Reading in the input data. This part of the program reads in and prints out the input data to maintain a permanent record of each run.

2) Initialization of the system. This part of the program performs the following functions:

- a) Converts input data from dimensional values to non dimensional values; and
- b) Sets time independent boundary conditions.

3) Iterative operations performed after each time increment. This part of the program performs the following tasks:

- a) Checks the stability conditions, if these conditions are not satisfied then OUTP subroutine is called and program terminated.

- b) Sets time - dependent boundary conditions.
- c) Calculates vertical velocity components by a spatial integration of equation (61) using updated values of horizontal velocity components.
- d) Calls subroutine OUTF to print out updated values of the important variables.
- e) Calls subroutine DERIV and SUM in the process of finding the updated values.
- f) Compares the current time TIME with the final problem time TEND, in order to stop the problem if TEND is exceeded.

1.2 DERIV Subroutine

A detailed description of the subroutine DERIV is shown in Figs. 25a to 25r. This subroutine calculates the first and the second spatial derivatives of the variables. These spatial derivatives are used along with the pressure gradients calculated in subroutine PRESS to find the time derivatives of the horizontal velocity components, temperature, and level according to equation (78) through (84). Communication among DERIV, MAIN and other subroutine take place by means of a COMMON block. The calling sequence of this subroutine is: CALL DERIV.

1.3 PRESS Subroutine

A detailed description of the subroutine PRESS is shown in Figs. 26a to 26k. This subroutine solves for the pressure according to

equations (73) and (74) by an iterative method, based on a modified Gauss-Seidel iteration technique called successive overrelaxation technique. The calculation of the pressure for the entire flow field is repeated until the values of pressure at every grid point converge within a prescribed error. This subroutine further sets the boundary condition on pressure and calculates the overrelaxation factor ω according to equations (75) through (77). This factor expedites the convergence of the pressure calculations. The variables are passed into and returned from PRESS by means of a COMMON block. The calling sequence of this subroutine is: CALL PRESS.

1.4 SUM Subroutine

A detailed description of the subroutine SUM is shown in Figs. 27a to 27b. This subroutine performs the time integration for the horizontal velocity components, temperature, and water-level. The communications among SUM, MAIN and other subroutines take place by means of a COMMON block. The calling sequence of this subroutine is: CALL SUM.

1.5 SI Function

This function is used to interpolate values from the table of temperature versus density. It is a standard subroutine:

SI(XTBL,YTBL,X,N)

XTBL=Tabular values of the independent variable.

YTBL=Tabular values of the dependent variable.

X=Value of independent variable.

N=Number of points in the table.

1.6 OUTF Subroutine

A detailed description of subroutine OUTF is shown in Figs. 28a to 28c. This subroutine prints out all variables encompassing the problem solution at any desired multiple of integration time step. These variables are passed into OUTF from the other subroutines, including the MAIN program, by means of a COMMON block. The calling sequence of this subroutine is: CALL OUTF.

2. DESCRIPTION OF INPUT DATA

The input data can be divided into two groups:

- 1) Non-subscripted variables
- 2) Subscripted variables.

A brief description of each input variable appears in the program listing to aid in entering the data. All input data are entered using FORMAT E10.3.

2.1 Non-Subscripted Variables

These variables are floating point numbers and are entered into the computer storage by means of cards as specified hereunder:

Card No. 1

1) ENRUN, Run designation number. This number is used to identify each run for record keeping purposes.

Card No. 2

1) TBEG, Starting time, sec. This quantity specifies the initial problem time at which the dynamic problem starts.

2) TEND, End of time, sec. This variable defines the final problem time at which the dynamic problem stops automatically. The dynamic analysis may be aborted, prior to TEND, due to the occurrence of computational errors.

3) ENITER, Number of integration steps. This quantity defines the number of integration steps to be used between starting time

TBEG, and end of time TEND. The program calculates the integration time increment by

$$\Delta t = (TEND - TBEG) / ENITER$$

4) ENPRNT, Frequency of printout. This variable sets the number of integration steps between two successive printouts.

Card No. 3

1) UI, Initial velocity component in x-direction, ft/sec.

This is the initial value of the x-component of the fluid velocity assumed constant throughout the water body. It is used to obtain the dynamic response of the water body from a set of uniform velocity distributions as discussed in Part 1 Section 5.

2) VI, Initial velocity component in y-direction, ft/sec.

This is the initial value of the y-component of the fluid velocity assumed constant throughout the water body. The purpose of the introduction of this variable is described under variable UI above.

3) TI, Initial temperature of water body and inflow, degrees Fahrenheit. This variable defines the initial value of the fluid temperature assumed constant throughout the water body. The purpose of the introduction of this variable is described under variable UI above. Furthermore, this variable defines the inflow temperature into the water body assumed constant during the dynamic analysis.

Card No. 4

1) UO, Thermal discharge velocity, ft/sec. This quantity specifies the water velocity in the thermal discharge channel from the power plant entering the water body. This variable is used to non-

dimensionalize all velocity components involved in the program.

2) ANGLE, Thermal discharge angle, radian. This variable defines the angle at which the thermal discharge enters the water body measured from the x-axis. The x and y velocities of the thermal discharge entering the water body are calculated by $U_0 \cos(\text{ANGLE})$ and $U_0 \sin(\text{ANGLE})$ respectively.

3) TO, Thermal discharge temperature, degrees Fahrenheit. This variable specifies the water temperature in the thermal channel from the power plant entering the water body. This variable is used to nondimensionalize all temperature quantities involved in the program and should be always entered regardless of the value of HEAT on Card No. 9 to be discussed later.

4) DO, The half width of thermal discharge channel, ft. This variable is the half width of the thermal discharge channel. This variable is used to nondimensionalize all geometric dimensions involved in the program.

5) XNUZ, Kinematic viscosity of thermal discharge, ft²/sec. This variable defines the kinematic viscosity of the thermal discharge water. It is used to nondimensionalize the thermal eddy diffusivities and momentum eddy viscosities.

Card No. 5

1) WINDX, x-component of wind velocity, ft/sec. This quantity is the wind velocity component in x-direction assumed to be blowing horizontally over the water body.

2) WINDY, y-component of wind velocity, ft/sec. This quantity

is the wind velocity component in y-direction. This variable together with WINDX provides the means of analyzing wind effects blowing at an oblique angle with respect to the Cartesian coordinate system.

3) RHOA, Air density, lbm/ft³. This variable specifies the air density above the water body. It affects the water body dynamics because the wind shear effect at the water surface is proportional to air density as described in Part 1, Section 2.3.

4) E, Water body equilibrium temperature, degrees Fahrenheit. This variable specifies the water equilibrium temperature. It is used in the calculation of heat dissipation from water body surface to atmosphere by employing equation (54).

5) XK, Heat exchange coefficient, Btu/(ft²-day-°F). This variable specifies the coefficient of heat transfer between water body surface and atmosphere. It is used in the calculation of heat dissipation from water body to atmosphere by employing equation (54).

6) CFX, The skin coefficient in x-direction. This variable specifies the skin coefficient in x-direction. It is used to calculate the wind shear effect in x-direction from equation (52).

7) CFY, The skin coefficient in y-direction. This variable specifies the skin coefficient in y-direction. It is used to calculate the wind shear effect in y-direction from equation (53).

Card No. 6

1) DH, Horizontal eddy diffusivity of heat, ft²/sec. This variable is the eddy diffusivity of heat for the water body in the

horizontal direction. It is used to calculate the turbulent heat exchange in the water in both x and y directions.

2) Dv, Vertical eddy diffusivity of heat, ft²/sec. This variable is the eddy diffusivity of heat for the water body in the vertical direction. It is used to calculate the turbulent heat exchange in the water in the z direction.

3) XNUH, Horizontal eddy viscosity, ft²/sec. This variable defines the eddy diffusivity of momentum for the water body in the horizontal direction. It is used to calculate the turbulent momentum transfer in the water in both x and y directions.

4) XNUV, Vertical eddy viscosity, ft²/sec. This variable defines the eddy diffusivity of momentum for the water body in the vertical direction. It is used to calculate the turbulent momentum transfer in the water in the z-direction.

5) CP, Specific heat of water, Btu/lbm-^oF. This quantity is the specific heat of water in the water body and is used in the calculation of heat transfer from the water to the atmosphere from equation (54).

Card No. 7

1) EIMAX, Number of cells in x-direction. This variable specifies the number of space increments, in the flow field, along x-axis.

2) EJMAX, Number of cells in y-direction. This variable represents the number of space increments, in the flow field, along the y-axis.

3) EKMAX, Number of cells in z-direction. This variable defines the number of space increments, in the flow field, along the z-axis.

Card No. 81) DEBUG1, Debugging printout controller in the MAIN program.

This quantity controls the printout of the calculated variable in the MAIN program for debugging purposes. Calculated results are printed out when $DEBUG1 = 1$. No printout occurs when $DEBUG1 = 0$.

2) DEBUG2, Debugging printout controller in subroutine DERIV.

This quantity controls the printout of the calculated variable in subroutine DERIV for debugging purposes. Calculated results are printed out when $DEBUG2 = 1$. No printout occurs when $DEBUG2 = 0$.

3) DEBUG3, Unused.4) DEBUG4, Debugging printout controller in subroutine PRESS.

This quantity controls the printout of the calculated variables in subroutine PRESS for debugging purposes. Calculated results are printed out when $DEBUG4 = 1$. No printout occurs when $DEBUG4 = 0$.

5) EIRUN, Restart switch. This quantity controls the initialization of the program. When $EIRUN = 0$, program will start from initial condition described earlier in this section. When $EIRUN = 1$, program will skip the initialization and will use previous output already stored on tape as initial values. The latter tape storage is achieved by setting $EIEND = 1$ in the previous run as discussed next.

6) EIEND, Restart controller. This quantity controls the output at the completion of a dynamic run. When $EIEND = 1$, the final program output will be stored on tape for restarting the run at a later date. When $EIEND = 0$, no storage of final program output will be made.

Card No. 91) EIM, Number of thermal discharge cells in vertical direction.

This parameter indicates the number of grid cells in the vertical direction in the thermal discharge channel.

2) EML, Starting location of thermal discharge. This parameter indicates the grid cell number in the x-direction where the thermal discharge channel begins.

3) EMK, End location of thermal discharge. This parameter indicates the grid cell number in the x-direction where the thermal discharge channel ends.

4) SLIP, Slip condition controller. This quantity indicates whether boundary conditions on velocity components are calculated using slip or non-slip conditions. When $SLIP = 1.$, the boundary conditions are calculated using slip conditions. When $SLIP = 0.$, the boundary conditions are calculated with non-slip conditions.

5) HYD, Hydrostatic pressure controller. This quantity indicates whether pressure is calculated by hydrostatic approximation or from equations of motion. When $HYD = 1.$, the pressure is calculated by hydrostatic approximation. When $HYD = 0.$, the pressure is not hydrostatic and is calculated from the equations of motion indicated by equations (73) and (74).

6) HEAT, Temperature controller. This quantity controls whether the thermal discharge is heated or not. When $HEAT = 1.$, it indicates that the thermal discharge has a higher temperature than surrounding fluid. When $HEAT = 0.$, it means that the thermal discharge is at same temperature as the surrounding fluid.

7) XIIP, Off line storage controller. This parameter controls the off-line storage of the output data for subsequent restart. At

the appropriate iteration, indicated by XIIP, the output are stored on a disc, and recalled in the restart mode.

2.2 Subscripted Variables

These variables involve fixed size arrays and are entered into the computer storage using FORMAT 7E10.3.

1) DELX(I), x-increments, ft. This array defines the spatial increments, in the flow field, along the x-axis. The number of cards needed to enter this array is equal to EIMAX/7.

2) DELY(J), y-increments. This array defines the spatial increments, in the flow field, along the y-axis. The number of cards needed to enter this array is equal to EJMAX/7.

3) DELZ(K), z-increments. This array defines the spatial increments, in the flow field, along the z-axis. The number of cards needed to enter this array is equal to EKMAX/7.

4) Density versus Temperature, lbm/ft³ and degree Fahrenheit respectively. The densities and their corresponding temperatures, given in Table 1, are entered in a BLOCK DATA routine. The following format is used in entering the data:

```
BLOCK DATA
```

```
COMMON/TABLE/TTBL,RHOTBL
```

```
REAL * 4 TTBL(26)/26 values of temperature/
```

```
REAL * 4 RHOTBL(26)/26 values of density/
```

3. DESCRIPTION OF OUTPUT DATA

The output data can be conveniently divided into the following groups:

1) Input Data Printout

All input data are printed out and labeled using the variable names defined in the description of input data. The temperature and density relation, entered by BLOCK DATA, are printed in a tabular form with one column labeled "Temperature" and the other column labeled "Density", followed by the appropriate values.

2) Debugging Printout

At the user's option, under the control of DEBUG1, DEBUG2, DEBUG3, and DEBUG4 as described in the input data, the intermediate calculations may be printed out for debugging or checking purposes. Labels used for these printouts are described in Part 2, Section 5 in program nomenclature.

When DEBUG1 = 1., the following variables will be printed out:

I, J, K, DT, DTT, DTMIN, DTTMIN, DTS, DTTS, DTSM, DTTSM, XIP1, XIM1, XIPM1, XI123, XIPM, YJP1, YJM1, YJPM1, YJ123, YJPM, ZKP1, ZKM1, ZKPM1, ZK123, SIGX, SIGX1, SIGX2, SIGX3, SIGX4, SIGY1, SIGY2, SIGY3, SIGY4, SIGZ, SIGZ1, SIGZ2, SIGZ3, SIGZ4, SIGMA1, SIGMA2, SIGMA3, SIGMA4, SIGMA5, SIGMA6, XN, XM, XL, YN, YL, ZN, ZL, DUDX, DUDY.

When DEBUG2 = 1., the following variables are printed out:

I, J, K, IP1, IM1, JP1, JM1, KP1, KM1, DUDT, DVDT, DWDT, DTDY, DIV, DQDX, DQDY, DQDZ, RHS.

DEBUG3 is presently unused and is reserved for future applications.

For DEBUG4 = 1., the following variables are printed out :

I,J,K,ZK123,ZKP1,ZKM1,YJ123,YJP1,YJM1,XI123,XIP1,XIM1,X,X1,X2,X3,X4,
X5,X6,X7,Y1,Y2,LAMB2,OMEGA,ITN

3) Variables Defining the Problem Solution

At user's option, under the control of ENPRNT described in the input data, the solution variables can be printed out at any integration time increment or multiple thereof. These variables are printed in two columns, each column containing the variables:

I,J,K,U,V,W,T,P,and H.

4. OPERATING PROCEDURE

To employ the THERMA computer program effectively, the user should first familiarize himself with Part 1, Section 2, particularly with the mathematical formulation. The user should then draw a sketch of the receiving water similar to Fig. 7, and determine the space and time increments to be used in the analysis as indicated in Section 4. The input data is then prepared in accordance with the format and description given in Section 2 of this part. These input data are then punched on cards and assembled with the THERMA program deck and the necessary control cards. The control cards vary from one computer organization to the other. A sample of the deck assembly for the UNIVAC SERIES 70 is given below:

```

/LOGONbuserid,acct.no.,bTIME=in second

/PARAMb LIST=YES,MAP=YES,DEBUG=YES

/EXECbBGFOR

        PROGRAM Name

        (Fortran Source Statement)

        END

/FILE THERMA.RESTART1,LINK=DSFT24,FCBTYPE=SAM1
/FILE THERMA.RESTART2,LINK=DSFT25,FCBTYPE=SAM2

/EXECb *

        (Data)

/LOGOFF

```

¹RESTART 1 designates the input offline storage

²RESTART 2 designates the output offline storage

The computer program is then run for several integration steps and the user should then verify the following points:

- 1) Correctness of input data. The input data printed out should be verified for possible error in punching.
- 2) Correctness of integration results. The integration results should be verified for consistency and the appropriateness of the initial conditions.

The program could then be run for maximum allowable computer time, at the end of which the results may be written on disc. This off-line storage may be used as input data in a later run to continue the computations. This procedure may be repeated until the results have reached steady state. After the program has been fully run, the user should verify the convergence of the solution by halving the time step Δt and the space increments Δx , Δy , and Δz . The program is then resubmitted and results are checked to verify the agreement between the two solutions. If the solutions agree, the convergence in time and space has been established. If the solutions do not agree, it would then be necessary to choose a smaller time step or further refine the mesh, and perform the above operations again until convergence is established.

5. PROGRAM NOMENCLATURE

To assist the user in the revision of the THERMA program, a list of some key variables with their corresponding analysis notations and definitions is presented in this section.

<u>Program</u>	<u>Analysis</u>	
<u>Notation</u>	<u>Notation</u>	<u>DESCRIPTION</u>
CFX	C_{fx}	Skin friction in x-direction
CFY	C_{fy}	Skin friction in y-direction
CP	C_p	Specific heat of water
DH	D_h	Horizontal eddy diffusivity of heat
DV	D_v	Vertical eddy diffusivity of heat
DO	D_0	Half width of thermal discharge channel
DT	Δt	Time increment
DELX(I)	Δx_j	Space increment in x-direction
DELY(J)	Δy_j	Space increment in y-direction
DELZ(K)	Δz_k	Space increment in z-direction
DEBUG1	-	Control for debugging of MAIN program
DEBUG2	-	Control for debugging of subroutine DERIV
DEBUG3	-	Unused
DEBUG4	-	Control for debugging of subroutine PRESS

<u>Program</u>	<u>Analysis</u>	
<u>Notation</u>	<u>Notation</u>	<u>Description</u>
DUDT	$\partial u / \partial t$	Time derivative of u component of velocity
DVDT	$\partial v / \partial t$	Time derivative of v component of velocity
DWDT	$\partial w / \partial t$	Time derivative of w component of velocity
DTDT	$\partial T / \partial t$	Time derivative of temperature
DHDT	$\partial H / \partial t$	Time derivative of water-level
E	E	Equilibrium temperature of water body
ENRUN	-	Run designation number
EPS	-	Pressure convergence error
FO	F_0	Froude number
G	g	Gravitational acceleration
H	H	Water level
IMAX	i_{\max}	Maximum number of elements in x-direction
IP	-	Current number of iterations
IPRINT	-	Printout control; used in conjunction with NPRINT and IP to determine when printout occurs
ITMAX	-	Maximum number of iterations for the convergence of pressure
ITN	-	Current iteration number used to check against ITMAX

<u>Program</u>	<u>Analysis</u>	
<u>Notation</u>	<u>Notation</u>	<u>Description</u>
JMAX	j_{\max}	Maximum number of elements in y-direction
KMAX	k_{\max}	Maximum number of elements in z-direction
LAMB2	λ^2	Norm ratio
NITER	-	Number of integration steps
NPRNT	-	Frequency of printout
OMEGA	ω	Relaxation factor
P	p	Pressure
RO	R_0	Reynolds number
RHO	ρ	Density of water
RHOA	ρ_a	Density of air
RHOZ	ρ_0	Density of thermal discharge
RHOTBL	-	Density table
RHS(I,J,K)	-	Defined by Eq. (66b)
SO	S_0	Stanton number
SIGMA1(I)	a_x	Weighting factor for first derivative in flow field defined by Eq. (67)
SIGMA2(I)	b_x	Weighting factor for first derivative in flow field defined by Eq. (67)
SIGMA3(J)	a_y	Weighting factor for first derivative in flow field defined by Eq.

<u>Program</u>	<u>Analysis</u>	<u>Description</u>
<u>Notation</u>	<u>Notation</u>	
		(67)
SIGMA4(J)	b_y	Weighting factor for first derivative in flow field defined by Eq. (67)
SIGMA5(K)	a_z	Weighting factor for first derivative in flow field defined by Eq. (67)
SIGMA6(K)	b_z	Weighting factor for first derivative in flow field defined by Eq. (67)
SIGX	σ	Ratio of space increments in x-direction defined by Eq. (C-12)
SIGX1	$2\sigma + 1$	Coefficient of q_1 in Eq. (C-13)
SIGX2	$(1+\sigma)^2$	Coefficient of q_2 in Eq. (C-13)
SIGX3	σ^2	Coefficient of q_3 in Eq. (C-13)
SIGX4	$\frac{1}{2}\Delta n_{12}(1+\sigma)$	Coefficient of Eq. (C-13)
SIGY	-	Analogous to SIGX in y-direction
SIGY1	-	Analogous to SIGX1 in y-direction
SIGY2	-	Analogous to SIGX2 in y-direction
SIGY3	-	Analogous to SIGX3 in y-direction
SIGY4	-	Analogous to SIGX4 in y-direction
SIGPY	σ'	Analogous to $1/\sigma$, ratio of spatial increments in y-direction defined by Eq. (C-27)

<u>Program</u>	<u>Analysis</u>	
<u>Notation</u>	<u>Notation</u>	<u>Description</u>
SIGPY1	$1+2 \sigma'$	Coefficient of q_3 in Eq. (C-28)
SIGPY2	$(1+\sigma')^2$	Coefficient of q_2 in Eq. (C-28)
SIGPY3	σ'^2	Coefficient of q_1 in Eq. (C-28)
SIGPY4	$\frac{1}{2} \Delta n_{23} (1+\sigma')$	Coefficient of Eq. (C-28)
SIGZ	-	Analogous to SIGX in z-direction
SIGZ1	-	Analogous to SIGX1 in z-direction
SIGZ2	-	Analogous to SIGX2 in z-direction
SIGZ3	-	Analogous to SIGX3 in z-direction
SIGZ4	-	Analogous to SIGX4 in z-direction
T	T	Temperature
TO	T_0	Thermal discharge temperature
TIME	t	Time
TTBL	-	Temperature table
U	u	Velocity in x-direction
UO	u_0	Velocity of thermal discharge
V	v	Velocity in y-direction
W	w	Velocity in z-direction
WINDX	\dot{W}_x	Wind velocity in x-direction
WINDY	\dot{W}_y	Wind velocity in y-direction
XIMP1(I)	c_x	A quantity defined by Eq. (67)
XNUZ	ν_0	Kinematic viscosity of thermal discharge
XNUH	ν_h	Horizontal eddy viscosity

<u>Program</u>	<u>Analysis</u>	
<u>Notation</u>	<u>Notation</u>	<u>Description</u>
XNUV	v_v	Vertical eddy viscosity
YJPM1(J)	c_y	A quantity defined by Eq. (67)
ZKPM1(K)	c_z	A quantity defined by Eq. (67)

6. PROGRAM LISTING AND SAMPLE RUN

In this section, the FORTRAN program listing of the THERMA computer program is presented. This listing is followed by a sample run for non-buoyant jet discussed in Part 1, Section 5.3. To reduce the bulk of computer printout, the output data are supplied for iterations number 0, 1 and 50 corresponding to times 0, 0.2 and 10 seconds respectively.

1	PROGRAM THERMA			
2 C	CARD NO.	SYMBOL	DESCRIPTION	
3 C	1-1	ENRUN	RUN DESIGNATION NUMBER	
4 C	2-1	TBEG	STARTING TIME	SEC
5 C	2-2	TEND	END OF TIME	SEC
6 C	2-3	ENITER	NUMBER OF INTEGRATION STEPS	
7 C	2-4	ENPRINT	FREQUENCY OF PRINTOUT	
8 C	3-1	UI	INITIAL VELOCITY COMPONENT IN X-DIRECTION	FT/SEC
9 C				
10 C	3-2	VI	INITIAL VELOCITY COMPONENT IN Y-DIRECTION	FT/SEC
11 C				
12 C	3-3	TI	INITIAL TEMPERATURE OF LAKE AND THERMAL DISCHARGE VELOCITY	FT/SEC
13 C	4-1	UO	THERMAL DISCHARGE VELOCITY	FT/SEC
14 C	4-2	ANGLE	THERMAL DISCHARGE ANGLE	RADIANS
15 C	4-3	TO	THERMA DISCHARGE TEMPERATURE	DEGREE F
16 C	4-4	DO	HYDRAULIC DIAMETER OF THERMAL DISCHARGE CHANNEL	
17 C				
18 C	4-5	XNUZ	KINEMATIC VISCOSITY OF THERMAL DISCHARGE	FT2/SEC
19 C				
20 C	5-1	WINDX	X-COMPONENT OF WIND VELOCITY	FT/SEC
21 C	5-2	WINDY	Y-COMPONENT OF WIND VELOCITY	FT/SEC
22 C	5-3	RHOA	AIR DENSITY	LBM/FT3
23 C	5-4	E	EQUILIBRIUM TEMPERATURE OF WATER	DEGREE F
24 C	5-5	XK	HEAT EXCHANGE COEFFICIENT	BTU/FT2-DAY-F
25 C	5-6	CFX	SKIN COEFFICIENT IN X-DIRECTION	
26 C	5-7	CFY	SKIN COEFFICIENT IN Y-DIRECTION	
27 C	6-1	DH	HORIZONTAL EDDY DIFFUSIVITY OF HEAT	FT2/SEC
28 C	6-2	DV	VERTICAL EDDY DIFFUSIVITY OF HEAT	FT2/SEC
29 C	6-3	XNUH	HORIZONTAL EDDY VISCOSITY	FT2/SEC
30 C	6-4	XNUV	VERTICAL EDDY VISCOSITY	FT2/SEC
31 C	6-5	CP	SPECIFIC HEAT OF WATER	BTU/LBM-F
32 C	7-1	EIMAX	NUMBER OF CELLS IN X-DIRECTION	
33 C	7-2	EJMAX	NUMBER OF CELLS IN Y-DIRECTION	
34 C	7-3	EKMAX	NUMBER OF CELLS IN Z-DIRECTION	
35 C	8-1	DEBUG1	CONTROL FOR DEBUGGING OF MAIN PROGRAM	
36 C	8-2	DEBUG2	CONTROL FOR DEBUGGING OF SUBROUTINE DERIV	
37 C	8-3	DEBUG3	UNUSED	
38 C	8-4	DEBUG4	CONTROL FOR DEBUGGING OF SUBROUTINE PRESS	
39 C	8-5	EIRUN	RESTART SWITCH	
40 C	8-6	EIEND	RESTART CONTROLLER	
41 C	9-1	EIM	NUMBER OF THERMAL DISCHARGE CELLS IN VERTICAL-DIRECTION	
42 C				
43 C	9-2	EML	STARTING LOCATION OF THERMAL DISCHARGE	
44 C	9-3	EMK	END LOCATION OF THERMAL DISCHARGE	
45 C	9-4	SLIP	SLIP CONDITION THERMAL DISCHARGE	
46 C	9-5	HYD	HYDROSTATIC PRESSURE CONTROLLER	
47 C	9-6	HEAT	TEMPERATURE CONTROLLER	
48 C	9-7	XIIP	OFF LINE STORAGE CONTROLLER	
49 C				
50 C				

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51 C	SUBSCRIPTED VARIABLES				
52 C	VARIABLE NAME	DESCRIPTION			
53 C	DELX(I)	DIMENSION OF ELEMENT I IN X-DIRECTION			FT
54 C	DELY(J)	DIMENSION OF ELEMENT J IN Y-DIRECTION			FT
55 C	DELZ(K)	DIMENSION OF ELEMENT K IN Z-DIRECTION			FT
56 C	V(I,J,K)	=Y-COMPONENT OF VELOCITY			
57 C	W(I,J,K)	=Z-COMPONENT OF VELOCITY			
58 C	T(I,J,K)	=TEMPERATURE			
59 C	H(I,J)	= WATER LEVEL HEIGHT			
60 C	P(I,J,K)	=PRESSURE			
61 C	RHO(I,J,K)	=DENSITY			
62 C	IMAX	=NUMBER OF CELLS IN X-DIRECTION			
63 C	JMAX	=NUMBER OF CELLS IN Y-DIRECTION			
64 C	KMAX	=NUMBER OF CELLS IN Z-DIRECTION			
65 C	X	=COEFFICIENT OF P(I,J,K)			
66 C	X1	=COEFFICIENT OF P(I+1,J,K)			
67 C	X2	=COEFFICIENT OF P(I-1,J,K)			
68 C	X3	=COEFFICIENT OF P(I,J+1,K)			
69 C	X4	=COEFFICIENT OF P(I,J-1,K)			
70 C	X5	=COEFFICIENT OF P(I,J,K+1)			
71 C	X6	=COEFFICIENT OF P(I,J,K-1)			
72 C	X7	=COEFFICIENT OF RHS(I,J,K)			
73 C		COMMON P(20,14,10), DELX(20), DELY(14), DELZ(10), RHO(20,14,10),			
74 C		1U(20,14,10), V(20,14,10), W(20,14,10), T(20,14,10),			
75		2DUDT(20,14,10), DVDI(20,14,10), DHDT(20,14,10), DTDI(20,14,10),			
76		3XNUH, XNUV, RO, FO, SO, DH, DV, ENRUN, TBEG, TEND, IP, IJ, CFY, CFZ,			
77		4WINDX, WINDY, E, RHOA, IMAX, JMAX, KMAX, OMEGA, IM, MH, ML, MK,			
78		5DEBUG1, DEBUG2, DEBUG3, DEBUG4, DEBUG5, XNUR, ZNUR, XDHR, XDVR, G, DT,			
79		6NITER, NPRINT, I1, I2, J1, J2, K1, K2, TIME, IPRINT, RFS(20,14,10),			
80		7XIP1(20), XIM1(20), XIPM1(20), XI123(20), YJP1(14), XIP2(20),			
81		8YJM1(14), YJPM1(14), YJ123(14), H(20,14), DZ, YJP2(14),			
82		9ZKP1(10), ZKP2(10), ZKX1(10), ZKM2(10), ZKPM1(10), ZK123(10)			
83		A, I1, JJ, KK, I3, J3, K3, F01, I1T, B, EPS, ITNMAX, SIGMA1(20), SIGMA2(20),			
84		BSIGMA3(14), SIGMA4(14), SIGMA5(10), SIGMA6(10), DHDT(20,14)			
85		COMMON SIGX1, SIGX2, SIGX3, SIGX4, SIGY1, SIGY2, SIGY3, SIGY4, SIGPY1,			
86		1SIGPY2, SIGPY3, SIGPY4, SIGZ1, SIGZ2, SIGZ3, SIGZ4			
87		COMMON/TABLE/TTBL, RHO(TBL)			
88		DIMENSION TTBL(26), RHO(TBL(26)			
89		DIMENSION XM(20), YM(14), ZM(14)			
90		DIMENSION XN(20), YN(14), ZN(10), XL(20), YL(14), ZL(10), CLEVEL(20,14)			
91		FORMAT(E10.3)			
92		FORMAT(2E10.3)			
93		FORMAT(3E10.3)			
94		FORMAT(4E10.3)			
95		FORMAT(5E10.3)			
96		FORMAT(6E10.3)			
97		FORMAT(7E10.3)			
98		FORMAT(8E10.3)			
99		FORMAT(9E10.3)			
100		FORMAT(10E10.3)			

```

101 604 FORMAT(1H0,10X,'I',4X,'XN',15X,'XM')
102 608 FORMAT(1H0,10X,3(I3,2X,2(E11.4)))
103 609 FORMAT(1H0,10X,'J',4X,'YN(J)',15X,'YM(J)')
104 610 FORMAT(1H0,10X,'K',4X,'ZN(K)',15X,'ZM(K)')
105 614 FORMAT(1H0,10X,3(I2,1X),2(E11.4,2X))
106 615 FORMAT(1H0,10X,'I',2X,'J',2X,'K',2X,'DT',5X,'DTT')
107 616 FORMAT(1H ,10X,3(I2,2X),E11.4,12X,3(I2,2X),E11.4,/)
108 621 FORMAT(1H0,10X,3(I2,2X),'STABILITY CONDITION FAILS IN X-DIRECTION
109 1')
110 622 FORMAT(1H0,10X,3(I2,2X),'STABILITY CONDITION FAILS IN Y-DIRECTION'
111 1)
112 623 FORMAT(1H0,10X,3(I2,2X),'STABILITY CONDITION FAILS IN Z-DIRECTION
113 1')
114 627 FORMAT(1H0,10X,6(I2,2X,E11.4,2X)/)
115 629 FORMAT(1H0,10X,'I',4X,'XIMU(I)')
116 630 FORMAT(1H0,10X,'J',4X,'YJM1(J)')
117 631 FORMAT(1H0,10X,'K',4X,'ZKM1(K)')
118 640 FORMAT(1H0,10X,'I',2X,'XL(I)')
119 641 FORMAT(1H0,10X,5(I2,1X,E11.4))
120 642 FORMAT(1H0,10X,'J',2X,'YL(J)')
121 643 FORMAT(1H0,10X,'K',2X,'ZL(K)')
122 644 FORMAT(1H0,10X,'I',5X,'SIGMA1',5X,'SIGMA2')
123 645 FORMAT(1H0,10X,I2,5X,2(E11.4,2X))
124 646 FORMAT(1H0,10X,'J',5X,'SIGMA3',5X,'SIGMA4')
125 647 FORMAT(1H0,10X,'K',5X,'SIGMA5',5X,'SIGMA6')
126 705 FORMAT(1H0,11X,'I',3X,'J',3X,'K',3X,'SUGGESTED DT',10X,
127 X'I',3X,'J',3X,'K',3X,'SUGGESTED DTT',/)
128 709 FORMAT(1H0,10X,'I',2X,'J',2X,'KMAX',2X,'DTS',5X,'DTTS')
129 710 FORMAT(1H0,10X,3(I2,1X),2(E11.4,2X))
130 714 FORMAT(1H0,11X,'I',3X,'J',3X,'KMAX',3X,'SUGGESTED DTS',10X,
131 X'I',3X,'J',3X,'KMAX',3X,'SUGGESTED DTTS',/)
132 715 FORMAT(1H0,10X,3(I2,2X),E11.4,12X,3(I2,2X),E11.4,/)
133 720 FORMAT(1H0,10X,'DT BECAME TOO SMALL AT',2X,'I=',I2,2X,'J=',I2,
134 12X,'HLEVEL=',E14.7,2X,'CLEVEL(I,J)=',E14.7)
135 1025 FORMAT(1H1)
136 1026 FORMAT(1H0,10X,'ENRUN =',E11.4)
137 1027 FORMAT(1H0,10X,'TBEG =',E11.4,5X,'TEND =',E11.4,5X,'ENITER='
138 1,E11.4,5X,'NPRNT=',E11.4)
139 1028 FORMAT(1H0,10X,'UI =',E11.4,5X,'VI =',E11.4,5X,
140 1 'TI =',E11.4)
141 1029 FORMAT(1H0,10X,'WINDX =',E11.4,5X,'WINDY =',E11.4,5X,'RHOA =',
142 1,E11.4,5X,'E =',E11.4,9X,'XK =',E11.4,/,/,11X,'CFX =',E11.4,5X,
143 2'CFY =',E11.4)
144 1030 FORMAT(1H0,10X,'DH =',E11.4,5X,'DV =',E11.4,5X,'XNUH =',
145 1,E11.4,5X,'XNUV =',E11.4,5X,'CP=',E11.4)
146 1031 FORMAT(1H0,10X,'EIMAX =',E11.4,5X,'EJMAX =',E11.4,5X,'EKMAX =',
147 1,E11.4)
148 1032 FORMAT(1H0,10X,'RO =',E11.4,5X,'FO =',E11.4,5X,'SO =',
149 1,E11.4)
150 1033 FORMAT(1H0,10X,'DEBUG1=',E11.4,5X,'DEBUG2=',E11.4,5X,'4EBU73=',

```

```

151      1E11.4,5X,'DEBUG4=',E11.4,/,11X,'EIRUN=',E11.4,5X,'EIEND=',E11.4)
152 1034 FORMAT(1H0,10X,'IMAX  =',I5  ,5X,'JMAX  =',I5  ,5X,'KMAX  ='
153      1,I5  )
154 1035 FORMAT(1H0,10X,'I1   =',I5  ,5X,'J1   =',I5  ,5X,'K1   ='
155      1,I5  )
156 1036 FORMAT(1H0,10X,'I2   =',I5  ,5X,'J2   =',I5  ,5X,'K2   ='
157      1,I5  )
158 1037 FORMAT(1H0,10X,'G     =',E11.4,5X,'DT    =',E11.4,5X,'E=',E14.7)
159 1038 FORMAT(1H0,10X,'I',3X,'DELX(I)')
160 1039 FORMAT(2(1H0,10X,4(I2,2X,E11.4,5X)/))
161 1040 FORMAT(1H0,10X,'J',3X,'DELY(J)')
162 1041 FORMAT(1H0,10X,'K',3X,'DELZ(K)')
163 1042 FORMAT(1H0,10X,'XNCR  =',E11.4,5X,'ZNUR   =',E11.4,5X,'XDHR  ='
164      1,E11.4,5X,'XDVR   =',E11.4)
165 1046 FORMAT(1H0,10X,'I3   =',I5  ,5X,'J3   =',I5  ,5X,'K3   ='
166      1,I5  )
167 1048 FORMAT(1H0,'K=',I2,1X,'ZKP1(K)=' ,E11.4,1X,
168      1'ZKM1(K)=' ,E11.4,1X,'ZKPM1(K)=' ,E11.4,1X,
169      2'ZK123(K)=' ,E11.4,/,1X,'J=' ,I2,1X,'YJP1(J)=' ,E11.4,5X,'YJM1(J)=' ,
170      3E11.4,5X,'YJPM1(J)=' ,E11.4,5X,'YJ123(J)=' ,E11.4,/,1X,'I=' ,I2,
171      51X,'XIP1(I)=' ,E11.4,1X,'XIM1(I)=' ,E11.4,1X,'XIPM1(I)=' ,E11.4,1X,
172      6'XI123(I)=' ,E11.4)
173 1054 FORMAT(1H ,2X,'I=' ,I3,2X,'J=' ,I3,2X,'K=' ,I3,'DUDX=' ,E14.7,2X,
174      1'DVDY=' ,E14.7,2X,'CS(K)=' ,E14.7,2X,'DS(I,J,K)=' ,E14.7)
175 1058 FORMAT(1H0,10X,'UO=' ,E11.4,9X,'ANGLE=' ,E11.4,6X,'T0=' ,E11.4,
176      19X,'DO=' ,E11.4,9X,'XNUZ=' ,E11.4)
177 1062 FORMAT(1H0,20X,20('*'),'NON-DIMENSIONAL INPUT VALUES ',20('*'))
178 1063 FORMAT(1H0,20X,20('*'),'END OF NON-DIMENSIONAL INPUT VALUES',
179      12C('*'))
180 1064 FORMAT(1H ,10X,2(E14.7,2X))
181 1065 FORMAT(1H0,15X,'TTBL',10X,'RHOTBL')
182 1066 FORMAT(1H0,10X,'RHCZ=' ,E15.7)
183 1067 FORMAT(1H0,10X,'TBEG  =',E11.4,5X,'TEND  =',E11.4,5X,' NITER='
184      1,I6  ,5X,'NPRINT=' ,I6)
185 1068 FORMAT(1H0,10X,'DH    =',E11.4,5X,'DV    =',E11.4,5X,'XNUH  ='
186      1,E11.4,5X,'XNUV  =',E11.4)
187 1069 FORMAT(1H0,10X,'WINDX =',E11.4,5X,'WINDY =',E11.4,5X,'RHOA  ='
188      1,E11.4)
189 1071 FORMAT(1H0,10X,'EIM=' ,E11.4,5X,'EML=' ,E11.4,5X,'EMK=' ,E11.4,
190      15X,'SLIP=' ,E11.4,5X,'HYD=' ,E11.4,/,11X,'HEAT=' ,E11.4,5X,'XILP=' ,
191      2E11.4)
192 1072 FORMAT(1H0,10X,'Z  =' ,E11.4)
193 1100 FORMAT(1H0,5(E14.7,4X),/,2X,5(E14.7,4X))
194 1110 FORMAT(1H0,5(E14.7,4X))
195 2005 FORMAT(1H ,3(I3,2X),7(E14.7,2X))
196 2006 FORMAT(1H0,5X,'TIME=' ,E14.7,5X,'DT=' ,E14.7,5X,'IP=' ,I5,5X,'KMAX=' ,
197      1I5,5X,'JMAX=' ,I5,5X,'IMAX=' ,I5,5X,'IPRINT=' ,I5)
198 2007 FORMAT(1H0,1X,'I',2X,'J',2X,'K',9X,'U',12X,'V',12X,'W',12X,'T',
199      112X,'P',12X,'RHO',12X,'H')
200 C      READ IN NON SUBSCRIPTED VARIABLES

```

```

201      READ(5,1001) ENRUN
202      WRITE(6,1026) ENRUN
203      READ(5,1004) TBEG,TEND,ENITER,ENPRNT
204      WRITE(6,1027) TBEG,TEND,ENITER,ENPRNT
205      READ(5,1003) UI,VI,TI
206      WRITE(6,1028) UI,VI,TI
207      READ(5,1005) UO,ANGLE,TO,DO,XNUZ
208      WRITE(6,1058) UO,ANGLE,TO,DO,XNUZ
209      READ(5,1007) WINDX,WINDY,RHOA,E,XK,CFX,CFY
210      WRITE(6,1029) WINDX,WINDY,RHOA,E,XK,CFX,CFY
211      READ(5,1005) DH,DV,XNUH,XNUV,CP
212      WRITE(6,1030) DH,DV,XNUH,XNUV,CP
213      READ(5,1003) EIMAX,EJMAX,EKMAX
214      WRITE(6,1031) EIMAX,EJMAX,EKMAX
215      READ (5,1006)  DEBUG1,DEBUG2,DEBUG3,DEBUG4,EIRUN,EIEND
216      WRITE(6,1033) DEBUG1,DEBUG2,DEBUG3,DEBUG4,EIRUN,EIEND
217      READ(5,1007) EIM,EML,EMK,SLIP,HYD,HEAT,XIIP
218      WRITE(6,1071) EIM,EML,EMK,SLIP,HYD,HEAT,XIIP
219      IMAX=EIMAX
220      JMAX=EJMAX
221      KMAX=EKMAX
222      NITER=ENITER
223      NPRINT=ENPRNT
224      IIP=XIIP
225 C    READ IN SUBSCRIPTED VARIABLES
226      READ(5,1007) (DELX(I),I=1,IMAX)
227      WRITE(6,1038)
228      WRITE(6,1039) (I,DELX(I),I=1,IMAX)
229      READ(5,1007) (DELY(J),J=1,JMAX)
230      WRITE(6,1040)
231      WRITE(6,1039) (J,DELY(J),J=1,JMAX)
232      READ(5,1007) (DELZ(K),K=1,KMAX)
233      WRITE(6,1041)
234      WRITE(6,1039) (K,DELZ(K),K=1,KMAX)
235      WRITE(6,1065)
236      DC 505 L=1,26
237      WRITE(6,1064) TTBL(L),RHOTBL(L)
238 505  CONTINUE
239      IM=EIM
240      ML=EML
241      MK=EMK
242      MH=MK+1
243      I1=IMAX-1
244      I2=IMAX-2
245      I3=IMAX-3
246      J1=JMAX-1
247      J2=JMAX-2
248      J3=JMAX-3
249      K1=KMAX-1
250      K2=KMAX-2

```

```
251      K3=KMAX-3
252      ITNMAX=500
253      EPS=.00001
254      TIME      =0.
255      IP=0
256      IPRINT=0
257      IJ=0
258      G=32.2
259      DTSM=1000.
260      DTTSM=1000.
261      DTMIN=1000.
262      DTTMIN=1000.
263      WRITE(6,1034) IMAX,JMAX,KMAX
264      WRITE(6,1035) I1,J1,K1
265      WRITE(6,1036) I2,J2,K2
266      WRITE(6,1046) I3,J3,K3
267      DO 624 I=2,I1
268      XIM1(I)=DELX(I-1)+DELX(I)
269      XN(I)=4.*XNUH/XIM1(I)
270      XM(I)=4.*DH/XIM1(I)
271 624  CONTINUE
272      DO 625 J=2,J1
273      YJM1(J)=DELY(J-1)+DELY(J)
274      YN(J)=4.*XNUH/YJM1(J)
275      YM(J)=4.*DH/YJM1(J)
276 625  CONTINUE
277      DO 626 K=2,K1
278      ZKM1(K)=DELZ(K-1)+DELZ(K)
279      ZN(K)=4.*XNUV/ZKM1(K)
280      ZM(K)=4.*DV/ZKM1(K)
281 626  CONTINUE
282 C    CALCULATE DENSITY OF THERMAL DISCHARGE
283      RHOZ=SI(TTBL,RHOTBL,T0,26)
284      WRITE(6,1066) RHOZ
285      IF(DEBUG1.EQ.0.) GO TO 51
286      WRITE(6,629)
287      WRITE(6,627) (I,XIM1(I),I=2,I1)
288      WRITE(6,630)
289      WRITE(6,627) (J,YJM1(J),J=2,J1)
290      WRITE(6,631)
291      WRITE(6,627) (K,ZKM1(K),K=2,K1)
292      WRITE(6,604)
293      WRITE(6,608)(I,XN(I),XM(I),I=2,I1)
294      WRITE(6,609)
295      WRITE(6,608)(J,YN(J),YM(J),J=2,J1)
296      WRITE(6,610)
297      WRITE(6,608)(K,ZN(K),ZM(K),K=2,K1)
298 51  CONTINUE
299 C    INITIALIZE WATER LEVEL HEIGHT
300      DO 706 I=1,IMAX
```



```

301      DO 706 J=1,JMAX
302      H(I,J)=DELZ(KMAX)/2.
303 706  CONTINUE
304 C    CALCULATE DT
305      IF(DEBUG1.EQ.0.) GO TO 655
306      WRITE(6,615)
307 655  CONTINUE
308      DO 613 K=2,K1
309      DO 613 J=2,J1
310      DO 613 I=2,I1
311      DT=1/(2.*(XM(I)/XIM1(I)+YN(J)/YJM1(J)+ZN(K)/ZKM1(K)))
312      DTT=1/(2.*(XM(I)/XIM1(I)+YM(J)/YJM1(J)+ZM(K)/ZKM1(K)))
313      IF(DEBUG1.EQ.0.) GO TO 656
314      WRITE(6,614) I,J,K,DT,DTT
315 656  CONTINUE
316      IF(DT.LT.DTMIN) GO TO 701
317      GO TO 702
318 701  DTMIN=DT
319      IIM=I
320      JJM=J
321      KKM=K
322 702  IF(DTT.LT.DTTMIN) GO TO 703
323      GO TO 613
324 703  DTTMIN=DTT
325      IIT=I
326      JJT=J
327      KKT=K
328 613  CONTINUE
329      WRITE(6,705)
330 C*****
331      WRITE(6,616) IIM,JJM,KKM,DTMIN,IIT,JJT,KKT,DTTMIN
332      IF(DEBUG1.EQ.0.) GO TO 717
333      WRITE(6,709)
334 717  CONTINUE
335      DO 707 I=2,I1
336      DO 707 J=2,J1
337      DTS=1/(2.*(XM(I)/XIM1(I)+YN(J)/YJM1(J)+XNUV/(H(I,J)*(H(I,J)+
338 1DELZ(K1))))
339      DTTS=1/(2.*(XM(I)/XIM1(I)+YM(J)/YJM1(J)+DV/(H(I,J)*(H(I,J)+
340 1DELZ(K1))))
341      IF(DEBUG1.EQ.0.) GO TO 708
342      WRITE(6,710) I,J,KMAX,DTS,DTTS
343 708  CONTINUE
344      IF(DTS.LT.DTSM) GO TO 711
345      GO TO 712
346 711  DTSM=DTS
347      IIM=I
348      JJM=J
349 712  IF(DTTS.LT.DTTSM) GO TO 713
350      GO TO 707

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351 713 DTTSM=DTTS
352 IIT=I
353 JJT=J
354 707 CONTINUE
355 WRITE(6,714)
356 WRITE(6,715) IIM,JJM,KMAX,DTSM,IIT,JJT,KMAX,DTTSM
357 DT=DTMIN
358 C
359 C*****BEGIN NONDIMENSIONALIZATION
360 C
361 WRITE(6,1062)
362 TBEG=TBEG*U0/DO
363 TEND=TEND*U0/DO
364 DT=(TEND-TBEG)/ENITER
365 UI=UI/U0
366 VI=VI/U0
367 TI=TI/T0
368 WINDX=WINDX/U0
369 WINDY=WINDY/U0
370 RHOA=RHOA/RHOZ
371 TD=TD/T0
372 DH=DH/XNUZ
373 DV=DV/XNUZ
374 XNUH=XNUH/XNUZ
375 XNUV=XNUV/XNUZ
376 DO 13 I=1,IMAX
377 DO 13 J=1,JMAX
378 H(I,J)=H(I,J)/DO
379 13 CONTINUE
380 DO 504 LL=1,26
381 TTBL(LL)=TTBL(LL)/T0
382 RHOTBL(LL)=RHOTBL(LL)/RHOZ
383 504 CONTINUE
384 DO 601 I=1,IMAX
385 DELX(I)=DELX(I)/DO
386 601 CONTINUE
387 DO 602 J=1,JMAX
388 DELY(J)=DELY(J)/DO
389 602 CONTINUE
390 DO 603 K=1,KMAX
391 DELZ(K)=DELZ(K)/DO
392 603 CONTINUE
393 RO=DO*U0/XNUZ
394 FO=U0/((G*DO)**.5)
395 E=E/T0
396 XK=XK/(3600*24)
397 SO=XK/(RHOZ*CP*U0)
398 U0=U0/U0
399 T0=T0/T0
400 XNUZ=XNUZ/XNUZ

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401 WRITE(6,1026) ENRUK
402 WRITE(6,1067) TBEG,TEND,NITER,NPRINT
403 WRITE(6,1028) UI,VI,TI
404 WRITE(6,1058) UO,ANGLE,TO,DO,XNUZ
405 WRITE(6,1069) WINDX,WINDY,RHOA
406 WRITE(6,1068) DH,DV,XNUH,XNUV
407 WRITE(6,1037) G,DT,E
408 WRITE(6,1038)
409 WRITE(6,1039) (I,DELX(I),I=1,IMAX)
410 WRITE(6,1040)
411 WRITE(6,1039) (J,DELY(J),J=1,JMAX)
412 WRITE(6,1041)
413 WRITE(6,1039) (K,DELZ(K),K=1,KMAX)
414 WRITE(6,1032) RO,FO,SO
415 WRITE(6,1065)
416 DO 507 L=1,26
417 WRITE(6,1064) TTBL(L),RHOTBL(L)
418 507 CONTINUE
419 WRITE(6,1063)
420 C
421 C*****END NONDIMENSIONALIZATION *****
422 C
423 DO 33 K=2,K1
424 ZKP1(K)=DELZ(K)+DELZ(K-1)
425 ZKM1(K) =DELZ(K)-DELZ(K-1)
426 ZKPM1(K) =ZKP1(K)+ZKM1(K)
427 ZK123(K)=ZKP1(K)*ZKM1(K)*ZKPM1(K)
428 33 CONTINUE
429 DO 34 J=2,J1
430 YJP1(J) =DELY(J)+DELY(J-1)
431 YJM1(J) =DELY(J)-DELY(J-1)
432 YJPM1(J) =YJP1(J)+YJM1(J)
433 YJ123(J) =YJP1(J)*YJM1(J)*YJPM1(J)
434 34 CONTINUE
435 DO 35 I=2,I1
436 XIP1(I) =DELX(I)+DELX(I-1)
437 XIM1(I) =DELX(I)-DELX(I-1)
438 XIPM1(I) =XIP1(I)+XIM1(I)
439 XI123(I) =XIP1(I)*XIM1(I)*XIPM1(I)
440 35 CONTINUE
441 IF(DEBUG1.EQ.0.) GO TO 18
442 DO 17 J=2,J1
443 DO 17 K=2,K1
444 DO 17 I=2,I1
445 WRITE(6,1048) K,ZKP1(K),ZKM1(K),ZKPM1(K),ZK123(K),
446 1J,YJP1(J),YJM1(J),YJPM1(J),YJ123(J),
447 2I,XIP1(I),XIM1(I),XIPM1(I),XI123(I)
448 17 CONTINUE
449 18 CONTINUE
450

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```
451 C   COEFFICIENT OF VISCOUS TERMS
452     XNUR      =XNUH/RC
453     ZNUR      =XNUV/RC
454     XDHR      =  DH/RC
455     XDVR      =  DV/RC
456     WRITE(6,1042) XNUR,ZNUR,XDHR,XDVR
457     DD=2,*DT
458 26  FORMAT(1H0,3(1X,I2),9(1X,E11.4))
459     F01=F0**2
460 C
461 C   Z=TOTAL HEIGHT OF LAKE
462     Z=0.
463     DO 110 L=1,K1
464     Z=Z-ZKP1(L)/2.
465     IF(DEBUG1.EQ.0.) GO TO 58
466     WRITE(6,1072) Z
467 58  CONTINUE
468 110 CONTINUE
469 55  FORMAT(1H 5X,'Z=',F5.2)
470 C   FIRST DERIVATIVE CORRECTION
471     DO 311 K=2,K1
472     SIGMA5(K)=2.*ZKM1(K)/ZKPM1(K)
473     SIGMA6(K)=2.*ZKP1(K)/ZKPM1(K)
474 311 CONTINUE
475     DO 312 J=2,J1
476     SIGMA3(J)=2.*YJM1(J)/YJPM1(J)
477     SIGMA4(J)=2.*YJP1(J)/YJPM1(J)
478 312 CONTINUE
479     DO 313 I=2,I1
480     SIGMA1(I)=2.*XIM1(I)/XIPM1(I)
481     SIGMA2(I)=2.*XIP1(I)/XIPM1(I)
482 313 CONTINUE
483 C
484     IF(DEBUG1.EQ.0.) GO TO 52
485     WRITE(6,644)
486 C*****
487     DO 657 I=2,I1
488     WRITE(6,645) I,SIGMA1(I),SIGMA2(I)
489 657 CONTINUE
490 C*****
491     WRITE(6,646)
492     DO 658 J=2,J1
493     WRITE(6,645) J,SIGMA3(J),SIGMA4(J)
494 658 CONTINUE
495 C*****
496     WRITE(6,647)
497     DO 659 K=2,K1
498     WRITE(6,645) K,SIGMA5(K),SIGMA6(K)
499 659 CONTINUE
500 C*****
```

```

501 648 CONTINUE
502 52 CONTINUE
503 C*****
504 C CALCULATE COEFFICIENT OF FIRST DERIVATIVE AT BOUNDARY
505 C*****
506 SIGY=(DELY(1)+DELY(2))/(DELY(2)+DELY(3))
507 SIGY1=(1.+2.*SIGY)
508 SIGY2=(1.+SIGY)*(1.+SIGY)
509 SIGY3=SIGY*SIGY
510 SIGY4=(DELY(1)+DELY(2))*(1.+SIGY)/2.
511 SIGPY=(DELY(JMAX)+DELY(J1))/(DELY(J1)+DELY(J2))
512 SIGPY1=(1.+2.*SIGPY)
513 SIGPY2=(1.+SIGPY)*(1.+SIGPY)
514 SIGPY3=SIGPY*SIGPY
515 SIGPY4=(DELY(JMAX)+DELY(J1))*(1.+SIGPY)/2.
516 IF(DEBUG1.EQ.0.) GC TO 53
517 WRITE(6,1100) SIGY,SIGY1,SIGY2,SIGY3,SIGY4,SIGPY,SIGPY1,SIGPY2,
518 2SIGPY3,SIGPY4
519 53 CONTINUE
520 SIGX=(DELX(1)+DELX(2))/(DELX(2)+DELX(3))
521 SIGX2=(1.+SIGX)*(1.+SIGX)
522 SIGX1=(1.+2.*SIGX)
523 SIGX3=SIGX*SIGX
524 SIGX4=(DELX(1)+DELX(2))*(1.+SIGX)/2.
525 IF(DEBUG1.EQ.0.) GC TO 54
526 WRITE(6,1110) SIGX,SIGX1,SIGX2,SIGX3,SIGX4
527 54 CONTINUE
528 SIGZ=(DELZ(1)+DELZ(2))/(DELZ(2)+DELZ(3))
529 SIGZ1=(1.+2.*SIGZ)
530 SIGZ2=(1.+SIGZ)*(1.+SIGZ)
531 SIGZ3=SIGZ*SIGZ
532 SIGZ4=(DELZ(1)+DELZ(2))*(1.+SIGZ)/2.
533 IF(DEBUG1.EQ.0.) GC TO 56
534 WRITE(6,1110) SIGZ,SIGZ1,SIGZ2,SIGZ3,SIGZ4
535 56 CONTINUE
536 C END CALCULATION OF COEFFICIENT OF FIRST DERIVATIVE
537 DMH=AMAX1(XDHR,XNUR)
538 DMV=AMAX1(XDVR,ZNUR)
539 DO 718 I=2,I1
540 DO 718 J=2,J1
541 CLEVEL(I,J)=(1./DT)-8.*DMH/(XIM1(I)*XIM1(I))-8.*DMH/(YJM1(J)*
542 1YJM1(J))
543 718 CONTINUE
544 C*****
545 C CHECK FOR RESTART ENTRY FOR DYNAMIC ANALYSIS
546 C*****
547 C BEGIN INITIALIZATION
548 IF(EIRUN.EQ.1.) GO TO 75
549 C*****
550 DO 5 K=1,KMAX

```

```

551      DO 5 J=1,JMAX
552      DO 5 I=1,IMAX
553      U(I,J,K)=0.
554      V(I,J,K)=0.
555      W(I,J,K)=0.
556      T(I,J,K)=0.
557      RHS(I,J,K)=0.
558      DUDT(I,J,K)=0.
559      DVDT(I,J,K)=0.
560      DWDT(I,J,K)=0.
561      DTDI(I,J,K)=0.
562      DHDT(I,J)=0.
563      5 CONTINUE
564 C*****INITIALIZATION OF HORIZONTAL VELOCITY COMPONENTS U AND V
565      DO 25 K=2,KMAX
566      DO 25 J=2,J1
567      DO 25 I=1,IMAX
568      U(I,J,K) =UI
569      V(I,J,K) =VI
570      25 CONTINUE
571      Z1=Z
572 C*****INITIALIZATION OF TEMPERATURE
573 C*****FLUID-FLUID BOUNDARY ON PRESSURE
574      DO 502 K=1,KMAX
575      DO 506 J=1,JMAX
576      DO 506 I=1,IMAX
577      T(I,J,K) =TI
578      RHO(I,J,K)=SI(TTBL,RHOTBL,T(I,J,K),26)
579      P(I,J,K)=-RHO(I,J,K)*Z
580      506 CONTINUE
581      IF(K.EQ.KMAX) GO TO 502
582      Z=Z+ZKP1(K)/2.
583      IF(K.EQ.K1) Z=-DELZ(KMAX)/4.
584      IF(DEBUG1.EQ.0.) GO TO 57
585      WRITE(6,1072) Z
586      57 CONTINUE
587      502 CONTINUE
588      Z=Z1
589 C      END INITIALIZATION
590 C      SET THE BOUNDARY CONDITIONS
591 C      FIXED BOUNDARY CONDITIONS
592 C*****FIXED BOUNDARY CONDITION ON TEMPERATURE IS BEEN SET DURING INITIALIZATION*
593 C      FIXED VELOCITY BOUNDARY CONDITION
594 C
595      DO 4 K=2,KMAX
596      DO 4 J=2,J1
597      U(1,J,K)=UI
598      4 CONTINUE
599 C*****
600 C      SET VELOCITY OF THERMAL DISCHARGE

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```

601      DO 63 I=ML,MK
602      DO 63 K=IM,KMAX
603      U(I,1,K)=U0*COS(ANGLE)
604      V(I,1,K)=U0*SIN(ANGLE)
605 63   CONTINUE
606      GO TO 79
607 75   CONTINUE
608 C*****READING FROM DISC*****
609      REWIND 24
610      READ(24) TIME,DT,IP,KMAX,JMAX,IMAX,IPRINT
611 C*****DISC OUTPUT*****
612      WRITE(6,2006) TIME,DT,IP,KMAX,JMAX,IMAX,IPRINT
613      WRITE(6,2007)
614      DO 76 K=1,KMAX
615      DO 76 J=1,JMAX
616      DO 76 I=1,IMAX
617      READ(24) U(I,J,K),V(I,J,K),W(I,J,K),T(I,J,K),P(I,J,K),H(I,J),
618      1RHO(I,J,K)
619      WRITE(6,2005) I,J,K,U(I,J,K),V(I,J,K),W(I,J,K),T(I,J,K),P(I,J,K),
620      1RHO(I,J,K),H(I,J)
621 76   CONTINUE
622 C*****END OF DISC OUTPUT*****
623      IPRINT=0
624 79   CONTINUE
625 C
626      DO 635 I=2,I1
627      XN(I)=4.*XNUR/XIM1(I)
628      XM(I)=4.*XDHR/XIM1(I)
629      XL(I)=AMIN1(XN(I),XM(I))
630 635  CONTINUE
631      DO 636 J=2,J1
632      YN(J)=4.*XNUR/YJM1(J)
633      YM(J)=4.*XDHR/YJM1(J)
634      YL(J)=AMIN1(YN(J),YM(J))
635 636  CONTINUE
636      DO 637 K=2,K1
637      ZN(K)=4.*ZNUR/ZKM1(K)
638      ZM(K)=4.*XDVR/ZKM1(K)
639      ZL(K)=AMIN1(ZN(K),ZM(K))
640 637  CONTINUE
641      IF (DEBUG1.EQ.0.) GO TO 654
642      WRITE(6,604)
643      WRITE(6,608)(I,XN(I),XM(I),I=2,I1)
644      WRITE(6,609)
645      WRITE(6,608)(J,YN(J),YM(J),J=2,J1)
646      WRITE(6,610)
647      WRITE(6,608)(K,ZN(K),ZM(K),K=2,K1)
648      WRITE(6,640)
649      WRITE(6,641) (I,XL(I),I=2,I1)
650      WRITE(6,642)

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651      WRITE(6,641) (J,YL(J),J=2,J1)
652      WRITE(6,643)
653      WRITE(6,641) (K,ZL(K),K=2,K1)
654 654  CONTINUE
655      WRITE(6,1025)
656      IF(HEAT.EQ.0.) GO TO 1
657 C     SET BOUNDARY AT THERMAL DISCHARGE INLET
658 C
659 C     TEMPERATURE
660      DO 61 K=IM,KMAX
661      DO 61 I=ML,MK
662      T(I,1,K)=1.
663 61    CONTINUE
664 C
665 C     PRESSURE
666 C
667      Z1=Z
668      DO 43 K=1,KMAX
669      DO 42 I=ML,MK
670      P(I,1,K)=-Z
671 42    CONTINUE
672      Z=Z+ZKP1(K)/2.
673      IF(K.EQ.K1) Z=-DELZ(KMAX)/4.
674 43    CONTINUE
675      Z=Z1
676 C
677 C ***** BEGINNING OF THE DYNAMIC LOOP *****
678 C     VARIABLE BOUNDARY CONDITIONS
679 C
680 1     CONTINUE
681 C     CHECK STABILITY FOR U,V, AND W
682      DO 617 K=2,KMAX
683      DO 617 I=2,I1
684      DO 617 J=2,J1
685      IF(ABS(U(I,J,K)).GT.XL(I)) GO TO 618
686      IF(ABS(V(I,J,K)).GT.YL(J)) GO TO 619
687      IF(K.EQ.KMAX) GO TO 716
688      IF(ABS(W(I,J,K)).GT.ZL(K)) GO TO 620
689 716   CONTINUE
690      GO TO 617
691 618   WRITE(6,621) I,J,K
692      IPRINT=NPRINT
693      CALL OUTP
694      CALL EXIT
695 619   WRITE(6,622) I,J,K
696      IPRINT=NPRINT
697      CALL OUTP
698      CALL EXIT
699 620   WRITE(6,623) I,J,K
700      IPRINT=NPRINT

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701      CALL OUTP
702      CALL EXIT
703 617  CONTINUE
704      DO 719 I=2,I1
705      DO 719 J=2,J1
706      HLEVEL=2.*DMV/(H(I,J)*(H(I,J)+DELZ(K1)))
707      IF(HLEVEL.LT.CLEVEL(I,J)) GO TO 719
708      WRITE(6,720) I,J,HLEVEL,CLEVEL(I,J)
709      IPRINT=NPRINT
710      CALL OUTP
711      CALL EXIT
712 719  CONTINUE
713 C    END STABILITY ANALYSIS
714 C
715 C    OUTGOING BOUNDARY ON VELOCITY
716      DO 70 J=1,JMAX
717      DO 70 K=2,KMAX
718      U(IMAX,J,K)=U(I1,J,K)
719      V(IMAX,J,K)=V(I1,J,K)
720 70   CONTINUE
721 C*****BOTTOM BOUNDARY ON HORIZONTAL VELOCITY COMPONENTS
722      DO 236 I=1,IMAX
723      DO 236 J=1,JMAX
724      V(I,J,1)=0.
725      U(I,J,1)=0.
726 236  CONTINUE
727 C*****SOLID-FLUID BOUNDARY ON HORIZONTAL VELOCITY COMPONENT U
728      IF(SLIP.EQ.0.) GO TO 950
729      DO 6 K=2,KMAX
730      DO 6 I=1,IMAX
731      U(I,JMAX,K)=(SIGPY2*U(I,J1,K)-SIGPY3*U(I,J2,K))/SIGPY1
732 6    CONTINUE
733      DO 135 K=2,KMAX
734      DO 137 I=1,IMAX
735      IF(K.LT.IM) GO TO 134
736      IF( I .LT. MH .AND. I .GE. ML ) GO TO 137
737 134  CONTINUE
738      U(I,1,K)=(SIGY2*U(I,2,K)-SIGY3*U(I,3,K))/SIGY1
739 137  CONTINUE
740 135  CONTINUE
741 950  CONTINUE
742 C*****SOLID-FLUID BOUNDARY ON TEMPERATURE
743      DO 130 K=2,KMAX
744      DO 130 I=1,IMAX
745      T(I,JMAX,K)=(SIGPY2*T(I,J1,K)-SIGPY3*T(I,J2,K))/SIGPY1
746 130  CONTINUE
747      DO 125 K=2,KMAX
748      DO 126 I=1,IMAX
749      IF(K.LT.IM) GO TO 124
750      IF( I .LT. MH .AND. I .GE. ML ) GO TO 126
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751 124 CONTINUE
752 T(I,1,K)=(SIGY2*T(I,2,K)-SIGY3*T(I,3,K))/SIGY1
753 126 CONTINUE
754 125 CONTINUE
755 C****SURFACE BOUNDARY ON VERTICAL VELOCITY COMPONENT IS ZEROED DURING
756 C INITIALIZATION
757 C****SURFACE BOUNDARY ON HORIZONTAL VELOCITY COMPONENTS U AND V
758 C
759 DO 7 I=1,IMAX
760 DO 7 J=1,JMAX
761 T(I,J,1)=(SIGZ2*T(I,J,2)-SIGZ3*T(I,J,3))/SIGZ1
762 7 CONTINUE
763 C
764 C
765 DO 73 I=1,IMAX
766 DO 73 J=1,JMAX
767 T(IMAX,J,K)=T(I1,J,K)
768 73 CONTINUE
769 C****CALCULATION OF DHDT(I,J) FOR THE TOP ELEMENTS *****
770 800 DO 661 I=2,I1
771 IM1=I-1
772 1120 DO 661 J=2,J1
773 JM1=J-1
774 DHDT(I,J)=-2.*(H(I,J)*(U(I,J,KMAX)-U(IM1,J,KMAX))*
775 1*U(I,J,KMAX)*(H(I,J)-H(IM1,J)))/XIM1(I)
776 2-2.*(H(I,J)*(V(I,J,KMAX)-V(I,JM1,KMAX))+
777 3*V(I,J,KMAX)*(H(I,J)-H(I,JM1)))/YJM1(J)
778 661 CONTINUE
779 C
780 C CALCULATE VERTICAL VELOCITY COMPONENT FROM CONTINUITY
781 DO 805 I=2,I1
782 IP1=I+1
783 IM1=I-1
784 DO 805 J=2,J1
785 JP1=J+1
786 JM1=J-1
787 DO 805 K=2,K1
788 KM1=K-1
789 DUDX=(U(IP1,J,K)-U(I,J,K))*SIGMA1(I)/XIP1(I)+(U(I,J,K)-U(IM1,J,K))
790 1*SIGMA2(I)/XIM1(I)
791 DVDY=(V(I,JP1,K)-V(I,J,K))*SIGMA3(J)/YJP1(J)+(V(I,J,K)-V(I,JM1,K))
792 1*SIGMA4(J)/YJM1(J)
793 W(I,J,K)=W(I,J,KM1)-(DUDX+DVDY)*ZKM1(K)/2.
794 805 CONTINUE
795 DO 72 K=2,KMAX
796 DO 72 J=2,J1
797 W(IMAX,J,K)=W(I1,J,K)
798 72 CONTINUE
799 DO 232 I=2,I1
800 DO 232 J=2,J1

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801      W(I,J,KMAX)=W(I,J,K1)+DHDT(I,J)
802 232  CONTINUE
803 C    END OF W CALCULATION
804 C    SET THE BOUNDARY ON W
805 C***** SOLID-FLUID BOUNDARY ON VERTICAL VELOCITY COMPONENT W
806      IF(SLIP.EQ.0.) GO TO 940
807      DO 145 K=2,KMAX
808      DO 146 I=1,IMAX
809      IF(K.LT.IM) GO TO 144
810      IF( I .LT. MH .AND. I .GE. ML ) GO TO 146
811 144  CONTINUE
812      W(I,1,K)=(SIGY2*W(I,2,K)-SIGY3*W(I,3,K))/SIGY1
813 146  CONTINUE
814 145  CONTINUE
815      DO 147 K=2,KMAX
816      DO 147 I=1,IMAX
817      W(I,JMAX,K)=(SIGPY2*W(I,J1,K)-SIGPY3*W(I,J2,K))/SIGPY1
818 147  CONTINUE
819 940  CONTINUE
820 C
821 C*****SET DHDT(I,J)=W(I,J,KMAX)*****
822 C
823      DO 233 I=1,IMAX
824      DO 233 J=1,JMAX
825      DHDT(I,J)=W(I,J,KMAX)
826 233  CONTINUE
827 C
828 C*****BOTTOM BOUNDARY ON DENSITY
829 C
830      DO 12 J=1,JMAX
831      DO 12 I=1,IMAX
832      RHO(I,J,1)=SI(TTBL,RHOTBL,T(I,J,1),26)
833      RHO(I,J,KMAX)=SI(TTBL,RHOTBL,T(I,J,KMAX),26)
834 12  CONTINUE
835 C
836 C    END OF BOUNDARY CONDITIONS
837 C
838      IF(TIME.GT.DQ) GO TO 32
839      IPRINT=NPRINT
840      CALL OUTP
841      IPRINT=IPRINT+2
842      IJ=IJ+2
843 32  CONTINUE
844      CALL OUTP
845      IJ=IJ+1
846      CALL DERIV
847      CALL SUM
848 C    DENSITY CALCULATION
849      DO 40 K=1,KMAX
850      DO 40 I=2,I1

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851      DO 40 J=2,J1
852      RHO(I,J,K)=SI(TTBL,RHOTBL,T(I,J,K),26)
853  40   CONTINUE
854      TIME=TIME+DT
855      IPRINT=IPRINT+1
856      IF(EIEND.EQ,0.) GO TO 2002
857      IF(IP,EQ,IIP) GO TO 2001
858  2002 CONTINUE
859      IF(NITER.EQ,IP) GO TO 20
860      IP=IP+1
861      GO TO 1
862  20   CALL OUTP
863  2001 CONTINUE
864  C*****WRITING OUTPUT ON THE DISC*****
865      WRITE(6,1025)
866      REWIND 25
867      WRITE(25) TIME,DT,IP,KMAX,JMAX,IMAX,IPRINT
868      WRITE(6,2006) TIME,DT,IP,KMAX,JMAX,IMAX,IPRINT
869      WRITE(6,2007)
870      DO 2003 K=1,KMAX
871      DO 2003 J=1,JMAX
872      DO 2003 I=1,IMAX
873      WRITE(25) U(I,J,K),V(I,J,K),W(I,J,K),T(I,J,K),P(I,J,K),H(I,J),
874      1RHO(I,J,K)
875      WRITE(6,2005) I,J,K,U(I,J,K),V(I,J,K),W(I,J,K),T(I,J,K),P(I,J,K),
876      1RHO(I,J,K),H(I,J)
877  2003 CONTINUE
878      STOP
879      END

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1  SUBROUTINE OUTP
2  COMMON P(20,14,10), DELX(20), DELY(14), DELZ(10), RHO(20,14,10),
3  1U(20,14,10), V(20,14,10), W(20,14,10), T(20,14,10),
4  2DUDT(20,14,10), DVDT(20,14,10), DWDI(20,14,10), DTDI(20,14,10),
5  3XNUH, XNUV, RO, FO, SO, DH, DV, ENRUN, TBEG, TEND, IP, IJ, CFY, Z,
6  4WINDX, WINDY, E, RHOA, IMAX, JMAX, KMAX, OMEGA, IM, MH, ML, MK,
7  5DEBUG1, DEBUG2, DEBUG3, DEBUG4, DEBUG5, XNUR, ZNUR, XDHR, XDVR, G, DT,
8  6NITER, NPRINT, I1, I2, J1, J2, K1, K2, TIME, IPRINT, RHS(20,14,10),
9  7XIP1(20), XIN1(20), XIPM1(20), XI123(20), YJP1(14), XIP2(20),
10 8YJM1(14), YJPM1(14), YJ123(14), H(20,14), DZ, YJP2(14),
11 9ZKP1(10), ZKP2(10), ZKM1(10), ZKM2(10), ZKPM1(10), ZK123(10)
12 A, I1, JJ, KK, I3, J3, K3, F01, I1T, B, EPS, ITNMAX, SIGMA1(20), SIGMA2(20),
13 B, SIGMA3(14), SIGMA4(14), SIGMA5(10), SIGMA6(10), DHDT(20,14)
14 COMMON SIGX1, SIGX2, SIGX3, SIGX4, SIGY1, SIGY2, SIGY3, SIGY4, SIGPY1,
15 1SIGPY2, SIGPY3, SIGPY4, SIGZ1, SIGZ2, SIGZ3, SIGZ4
16 COMMON/TABLE/TTBL, RHOTBL
17 DIMENSION TTBL(26), RHOTBL(26)
18 IF(IPRINT.LT.NPRINT.OR.DEBUG3.NE.0.) RETURN
19 IPRINT=0
20 IJ=0
21 10 CONTINUE
22 WRITE(6,1022) TIME, DT, IP
23 1022 FORMAT(1H0,40X,'TIME=',E11.4,5X,'DT ',E11.4,5X,' ITERATION NUMBER
24 1= ', I4/)
25 WRITE(6,1020)
26 DO 15 K=1, KMAX
27 DO 15 J=1, JMAX
28 MJ=1
29 MJJ=MJ+1
30 IF(MJ.EQ. IMAX) MJJ=MJ
31 IF(K.EQ. KMAX) GO TO 25
32 WRITE(6,1021) (I, J, K, U(I, J, K), V(I, J, K), W(I, J, K), T(I, J, K), P(I, J, K)
33 1, I=MJ, MJJ)
34 MJ=MJJ+2
35 IF(MJJ.EQ. IMAX) GO TO 15
36 GO TO 20
37 25 CONTINUE
38 MJ=1
39 30 CONTINUE
40 MJJ=MJ+1
41 IF(MJJ.EQ. IMAX) MJJ=MJ
42 WRITE(6,1023) (I, J, K, U(I, J, K), V(I, J, K), W(I, J, K), T(I, J, K), P(I, J, K)
43 1, H(I, J), I=MJ, MJJ)
44 MJ=MJJ+2
45 IF(MJJ.EQ. IMAX) GO TO 15
46 GO TO 30
47 15 CONTINUE
48 1021 FORMAT(1H ,2(3(12,1X),3(E11.4),2(F6.4),11X))
49 1023 FORMAT(1H ,2(3(12,1X),3(E11.4),2(F6.4),E11.4))
50 1020 FORMAT(1H ,2(1X,'I',2X,'J',2X,'K',7X,'U',9X,'V',9X,'W',7X,'T',7X,

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51        1('P',7X,'H',5X))  
52        RETURN  
53        END
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1  SUBROUTINE SUM
2  COMMON P(20,14,10),DELX(20),DELY(14),DELZ(10),RHO(20,14,10),
3  1U(20,14,10),V(20,14,10),W(20,14,10),T(20,14,10),
4  2DUDT(20,14,10),DVDT(20,14,10),DWDT(20,14,10),DTDT(20,14,10),
5  3XNUH,XNUV,RO,FO,SO,DH,DV,ENRUN,TBEG,TEND,IP,IJ,CFX,CFY,Z,
6  4WINDX,WINDY,E,RHOA,IMAX,JMAX,KMAX,OMEGA,IM,MH,ML,MK,
7  5DEBUG1,DEBUG2,DEBUG3,DEBUG4,DEBUG5,XNUR,ZNUR,XDHR,XDVR,G,DT,
8  6NITER,NPRINT,I1,I2,J1,J2,K1,K2,TIME,IPRINT,RHS(20,14,10),
9  7XIP1(20),XIM1(20),XIPM1(20),XI123(20),YJP1(14),XIP2(20),
10 8YJM1(14),YJPM1(14),YJ23(14),H(20,14),DZ,YJP2(14),
11 9ZKP1(10),ZKP2(10),ZKM1(10),ZKM2(10),ZKPM1(10),ZK123(10)
12 A,I1,JJ,KK,I3,J3,K3,FO1,IIT,B,EPS,IINMAX,SIGMA1(20),SIGMA2(20),
13 B,SIGMA3(14),SIGMA4(14),SIGMA5(10),SIGMA6(10),DHDT(20,14)
14 COMMON SIGX1,SIGX2,SIGX3,SIGX4,SIGY1,SIGY2,SIGY3,SIGY4,SIGPY1,
15 1SIGPY2,SIGPY3,SIGPY4,SIGZ1,SIGZ2,SIGZ3,SIGZA
16 COMMON/TABLE/TTBL,RHOTBL
17 DIMENSION ITBL(26),RHOTBL(26)
18 C INTEGRATE FOR HORIZONTAL VELOCITY COMPONENTS AND TEMPERATURE
19 DO 22 K=2,KMAX
20 DO 21 J=2,J1
21 DO 20 I=2,I1
22 U(I,J,K)=U(I,J,K)+DT*DUDT(I,J,K)
23 V(I,J,K)=V(I,J,K)+DT*DVDT(I,J,K)
24 T(I,J,K)=T(I,J,K)+DT*DTDT(I,J,K)
25 CONTINUE
26 CONTINUE
27 20 CONTINUE
28 C INTEGRATE FOR WATER LEVEL HEIGHT H
29 DO 23 I=1,IMAX
30 DO 23 J=1,JMAX
31 H(I,J)=H(I,J)+DT*DHD(I,J)
32 CONTINUE
33 RETURN
34 END

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1      SUBROUTINE DERIV
2      COMMON P(20,14,10),DELX(20),DELY(14),DELZ(10),RHO(20,14,10),
3      1U(20,14,10),V(20,14,10),W(20,14,10),T(20,14,10),
4      2DUDT(20,14,10),DVDT(20,14,10),DWD(20,14,10),DTDT(20,14,10),
5      3XNUH,XNUV,RO,FO,SO,DH,DV,ENRUN,TBEG,TEND,IP,IJ,CFX,CFY,Z,
6      4WINDX,WINDY,E,RHOA,IMAX,JMAX,KMAX,OMEGA,IM,MH,ML,MK,
7      5DEBUG1,DEBUG2,DEBUG3,DEBUG4,DEBUG5,XNUR,ZNUR,XDWR,XDVR,G,DT,
8      6NITER,NPRINT,I1,I2,J1,J2,K1,K2,TIME,IPRINT,RHS(20,14,10),
9      7XIP1(20),XIM1(20),XJPM1(20),XI123(20),YJP1(14),XIP2(20),
10     8YJM1(14),YJPM1(14),YJ123(14),H(20,14),DZ,YJP2(14),
11     9ZKP1(10),ZKP2(10),ZKM1(10),ZKM2(10),ZKPM1(10),ZK123(10)
12     A,I1,JJ,KK,I3,J3,K3,FO1,IIT,B,EPS,ITNMAX,SIGMA1(20),SIGMA2(20),
13     B,SIGMA3(14),SIGMA4(14),SIGMA5(10),SIGMA6(10),DHDT(20,14)
14     COMMON SIGX1,SIGX2,SIGX3,SIGX4,SIGY1,SIGY2,SIGY3,SIGY4,SIGPY1,
15     1SIGPY2,SIGPY3,SIGPY4,SIGZ1,SIGZ2,SIGZ3,SIGZ4
16     COMMON/TABLE/TTBL,RHOTBL
17     DIMENSION TTBL(26),RHOTBL(26)
18 C
19 C      CALCULATE DERIVATIVES FOR WATER LEVEL ELEMENTS
20 C
21     DO 100 I=2,I1
22     DO 100 J=2,J1
23     IP1=I+1
24     IM1=I-1
25     JP1=J+1
26     JM1=J-1
27 C
28 C      ADJUST WATER LEVEL VARIABLE
29 C
30     ZKP1(K1)=DELZ(K1)+F(I,J)
31     DUDXR=(U(IP1,J,KMAX)-U(I,J,KMAX))/XIP1(I)
32     DUDXL=(U(I,J,KMAX)-U(IM1,J,KMAX))/XIM1(I)
33     DUDYR=(U(I,JP1,KMAX)-U(I,J,KMAX))/YJP1(J)
34     DUDYL=(U(I,J,KMAX)-U(I,JM1,KMAX))/YJM1(J)
35     DUDZL=2.*(U(I,J,KMAX)-U(I,J,K1))/ZKP1(K1)
36     DHDXR=(H(IP1,J)-H(I,J))/XIP1(I)
37     DHDXL=(H(I,J)-H(IM1,J))/XIM1(I)
38     DHDYL=(H(I,J)-H(I,JM1))/YJM1(J)
39     DHDYR=(H(I,JP1)-H(I,J))/YJP1(J)
40     DRHDXR=(RHO(IP1,J,KMAX)-RHO(I,J,KMAX))/XIP1(I)
41     DRHDXL=(RHO(I,J,KMAX)-RHO(IM1,J,KMAX))/XIM1(I)
42     DRHDYR=(RHO(I,JP1,KMAX)-RHO(I,J,KMAX))/YJP1(J)
43     DRHDYL=(RHO(I,J,KMAX)-RHO(I,JM1,KMAX))/YJM1(J)
44 C
45 C      CALCULATE HORIZONTAL MOMENTUM IN X-DIRECTION FOR WATER LEVEL ELEMENTS
46 C
47     DUDT(I,J,KMAX)=-U(I,J,KMAX)*(SIGMA1(I)*DUDXR+SIGMA2(I)*DUDXL)
48     1-V(I,J,KMAX)*(SIGMA3(J)*DUDYR+SIGMA4(J)*DUDYL)
49     2-H(I,J)*(SIGMA1(I)*DRHDXR+SIGMA2(I)*DRHDXL)/FO1
50     3*((SIGMA1(I)*DHDXR+SIGMA2(I)*DHDXL)

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51      4*(XNUR*(SIGMA1(I)*DUDXR+SIGMA2(I)*DUDXL)/H(I,J)
52      5-2.*RHO(I,J,KMAX)/FO1))
53      6+XNUR*8.*((DUDXR-DUDXL)/XIPM1(I)+(DUDYR-DUDYL)/YJPM1(J))
54      7+XNUR*((SIGMA3(J)*DUDYR+SIGMA4(J)*DUDYL)
55      8*(SIGMA3(J)*DHDYR+SIGMA4(J)*DHDYL))/H(I,J)
56      A-ZNUR*DUDZL/H(I,J)
57      B+CFX*RHOA*WINDX*WINDX/(2.*H(I,J))
58 C
59 C      CALCULATE VELOCITY GRADIENT FOR Y-COMPONENT OF VELOCITY FOR WATER LEVEL
60 C
61      DVDXR=(V(IP1,J,KMAX)-V(I,J,KMAX))/XIP1(I)
62      DVDXL=(V(I,J,KMAX)-V(IM1,J,KMAX))/XIM1(I)
63      DVDYR=(V(I,JP1,KMAX)-V(I,J,KMAX))/YJP1(J)
64      DVDYL=(V(I,J,KMAX)-V(I,JM1,KMAX))/YJM1(J)
65      DVDZL=2.*(V(I,J,KMAX)-V(I,J,K1))/ZKP1(K1)
66 C
67 C      CALCULATE HORIZONTAL MOMENTUM IN Y-DIRECTION FOR WATER LEVEL ELEMENTS
68 C
69      DVDT(I,J,KMAX)=-U(I,J,KMAX)*(SIGMA1(I)*DVDXR+SIGMA2(I)*DVDXL)
70      1-V(I,J,KMAX)*(SIGMA3(J)*DVDYR+SIGMA4(J)*DVDYL)
71      2-H(I,J)*(SIGMA3(J)*DRHDYR+SIGMA4(J)*DRHDYL)/FO1
72      3+XNUR*((SIGMA1(I)*DHDXR+SIGMA2(I)*DHDXL)
73      4*(SIGMA1(I)*DVDXR+SIGMA2(I)*DVDXL))/H(I,J)
74      5*((SIGMA3(J)*DHDYR+SIGMA4(J)*DHDYL)
75      6*(XNUR*(SIGMA3(J)*DVDYR+SIGMA4(J)*DVDYL)/H(I,J)
76      7-2.*RHO(I,J,KMAX)/FO1))
77      8+8.*XNUR*((DVDXR-DVDXL)/XIPM1(I)+(DVDYR-DVDYL)/YJPM1(J))
78      A-ZNUR*DVDZL/H(I,J)
79      B+CFY*RHOA*WINDY*WINDY/(2.*H(I,J))
80 C
81 C      CALCULATE TEMPERATURE GRADIENT FOR WATER LEVEL ELEMENTS
82 C
83      DTDXR=(T(IP1,J,KMAX)-T(I,J,KMAX))/XIP1(I)
84      DTDXL=(T(I,J,KMAX)-T(IM1,J,KMAX))/XIM1(I)
85      DTDYR=(T(I,JP1,KMAX)-T(I,J,KMAX))/YJP1(J)
86      DTDYL=(T(I,J,KMAX)-T(I,JM1,KMAX))/YJM1(J)
87      DTDZL=2.*(T(I,J,KMAX)-T(I,J,K1))/ZKP1(K1)
88 C
89 C      CALCULATE ENERGY EQUATION FOR WATER LEVEL ELEMENTS
90 C
91      DTD(I,J,KMAX)=-U(I,J,KMAX)*(SIGMA1(I)*DTDXR+SIGMA2(I)*DTDXL)
92      1-V(I,J,KMAX)*(SIGMA3(J)*DTDYR+SIGMA4(J)*DTDYL)
93      2*XDHR*((SIGMA1(I)*DHDXR+SIGMA2(I)*DHDXL)
94      3*(SIGMA1(I)*DTDXR+SIGMA2(I)*DTDXL))/H(I,J)
95      4*XDHR*8.*((DTDXR-DTDXL)/XIPM1(I)+(DTDYR-DTDYL)/YJPM1(J))
96      5*XDHR*((SIGMA3(J)*DHDYR+SIGMA4(J)*DHDYL)
97      6*(SIGMA3(J)*DTDYR+SIGMA4(J)*DTDYL))/H(I,J)
98      8-XDVR*DTDZL/H(I,J)
99      9-SO*(T(I,J,KMAX)-E)/H(I,J)
100 100 CONTINUE

```

```

101      DO 7 K=2,K1
102      DO 6 J=2,J1
103      DO 5 I=2,I1
104      IP1=I+1
105      IM1=I-1
106      JP1=J+1
107      JM1=J-1
108      KP1=K+1
109      KM1=K-1
110 C
111 C      ADJUST WATER LEVEL VARIABLE
112 C
113      IF(K.NE.K1) GO TO 506
114      ZKP1(K1)=DELZ(K1)*H(I,J)
115      ZKPM1(K1)=ZKP1(K1)+ZKM1(K1)
116      SIGMA5(K)=2.*ZKM1(K)/ZKPM1(K)
117      SIGMA6(K)=2.*ZKP1(K)/ZKPM1(K)
118 506 CONTINUE
119      IF(DEBUG2.EQ.0.) GO TO 11
120      WRITE(6,12) I,J,K
121 12  FORMAT(1H0,2X,'I=',I3,2X,'J=',I3,2X,'K=',I3,/)
122      WRITE(6,10) IP1,IM1,JP1,JM1,KP1,KM1
123 10  FORMAT(1H ,2X,'IP1=',I3,2X,'IM1=',I3,2X,'JP1=',I3,2X,'JM1=',I3,2X,
124      1'KP1=',I3,2X,'KM1=',I3)
125 11  CONTINUE
126 C
127 C*****CALCULATION OF HORIZONTAL MOMENTUM IN X-DIRECTION
128 C
129      DUDXR=(U(IP1,J,K)-U(I,J,K))/XIP1(I)
130      DUDXL=(U(I,J,K)-U(IM1,J,K))/XIM1(I)
131      DUDYR=(U(I,JP1,K)-U(I,J,K))/YJP1(J)
132      DUDYL=(U(I,J,K)-U(I,JM1,K))/YJM1(J)
133      DUDZR=(U(I,J,KP1)-U(I,J,K))/ZKP1(K)
134      DUDZL=(U(I,J,K)-U(I,J,KM1))/ZKM1(K)
135      DUDT(I,J,K)=-U(I,J,K)*(DUDXR*SIGMA1(I)+DUDXL*SIGMA2(I))
136      1-V(I,J,K)*(DUDYR*SIGMA3(J)+DUDYL*SIGMA4(J))
137      2-W(I,J,K)*(DUDZR*SIGMA5(K)+DUDZL*SIGMA6(K))
138      3+S.*(XNUR*((DUDXR-DUDXL)/XIPM1(I)+(DUDYR-DUDYL)/YJPM1(J))
139      4+ZNUR*(DUDZR-DUDZL)/ZKPM1(K))
140 C
141 C*****CALCULATION OF HORIZONTAL MOMENTUM IN Y-DIRECTION
142 C
143      DVDXR=(V(IP1,J,K)-V(I,J,K))/XIP1(I)
144      DVDXL=(V(I,J,K)-V(IM1,J,K))/XIM1(I)
145      DVDYR=(V(I,JP1,K)-V(I,J,K))/YJP1(J)
146      DVDYL=(V(I,J,K)-V(I,JM1,K))/YJM1(J)
147      DVDZR=(V(I,J,KP1)-V(I,J,K))/ZKP1(K)
148      DVDZL=(V(I,J,K)-V(I,J,KM1))/ZKM1(K)
149      DVDYD(I,J,K)=-U(I,J,K)*(DVDXR*SIGMA1(I)+DVDXL*SIGMA2(I))
150      1-V(I,J,K)*(DVDYR*SIGMA3(J)+DVDYL*SIGMA4(J))

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151      2-W(I,J,K)*(DVDZR*SIGMA5(K)+DVDZL*SIGMA6(K))
152      3+8.*(XNUR*((DVDXR-DVDXL)/XIPM1(I)
153      4+(DVDYR-DVDYL)/YJPM1(J))
154      5+ZNUR*(DVDZR-DVDZL)/ZKPM1(K))
155 C
156 C*****CALCULATION OF VERTICAL MOMENTUM IN Z-DIRECTION
157 C
158      DWDXR=(W(IP1,J,K)-W(I,J,K))/XIP1(I)
159      DWDXL=(W(I,J,K)-W(IM1,J,K))/XIM1(I)
160      DWDYR=(W(I,JP1,K)-W(I,J,K))/YJP1(J)
161      DWDYL=(W(I,J,K)-W(I,JM1,K))/YJM1(J)
162      DWDZR=(W(I,J,KP1)-W(I,J,K))/ZKP1(K)
163      DWDZL=(W(I,J,K)-W(I,J,KM1))/ZKM1(K)
164      DWDT(I,J,K)=-U(I,J,K)*(DWDXR*SIGMA1(I)+DWDXL*SIGMA2(I))
165      1-V(I,J,K)*(DWDYR*SIGMA3(J)+DWDYL*SIGMA4(J))
166      2-W(I,J,K)*DWDZL*2.
167      3+8.*(XNUR*((DWDXR-DWDXL)/XIPM1(I)
168      4+(DWDYR-DWDYL)/YJPM1(J))
169      5+ZNUR*(DWDZR-DWDZL)/ZKPM1(K))
170      6-RHO(I,J,K)/FO1
171 C
172 C*****CALCULATION OF ENERGY EQUATION
173 C
174      DTDXR=(T(IP1,J,K)-T(I,J,K))/XIP1(I)
175      DTDXL=(T(I,J,K)-T(IM1,J,K))/XIM1(I)
176      DTDYR=(T(I,JP1,K)-T(I,J,K))/YJP1(J)
177      DTDYL=(T(I,J,K)-T(I,JM1,K))/YJM1(J)
178      DTDZR=(T(I,J,KP1)-T(I,J,K))/ZKP1(K)
179      DTDZL=(T(I,J,K)-T(I,J,KM1))/ZKM1(K)
180      DTDI(I,J,K)=-U(I,J,K)*(DTDXR*SIGMA1(I)+DTDXL*SIGMA2(I))
181      1-V(I,J,K)*(DTDYR*SIGMA3(J)+DTDYL*SIGMA4(J))
182      2-W(I,J,K)*(DTDZR*SIGMA5(K)+DTDZL*SIGMA6(K))
183      3+8.*(XDHR*((DTDXR-DTDXL)/XIPM1(I)
184      4+(DTDYR-DTDYL)/YJPM1(J))
185      5+XDVR*((DTDZR-DTDZL)/ZKPM1(K))
186 C
187 C*****CALCULATION OF DIVERGENCE
188 C
189      IF(DEBUG2.EQ.0.) GO TO 90
190      IF(IJ.NE.NPRINT) GO TO 90
191      DIV=(DUDXR*SIGMA1(I)+DUDXL*SIGMA2(I))+
192      (DVDYR*SIGMA3(J)+
193      1DVDYL*SIGMA4(J)+DWDZL*2.
194      WRITE(6,80) I,J,K,DUDT(I,J,K),DVDT(I,J,K),DWDT(I,J,K),DIV
195      90 CONTINUE
196      80 FORMAT(1H ,3(I2,2X),2X,'DUDT(I,J,K)=',E15.7,2X,'DVDT(I,J,K)=',
197      1E15.7,2X,'DWDT(I,J,K)=',E15.7,2X,'DIV=',E15.7)
198      5 CONTINUE
199      6 CONTINUE
200      7 CONTINUE
200 C      SET QX AT THE BOUNDARIES

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```

201      DO 200 J=2,J1
202      DO 200 K=2,K1
203      DUDT(1,J,K)=0
204      DUDT(IMAX,J,K)=DUDT(1,J,K)
205 200  CONTINUE
206 C    SET QY AT THE BOUNDARIES
207      DO 210 I=2,I1
208      DO 210 K=2,K1
209      DVDYRO=(V(I,3,K)-V(I,2,K))/YJPM1(2)
210      DVDYLO=(V(I,2,K)-V(I,1,K))/YJPM1(2)
211      DVDYRM=(V(I,JMAX,K)-V(I,J1,K))/YJPM1(J1)
212      DVDYLM=(V(I,J1,K)-V(I,J2,K))/YJPM1(J1)
213      DVDT(I,JMAX,K)=8.*XNUR*(DVDYRM-DVDYLM)/YJPM1(J1)
214      IF(K.LT.IM) GO TO 205
215      IF( I .LT. MH .AND. I .GE. ML ) GO TO 220
216 205  CONTINUE
217      DVDT(I,1,K)=8.*XNUR*(DVDYRO-DVDYLO)/YJPM1(2)
218      GO TO 210
219 220  CONTINUE
220      DVDT(I,1,K)=0.
221 210  CONTINUE
222 C    SET QZ AT THE BOUNDARIES
223      DO 230 I=2,I1
224      DO 230 J=2,J1
225      DWDZTO=(W(I,J,3)-W(I,J,2))/ZKPM1(2)
226      DWDZLO=(W(I,J,2)-W(I,J,1))/ZKPM1(2)
227      DWDT(I,J,1)=-RHO(I,J,1)/FO1+8.*ZNUR*(DWDZTO-DWDZLO)/ZKPM1(2)
228      DWDT(I,J,KMAX)=-RHO(I,J,KMAX)/FO1
229 230  CONTINUE
230      DO 504 K=2,K1
231      KP1=K+1
232      KM1=K-1
233      DO 504 J=2,J1
234      JP1=J+1
235      JM1=J-1
236      DO 504 I=2,I1
237      IP1=I+1
238      IM1=I-1
239 C
240 C    ADJUST WATER LEVEL VARIABLE
241 C
242      IF(K.NE.K1) GO TO 505
243      ZKPM1(K1)=DELZ(K1)+H(I,J)
244      ZKPM1(K1)=ZKPM1(K1)+ZKM1(K1)
245      SIGMA5(K)=2.*ZKM1(K)/ZKPM1(K)
246      SIGMA6(K)=2.*ZKP1(K)/ZKPM1(K)
247 505  CONTINUE
248      DQDZ=(DWDT(I,J,KP1)-DWDT(I,J,K))*SIGMA5(K)/ZKPM1(K)+(DWDT(I,J,K)-
249      1DWDT(I,J,KM1))*SIGMA6(K)/ZKM1(K)
250      RHS(I,J,K)=DQDZ

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251      DQDY=(DVDT(I,JP1,K)-DVDT(I,J,K))*SIGMA3(J)/YJP1(J)+(DVDT(I,J,K)-
252      1DVDT(I,JM1,K))*SIGMA4(J)/YJM1(J)
253      RHS(I,J,K)=RHS(I,J,K)+DQDY
254      DQDX=(DUDT(IP1,J,K)-DUDT(I,J,K))*SIGMA1(I)/XIP1(I)+(DUDT(I,J,K)-
255      1DUDT(IM1,J,K))*SIGMA2(I)/XIM1(I)
256      RHS(I,J,K)=(RHS(I,J,K)+DQDX)*FO1
257      IF(DEBUG2.EQ.0.) GO TO 504
258      WRITE(6,1000) I,J,K,DQDX,DQDY,DQDZ
259 1000  FORMAT(1H,2X,'I=',I2,2X,'J=',I2,2X,'K=',I2,2X,'DQDX=',E14.7,2X,
260      1'DQDY=',E14.7,2X,'DQDZ=',E14.7)
261 504  CONTINUE
262 600  CONTINUE
263      CALL PRESS
264      DO 15 K=2,K1
265      DO 15 J=2,J1
266      DO 15 I=2,I1
267      IP1=I+1
268      IM1=I-1
269      JP1=J+1
270      JM1=J-1
271      KP1=K+1
272      KM1=K-1
273 C
274 C      CALCULATE PRESSURE GRADIENT IN X-DIRECTION
275 C
276      DPDX=(P(IP1,J,K)-P(I,J,K))*SIGMA1(I)/XIP1(I)+(P(I,J,K)-P(IM1,J,K))
277      1*SIGMA2(I)/XIM1(I)
278      DUDT(I,J,K)=DUDT(I,J,K)-DPDX/FO1
279 C
280 C      CALCULATE PRESSURE GRADIENT IN Y-DIRECTION
281 C
282      DPDY=(P(I,JP1,K)-P(I,J,K))*SIGMA3(J)/YJP1(J)+(P(I,J,K)-P(I,JM1,K))
283      1*SIGMA4(J)/YJM1(J)
284      DVDT(I,J,K)=DVDT(I,J,K)-DPDY/FO1
285 15  CONTINUE
286      RETURN
287      END

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1     SUBROUTINE PRESS
2     REAL LAMB1,LAMB2
3     COMMON P(20,14,10),DELX(20),DELY(14),DELZ(10),RHO(20,14,10),
4     1U(20,14,10),V(20,14,10),W(20,14,10),T(20,14,10),
5     2DUDT(20,14,10),DVDT(20,14,10),DWDT(20,14,10),DTDT(20,14,10),
6     3XNUH,XNUV,RO,FO,SO,DH,DV,ENRUN,TEG,TEND,IP,IJ,CFX,CFY,Z,
7     4WINDX,WINDY,E,RHOA,IMAX,JMAX,KMAX,OMEGA,IM,MH,ML,MK,
8     5DEBUG1,DEBUG2,DEBUG3,DEBUG4,DEBUG5,XNUR,ZNUR,XDHR,XDVR,G,DT,
9     6NITER,NPRINT,I1,I2,J1,J2,K1,K2,TIME,IPRINT,RHS(20,14,10),
10    7XIP1(20),XIM1(20),XIPM1(20),XI123(20),YJP1(14),XIP2(20),
11    8YJM1(14),YJPM1(14),YJ123(14),H(20,14),DZ,YJP2(14),
12    9ZKP1(10),ZKP2(10),ZKM1(10),ZKM2(10),ZKPM1(10),ZK123(10)
13    A,I,J,K,I3,J3,K3,FO1,IIT,B,EPS,ITNMAX,SIGMA1(20),SIGMA2(20),
14    B,SIGMA3(14),SIGMA4(14),SIGMA5(10),SIGMA6(10),DHDT(20,14)
15    COMMON SIGX1,SIGX2,SIGX3,SIGX4,SIGY1,SIGY2,SIGY3,SIGY4,SIGPY1,
16    1SIGPY2,SIGPY3,SIGPY4,SIGZ1,SIGZ2,SIGZ3,SIGZ4
17    COMMON/TABLE/TTBL,RHOTBL
18    DIMENSION ITBL(26),RHOTBL(26)
19    IF(HYD.EQ.0.) GO TO 300
20 C   CALCULATE PRESSURE FROM HYDROSTATIC APPROXIMATION
21     DO 110 L=1,KMAX
22     K=KMAX+1-L
23     KP1=K+1
24     DO 110 J=1,JMAX
25     DO 110 I=1,IMAX
26     IF( I .LT. MH .AND. I .GE. ML ) GO TO 110
27     IF(K.NE.KMAX) GO TO 120
28     P(I,J,K)=RHO(I,J,K)*H(I,J)/2.
29     GO TO 110
30 120 CONTINUE
31     IF(K.NE.K1) GO TO 130
32     ZKP1(K1)=DELZ(K1)+H(I,J)
33 130 CONTINUE
34     P(I,J,K)=P(I,J,KP1)+RHO(I,J,K)*ZKP1(K)/2.
35 110 CONTINUE
36     GO TO 100
37 300 CONTINUE
38 C   END PRESSURE CALCULATION FROM HYDROSTATIC APPROXIMATION
39 33  FORMAT(1H0,2X,'OMEGA=',E14.6,/)
40 C   CALCULATE FROM PRESSURE EQUATION
41     ITN     =1
42     YI=1.
43     LAMB1=1.
44     OMEGA=1.
45     IF(IP.NE.0) GO TO 10
46     IF(.NOT.(DEBUG4.GE.2.)) GO TO 10
47     WRITE(6,33) OMEGA
48     DO 6 K=2,K1
49     DO 6 J=2,J1
50     DO 6 I=2,I1

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51      WRITE(6,23) K,ZK123(K),ZKP1(K),ZKP2(K),ZKM1(K),J,YJ123(J),
52      1YJP1(J),YJM1(J),I,XI123(I),XIP1(I),XIM1(I)
53      6  CONTINUE
54      10  D      =0.
55      B=1,-OMEGA
56      Y2=0.
57 C
58 C      PRESSURE BOUNDARY CONDITIONS
59 C ****SOLID-FLUID BOUNDARY ON PRESSURE
60 C
61      DO 200 I=2,I1
62      DO 200 K=2,K1
63      IF(K,LT,IM) GO TO 225
64      IF( I .LT. MH .AND. I .GE. ML ) GO TO 210
65      225  CONTINUE
66      P(I,1,K)=(SIGY2*P(I,2,K)-SIGY3*P(I,3,K)-SIGY4*DVRT(I,1,K)*F01)/SIG
67      AY1
68      210  CONTINUE
69      P(I,JMAX,K)=(SIGPY2*P(I,J1,K)-SIGPY3*P(I,J2,K)+SIGPY4*DVRT(I,JMAX,
70      AK)*F01)/SIGPY1
71      200  CONTINUE
72 C
73 C      CALCULATION OF PRESSURE IN THE WATER LEVEL ELEMENTS
74 C
75      DO 50 I=2,I1
76      DO 50 J=2,J1
77      P(I,J,KMAX)=H(I,J)*RHO(I,J,KMAX)/2.
78      50  CONTINUE
79 C
80 C      FLUID-FLUID BOUNDARY CONDITION IS SET DURING INITIALIZATION
81 C *****BOTTOM BOUNDARY ON PRESSURE
82 C
83      DO 220 I=2,I1
84      DO 220 J=2,J1
85      P(I,J,1)=(SIGZ2*P(I,J,2)-SIGZ3*P(I,J,3)-SIGZ4*DVRT(I,J,1)*F01)/SIG
86      AZ1
87      220  CONTINUE
88 C
89 C      OUTGOING BOUNDARY ON PRESSURE
90 C
91      DO 55 K=2,KMAX
92      DO 55 J=2,J1
93      P(IMAX,J,K)=P(I1,J,K)
94      55  CONTINUE
95 C
96 C      CALCULATE PRESSURE IN FLOW REGION
97 C
98      DO 5 K=2,K1
99      DO 5 J=2,J1
100     DO 5 I=2,I1

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101 C      ADJUST WATER LEVEL VARIABLE
102 C
103 C
104      IF(K.NE.K1) GO TO 18
105      ZKPI(K1)=DELZ(K1)+H(I,J)
106      ZKPM1(K1)=ZKPI(K1)+ZKM1(K1)
107      ZK123(K1)=ZKPI(K1)*ZKPM1(K1)*ZKPM1(K1)
108      CONTINUE
109      X1 =YJ123(J)*ZK123(K)*XJM1(I)
110      X2 =YJ123(J)*ZK123(K)*XIP1(I)
111      X3 =XI123(I)*ZK123(K)*YJM1(J)
112      X4 =XI123(I)*ZK123(K)*YJP1(J)
113      X5 =XI123(I)*YJ123(J)*ZKPM1(K)
114      X6 =XI123(I)*YJ123(J)*ZKP1(K)
115      X7 =XI123(I)*YJ123(J)*ZK123(K)
116      X  =X1+X2+X3+X4+X5+X6
117      A  =OMEGA/X
118      PNEW=A*(X1*P(I+1,J,K)+X2*P(I-1,J,K)+X3*P(I,J+1,K)+X4*P(I,J-1,K)+
119      X5*P(I,J,K+1)+X6*P(I,J,K-1)-X7*RHS(I,J,K)/8,)+B*P(I,J,K)
120      RESID =ABS(PNEW-P(I,J,K))
121      IF(RESID-D) 15,15,16
122      D  =RESID
123      P(I,J,K) =PNEW
124      IF(OMEGA.NE.1.) GO TO 9
125      Y2=Y2+RESID
126      CONTINUE
127      IF(IP.NE.0) GO TO 5
128      IF(.NOT.(DEBUG4.EQ.1.)) GO TO 5
129      WRITE(6,26) X,X1,X2,X3,X4,X5,X6,X7,A
130      CONTINUE
131 C
132 C      CALCULATE RELAXATION FACTOR OMEGA
133 C
134      IF(OMEGA.NE.1.) GO TO 28
135      LAMB2=Y2/Y1
136      IF(DEBUG4.EQ.0.) GO TO 60
137      WRITE(6,42)Y1,Y2,LAMB2
138      CONTINUE
139      IF(LAMB2.GE.1.) GO TO 28
140      IF(ABS(LAMB2-LAMB1).GT. .01) GO TO 28
141      OMEGA=2./((1+SQRT(1-LAMB2))
142      IF(DEBUG4.EQ.0.) GO TO 28
143      WRITE(6,33) OMEGA
144      CONTINUE
145      Y1=Y2
146      LAMB1=LAMB2
147 C
148      42  FORMAT(1H ,Y1=',E14,7,2X ,Y2=',E14,7,2X ,OMEGAS=',E14,7)
149      IF(D-EPS) 14,12,12
150      12  ITN =ITN+1

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151     IF(ITN-ITNMAX) 10,10,21
152 21  WRITE(6,22)
153     CALL EXIT
154 22  FORMAT(1H0,4X,' FAILS TO CONVERGE IN GIVEN ITNMAX ',/)
155 26  FORMAT(1H0,2X,'I',2X,'J',3X,'K',6X,'X',6X,'X1',6X,'X2',6X,'X3',
156     16X,'X4',6X,'X5',6X,'X6',6X,'X7',6X,'A')
157 14  CONTINUE
158     IF(DEBUG4.EQ.0.) GO TO 40
159     WRITE(6,20) ITN
160 20  FORMAT(1H0,10X,'ITN=',I3)
161 23  FORMAT(1H0,'K=',I2,2X,'ZK123(K)=',E11.4,2X,'ZKP1(K)=',E11.4,2X,
162     1'ZKP2(K)=',E11.4,2X,'ZKM1(K)=',E11.4,/,2X,'J=',I2,2X,'YJ123(J)=',
163     2E11.4,2X,'YJP1(J)=',E11.4,2X,'YJM1(J)=',E11.4,2X,/,2X,'I=',I2,
164     32X,'XI123(I)=',E11.4,2X,'XIP1(I)=',E11.4,2X,'XIM1(I)=',E11.4)
165 24  FORMAT(1H , 'I=',I2,1X,'J=',I2,1X,'K=',I2,1X,
166     1'P(I,J,K)=',E11.4,1X,'P(I-1,J,K)=',E11.4,1X,'P(I+1,J,K)=',E11.4,
167     21X,'P(I,J-1,K)=',E11.4,/,7X,'P(I,J+1,K)=',E11.4,1X,'P(I,J,K-1)=',
168     3E11.4,1X,'P(I,J,K+1)=',E11.4,1X,'RHS(I,J,K)=',E11.4,1X,'A(I,J,K)=',
169     4,E11.4,/,7X,'B=',E11.4,2X,'D=',E11.4)
170 100 CONTINUE
171 40  RETURN
172     END

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```

1      FUNCTION SI(XTBL,YTBL,X,N)
2 C      LINEAR INTERPOLATION OR EXTRAPOLATION OF SINGLE VARIABLE FUNCTION
3 C      XTBL = INDEPENDENT VARIABLE TABLE
4 C      YTBL = DEPENDENT VARIABLE TABLE
5 C      X      =VALUE OF THE INDEPENDENT VARIABLE
6 C      N      = NO. OF POINTS IN TABLE
7 C      0=NO EXTRAPOLATION, 1=LOWER EXTRAPOLATION, 2=UPPER EXTRAPOLATION
8      DIMENSION XTBL(30),YTBL(30)
9 C      CHECK TO SEE IF EXTRAPOLATION IS NEEDED
10      IF(X-XTBL(1)) 120,130,150
11      120 WRITE(6,202)
12      202 FORMAT(1H0,5X,' DENSITY TABLE OUT OF RANGE AT THE LOWER END')
13      130 II = 2
14      GO TO 254
15      150 IF(XTBL(N)-X) 160,180,210
16      160 WRITE(6,201)
17      201 FORMAT(1H0, 5X,' DENSITY TABLE OUT OF RANGE AT THE UPPER END ')
18      180 II = N
19      GO TO 254
20 C      FIND X IN TABLE
21      210 DO 220 IK=2,N
22           II = IK
23           IF(XTBL(IK)-X) 220,254,254
24      220 CONTINUE
25      254 X1 = XTBL(II-1)
26           X2 = XTBL(II)
27           Y1 = YTBL(II-1)
28           Y2 = YTBL(II)
29           SI = Y1+(Y2-Y1)*(X-X1)/(X2-X1)
30      RETURN
31      END

```

```
1      BLOCK DATA
2      COMMON/TABLE/TTBL,RHOTBL
3      REAL * 4 TTBL(26)/70.,72.,74.,76.,78.,80.,82.,84.,86.,88.,90.,92.,
4      *94.,
5      *96.,98.,100.,102.,104.,106.,108.,110.,112.,114.,116.,118.,120./
6      REAL * 4 RHOTBL(26)/62.3029595,62.28977202,62.27425582,62.25487144
7      *.62,23937263,62.22000969,62.20065933,62.18132073,62.16199416,
8      *62.13881812,62.11565936,62.09251785,62.06939358,62.04628653,
9      *62.02319668,61.99628022,61.96938712,61.94251734,61.91567086,
10     *61.88884763,61.86204763,61.83144747,61.80469716,61.77415369,
11     *61.74364041,61.71315725/
12     END
```

ENRUN = 0.2000E 01

TBEG = 0.0000E 00 TEND = 0.1000E 04 ENITER= 0.5000E 04 NPRNT= 0.2500E 02

UI = 0.4000E 00 VI = 0.0000E 00 TI = 0.7500E 02

UQ= 0.2000E 01 ANGLE= 0.1571E 01 T0= 0.9000E 02 D0= 0.5000E 02 XNUZ= 0.6250E-05

WINDX = 0.0000E 00 WINDY = 0.0000E 00 RHOA = 0.7630E-01 E = 0.7400E 02 XK = 0.1000E 03

CFX = 0.0000E 00 CFY = 0.0000E 00

DH = 0.1350E 04 DV = 0.4000E 00 XNUH = 0.1350E 04 XNUV = 0.4000E 00 CP= 0.1000E 01

EIMAX = 0.1000E 02 EJMAX = 0.7000E 01 EKMAX = 0.5000E 01

DEBUG1= 0.0000E 00 DEBUG2= 0.0000E 00 DEBUG3= 0.0000E 00 DEBUG4= 0.0000E 00

EIRUN= 0.0000E 00 EIEND= 0.1000E 01

EIM= 0.4000E 01 EML= 0.4000E 01 EMK= 0.5000E 01 SLIP= 0.0000E 00 HYD= 0.0000E 00

HEAT= 0.0000E 00 XIIP= 0.5200E 02

I DELX(I)

1 0.6500E 02 2 0.6000E 02 3 0.5500E 02 4 0.5000E 02

5 0.5000E 02 6 0.5500E 02 7 0.6000E 02 8 0.6500E 02

9 0.7000E 02 10 0.7000E 02

J DELY(J)

1 0.5000E 02 2 0.5000E 02 3 0.5500E 02 4 0.6000E 02

5 0.6500E 02 6 0.7000E 02 7 0.7000E 02

K DELZ(K)

1 0.3500E 01 2 0.4000E 01 3 0.5000E 01 4 0.4000E 01

5 0.3500E 01

TTBL	RHOTRL
0.7000000E 02	0.6230296E 02
0.7200000E 02	0.6228976E 02
0.7400000E 02	0.6227426E 02
0.7600000E 02	0.6225487E 02
0.7800000E 02	0.6223938E 02
0.8000000E 02	0.6222002E 02
0.8200000E 02	0.6220665E 02
0.8400000E 02	0.6218132E 02
0.8600000E 02	0.6216199E 02

4

0.8800000E 02	0.6213882E 02
0.9000000E 02	0.6211566E 02
0.9200000E 02	0.6209251E 02
0.9400000E 02	0.6206940E 02
0.9600000E 02	0.6204628E 02
0.9800000E 02	0.6202319E 02
0.1000000E 03	0.6199628E 02
0.1020000E 03	0.6196939E 02
0.1040000E 03	0.6194252E 02
0.1060000E 03	0.6191566E 02
0.1080000E 03	0.6188885E 02
0.1100000E 03	0.6186205E 02
0.1120000E 03	0.6183145E 02
0.1140000E 03	0.6180470E 02
0.1160000E 03	0.6177415E 02
0.1180000E 03	0.6174364E 02
0.1200000E 03	0.6171315E 02

IMAX = 10 JMAX = 7 KMAX = 5

I1 = 9 J1 = 6 K1 = 4

I2 = 8 J2 = 5 K2 = 3

I3 = 7 J3 = 4 K3 = 2

RHOZ= 0.6211566E 02

I	J	K	SUGGESTED DT	I	J	K	SUGGESTED DTT
5	2	2	0.4511E 00	5	2	2	0.4511E 00

I	J	KMAX	SUGGESTED DTS	I	J	KMAX	SUGGESTED DTTS
5	2	5	0.4465E 00	5	2	5	0.4465E 00

*****NON-DIMENSIONAL INPUT VALUES*****

ENRUN = 0.2000E 01

TBEG = 0.0000E 00 TEND = 0.4000E 02 NITER= 5000 NPRINT= 25

UI = 0.2000E 00 VI = 0.0000E 00 TI = 0.8333E 00

U0= 0.1000E 01 ANGLE= 0.1571E 01 T0= 0.1000E 01 D0= 0.5000E 02 XNUZ= 0.1000E 01

WINDX = 0.0000E 00 WINDY = 0.0000E 00 RHOA = 0.1228E-02

DH = 0.1636E 09 DV = 0.4848E 05 XNUH = 0.1636E 09 XNUV = 0.4848E 05

G = 0.3220E 02 DT = 0.8000E-02 E= 0.8222222E 00

I DELX(I)

1 0.1300E 01 2 0.1200E 01 3 0.1100E 01 4 0.1000E 01

5 0.1000E 01 6 0.1100E 01 7 0.1200E 01 8 0.1300E 01

9 0.1400E 01 10 0.1400E 01

J DELY(J)

1 0.1000E 01 2 0.1000E 01 3 0.1100E 01 4 0.1200E 01

5 0.1300E 01 6 0.1400E 01 7 0.1400E 01

K DELZ(K)

1 0.7000E-01 2 0.8000E-01 3 0.1000E 00 4 0.8000E-01

5 0.7000E-01

RO = 0.1212E 08 FO = 0.4984E-01 SO = 0.9317E-05

TTBL	RFOTBL
0.7777777E 00	0.1003015E 01
0.8000000E 00	0.1002803E 01
0.8222222E 00	0.1002553E 01
0.8444444E 00	0.1002240E 01
0.8666666E 00	0.1001991E 01
0.8888888E 00	0.1001679E 01
0.9111111E 00	0.1001368E 01
0.9333333E 00	0.1001057E 01
0.9555555E 00	0.1000746E 01
0.9777777E 00	0.1000373E 01
0.1000000E 01	0.1000000E 01
0.1022222E 01	0.9996273E 00
0.1044444E 01	0.9992552E 00
0.1066667E 01	0.9988830E 00
0.1088888E 01	0.9985113E 00
0.1111111E 01	0.9980780E 00
0.1133333E 01	0.9976451E 00
0.1155555E 01	0.9972126E 00
0.1177777E 01	0.9967802E 00
0.1200000E 01	0.9963486E 00
0.1222221E 01	0.9959170E 00
0.1244444E 01	0.9954244E 00
0.1266666E 01	0.9949939E 00
0.1288888E 01	0.9945321E 00
0.1311110E 01	0.9940107E 00
0.1333333E 01	0.9935200E 00

*****END OF NON-DIMENSIONAL INPUT VALUES*****

XNUR = 0.1350E 02 ZNUR = 0.4000E-02 XDHR = 0.1350E 02 XDVR = 0.4000E-02

1	4	1	0.0000E	00	0.0000E	00	0.0000E	000,83330,3308	2	4	1	0.0000E	00	0.0000E	00	0.0000E	000,83330,3308
3	4	1	0.0000E	00	0.0000E	00	0.0000E	000,83330,3308	4	4	1	0.0000E	00	0.0000E	00	0.0000E	000,83330,3308
5	4	1	0.0000E	00	0.0000E	00	0.0000E	000,83330,3308	6	4	1	0.0000E	00	0.0000E	00	0.0000E	000,83330,3308
7	4	1	0.0000E	00	0.0000E	00	0.0000E	000,83330,3308	8	4	1	0.0000E	00	0.0000E	00	0.0000E	000,83330,3308
9	4	1	0.0000E	00	0.0000E	00	0.0000E	000,83330,3308	10	4	1	0.0000E	00	0.0000E	00	0.0000E	000,83330,3308
1	5	1	0.0000E	00	0.0000E	00	0.0000E	000,83330,3308	2	5	1	0.0000E	00	0.0000E	00	0.0000E	000,83330,3308
3	5	1	0.0000E	00	0.0000E	00	0.0000E	000,83330,3308	4	5	1	0.0000E	00	0.0000E	00	0.0000E	000,83330,3308
5	5	1	0.0000E	00	0.0000E	00	0.0000E	000,83330,3308	6	5	1	0.0000E	00	0.0000E	00	0.0000E	000,83330,3308
7	5	1	0.0000E	00	0.0000E	00	0.0000E	000,83330,3308	8	5	1	0.0000E	00	0.0000E	00	0.0000E	000,83330,3308
9	5	1	0.0000E	00	0.0000E	00	0.0000E	000,83330,3308	10	5	1	0.0000E	00	0.0000E	00	0.0000E	000,83330,3308
1	6	1	0.0000E	00	0.0000E	00	0.0000E	000,83330,3308	2	6	1	0.0000E	00	0.0000E	00	0.0000E	000,83330,3308
3	6	1	0.0000E	00	0.0000E	00	0.0000E	000,83330,3308	4	6	1	0.0000E	00	0.0000E	00	0.0000E	000,83330,3308
5	6	1	0.0000E	00	0.0000E	00	0.0000E	000,83330,3308	6	6	1	0.0000E	00	0.0000E	00	0.0000E	000,83330,3308
7	6	1	0.0000E	00	0.0000E	00	0.0000E	000,83330,3308	8	6	1	0.0000E	00	0.0000E	00	0.0000E	000,83330,3308
9	6	1	0.0000E	00	0.0000E	00	0.0000E	000,83330,3308	10	6	1	0.0000E	00	0.0000E	00	0.0000E	000,83330,3308
1	7	1	0.0000E	00	0.0000E	00	0.0000E	000,83330,3308	2	7	1	0.0000E	00	0.0000E	00	0.0000E	000,83330,3308
3	7	1	0.0000E	00	0.0000E	00	0.0000E	000,83330,3308	4	7	1	0.0000E	00	0.0000E	00	0.0000E	000,83330,3308
5	7	1	0.0000E	00	0.0000E	00	0.0000E	000,83330,3308	6	7	1	0.0000E	00	0.0000E	00	0.0000E	000,83330,3308
7	7	1	0.0000E	00	0.0000E	00	0.0000E	000,83330,3308	8	7	1	0.0000E	00	0.0000E	00	0.0000E	000,83330,3308
9	7	1	0.0000E	00	0.0000E	00	0.0000E	000,83330,3308	10	7	1	0.0000E	00	0.0000E	00	0.0000E	000,83330,3308
1	1	2	0.0000E	00	0.0000E	00	0.0000E	000,83330,2556	2	1	2	0.0000E	00	0.0000E	00	0.0000E	000,83330,2557
3	1	2	0.0000E	00	0.0000E	00	0.0000E	000,83330,2556	4	1	2	0.0000E	00	0.0000E	00	0.0000E	000,83330,2557
5	1	2	0.0000E	00	0.0000E	00	0.0000E	000,83330,2557	6	1	2	0.0000E	00	0.0000E	00	0.0000E	000,83330,2556
7	1	2	0.0000E	00	0.0000E	00	0.0000E	000,83330,2556	8	1	2	0.0000E	00	0.0000E	00	0.0000E	000,83330,2556
9	1	2	0.0000E	00	0.0000E	00	0.0000E	000,83330,2556	10	1	2	0.0000E	00	0.0000E	00	0.0000E	000,83330,2556
1	2	2	0.2000E	00	0.0000E	00	0.0000E	000,83330,2556	2	2	2	0.1764E	00	0.1169E-03	0.6792E-030	0.83330,2557	
3	2	2	0.1763E	00	-0.5988E-04	0.1119E-040	0.83330,2556	4	2	2	0.1761E	00	0.1910E-03	0.1785E-040	0.83330,2557		
5	2	2	0.1765E	00	0.1906E-03	0.2106E-040	0.83330,2557	6	2	2	0.1765E	00	-0.6188E-04	0.1025E-040	0.83330,2556		
7	2	2	0.1763E	00	0.1966E-06	0.6213E-050	0.83330,2556	8	2	2	0.1763E	00	-0.5441E-06	0.1428E-050	0.83330,2556		
9	2	2	0.1763E	00	-0.1731E-05	0.7238E-070	0.83330,2556	10	2	2	0.1763E	00	-0.1731E-05	0.7238E-070	0.83330,2556		
1	3	2	0.2000E	00	0.0000E	00	0.0000E	000,83330,2556	2	3	2	0.1974E	00	0.1038E-03	0.7779E-040	0.83330,2556	
3	3	2	0.1974E	00	-0.5242E-04	0.1770E-050	0.83330,2556	4	3	2	0.1974E	00	0.1653E-03	0.6504E-050	0.83330,2556		
5	3	2	0.1974E	00	0.1651E-03	0.6236E-050	0.83330,2556	6	3	2	0.1974E	00	-0.5454E-04	0.1947E-050	0.83330,2556		
7	3	2	0.1974E	00	-0.7965E-06	0.6880E-070	0.83330,2556	8	3	2	0.1974E	00	-0.2788E-05	0.8248E-070	0.83330,2556		
9	3	2	0.1974E	00	-0.1513E-05	0.3083E-070	0.83330,2556	10	3	2	0.1974E	00	-0.1513E-05	0.3083E-070	0.83330,2556		
1	4	2	0.2000E	00	0.0000E	00	0.0000E	000,83330,2556	2	4	2	0.1974E	00	-0.1057E-05	0.7585E-040	0.83330,2556	
3	4	2	0.1974E	00	-0.2474E-06	0.1546E-050	0.83330,2556	4	4	2	0.1974E	00	-0.1103E-05	0.5816E-050	0.83330,2556		
5	4	2	0.1974E	00	-0.1277E-05	0.5578E-050	0.83330,2556	6	4	2	0.1974E	00	-0.6820E-06	0.1720E-050	0.83330,2556		
7	4	2	0.1974E	00	-0.2341E-05	0.3409E-070	0.83330,2556	8	4	2	0.1974E	00	-0.2226E-05	0.1873E-060	0.83330,2556		
9	4	2	0.1974E	00	-0.6755E-06	0.8487E-070	0.83330,2556	10	4	2	0.1974E	00	-0.6755E-06	0.8487E-070	0.83330,2556		
1	5	2	0.2000E	00	0.0000E	00	0.0000E	000,83330,2556	2	5	2	0.1974E	00	-0.4116E-04	0.7551E-040	0.83330,2556	
3	5	2	0.1974E	00	-0.1857E-05	0.2295E-060	0.83330,2556	4	5	2	0.1974E	00	-0.5352E-06	0.3336E-060	0.83330,2556		
5	5	2	0.1974E	00	0.5352E-06	0.8313E-070	0.83330,2556	6	5	2	0.1974E	00	-0.1105E-05	0.8852E-070	0.83330,2556		
7	5	2	0.1974E	00	-0.7059E-06	0.2049E-060	0.83330,2556	8	5	2	0.1974E	00	0.3872E-06	0.1524E-060	0.83330,2556		
9	5	2	0.1974E	00	-0.2960E-06	0.2647E-070	0.83330,2556	10	5	2	0.1974E	00	-0.2960E-06	0.2647E-070	0.83330,2556		

5	6	2	0.5283E-01	0.2369E-01	0.1908E-02	0.83330,2573	6	6	2	0.5109E-01	0.2170E-01	0.1227E-02	0.83330,2573
7	6	2	0.4627E-01	0.1169E-01	0.8584E-03	0.83330,2563	8	6	2	0.4119E-01	0.6343E-02	0.5409E-03	0.83330,2560
9	6	2	0.3697E-01	0.3273E-02	0.2498E-03	0.83330,2558	10	6	2	0.3697E-01	0.3273E-02	0.2498E-03	0.83330,2558
1	7	2	0.0000E 00	0.0000E 00	0.0000E 00	0.83330,2556	2	7	2	0.0000E 00	0.0000E 00	0.0000E 00	0.83330,2605
3	7	2	0.0000E 00	0.0000E 00	0.0000E 00	0.83330,2577	4	7	2	0.0000E 00	0.0000E 00	0.0000E 00	0.83330,2575
5	7	2	0.0000E 00	0.0000E 00	0.0000E 00	0.83330,2571	6	7	2	0.0000E 00	0.0000E 00	0.0000E 00	0.83330,2565
7	7	2	0.0000E 00	0.0000E 00	0.0000E 00	0.83330,2561	8	7	2	0.0000E 00	0.0000E 00	0.0000E 00	0.83330,2559
9	7	2	0.0000E 00	0.0000E 00	0.0000E 00	0.83330,2557	10	7	2	0.0000E 00	0.0000E 00	0.0000E 00	0.83330,2556
1	1	3	0.0000E 00	0.0000E 00	0.0000E 00	0.83330,1654	2	1	3	0.0000E 00	0.0000E 00	0.0000E 00	0.83330,1720
3	1	3	0.0000E 00	0.0000E 00	0.0000E 00	0.83330,1644	4	1	3	0.0000E 00	0.0000E 00	0.0000E 00	0.83330,1863
5	1	3	0.0000E 00	0.0000E 00	0.0000E 00	0.83330,1835	6	1	3	0.0000E 00	0.0000E 00	0.0000E 00	0.83330,1629
7	1	3	0.0000E 00	0.0000E 00	0.0000E 00	0.83330,1668	8	1	3	0.0000E 00	0.0000E 00	0.0000E 00	0.83330,1653
9	1	3	0.0000E 00	0.0000E 00	0.0000E 00	0.83330,1654	10	1	3	0.0000E 00	0.0000E 00	0.0000E 00	0.83330,1654
1	2	3	0.2000E 00	0.0000E 00	0.0000E 00	0.83330,1654	2	2	3	0.7214E-01	0.3074E-01	0.1179E-01	0.83330,1711
3	2	3	-0.1406E-01	0.4601E-01	0.7703E-03	0.83330,1648	4	2	3	-0.3996E-02	0.1474E 00	-0.2324E-01	0.83330,1805
5	2	3	0.1249E 00	0.1398E 00	-0.2234E-01	0.83330,1783	6	2	3	0.1315E 00	0.3911E-01	0.5003E-03	0.83330,1634
7	2	3	0.6143E-01	0.1772E-01	0.3719E-02	0.83330,1667	8	2	3	0.5345E-01	0.4376E-02	0.5023E-03	0.83330,1655
9	2	3	0.4335E-01	0.1654E-02	0.3579E-03	0.83330,1655	10	2	3	0.4335E-01	0.1654E-02	0.3579E-03	0.83330,1655
1	3	3	0.2000E 00	0.0000E 00	0.0000E 00	0.83330,1654	2	3	3	0.9961E-01	0.4083E-01	0.1012E-01	0.83330,1699
3	3	3	0.5128E-01	0.6376E-01	0.1209E-02	0.83330,1682	4	3	3	0.6225E-01	0.1608E 00	-0.4140E-02	0.83330,1714
5	3	3	0.1323E 00	0.1527E 00	-0.4567E-02	0.83330,1704	6	3	3	0.1444E 00	0.5620E-01	-0.1836E-03	0.83330,1668
7	3	3	0.1081E 00	0.2944E-01	0.2302E-02	0.83330,1669	8	3	3	0.9259E-01	0.1042E-01	0.9472E-03	0.83330,1660
9	3	3	0.7921E-01	0.4641E-02	0.3848E-03	0.83330,1656	10	3	3	0.7921E-01	0.4641E-02	0.3848E-03	0.83330,1656
1	4	3	0.2000E 00	0.0000E 00	0.0000E 00	0.83330,1654	2	4	3	0.1189E 00	0.3434E-01	0.9354E-02	0.83330,1693
3	4	3	0.9065E-01	0.6362E-01	0.2776E-02	0.83330,1683	4	4	3	0.9593E-01	0.1077E 00	0.2509E-02	0.83330,1701
5	4	3	0.1338E 00	0.1044E 00	0.1573E-02	0.83330,1694	6	4	3	0.1432E 00	0.6081E-01	0.8266E-03	0.83330,1671
7	4	3	0.1234E 00	0.3519E-01	0.2080E-02	0.83330,1667	8	4	3	0.1091E 00	0.1603E-01	0.1157E-02	0.83330,1660
9	4	3	0.9705E-01	0.7677E-02	0.4613E-03	0.83330,1656	10	4	3	0.9705E-01	0.7677E-02	0.4613E-03	0.83330,1656
1	5	3	0.2000E 00	0.0000E 00	0.0000E 00	0.83330,1654	2	5	3	0.1248E 00	0.1697E-01	0.8705E-02	0.83330,1689
3	5	3	0.1067E 00	0.4344E-01	0.4353E-02	0.83330,1679	4	5	3	0.1022E 00	0.6853E-01	0.3885E-02	0.83330,1683
5	5	3	0.1144E 00	0.6846E-01	0.2936E-02	0.83330,1679	6	5	3	0.1175E 00	0.4686E-01	0.2300E-02	0.83330,1669
7	5	3	0.1080E 00	0.2932E-01	0.2225E-02	0.83330,1664	8	5	3	0.9795E-01	0.1504E-01	0.1436E-02	0.83330,1659
9	5	3	0.8937E-01	0.7659E-02	0.6154E-03	0.83330,1656	10	5	3	0.8937E-01	0.7659E-02	0.6154E-03	0.83330,1656
1	6	3	0.2000E 00	0.0000E 00	0.0000E 00	0.83330,1654	2	6	3	0.1009E 00	0.1630E-02	0.8830E-02	0.83330,1698
3	6	3	0.3498E-01	0.1827E-01	0.4619E-02	0.83330,1675	4	6	3	0.7177E-01	0.2967E-01	0.4783E-02	0.83330,1675
5	6	3	0.7142E-01	0.3101E-01	0.3835E-02	0.83330,1671	6	6	3	0.6997E-01	0.2384E-01	0.2910E-02	0.83330,1665
7	6	3	0.6478E-01	0.1592E-01	0.2225E-02	0.83330,1661	8	6	3	0.5906E-01	0.8799E-02	0.1398E-02	0.83330,1658
9	6	3	0.5428E-01	0.4690E-02	0.6608E-03	0.83330,1656	10	6	3	0.5428E-01	0.4690E-02	0.6608E-03	0.83330,1655
1	7	3	0.0000E 00	0.0000E 00	0.0000E 00	0.83330,1654	2	7	3	0.0000E 00	0.0000E 00	0.0000E 00	0.83330,1704
3	7	3	0.0000E 00	0.0000E 00	0.0000E 00	0.83330,1675	4	7	3	0.0000E 00	0.0000E 00	0.0000E 00	0.83330,1674
5	7	3	0.0000E 00	0.0000E 00	0.0000E 00	0.83330,1669	6	7	3	0.0000E 00	0.0000E 00	0.0000E 00	0.83330,1664
7	7	3	0.0000E 00	0.0000E 00	0.0000E 00	0.83330,1659	8	7	3	0.0000E 00	0.0000E 00	0.0000E 00	0.83330,1657
9	7	3	0.0000E 00	0.0000E 00	0.0000E 00	0.83330,1655	10	7	3	0.0000E 00	0.0000E 00	0.0000E 00	0.83330,1654

1	1	4	0.0000E	00	0.0000E	00	0.0000E	000,83330,0752	2	1	4	0.0000E	00	0.0000E	00	0.0000E	000,83330,0752		
3	1	4	0.0000E	00	0.0000E	00	0.0000E	000,83330,0752	4	1	4	0.6036E-05	0.1000E	01	0.0000E	000,83330,0752			
5	1	4	0.6036E-05	0.1000E	01	0.0000E	000,83330,0752	6	1	4	0.0000E	00	0.0000E	00	0.0000E	000,83330,0752			
7	1	4	0.0000E	00	0.0000E	00	0.0000E	000,83330,0752	8	1	4	0.0000E	00	0.0000E	00	0.0000E	000,83330,0752		
9	1	4	0.0000E	00	0.0000E	00	0.0000E	000,83330,0752	10	1	4	0.0000E	00	0.0000E	00	0.0000E	000,83330,0752		
1	2	4	0.2000E	00	0.0000E	00	0.0000E	000,83330,0752	2	2	4	0.1789E	00	0.5240E-04	0.2131E-020	0.83330,0752			
3	2	4	0.1790E	00	0.1123E-04	0.2668E-040	0.83330,0752	4	2	4	0.1790E	00	0.1053E-00	0.4560E-010	0.83330,0752				
5	2	4	0.1789E	00	0.1053E-00	0.4560E-010	0.83330,0752	6	2	4	0.1789E	00	0.1088E-04	0.2447E-040	0.83330,0752				
7	2	4	0.1789E	00	0.8916E-07	0.1487E-040	0.83330,0752	8	2	4	0.1789E	00	0.8916E-07	0.3423E-050	0.83330,0752				
9	2	4	0.1789E	00	0.4641E-06	0.1426E-060	0.83330,0752	10	2	4	0.1789E	00	0.4641E-06	0.1426E-060	0.83330,0752				
1	3	4	0.2000E	00	0.0000E	00	0.0000E	000,83330,0752	2	3	4	0.2000E	00	0.4446E-04	0.8777E-040	0.83330,0752			
3	3	4	0.1999E	00	0.8090E-05	0.6413E-060	0.83330,0752	4	3	4	0.1999E	00	0.4913E-04	0.4721E-020	0.83330,0752				
5	3	4	0.2001E	00	0.4874E-04	0.4721E-020	0.83330,0752	6	3	4	0.2001E	00	0.7548E-05	0.1054E-050	0.83330,0752				
7	3	4	0.2000E	00	0.4142E-06	0.4500E-050	0.83330,0752	8	3	4	0.2000E	00	0.7328E-06	0.1674E-060	0.83330,0752				
9	3	4	0.2000E	00	0.3663E-06	0.8379E-070	0.83330,0752	10	3	4	0.2000E	00	0.3663E-06	0.8379E-070	0.83330,0752				
1	4	4	0.2000E	00	0.0000E	00	0.0000E	000,83330,0752	2	4	4	0.2000E	00	0.3145E-06	0.8594E-040	0.83330,0752			
3	4	4	0.2000E	00	0.1738E-06	0.1738E-06	0.83330,0752	4	4	4	0.2000E	00	0.9346E-04	0.1244E-040	0.83330,0752				
5	4	4	0.2000E	00	0.9346E-04	0.1197E-040	0.83330,0752	6	4	4	0.2000E	00	0.0000E	00	0.4100E-050	0.83330,0752			
7	4	4	0.2000E	00	0.6420E-06	0.3825E-070	0.83330,0752	8	4	4	0.2000E	00	0.6954E-06	0.4044E-060	0.83330,0752				
9	4	4	0.2000E	00	0.3076E-06	0.1730E-060	0.83330,0752	10	4	4	0.2000E	00	0.3076E-06	0.1730E-060	0.83330,0752				
1	5	4	0.2000E	00	0.0000E	00	0.0000E	000,83330,0752	2	5	4	0.2000E	00	0.1739E-04	0.7751E-040	0.83330,0752			
3	5	4	0.2000E	00	0.6152E-06	0.5002E-060	0.83330,0752	4	5	4	0.2000E	00	0.2619E-05	0.4345E-050	0.83330,0752				
5	5	4	0.2000E	00	0.2619E-06	0.3828E-050	0.83330,0752	6	5	4	0.2000E	00	0.2734E-06	0.1896E-060	0.83330,0752				
7	5	4	0.2000E	00	0.1138E-06	0.4365E-060	0.83330,0752	8	5	4	0.2000E	00	0.0000E	00	0.3199E-060	0.83330,0752			
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1	6	4	0.2000E	00	0.0000E	00	0.0000E	000,83330,0752	2	6	4	0.1888E	00	0.2217E-04	0.1164E-020	0.83330,0752			
3	6	4	0.1888E	00	0.8533E-06	0.1226E-050	0.83330,0752	4	6	4	0.1888E	00	0.1448E-06	0.4911E-050	0.83330,0752				
5	6	4	0.1888E	00	0.1448E-06	0.3656E-060	0.83330,0752	6	6	4	0.1888E	00	0.4241E-06	0.1136E-060	0.83330,0752				
7	6	4	0.1888E	00	0.3517E-06	0.1259E-050	0.83330,0752	8	6	4	0.1888E	00	0.6730E-07	0.4540E-070	0.83330,0752				
9	6	4	0.1888E	00	0.2121E-06	0.6036E-070	0.83330,0752	10	6	4	0.1888E	00	0.2121E-06	0.6036E-070	0.83330,0752				
1	7	4	0.0000E	00	0.0000E	00	0.0000E	000,83330,0752	2	7	4	0.0000E	00	0.0000E	00	0.0000E	000,83330,0752		
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5	7	4	0.0000E	00	0.0000E	00	0.0000E	000,83330,0752	6	7	4	0.0000E	00	0.0000E	00	0.0000E	000,83330,0752		
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5	1	5	0.6036E-05	0.1000E	01	0.0000E	000,83330,0175	0.3500E-01	6	1	5	0.0000E	00	0.0000E	00	0.0000E	000,83330,0175	0.3500E-01	
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7	3	5	0.2000E	00	0.0000E	00	0.4500E-050.83330.0175	0.3500E-01	8	3	5	0.2000E	00	0.0000E	00	0.1676E-060.83330.0175	0.3500E-01		
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1	4	5	0.2000E	00	0.0000E	00	0.0000E	000.83330.0175	0.3500E-01	2	4	5	0.2000E	00	0.0000E	00	0.8584E-040.83330.0175	0.3500E-01	
3	4	5	0.2000E	00	0.0000E	00	-0.3708E-050.83330.0175	0.3500E-01	4	4	5	0.2000E	00	0.0000E	00	0.1244E-040.83330.0175	0.3500E-01		
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5	5	5	0.2000E	00	0.0000E	00	0.3828E-050.83330.0175	0.3500E-01	6	5	5	0.2000E	00	0.0000E	00	0.1896E-060.83330.0175	0.3500E-01		
7	5	5	0.2000E	00	0.0000E	00	-0.4365E-060.83330.0175	0.3500E-01	8	5	5	0.2000E	00	0.0000E	00	-0.3189E-060.83330.0175	0.3500E-01		
9	5	5	0.2000E	00	0.0000E	00	-0.5757E-070.83330.0175	0.3500E-01	10	5	5	0.2000E	00	0.0000E	00	0.0000E	000.83330.0175	0.3500E-01	
1	6	5	0.2000E	00	0.0000E	00	0.0000E	000.83330.0175	0.3500E-01	2	6	5	0.1888E	00	0.0000E	00	0.1472E-020.83330.0175	0.3500E-01	
3	6	5	0.1888E	00	0.0000E	00	0.1226E-050.83330.0175	0.3500E-01	4	6	5	0.1888E	00	0.0000E	00	0.4911E-050.83330.0175	0.3500E-01		
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9	6	5	0.1888E	00	0.0000E	00	-0.6036E-070.83330.0175	0.3500E-01	10	6	5	0.1888E	00	0.0000E	00	0.0000E	000.83330.0175	0.3500E-01	
1	7	5	0.0000E	00	0.0000E	00	0.0000E	000.83330.0175	0.3500E-01	2	7	5	0.0000E	00	0.0000E	00	0.0000E	000.83330.0175	0.3500E-01
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5	7	5	0.0000E	00	0.0000E	00	0.0000E	000.83330.0175	0.3500E-01	6	7	5	0.0000E	00	0.0000E	00	0.0000E	000.83330.0175	0.3500E-01
7	7	5	0.0000E	00	0.0000E	00	0.0000E	000.83330.0175	0.3500E-01	8	7	5	0.0000E	00	0.0000E	00	0.0000E	000.83330.0175	0.3500E-01
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Wellington Business Pa

TIME= 0.4000E 00

DT = 0.8000E-02

ITERATION NUMBER= 50

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7	1	0.0000E 00	0.0000E 00	0.0000E 00	000.83330,3308			8	1	1	0.0000E 00	0.0000E 00	0.0000E 00	000.83330,3308		
9	1	0.0000E 00	0.0000E 00	0.0000E 00	000.83330,3308			10	1	1	0.0000E 00	0.0000E 00	0.0000E 00	000.83330,3308		
1	2	0.0000E 00	0.0000E 00	0.0000E 00	000.83330,3308			2	2	1	0.0000E 00	0.0000E 00	0.0000E 00	000.83330,3364		
3	2	0.0000E 00	0.0000E 00	0.0000E 00	000.83330,3304			4	2	1	0.0000E 00	0.0000E 00	0.0000E 00	000.83330,3455		
5	2	0.0000E 00	0.0000E 00	0.0000E 00	000.83330,3434			6	2	1	0.0000E 00	0.0000E 00	0.0000E 00	000.83330,3290		
7	2	0.0000E 00	0.0000E 00	0.0000E 00	000.83330,3320			8	2	1	0.0000E 00	0.0000E 00	0.0000E 00	000.83330,3309		
9	2	0.0000E 00	0.0000E 00	0.0000E 00	000.83330,3309			10	2	1	0.0000E 00	0.0000E 00	0.0000E 00	000.83330,3308		
1	3	0.0000E 00	0.0000E 00	0.0000E 00	000.83330,3308			2	3	1	0.0000E 00	0.0000E 00	0.0000E 00	000.83330,3352		
3	3	0.0000E 00	0.0000E 00	0.0000E 00	000.83330,3336			4	3	1	0.0000E 00	0.0000E 00	0.0000E 00	000.83330,3368		
5	3	0.0000E 00	0.0000E 00	0.0000E 00	000.83330,3358			6	3	1	0.0000E 00	0.0000E 00	0.0000E 00	000.83330,3322		
7	3	0.0000E 00	0.0000E 00	0.0000E 00	000.83330,3323			8	3	1	0.0000E 00	0.0000E 00	0.0000E 00	000.83330,3314		
9	3	0.0000E 00	0.0000E 00	0.0000E 00	000.83330,3310			10	3	1	0.0000E 00	0.0000E 00	0.0000E 00	000.83330,3308		
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3	4	0.0000E 00	0.0000E 00	0.0000E 00	000.83330,3337			4	4	1	0.0000E 00	0.0000E 00	0.0000E 00	000.83330,3354		
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1	5	0.0000E 00	0.0000E 00	0.0000E 00	000.83330,3308			2	5	1	0.0000E 00	0.0000E 00	0.0000E 00	000.83330,3343		
3	5	0.0000E 00	0.0000E 00	0.0000E 00	000.83330,3333			4	5	1	0.0000E 00	0.0000E 00	0.0000E 00	000.83330,3337		
5	5	0.0000E 00	0.0000E 00	0.0000E 00	000.83330,3332			6	5	1	0.0000E 00	0.0000E 00	0.0000E 00	000.83330,3323		
7	5	0.0000E 00	0.0000E 00	0.0000E 00	000.83330,3318			8	5	1	0.0000E 00	0.0000E 00	0.0000E 00	000.83330,3313		
9	5	0.0000E 00	0.0000E 00	0.0000E 00	000.83330,3310			10	5	1	0.0000E 00	0.0000E 00	0.0000E 00	000.83330,3308		
1	6	0.0000E 00	0.0000E 00	0.0000E 00	000.83330,3308			2	6	1	0.0000E 00	0.0000E 00	0.0000E 00	000.83330,3352		
3	6	0.0000E 00	0.0000E 00	0.0000E 00	000.83330,3329			4	6	1	0.0000E 00	0.0000E 00	0.0000E 00	000.83330,3329		
5	6	0.0000E 00	0.0000E 00	0.0000E 00	000.83330,3324			6	6	1	0.0000E 00	0.0000E 00	0.0000E 00	000.83330,3319		
7	6	0.0000E 00	0.0000E 00	0.0000E 00	000.83330,3315			8	6	1	0.0000E 00	0.0000E 00	0.0000E 00	000.83330,3311		
9	6	0.0000E 00	0.0000E 00	0.0000E 00	000.83330,3309			10	6	1	0.0000E 00	0.0000E 00	0.0000E 00	000.83330,3308		
1	7	0.0000E 00	0.0000E 00	0.0000E 00	000.83330,3308			2	7	1	0.0000E 00	0.0000E 00	0.0000E 00	000.83330,3308		
3	7	0.0000E 00	0.0000E 00	0.0000E 00	000.83330,3308			4	7	1	0.0000E 00	0.0000E 00	0.0000E 00	000.83330,3308		
5	7	0.0000E 00	0.0000E 00	0.0000E 00	000.83330,3308			6	7	1	0.0000E 00	0.0000E 00	0.0000E 00	000.83330,3308		
7	7	0.0000E 00	0.0000E 00	0.0000E 00	000.83330,3308			8	7	1	0.0000E 00	0.0000E 00	0.0000E 00	000.83330,3308		
9	7	0.0000E 00	0.0000E 00	0.0000E 00	000.83330,3308			10	7	1	0.0000E 00	0.0000E 00	0.0000E 00	000.83330,3308		
1	1	2	0.0000E 00	0.0000E 00	0.0000E 00	000.83330,2556		2	1	2	0.0000E 00	0.0000E 00	0.0000E 00	000.83330,2620		
3	1	2	0.0000E 00	0.0000E 00	0.0000E 00	000.83330,2545		4	1	2	0.0000E 00	0.0000E 00	0.0000E 00	000.83330,2753		
5	1	2	0.0000E 00	0.0000E 00	0.0000E 00	000.83330,2726		6	1	2	0.0000E 00	0.0000E 00	0.0000E 00	000.83330,2529		
7	1	2	0.0000E 00	0.0000E 00	0.0000E 00	000.83330,2568		8	1	2	0.0000E 00	0.0000E 00	0.0000E 00	000.83330,2555		
9	1	2	0.0000E 00	0.0000E 00	0.0000E 00	000.83330,2556		10	1	2	0.0000E 00	0.0000E 00	0.0000E 00	000.83330,2556		
1	2	2	0.2000E 00	0.0000E 00	0.0000E 00	000.83330,2556		2	2	2	0.6389E-01	0.2101E-01	0.5704E-02	000.83330,2613		
3	2	2	-0.2045E-01	0.2558E-01	0.8642E-03	000.83330,2552		4	2	2	-0.1300E-01	0.1095E 00	-0.9524E-02	000.83330,2714		
5	2	2	0.1069E 00	0.1020E 00	-0.9140E-02	000.83330,2683		6	2	2	0.1121E 00	0.1818E-01	0.6971E-03	000.83330,2538		
7	2	2	0.4603E-01	0.6755E-02	0.1893E-02	000.83330,2569		8	2	2	0.3902E-01	-0.4089E-03	0.3348E-03	000.83330,2557		
9	2	2	0.3006E-01	-0.6560E-03	0.2161E-03	000.83330,2557		10	2	2	0.3006E-01	-0.6560E-03	0.2161E-03	000.83330,2557		

1	3	2	0.2000E 00	0.0000E 00	0.0000E 000	83330,2556	2	3	2	0.8566E-01	0.2867E-01	0.4890E-020	83330,2601
3	3	2	0.3433E-01	0.3993E-01	0.6851E-030	83330,2584	4	3	2	0.4129E-01	0.1224E 00	0.1838E-020	83330,2616
5	3	2	0.1048E 00	0.1145E 00	0.2058E-020	83330,2606	6	3	2	0.1145E 00	0.3161E-01	0.5649E-040	83330,2570
7	3	2	0.8330E-01	0.1513E-01	0.1027E-020	83330,2571	8	3	2	0.6664E-01	0.3573E-02	0.4179E-030	83330,2562
9	3	2	0.5484E-01	0.1144E-02	0.1777E-030	83330,2558	10	3	2	0.5484E-01	0.1144E-02	0.1777E-030	83330,2558
1	4	2	0.2000E 00	0.0000E 00	0.0000E 000	83330,2556	2	4	2	0.1010E 00	0.2400E-01	0.4541E-020	83330,2595
3	4	2	0.6709E-01	0.4368E-01	0.1255E-020	83330,2585	4	4	2	0.6820E-01	0.7987E-01	0.9583E-030	83330,2603
5	4	2	0.1011E 00	0.7639E-01	0.5013E-030	83330,2596	6	4	2	0.1084E 00	0.3984E-01	0.2179E-030	83330,2573
7	4	2	0.3984E-01	0.2205E-01	0.8108E-030	83330,2569	8	4	2	0.7724E-01	0.9166E-02	0.4604E-030	83330,2562
9	4	2	0.6671E-01	0.4001E-02	0.1768E-030	83330,2558	10	4	2	0.6671E-01	0.4001E-02	0.1768E-030	83330,2558
1	5	2	0.2000E 00	0.0000E 00	0.0000E 000	83330,2556	2	5	2	0.1066E 00	0.1065E-01	0.4230E-020	83330,2591
3	5	2	0.2230E-01	0.3091E-01	0.1913E-020	83330,2582	4	5	2	0.7480E-01	0.5154E-01	0.1550E-020	83330,2585
5	5	2	0.2435E-01	0.5118E-01	0.1085E-020	83330,2581	6	5	2	0.8629E-01	0.3315E-01	0.8251E-030	83330,2571
7	5	2	0.7738E-01	0.2022E-01	0.8418E-030	83330,2566	8	5	2	0.6854E-01	0.9991E-02	0.5493E-030	83330,2561
9	5	2	0.6192E-01	0.4836E-02	0.2284E-030	83330,2558	10	5	2	0.6102E-01	0.4836E-02	0.2284E-030	83330,2558
1	6	2	0.2000E 00	0.0000E 00	0.0000E 000	83330,2556	2	6	2	0.8820E-01	0.7134E-03	0.4149E-020	83330,2600
3	6	2	0.6840E-01	0.1305E-01	0.2000E-020	83330,2578	4	6	2	0.5406E-01	0.2260E-01	0.1960E-020	83330,2577
5	6	2	0.5283E-01	0.2369E-01	0.1508E-020	83330,2573	6	6	2	0.5109E-01	0.1773E-01	0.1117E-020	83330,2567
7	6	2	0.4627E-01	0.1169E-01	0.8584E-030	83330,2563	8	6	2	0.4119E-01	0.6343E-02	0.5409E-030	83330,2560
9	6	2	0.3697E-01	0.3273E-02	0.2498E-030	83330,2558	10	6	2	0.3697E-01	0.3273E-02	0.2498E-030	83330,2558
1	7	2	0.0000E 00	0.0000E 00	0.0000E 000	83330,2556	2	7	2	0.0000E 00	0.0000E 00	0.0000E 000	83330,2605
3	7	2	0.0000E 00	0.0000E 00	0.0000E 000	83330,2577	4	7	2	0.0000E 00	0.0000E 00	0.0000E 000	83330,2575
5	7	2	0.0000E 00	0.0000E 00	0.0000E 000	83330,2571	6	7	2	0.0000E 00	0.0000E 00	0.0000E 000	83330,2565
7	7	2	0.0000E 00	0.0000E 00	0.0000E 000	83330,2561	8	7	2	0.0000E 00	0.0000E 00	0.0000E 000	83330,2559
9	7	2	0.0000E 00	0.0000E 00	0.0000E 000	83330,2557	10	7	2	0.0000E 00	0.0000E 00	0.0000E 000	83330,2556
1	1	3	0.0000E 00	0.0000E 00	0.0000E 000	83330,1654	2	1	3	0.0000E 00	0.0000E 00	0.0000E 000	83330,1720
3	1	3	0.0000E 00	0.0000E 00	0.0000E 000	83330,1644	4	1	3	0.0000E 00	0.0000E 00	0.0000E 000	83330,1863
5	1	3	0.0000E 00	0.0000E 00	0.0000E 000	83330,1835	6	1	3	0.0000E 00	0.0000E 00	0.0000E 000	83330,1629
7	1	3	0.0000E 00	0.0000E 00	0.0000E 000	83330,1668	8	1	3	0.0000E 00	0.0000E 00	0.0000E 000	83330,1653
9	1	3	0.0000E 00	0.0000E 00	0.0000E 000	83330,1654	10	1	3	0.0000E 00	0.0000E 00	0.0000E 000	83330,1654
1	2	3	0.2000E 00	0.0000E 00	0.0000E 000	83330,1654	2	2	3	0.7214E-01	0.3074E-01	0.1179E-010	83330,1711
3	2	3	0.1426E-01	0.4601E-01	0.7703E-030	83330,1648	4	2	3	0.3996E-02	0.1474E 00	0.2324E-010	83330,1805
5	2	3	0.1249E 00	0.1398E 00	0.2234E-010	83330,1783	6	2	3	0.1315E 00	0.3911E-01	0.5003E-030	83330,1634
7	2	3	0.6143E-01	0.1772E-01	0.3719E-020	83330,1667	8	2	3	0.5345E-01	0.4376E-02	0.5023E-030	83330,1655
9	2	3	0.4335E-01	0.1654E-02	0.3579E-030	83330,1655	10	2	3	0.4335E-01	0.1654E-02	0.3579E-030	83330,1655
1	3	3	0.2000E 00	0.0000E 00	0.0000E 000	83330,1654	2	3	3	0.9961E-01	0.4083E-01	0.1012E-010	83330,1699
3	3	3	0.5126E-01	0.6376E-01	0.1209E-020	83330,1682	4	3	3	0.6225E-01	0.1608E 00	0.4149E-020	83330,1714
5	3	3	0.1323E 00	0.1527E 00	0.4567E-020	83330,1704	6	3	3	0.1444E 00	0.5620E-01	0.1836E-030	83330,1668
7	3	3	0.1081E 00	0.2944E-01	0.2392E-020	83330,1669	8	3	3	0.9259E-01	0.1042E-01	0.9472E-030	83330,1660
9	3	3	0.7921E-01	0.4641E-02	0.3848E-030	83330,1656	10	3	3	0.7921E-01	0.4641E-02	0.3848E-030	83330,1656
1	4	3	0.2000E 00	0.0000E 00	0.0000E 000	83330,1654	2	4	3	0.1189E 00	0.3434E-01	0.9354E-020	83330,1693
3	4	3	0.9065E-01	0.6362E-01	0.2776E-020	83330,1683	4	4	3	0.9593E-01	0.1077E 00	0.2509E-020	83330,1701
5	4	3	0.1338E 00	0.1044E 00	0.1573E-020	83330,1694	6	4	3	0.1432E 00	0.6081E-01	0.8266E-030	83330,1671
7	4	3	0.1234E 00	0.3519E-01	0.2080E-020	83330,1667	8	4	3	0.1091E 00	0.1603E-01	0.1197E-020	83330,1660
9	4	3	0.9705E-01	0.7677E-02	0.4613E-030	83330,1656	10	4	3	0.9705E-01	0.7677E-02	0.4613E-030	83330,1656

5 3	0.2000E 00	0.0000E 00	0.0000E 00	0.83330,1654	2 5 3	0.1248E 00	0.1697E-01	0.8705E-02	0.83330,1689
5 3	0.1067E 00	0.4344E-01	0.4353E-02	0.83330,1679	4 5 3	0.1022E 00	0.6853E-01	0.3885E-02	0.83330,1683
5 3	0.1144E 00	0.6846E-01	0.2936E-02	0.83330,1679	6 5 3	0.1175E 00	0.4686E-01	0.2300E-02	0.83330,1669
5 3	0.1060E 00	0.2932E-01	0.2225E-02	0.83330,1664	8 5 3	0.9795E-01	0.1504E-01	0.1436E-02	0.83330,1659
5 3	0.8937E-01	0.7659E-02	0.6154E-03	0.83330,1656	10 5 3	0.8937E-01	0.7659E-02	0.6154E-03	0.83330,1656
6 3	0.2000E 00	0.0000E 00	0.0000E 00	0.83330,1654	2 6 3	0.1009E 00	0.1630E-02	0.8830E-02	0.83330,1698
6 3	0.3498E-01	0.1827E-01	0.4619E-02	0.83330,1675	4 6 3	0.7177E-01	0.2967E-01	0.4783E-02	0.83330,1675
6 3	0.7142E-01	0.3101E-01	0.3835E-02	0.83330,1671	6 6 3	0.6997E-01	0.2384E-01	0.2910E-02	0.83330,1665
6 3	0.6478E-01	0.1592E-01	0.2225E-02	0.83330,1661	8 6 3	0.5906E-01	0.8799E-02	0.1398E-02	0.83330,1658
6 3	0.5428E-01	0.4690E-02	0.6608E-03	0.83330,1656	10 6 3	0.5428E-01	0.4690E-02	0.6608E-03	0.83330,1655
7 3	0.0000E 00	0.0000E 00	0.0000E 00	0.83330,1654	2 7 3	0.0000E 00	0.0000E 00	0.0000E 00	0.83330,1704
7 3	0.0000E 00	0.0000E 00	0.0000E 00	0.83330,1675	4 7 3	0.0000E 00	0.0000E 00	0.0000E 00	0.83330,1674
7 3	0.0000E 00	0.0000E 00	0.0000E 00	0.83330,1669	6 7 3	0.0000E 00	0.0000E 00	0.0000E 00	0.83330,1664
7 3	0.0000E 00	0.0000E 00	0.0000E 00	0.83330,1659	8 7 3	0.0000E 00	0.0000E 00	0.0000E 00	0.83330,1657
7 3	0.0000E 00	0.0000E 00	0.0000E 00	0.83330,1655	10 7 3	0.0000E 00	0.0000E 00	0.0000E 00	0.83330,1654
1 4	0.0000E 00	0.0000E 00	0.0000E 00	0.83330,0752	2 1 4	0.0000E 00	0.0000E 00	0.0000E 00	0.83330,0927
1 4	0.0000E 00	0.0000E 00	0.0000E 00	0.83330,0768	4 1 4	0.6036E-05	0.1000E 01	0.0000E 00	0.83330,0752
1 4	0.6036E-05	0.1000E 01	0.0000E 00	0.83330,0752	6 1 4	0.0000E 00	0.0000E 00	0.0000E 00	0.83330,0755
1 4	0.0000E 00	0.0000E 00	0.0000E 00	0.83330,0776	8 1 4	0.0000E 00	0.0000E 00	0.0000E 00	0.83330,0754
1 4	0.0000E 00	0.0000E 00	0.0000E 00	0.83330,0753	10 1 4	0.0000E 00	0.0000E 00	0.0000E 00	0.83330,0752
2 4	0.2000E 00	0.0000E 00	0.0000E 00	0.83330,0752	2 2 4	0.7200E-01	0.6538E-01	0.1643E-01	0.83330,0810
2 4	0.1863E-01	0.1574E 00	0.3379E-02	0.83330,0743	4 2 4	0.5890E-02	0.4563E 00	0.6305E-03	0.83330,0907
2 4	0.1336E 00	0.4591E 00	0.1910E-02	0.83330,0883	6 2 4	0.1424E 00	0.1578E 00	0.3965E-02	0.83330,0730
2 4	0.6719E-01	0.6109E-01	0.3720E-02	0.83330,0766	8 2 4	0.5872E-01	0.1970E-01	0.1217E-03	0.83330,0753
2 4	0.4737E-01	0.7989E-02	0.1526E-03	0.83330,0752	10 2 4	0.4737E-01	0.7989E-02	0.1526E-03	0.83330,0752
3 4	0.2000E 00	0.0000E 00	0.0000E 00	0.83330,0752	2 3 4	0.9885E-01	0.7647E-01	0.1572E-01	0.83330,0797
3 4	0.4985E-01	0.1504E 00	0.3918E-02	0.83330,0779	4 3 4	0.6453E-01	0.3182E 00	0.2750E-02	0.83330,0811
3 4	0.1421E 00	0.3128E 00	0.2552E-02	0.83330,0801	6 3 4	0.1572E 00	0.1483E 00	0.2134E-02	0.83330,0766
3 4	0.1185E 00	0.7400E-01	0.4102E-02	0.83330,0757	8 3 4	0.1015E 00	0.2918E-01	0.1532E-02	0.83330,0757
3 4	0.6648E-01	0.1331E-01	0.5601E-03	0.83330,0754	10 3 4	0.8648E-01	0.1331E-01	0.5601E-03	0.83330,0754
4 4	0.2000E 00	0.0000E 00	0.0000E 00	0.83330,0752	2 4 4	0.1191E 00	0.6012E-01	0.1489E-01	0.83330,0791
4 4	0.9194E-01	0.1176E 00	0.6286E-02	0.83330,0780	4 4 4	0.1007E 00	0.1872E 00	0.8216E-02	0.83330,0799
4 4	0.1443E 00	0.1849E 00	0.6934E-02	0.83330,0793	6 4 4	0.1562E 00	0.1183E 00	0.3711E-02	0.83330,0769
4 4	0.1351E 00	0.6826E-01	0.4389E-02	0.83330,0765	8 4 4	0.1192E 00	0.3210E-01	0.2318E-02	0.83330,0758
4 4	0.1058E 00	0.1584E-01	0.8991E-03	0.83330,0754	10 4 4	0.1058E 00	0.1584E-01	0.8991E-03	0.83330,0754
5 4	0.2000E 00	0.0000E 00	0.0000E 00	0.83330,0752	2 5 4	0.1261E 00	0.3213E-01	0.1366E-01	0.83330,0788
5 4	0.1102E 00	0.7256E-01	0.8034E-02	0.83330,0777	4 5 4	0.1082E 00	0.1076E 00	0.8225E-02	0.83330,0781
5 4	0.1234E 00	0.1079E 00	0.6881E-02	0.83330,0777	6 5 4	0.1280E 00	0.7760E-01	0.5304E-02	0.83330,0767
5 4	0.1180E 00	0.4894E-01	0.4524E-02	0.83330,0762	8 5 4	0.1068E 00	0.2554E-01	0.2787E-02	0.83330,0757
5 4	0.9722E-01	0.1338E-01	0.1223E-02	0.83330,0754	10 5 4	0.9722E-01	0.1338E-01	0.1223E-02	0.83330,0754
6 4	0.2000E 00	0.0000E 00	0.0000E 00	0.83330,0752	2 6 4	0.1025E 00	0.9843E-02	0.1363E-01	0.83330,0796
6 4	0.8850E-01	0.3163E-01	0.8082E-02	0.83330,0773	4 6 4	0.7619E-01	0.4574E-01	0.8804E-02	0.83330,0773

6	4	0.7702E-01	0.4679E-01	0.7389E-020	0.83330,0768	6	6	4	0.7606E-01	0.3658E-01	0.5704E-020	0.83330,0763			
7	6	4	0.7058E-01	0.2450E-01	0.4269E-020	0.83330,0759	8	6	4	0.6425E-01	0.1366E-01	0.2639E-020	0.83330,0755		
9	6	4	0.5895E-01	0.7449E-02	0.1277E-020	0.83330,0753	10	6	4	0.5895E-01	0.7449E-02	0.1277E-020	0.83330,0753		
1	7	4	0.0000E 00	0.0000E 00	0.0000E 000	0.83330,0752	2	7	4	0.0000E 00	0.0000E 00	0.0000E 000	0.83330,0802		
3	7	4	0.0000E 00	0.0000E 00	0.0000E 000	0.83330,0773	4	7	4	0.0000E 00	0.0000E 00	0.0000E 000	0.83330,0773		
5	7	4	0.0000E 00	0.0000E 00	0.0000E 000	0.83330,0768	6	7	4	0.0000E 00	0.0000E 00	0.0000E 000	0.83330,0762		
7	7	4	0.0000E 00	0.0000E 00	0.0000E 000	0.83330,0758	8	7	4	0.0000E 00	0.0000E 00	0.0000E 000	0.83330,0755		
9	7	4	0.0000E 00	0.0000E 00	0.0000E 000	0.83330,0753	10	7	4	0.0000E 00	0.0000E 00	0.0000E 000	0.83330,0752		
1	1	5	0.0000E 00	0.0000E 00	0.0000E 000	0.83330,0175	0.3500E-01	2	1	5	0.0000E 00	0.0000E 00	0.0000E 000	0.83330,0175	0.3500E-01
3	1	5	0.0000E 00	0.0000E 00	0.0000E 000	0.83330,0175	0.3500E-01	4	1	5	0.6036E-05	0.1000E 01	0.0000E 000	0.83330,0175	0.3500E-01
5	1	5	0.4036E-05	0.1000E 01	0.0000E 000	0.83330,0175	0.3500E-01	6	1	5	0.0000E 00	0.0000E 00	0.0000E 000	0.83330,0175	0.3500E-01
7	1	5	0.0000E 00	0.0000E 00	0.0000E 000	0.83330,0175	0.3500E-01	8	1	5	0.0000E 00	0.0000E 00	0.0000E 000	0.83330,0175	0.3500E-01
9	1	5	0.0000E 00	0.0000E 00	0.0000E 000	0.83330,0175	0.3500E-01	10	1	5	0.0000E 00	0.0000E 00	0.0000E 000	0.83330,0175	0.3500E-01
1	2	5	0.2000E 00	0.0000E 00	0.0000E 000	0.83330,0175	0.3500E-01	2	2	5	0.6532E-01	0.1831E-01	0.1968E-010	0.83330,0206	0.4119E-01
3	2	5	-0.7848E-01	0.1496E 00	-0.4531E-020	0.83330,0169	0.3378E-01	4	2	5	-0.5003E-01	0.4579E 00	0.2038E-010	0.83330,0255	0.5097E-01
5	2	5	0.2067E 00	0.4685E 00	0.9457E-020	0.83330,0242	0.4841E-01	6	2	5	0.2190E 00	0.1688E 00	-0.6100E-020	0.83330,0163	0.3252E-01
7	2	5	0.7888E-01	0.4923E-01	0.6023E-020	0.83330,0183	0.3651E-01	8	2	5	0.7044E-01	0.1519E-01	0.3372E-030	0.83330,0176	0.3507E-01
9	2	5	0.5176E-01	0.5997E-02	0.4268E-030	0.83330,0176	0.3508E-01	10	2	5	0.5176E-01	0.5997E-02	0.4089E-030	0.83330,0176	0.3507E-01
1	3	5	0.2000E 00	0.0000E 00	0.0000E 000	0.83330,0175	0.3500E-01	2	3	5	0.7844E-01	0.8176E-01	0.1699E-010	0.83330,0199	0.3975E-01
3	3	5	0.1950E-01	0.1619E 00	0.4832E-020	0.83330,0189	0.3776E-01	4	3	5	0.4814E-01	0.4007E 00	0.7541E-020	0.83330,0205	0.4095E-01
5	3	5	0.1853E 00	0.3922E 00	0.3344E-020	0.83330,0200	0.3989E-01	6	3	5	0.2086E 00	0.1638E 00	0.1581E-020	0.83330,0182	0.3641E-01
7	3	5	0.1422E 00	0.8004E-01	0.5111E-020	0.83330,0183	0.3661E-01	8	3	5	0.1181E 00	0.2901E-01	0.1832E-020	0.83330,0176	0.3568E-01
9	3	5	0.9467E-01	0.1301E-01	0.9612E-030	0.83330,0176	0.3521E-01	10	3	5	0.9467E-01	0.1301E-01	0.9610E-030	0.83330,0176	0.3521E-01
1	4	5	0.2000E 00	0.0000E 00	0.0000E 000	0.83330,0175	0.3500E-01	2	4	5	0.9954E-01	0.7573E-01	0.1795E-010	0.83330,0195	0.3912E-01
3	4	5	0.7530E-01	0.1454E 00	0.7700E-020	0.83330,0189	0.3765E-01	4	4	5	0.9644E-01	0.2444E 00	0.1287E-010	0.83330,0200	0.3991E-01
5	4	5	0.1746E 00	0.2399E 00	0.9331E-020	0.83330,0196	0.3920E-01	6	4	5	0.1939E 00	0.1450E 00	0.4067E-020	0.83330,0184	0.3668E-01
7	4	5	0.1581E 00	0.8296E-01	0.5491E-020	0.83330,0182	0.3636E-01	8	4	5	0.1350E 00	0.3771E-01	0.2785E-020	0.83330,0178	0.3542E-01
9	4	5	0.1145E 00	0.1815E-01	0.1311E-020	0.83330,0177	0.3524E-01	10	4	5	0.1145E 00	0.1815E-01	0.1303E-020	0.83330,0177	0.3524E-01
1	5	5	0.2000E 00	0.0000E 00	0.0000E 000	0.83330,0175	0.3500E-01	2	5	5	0.1130E 00	0.4718E-01	0.1692E-010	0.83330,0194	0.3875E-01
3	5	5	0.1091E 00	0.1011E 00	0.9627E-020	0.83330,0188	0.3758E-01	4	5	5	0.1134E 00	0.1521E 00	0.1106E-010	0.83330,0190	0.3804E-01
5	5	5	0.1429E 00	0.1504E 00	0.8734E-020	0.83330,0188	0.3754E-01	6	5	5	0.1511E 00	0.1033E 00	0.6391E-020	0.83330,0183	0.3654E-01
7	5	5	0.1352E 00	0.6451E-01	0.5628E-020	0.83330,0180	0.3605E-01	8	5	5	0.1188E 00	0.3297E-01	0.3441E-020	0.83330,0178	0.3554E-01
9	5	5	0.1042E 00	0.1634E-01	0.1665E-020	0.83330,0176	0.3522E-01	10	5	5	0.1042E 00	0.1684E-01	0.1627E-020	0.83330,0176	0.3522E-01
1	6	5	0.2000E 00	0.0000E 00	0.0000E 000	0.83330,0175	0.3500E-01	2	6	5	0.9742E-01	0.5013E-01	0.1660E-010	0.83330,0198	0.3966E-01
3	6	5	0.9892E-01	0.6984E-01	0.9132E-020	0.83330,0186	0.3714E-01	4	6	5	0.8495E-01	0.8645E-01	0.1116E-010	0.83330,0186	0.3721E-01
5	6	5	0.3939E-01	0.8234E-01	0.9152E-020	0.83330,0184	0.3673E-01	6	6	5	0.8927E-01	0.6094E-01	0.6928E-020	0.83330,0181	0.3614E-01
7	6	5	0.8075E-01	0.3948E-01	0.5235E-020	0.83330,0179	0.3573E-01	8	6	5	0.7116E-01	0.2136E-01	0.3236E-020	0.83330,0177	0.3538E-01
9	6	5	0.6299E-01	0.1115E-01	0.1649E-020	0.83330,0176	0.3517E-01	10	6	5	0.6299E-01	0.1115E-01	0.1580E-020	0.83330,0176	0.3515E-01
1	7	5	0.0000E 00	0.0000E 00	0.0000E 000	0.83330,0175	0.3500E-01	2	7	5	0.0000E 00	0.0000E 00	0.0000E 000	0.83330,0175	0.3500E-01
3	7	5	0.0000E 00	0.0000E 00	0.0000E 000	0.83330,0175	0.3500E-01	4	7	5	0.0000E 00	0.0000E 00	0.0000E 000	0.83330,0175	0.3500E-01
5	7	5	0.0000E 00	0.0000E 00	0.0000E 000	0.83330,0175	0.3500E-01	6	7	5	0.0000E 00	0.0000E 00	0.0000E 000	0.83330,0175	0.3500E-01
7	7	5	0.0000E 00	0.0000E 00	0.0000E 000	0.83330,0175	0.3500E-01	8	7	5	0.0000E 00	0.0000E 00	0.0000E 000	0.83330,0175	0.3500E-01
9	7	5	0.0000E 00	0.0000E 00	0.0000E 000	0.83330,0175	0.3500E-01	10	7	5	0.0000E 00	0.0000E 00	0.0000E 000	0.83330,0175	0.3500E-01

VITA

The author was born in _____ in Iran. He began attending Fairleigh Dickinson University, Teaneck, New Jersey in 1966 obtaining the B.S.M.E. degree from that university in 1969. He continued at Fairleigh Dickinson University for the master's degree in Mechanical Engineering, which was completed in 1971. He was accepted into the doctoral program at Newark College of Engineering the same year. The development of this dissertation took place between April 1974 and April 1977 at New Jersey Institute of Technology.