Copyright Warning & Restrictions

The copyright law of the United States (Title 17, United States Code) governs the making of photocopies or other reproductions of copyrighted material.

Under certain conditions specified in the law, libraries and archives are authorized to furnish a photocopy or other reproduction. One of these specified conditions is that the photocopy or reproduction is not to be "used for any purpose other than private study, scholarship, or research." If a, user makes a request for, or later uses, a photocopy or reproduction for purposes in excess of "fair use" that user may be liable for copyright infringement,

This institution reserves the right to refuse to accept a copying order if, in its judgment, fulfillment of the order would involve violation of copyright law.

Please Note: The author retains the copyright while the New Jersey Institute of Technology reserves the right to distribute this thesis or dissertation

Printing note: If you do not wish to print this page, then select "Pages from: first page # to: last page #" on the print dialog screen



The Van Houten library has removed some of the personal information and all signatures from the approval page and biographical sketches of theses and dissertations in order to protect the identity of NJIT graduates and faculty.

EFFECT OF LIQUID VISCOSITY ON PERFORMANCE

OF WIRE MESH ENTRAINMENT SEPARATORS

BY

RAYMOND P. VOGEL

A THESIS

PRESENTED IN PARTIAL FULFILLMENT OF

THE REQUIREMENTS FOR THE DEGREE

OF

MASTER OF SCIENCE IN CHEMICAL ENGINEERING

AT

NEWARK COLLEGE OF ENGINEERING

This thesis is to be used only with due regard to the rights of the author(s). Bibliographical references may be noted, but passages must not be copied without permission of the College and without credit being given in subsequent written or published work.

> Newark, New Jersey 1964

...JSTANOT

The knitted wire mesh entrainment separator, componly referred to as a demister⁴, is in common use to separate entrained liquid from vapor streams. However, little quantitative information is available concerning the effect of physical properties of the entrained liquid on allowable gas velocities.

Studies were made on two demister styles to determine the effect of liquid viscosity on demister performance. Liquid viscosities from 0.9 to 12 centipoise were investigated using water, glycerine-water mixtures and heavy no. 2 fuel oil as the test liquids. Air was used as the gas medium in all cases.

Regression analysis of the test data indicates that by increasing viscosity from 0.9 to 12 centipoise the allowable vapor velocity is decreased by only 10%. The effects of other liquid and demister properties on flooding velocity can be approximated by the following proposed equation:

$$v_{FLOOD} = \frac{5.45 (P_L)^{0.47} (\gamma)^{0.20}}{(\alpha/\epsilon^3)^{0.30} (\mu_L)^{0.036} (\Im_L)^{0.11}}$$

* Registered trademark of Otto York Company.

APPROVAL OF THESIS

FOR

DEPARTMENT OF CHEMICAL ENGINEERING NEWARK COLLEGE OF ENGINEERING

ЗY

FACULTY COMMITTEE

APPROVED :

NEWARK, NEW JERSEY

FEBRUARY, 1964

The author expresses his appreciation to the following individuals for their contributions to this research project:

<u>Prof. J.J. Salamone</u>, Chemical Engineering Department, Newark College of Engineering, for his guidance and encouragement as well as the supply of various items of equipment.

<u>Hr. H.F. Schroeder</u>, Esso Research and Engineering Company, who constructed the original test apparatus for his earlier studies.

Messra. S.W. Clark, A.J. Schilling, A.A. Giacobbe, J.Keil California Oil Company, Perth Amboy, N.J., for the supply of additional items of test equipment, laboratory determination of test liquid physical properties and the processing of test data for regression analysis on the IBM 7094 digital computer of Standard Oil of California.

<u>Mr. E.W. Poppele</u>, Otto York Company, Inc., for supplying the essential test demisters as well as for his valuable comments on his earlier experiments and industrial experience.

Mrs. R.P. Vogel, for her many contributions, the latest of which is typing this report.

11

TABLE OF CONTENTS

	rake No.
abstract	11
Acknowledgements	TA
List of Figures	VII
List of Tables	VIII
Introduction	1
Theory	3
Description of Test Apparatus	6
Air Aandling Section	6
Liquid Handling System	8
Test Column	9
Experimental Procedure	LL
Test Column Preparation	11
Air Section Startup	12
Experimental Data	12
Letermination of Load and Flood Points	12
Liquid Properties	15
Experimental Results	17
Lata Correlation	27
Published Correlations	27
Linear Regression Analysis	30
Conclusions	37
Recommendations	39
Nomenclature	40

V

TABLE OF CONTENTS (Cont'd)

Page No.

Appendix A	Equipment Letails	41
Appendix B	Experimental Data	53
Appendix C	Physical Properties of Test Liquids	-95
Appendix D	Data Correlation	98
References		103

Vl

LIST OF FIGURES

Figure <u>No.</u>	Title	Page No.
1	Equipment Schematic Diagram	7
2	Flooding Velocity, 421 Demister	20
3	Loading Velocity, 421 Demister	21
4	Flooding Velocity, 931 Demister	22
5	Loading Velocity, 931 Demister	23
6	Demister Pressure Drop At Flood Point	25
7	Demister Entrainment Factor $(K_{\rm F})$ At The	28
	Flood Point	
8	Data Correlation Using Proposal Of Poppele	29
9	Regression Equation For Flooding Velocity	33
10	Regression Equation For Loading Velocity	34
A-1	Orifice Calibration	48
A-2	Air Density	49
A-3	Calculated Air Velocities	50
4-4	Rotometer Galibration, Water-Glycerine	51
A-5	Rotometer Calibration, Heavy No. 2 Fuel 011	52
B-1 Thru	Experimental Determination Of Demister	63-94
B-32	Load Point And Flood Point	
C-1	Surface Tension Of Test Liquids	96
C-2	Viscosity Of Glycerine-Water Mixtures	97

Vll

LIST OF TABLES

Table	1	Data Summary	<u>Page No.</u> 18 ,
Table	3-1	Experimental Data	54-62
Thru E	9-9		
Table	D-1	Calculated Data For Figures 7 And 8	99
Table	D-2	Regression Coefficients From Stepwise	100
		Linear Regression Analysis	
Table	D-3	Regression Output, Step 6, Flooding	101
		Velocity	:
Table	D-4	Regression Output, Step 6, Loading	102
		Velocity	

INTROLUCTION

Demisters are commonly used to separate entrained liquid from a vapor stream in equipment such as distillation columns, separator drums and evaporators. Although demisters have proven effective in this service, the effect of variables on demister performance have not been completely defined.

York and Poppele⁽⁴⁾ have summarized the variables which affect allowable gas velocities through mesh demisters. These factors are: (a) gas density, (b) liquid surface tension, viscosity and entrainment loading rate, and (c) specific surface of the wire mesh demister. York⁽³⁾ has proposed application of the Souders-Brown⁽⁶⁾ expression to define the effects of liquid and gas densities on allowable vapor velocities. This expression (equation 1) considers only the variables of gas and liquid densities and is inadequate to predict demister performance in systems where the other variables may have significant effects.

Poppele(2) established the effect of liquid entrainment loading on two demister styles using the water/air system. Schroeder(1) studied the effects of surface tension by using surfactants in the water/air system and verified the work of Poppele(2). Liquid viscosity was the remaining liquid property to evaluate. The purpose of this study was to study the effect of liquid viscosity on allowable demister vapor velocity. Viscosities ranging from 0.9 to 12 centipoises were studied using systems of water/air, glycerine-water/air and heavy no. 2 fuel oil/air. Two demister styles were used and the liquid loading rates were varied from 25 to 400 %/hr.-ft.². Since the density and surface tension of the test liquids were not constant, the effects of these variables as well as viscosity were determined by stepwise linear regression analysis of the experimental results. (2)

THEORY

The benefits derived from the use of a demister result from its ability to coalesce small entrained particles. The larger particles thus formed have entrainment velocities which are higher than the originally entrained particle and can, therefore, settle through the vapor stream to the bulk of the liquid phase.

A dry demister has a very high void fraction and can tolerate high velocities with only small pressure drop. As entrained liquid is coalesced, the free flow area for the vapor decreases and velocities within the demister are higher than the superficial velocity based on total column area. By increasing the vapor velocity through a demister we will eventually reach a point where the upward frictional force of the air on the particle is equal to the downward gravitational force on the particle. Further increase in velocity will result in a net upward force which will carry the particle through the top demister surface. This carryover is called flooding. The flooding velocity is defined as the superficial vapor velocity at which flooding starts.

Equation (1) shows the theoretical velocity at which the net vertical force is zero on a freely suspended liquid droplet.

 $V = K \sqrt{(P_L - P_G)/P_G}$

This equation has been applied by York(3) to define the effect of liquid and gas density on demister

(3)

 (\mathbf{i})

flooding velocity. This equation is not rigorous for direct application to demisters, and values of the entrainment factor (k) must be determined experimentally. The need for experimental evaluation of k is obvious when we consider the following:

- (1) Equation (1) was derived for freely falling suspended droplets. In comisters the droplets are suspended on wire mesh. Therefore, forces between the liquid and wire resulting from surface tension and viscosity will affect the velocity as which a particle will be entrained above the demister.
- (2) The velocity (V) in equation (1) is the superficial vapor velocity based on column diameter. Local velocity within the demister is higher than the superficial vapor velocity depending upon demister properties, liquid load and particle size of the coalesced liquid.
- (3) The value of the entrainment factor (A) is not constant as implied by the equation form. If is directly related to the diameter of the coalesced particle and inversely related to the drag coefficient. Particle diameter is a function of liquid surface tension, liquid density, to a lesser extent of liquid viscosity, and of the relative

(4)

volumes of vapor and liquid⁽⁷⁾. The drag coefficient is a function of Reynolds number and particle sphericity. Reynolds number, in turn, is dependent upon particle diameter.

Until a more rigorous equation is developed for demisters, the effects of variables on allowable vapor velocity must be determined experimentally. Actual performance of demisters and the procedure used to determine demister flooding velocities are discussed in detail under EXPERIMENTAL PROCEDURE.

DESCRIPTION OF TEST APPARATUS

A schematic drawing of the test apparatus is snown in Figure 1. The apparatus is comprised of three basic sections, namely, (1) air handling section, (2) liquid handling system and (3) the test column. Each of these sections is discussed separately below. Additional equipment details are given in Appendix A.

AIR HANDLING SECTION

The purpose of this equipment is to provide a measured quantity of air to the test column. Control of both the temperature and humidity of the air is necessary to minimize vaporization losses of the test liquid.

Air is compressed in turbo blowers (two in parallel) which results in substantial increase in temperature and decrease in humidity. Primary cooling and humidification is obtained by a Water spray into the blower discharge line. Secondary humidification is accomplished in the humidifier (55 gallon drum) where the air is contacted with additional water sprays. The air leaves the humidifier through a wire mesh demister to prevent water carryover into the test column. This treatment results in air to the test column at 80-100°F and 80% humidity.

Air flow is regulated by a butterfly valve in the



(7)

blower discharge line. Flow rate is calculated by standard orifice calculation methods⁽³⁾ from orifice pressure drop measurements. Letails of orifice plate and manometer are given in Appendix ... case 44.

The air lines are 4 inch diameter sheet metal pipe. All seams are sealed with furnace pipe coment to prevent any air leakage.

LIQUID HANDLING SYSTEM

The purpose of this equipment is to deliver the desired quantity of test liquid to the bottom surface of the test demister in a full cone spray.

The bottom section of the test column is used as a reservoir for the test liquid. The liquid is pumped through a nozzle so as to completely impinge upon the bottom surface of the test demister. Liquid temperature is measured with a momenty therefore to serve into the rotometer outlet line. The liquid which drains from the demister drops back to the reservoir and is recirculated.

A bypase line is available between the pump discharge and reservoir. Between runs this line is used to mix test liquid to maintain as uniform a composition as possible (particularly for water-glycerine mixtures). A connection is available for sampling between runs and draining the test liquid when required. (8)

Test liquid is prevented from draining into the air inlet line by a deflector cone mounted to the liquid line 3 inches above the air inlet line.

TEST COLUMN

The test column is comprised of a long bottom section of 9 inch diameter stove gips upon which is attached (air tight) a short 8 inch 0.D. lucite tube. The test demister is contained in the lucite tube about 28 inches above the end of the air inlet line. The liquid spray nozzle is about 8 inches below the test demister. Slight adjustment is made to get complete impingement of liquid on the bottom demister surface.

The demister performance is determined almost entirely on observation of pressure drop. A water manometer is used to measure demister pressure drop.

The major equipment in the air handling section and (1) test column are those constructed by H. F. Schroeder (1). Modifications to his equipment include:

- Piping a second blower in parallel to the first to obtain required air rates to test the 931 demister.
- (2) Installation of the water spray into the blower discharge.

(9)

- (3) Cementing seams in air system to replace tape which had deteriorated with time.
- (4) Revise humidifier water flow scheme.
- (5) Install deflector cone in test column.

EXPERIMENTAL PROCEDURE

TEST COLUMN PREPARATION

The spray nozzle was selected based on the desired liquid loading rate. It was found that the best liquid spray was obtained when the liquid pump was operated at full capacity and control valve wide open. Changes in liquid rate from run to run were therefore accomplished by nozzle changes. After installing the nozzle, liquid was circulated to check spray impingement on the pottom of the demister.

The test demister was then placed in position in the lucite column. To minimize nozzle changes the test runs on the 931 and 421 demisters were usually made consecutively at a given liquid rate. After each test run the demister was removed either to change the nozzle or to change to the other demister.

Test liquid circulation was started. Slight adjustment (closing)f the liquid control valve was usually required to stabilize rotometer reading. This also provided sufficient control to maintain constant liquid rates throughout the run to compensate for reduction in liquid rate due to slight fouling of the nozzle filter.

AIR SECTION STARTUP

Water rate to the secondary spray was adjusted to establish a constant water level in the bottom of the humidifier. The air blower was started with the air control value in the closed position. Air rate was increased to obtain a small orifice differential pressure. The wet bulb thermometer was moistened with water and the air system was prought to equilibrium (80-100°F dry bulb temperature and 80% humidity).

EXPERIMENTAL LATA

Air rate was varied from velocities below the load point to velocities beyond the flooding point. At each air rate the system was allowed to attain equilibrium. Data recorded includes (1) time, (2) liquid rate and temperature, (3) orifice and demister differential pressure, (4) air wet and dry bulb temperature. Experimental data are included in Appendix 3.

DETERMINATION OF LOAD AND FLOOD POINTS.

For each air rate the orifice pressure drop (function of air velocity) was plotted against the demister pressure drop on log-log paper. These plots are shown in Appendix B, Figures BL turu B32. Plots were made during the run to assure that sufficient data points were available to define

(12)

loading and flooding velocities. These plots were also a good indication whether or not equilibrium was obtained.

At air rates below the loading point the log-log plot yields a linear relation with gradual increase in demister pressure drop as air is increased up to loading velocity. The increase is attributed to that associated with increased velocity as well as the accumulation of a larger quantity of liquid within the demsiter.

Increasing velocity beyond the load point results in a more rapid increase of demister pressure drop as the quantity of liquid accumulation increases and the liquid level in the demister increases more rapidly. A linear relation is still obtained but with higher slope.

Futher increases in air rate eventually resulted in liquid carryover and reentrainment above the demister. A number of pressure drops were recorded beyond the point of liquid carryover. It was found that these points also formed a straight line plot but the slope of this line was much lower than the slope obtained below the load point. The quantity of liquid within the demister had reached an equilibrium value and the increased demister pressure drop was due to air velocity increases only.

It was decided to define this second break in the curve as the demister flooding point. This appeared to be (13)

a more quantitative method of defining the flood point which would result in greater repeatability. Earlier investigations by Schroeder (1) and Poppele (2) defined flooding as the inception of liquid break through. Inception of liquid breakthrough can mean different things to different experimenters whereas a break in a curve is quantitative.

After flooding the demister additional data points were obtained between load and flood points as well as below the load point. The data between load and flood points fell along the original curve and aided in defining the flood point.

At velocities below the load point there was a tendency for demister pressure drop to be slightly above the original curve. This is best illustrated in the curves for Runs 33 thru 36 (Figures 3-29 thru 3-32). This higher demister pressure drop is most likely due to the presence of a small quantity of liquid remaining in the demister resulting in less flow area for the air. The load point (velocity) could be affected by as much as 5% depending on whether the data was obtained on a wet or dry demister. Flooding velocity would not be significantly affected.

Worthy of mention is an observation on Run No. 1,

(14)

Figure B-1. At very low velocities (less than 5 ft/sec.) there seems to be a fourth linear portion to the demister pressure drop curve. The effect of air velocity on pressure drop is very small most likely due to the absense of liquid in the test demister.

Equilibrium conditions were reached quickly, usually within one minute, for air rates below the load point and above the flood point. However, demister pressure drop changed very slowly between flood and load points particularly for low liquid rates (55 $\#/hr-ft^2$) on the 931 demister. At times it was necessary to allow thirty minutes to attain steady state.

LIQUID PROPERTIES

Densities of test liquids were obtained between each run. Some evaporation of water from water glycerine mixtures was apparent.

Samples were taken occasionally for determination of surface tension and viscosity. Densities were also obtained, and checked favorably with the first determination.

Surface tension of water was found to decrease with time. Fresh water gave surface tension equivalent to that contained in standard reference books. However, prolonged (15)

testing of the same water sample resulted in decrease in surface tension.

The surface tensions of water-glycerine mixtures were below the surface tension stated in various reference books of physical properties. The most likely cause of lower surface tension is the presence of some unknown contamination. Measurements are considered accurate since checks were made on both distilled water and glycerine. Very good agreement was obtained on these checks with published surface tensions.

The measured surface tensions were plotted as a function of the time on a particular sample. Surface tensions for the other runs were then estimated from this curve (see Figure C-1)

Because of water evaporation from the water-glycerine mixtures, the density and viscosity increased slightly with time. Average densities were used and the viscosity of the mixtrue was determined from Figure C-2 which are published viscosities. Viscosities of samples agreed closely with published data. (16)

EXPERIMENTAL RESULTS

A total of 36 runs was performed covering the following range of variables:

Demister Style	York styles 421 and 931
Liquid Density	52-73 #/ft ³
Liquid Viscosity	0.9-12 centipoise
Liquid Loading Rates	$25-400 \ \#/hr-ft^2$
Liquid Surface Tension	31-72 dynes/cm.
Air Density	0.071 #/ft ³
Air Velocity	5-24 ft./sec.

Flooding velocities, loading velocities, and liquid properties were determined as previously discussed in "EXPERIMENTAL PROCEDURE". These variables for each test run are summarized in Table 1. Runs No. 3,4,7,15 and 29 are not included in the summary. Runs 3 and 7 were inconclusive because test liquid entered the demister manometer lead line. Run 4 was performed using a smaller orifirce which was not used in other runs and therefore was not calibrated. Run 15 was a partial run which checked run no. 10. Run 29 was not completed since it was not possible to obtain adequate spray with 85% glycerine (50 centipoise).

Dimensional units of FT./Sec. are used for flooding and loading velocities for the purpose of discussion of test results. Evaluation of the variables which affect demister

TABLE 1 DATA SURMARY

	D ES MT IY SL	TEST LIQUID						AIR		
RUN <u>NÚ.</u>	E R	LIQUID	TL	PL	ML	<u> </u>	Y	<u> </u>	<u> </u>	V _F
125	421 421 421	Water	78 76 79	62.3 62.3 62.3	0.89 0.92 0.88	400 380 385	70.5 69.1 68.6	.0702 .0714 .0713	9.8 10.1 11.0	11.3 11.0 12.0
6	931	17	78	62.3	0.89	385	70.4	.0716	9.1	15.5
8.	421	17	72	62.3	0.96	110	70.3	.0721	12.4	13.3
9	931	17	81	62.3	0.86	110	68.7	.0717	14.3	15.8
10	931	र्थ	80	62.3	0.87	410	67.3	.0715	13.8	14.3
11	931	१४	82	62.3	0.85	55	68.7	.0712	13.4	16.3
12	931	११	84	62.3	0.83	55	67.8	.0714	13.0	16.6
13 14 16	421 421 931	" 25% glyc.	85 76 80	62.3 62.3 66.1	0.82 0.92 1.8	55 400 390	65.0 69.8 58.0	.0708 .0719 .0718	12.0 10.4 13.1	14.1 10.7 14.6
17	421	33% "	81	67.1	2.3	360	58.6	.0715	9.3	9.9
18	421	33% "	84	67.1	2.2	85	57.2	.0713	10.6	11.7
19	931	33% "	86	67.2	2.1	85	55.5	.0710	12.1	15.1
20	931	41% "	86	68.8	2.8	107	53.8	.0720	14.5	17.0
21	421	43% "	81	69.4	3.4	142	52.8	.0716	10.9	12.2
22	421	52% "	82	70.4	5.1	135	58.1	.0715	11.3	11.9
23	931	57 / "	85	71.3	6.2	120	56.4	.0713	14.9	16.8
24	931	65% "	84	72.5	10.3	40	58.6	.0714	14.8	17.2
25	931	65% "	81	72.5	11.5	55	57.1	.0722	13.5	17.3
26	931	65% "	88	72.5	9.5	265	56.0	.0716	14.4	16.0
27	421	68% "	87	73.1	12.0	230	54.5	.0717	9.2	9.8
28	421	68% "	86	73.1	12.3	162	53.0	.0718	9.8	10.6
30	421	38% "	81	68.2	2.8	145	58.2	.0722	10.1	12.6
31	421	Hvy. 2 011	85	52.5	3.7	280	31.5	.0717	7.0	7.7
32	931	"	92	52.5	3.7	270	31.5	.0719	9.1	10.9
33	421	हरू	85	52.5	3.7	25	31.5	.0715	11.2	12.3
34	421	दह	91	52.5	3.7	95	31.5	.0711	8.4	9.3
35	931	भूट	88	52.5	3.7	95	31.5	.0713	11.0	12.8
36	421	8 9	90	52.5	3.7	107	31.5	.0712	8.6	10.3

performance should not be affected since all test runs were made at constant air density. However, it is recognized that vapor mass velocity may ultimately provide the best definition of demister performance for systems which either use other than air or are operated above or below atmospheric pressure.

The flooding and loading velocities for the 421 and 931 style demisters are plotted in Figures 2 thru 5 as a function of liquid loading. For convenience of presentation the waterglycerine data are presented in only two groups, 25-52 wt. % glycerine and 65-68 wt. % glycerine. The data of Poppele(2) is shown for comparison of results for the air/water system.

All plots show the same general trend in the effect of liquid properties on flooding and loading velocities. Velocities for water are 3-4 ft./sec. above those for Heavy No. 2 Fuel Oil. Velocities for water-glycerine on the 421 demister give intermediate velocities which decrease with increasing wt. % glycerine. On the 931 demister the water and water-glycerine velocities are about the same. The flooding and loading velocities decrease with increased liquid load at about the same slope for each test liquid and for each demister. The relative effects of the variables which affect flooding and loading velocities are discussed in

(19)

FIGURE 2 FLOODING VELOCITY 421 DEMISTER



LIQUID LOAD, #/HR-FT2

(20)



LIQUID LOAD, #/HR-FT2

FIGURE 4 FLOODING VELOCITY 931 DEMISTER



LIQUID LOAD, #/HR-FT2

(22)

FIGURE 5 LOADING VELOCITY 931 DEMISTER



LIQUID LOAD, #/HR-FT2

(23)

detail in the next section entitled "DATA COERELATION".

These results for air/water system show reasonable agreement with the work of Poppele(2). Velocities are within 5-15% of those of Poppele. The effect of liquid load on velocities are almost identical with the exception of the 421 flooding velocity (Figure 2) where Poppele shows a much higher effect of liquid load on velocity.

This difference in the effect of liquid load on flooding velocity is attributed to the difference in the methods used in determining flood points. Poppele relied on visual observation of incipient flood, whereas, the author studied velocities above the flood point and redefined flooding as discussed under EXPERIMENTAL PROCEDURE. This new definition of flood point is expected to give better agreement among experimental results. It is essential that velocities between load and flood point be redetermined after the flooding point is exceeded to assist in obtaining a more exact location of the flood point.

Demister pressure drop at the flood point was found to be dependent on the demister style and the test liquid. Liquid load had only a slight effect on demister pressure drop. This is illustrated in Figure 6. This relation provides additional guidance in determining flood point in future experiments. (24)



FLOODING VELOCITY, FT./SEC.

Flooding and loading velocities were relatively easy to detersine at algh liquid load rates on the denser demister (Style 421). Leterainations became increasingly more difficult as liquid load was decreased and as demister void fraction increased (Style 931). This accounts somewhat for the greater variability of data for the 931 style desister (Figure 4 and 5).

Both the 421 and 931 demister were very effective in removing liquid from the air. Below the load point there was no visible sign of liquid breakthrough. It very bign air velocities (low liquid load on 931 demister) a few drops of liquid were carried through as the flood point was approached. However, this is not the normal operating range of the demister (near flood). Flooding and loading velocities for the 931 demister were about 20% higher than those for the 421 demister.

(26)
PUBLISHED CORRELATIONS

The most popular method of expressing allowable demister vapor velocity was proposed by York(3). His equation is as follows:

$$I = K \sqrt{(P_L - P_G)/P_G}$$
(1)

Values of X at the flood point have been calculated from the test data in Table 1 and are plotted in Figures 7A and 7B as a function of demister liquid load. Calculated data are tabulated in Table D-1.

The values of $K_{\rm F}$, and therefore the value of flooding velocity, differ by as much as 30% on the 421 demister at constant liquid loading primarily as a result of variations in liquid properties. A difference of 30% is also obtained between the 421 and 931 demister at constant liquid rate and constant liquid properties. These differences illustrate the need to determine the effects of each variable on demister performance.

Poppele⁽⁴⁾ has proposed the following relation which is similar to the equations used in defining the flooding point in packed columns:

$$\frac{\sqrt{2a} \operatorname{R}_{G} (m_{L})^{0,2}}{g_{c} \operatorname{E}^{3} \operatorname{P}_{L}} = \int \left(\frac{G_{L}}{G_{G}} \sqrt{\frac{\operatorname{R}_{G}}{\operatorname{P}_{L}}} \right) \quad (2)$$

These variables have been calculated for the test data in Table D-1 and plotted in Figure 8.

Curves A and B in Figure S show a straight line relation between variables but there is a distinct line for each demister style. Lines A and B are made to coincide into line C in Figure 8 by using $(\alpha/\epsilon^3)^{0.5}$ as the correlating

FIGURE TA AND TB DEMISTER ENTRAINMENT FACTOR (KF) AT THE FLOOD POINT



LIQUID LOAD , #/HR-FT2

(28)





.0001



GL VegeL



variable rather than $(a/\epsilon^3)^{l,0}$. There is still considerable variation of the data either from variables not included or from variables not adequately represented.

Curve C in Figure 6 can be considerably simplified by assuming a linear relationship. The equation is as follows:

$$Log\left[\frac{V_F^2\left(\frac{\alpha}{E^3}\right)^{0.5} P_G\left(\mu_L\right)^{0.2}}{32.2 P_L}\right] = -0.251 Log\left(\frac{G_L}{G_G}\sqrt{\frac{P_G}{P_L}}\right) + Log(0.0122) \quad (3)$$

By substituting $G_{G} = 3,600 \ P_{G} V_{F}$ in equation (3) and solving for V_{F} we obtain:

$$V_{F} = \frac{1.90 (P_{L})^{0.64}}{(G_{L})^{0.14} (\frac{Q_{L}}{E^{3}})^{0.29} (M_{L})^{0.11} (P_{G})^{0.50}}$$
(4)

This equation is a clearer illustration of the effect of variables on flooding velocity and is a more convenient form for adding other variables such as surface tension.

LINEAR REGRESSION ANALYSIS

Stepwise Linear Regression analyses were run on the IBM 7094 Digital Computer of Standard Gil Company of California. $V_{\rm F}$ and $V_{\rm L}$ were dependent variables. The equation form was as follows:

$$\frac{\ln(V_{\rm F})}{\ln(V_{\rm L})} = \frac{\kappa_1 \ln(\alpha/\epsilon^3) + \kappa_2 \ln(\ell\epsilon) + \kappa_3 \ln(\mu_{\rm L})}{\kappa_5 \ln(\ell_{\rm L}) + \kappa_5 \ln(\ell_{\rm L}) + \kappa_6 \ln(\gamma) + \kappa_7 \ln(\epsilon)}$$
(5)

where; $n_7 \ln(e) = \text{constant}$ and $\ln(e) = 1.0$

Regression coefficients $(K_1 \text{ thru } K_7)$ are summarized for each regression step in Table D-2. Coefficient (K_2) for vapor density (ρ_G) was not determined since this variable was practically constant for all tests and could not be recognized as a significant variable. Coefficients for the remaining variables are complete in Step 6 of each regression analysis. The resultant equations for V_F and V_L are as follows:

$$V_{\rm F} = \frac{5.45 \ (P_{\rm L})^{0.47} \ (\gamma)^{0.20}}{(\alpha/\epsilon^3)^{0.30} \ (\mu_{\rm L})^{0.036} \ (G_{\rm L})^{0.11}} \tag{6}$$

Standard Error = ± 1 ft./sec. at V_F = 15 ft./sec.

$$V_{L} = \frac{2.34 \ (\ell_{L})^{0.61} \ (\gamma)^{0.12}}{(\alpha/\epsilon^{3})^{0.22} \ (\mu_{L})^{0.033} \ (g_{L})^{0.089}}$$
(7)

Standard Error = ± 1.3 ft./sec. at $V_F = 13$ ft./sec.

(31)

These equations can be modified to include the variable of vapor density by multiplying the constant term by $(0.0715)^{0.5}$ and including $(P_{\rm c})^{0.5}$ in the denominator. This inverse relation between $V_{\rm F}$ or $V_{\rm L}$ with the square root of vapor density was proposed by York and Poppele in equations (1) and (4). This relation however requires future substantiation.

Computer putput solutions for equations (6) and (7) are presented in Tables 1-3 and 2-4. The data are plotted in Figures 9 and 10. These equations predict the value of $V_{\rm p}$ within \pm 1 ft./sec. for 65% of the test data and within \pm 2 ft./sec. for 95% of the test data. The error in predicting $V_{\rm p}$ is 30% higher than that for $V_{\rm p}$. Note that the standard error of correlation in Table 2-2 is based on in $V_{\rm p}$ or $V_{\rm p}$ and must be converted to the corresponding velocity values.

Table 2-2 shows correlation coefficients for seven steps of the stepwise regression for J_F and six steps of the stepwise regression on V_{\perp} . After step no. 4 in each case there is little improvement (decrease) in the standard error of correlation. This indicates that for the test liquids used in these experiments the flooding and loading velocities can be defined by the variables of liquid density, demister

(32)



 $\frac{5.45 (P_L)^{0.47} (\gamma)^{0.20}}{(^{0}/E^{3})^{0.30} (\mathcal{M}_L)^{0.036} (G_L)^{0.11}} = CALCULATED V_F$





(24)

specific surface, liquid load and liquid viscosity. Addition of a constant term and surface tension effects add nothing to the accuracy of prediction. However, step no. 6 was chosen for presentation because it includes all variables and shows effects of liquid density similar to that proposed by York(3) and Roppele(4).

Note that in Table D-2 the standard error of the regression coefficients is small through step 5. However, the addition of the surface tension variable resulted in a large increase in the standard error of regression coefficients for liquid properties of viscosity, density and surface tension. The following are suspected reasons for the increase in coefficient standard errors:

(1) A high correlation existed between two of the three liquid properties or among all three properties. Flots of any two variables do not show a high correlation. However, it is likely that a high correlation could exist among all three liquid properties of the various test liquids. Selection of test liquids was limited by considerations of toxicity and flammability due to the nature of the test gas (air) and test apparatus (open system vented to laboratory atmosphere).

(35)

- (2) The effects of liquid properties on defister performance could vary with demister properties. The regression of data found the best fit of experimental data for two demisters. Therefore, any variation due to demister properties would result in greater error of the regression coefficients.
- (3) Nanges in physical properties were not large enough to determine effects with a high degree of certainty. For example, liquid density ranged from 52 to 73 //ft.³ or ± 17 % around the average of 62.5 %/ft.³. However, the ln (Q.) which was used in the regression analysis ranges from 3.95 to 4.3 or ± 4 % around the average value.

The effect of surface tension was previously studied by Schroeder by using surfactants in water. Schroeders results snow a considerably larger effect of surface tension on demister performance than those shown by the author. Schroeder showed that by reducing surface tension of water from 72 dynes/cm to 36 dynes/cm the flooding velocity at the lower surface tension was only 40% of the original value. In contrast, the regression correlation would predict only a 15% recuction in flooding velocity.

(36)

CONCLUSIONS

- Allowable demister velocities increase with increase in liquid density and liquid surface tension.
- 2. Allowable demister velocities decrease with increases in demister specific surface area, liquid viscosity and liquid load.
- 3. Liquid viscosity has only a small effect on allowable demister velocity. The allowable velocity is decreased by only 10% when viscosity is increased from 1 to 12 centipoise.
- 4. The effect of liquid loading on the performance of the 421 and 931 demisters show good agreement with the work of Poppele. It is expected that better agreement would have been obtained if the method had been standardized for determining the flood point.
- 5. The effect of liquid density on demister velocity is in close agreement with work of York and Poppele.
- 6. The effect of surface tension varies considerably from the work of Schroeder. The correlations developed in this study are obviously not applicable to the use of surfactants to reduce surface tension of liquids.

- 7. Velocities beyond the flood point were explored. This gives useful data for determining the flood point.
- 5. The quantitative effects of variables which affect demister performance have been determined with the exception of gas density. The application of these correlations should be limited to the ranges of experimental data used in these experiments.

RECOMMENDATIONS

The equations developed in this thesis represent the first step in defining a generalized correlation for demister performance. Most significant variables are included with the exception of vapor density. With slight modification this latter variable can be included based on earlier proposals. Application of these equations should be limited to the range of variables studied. Extrapolation beyond experimental data could result in significant error. It is expected that these equations are best suited for application to hydrocarbor liquids, water and liquids mixed with water.

The major shortcoming of these equations in application to other systems is the undetermened effect of vapor density. Host experiments have been performed with air at atmospheric pressure. No doubt there are numerous systems where vapor density is substantially different than the density of air. In future studies major emphasis should be placed on defining the effect of vapor density.

The method used by the author in determining flood points is considered an improvement over earlier definitions. This procedure should be used in future experiments to improve reproducability of results.

NOMENCLATURE

8.	Demister specific surface, ft.2/ft.3
Sc	Acceleration of gravity, $ft./(sec.)^2$
9 0	Vapor mass velocity, #/hrft.2
0 _L	Liquid entrainment mass velocity, $\#/hrft.^2$
K	Demister entrainment factor
₿ F	Demister entrainment factor at the flood point
K1 thru K7	Regression coefficients
Δ P _D	Demister pressure drop, inches of water
ΔP_0	Orifice pressure drop, inches of water
TD	Air temperature to demister, ^O F dry oulb
T	Liquid temperature to demister, ^O F
$\mathbf{T}_{\mathbf{W}}$	Air temperature to demister, ^o F wet bulb
V	Gas velocity, ft./sec.
٧ _F	Gas velocity at demister flood point, ft./sec.
V	Gas velocity at demister load point, ft./sec.
ε	Demister void fraction
୧ୢ	Gas density, #/ft.3
<i>९</i> .	Liquid density, #/ft.3
Ý	Liquid surface tension, dynes/cm.
KL.	Liquid viscosity, centipoise

(40)

APPENDIX A

EWUIPMENT DETAILS

AIR BLOWERS

No. of units	2 in parallel
Туре	Spencer Turbo Compressor,
	model no. 5010-H
Rated capacity	250 SOFM at 80 $oz./in^2$.
Horsepower	10 at 3500 RPM

Comments - These air blowers are equipped so that if the rated capacity is exceeded the motor is turned off automatically. Because of the low back pressure of the test apparatus it was necessary to operate the control butterfly valve in the 0-40% open range. If more than 40% control position was used the blower would stop. In order to run tests at low liquid loads on the 931 type demister it was necessary to use two blowers in parallel. All leaks in the test apparatus were cemented to obtain maximum blower output.

HUMIDIFIER

Chamber

Mesh demister

Secondary spray

Construction

Assembly

Level Gauge

Water supply

55 gallon drum (open top type).

Manufactured by Otto York Go, 22 inch diameter, 6 inch thick.

Stainless steel tube, 2 feet long with 1/32 inch holes along bottom of tube.

Two tubes mounted raidially at right angles just above air inlet line. One two foot section of semi-transparent, semiflexible plastic tubing (bottom attached to lead line below water level and top attached to lead line in humidifier vapor space). Water was supplied by hose from city water supply line. Water rate was limited by the drainage rate to the sewer. City water rate was

adjusted to maintain about an 8 inch water level in the level gage (above bottom of barrel). Air cooling and humidification were limited by the rate of water. This setup gave 100[°]F air temperature at 40% humidity. The addition of the spray into the compressor discharge resulted in temperature ^{of} 80-100[°]F and 80% humidity

THERMOMETERS

Three mercury thermometers $(0-130^{0}F)$ were used. Not build temperature was obtained by tieing gauze to one thermometer and moistening with water at required intervals.

Air temperature from the humidifier and liquid temperature to the test column were measured throughout the run. Temperatures of the air leaving the test demister were not obtained on a regular basis because it was found that this air temperature was very close to the test liquid temperature. Also, the air leaving the test demister was at 100% humidity for test liquids that contained water.

(43)

ANOMETTRS

Pressure drop across the orifice and static pressure below the demister were measured with a water filled manometer.

ORIFICE PLATE

The orifice plate is the same one used by H.F. Schroeder^(I) in his earlier experiments. The calibration curve (Figure A-1) is calculated by standard methods⁽⁸⁾ for an air density of $0.0695 \#/ \text{Ft.}^3$ upstream of the orifice. Figure A-1 includes corrections for (1) the effect of Reynolds Number on orifice flow coefficient and (2) gas expansion factor. However the combined effect of these corrections is small, being only one percent or less of the uncorrected flow rate at the load and flood points.

Details of the orifice plate are as follows:

Pipe diameter	4 inches
Orifice diameter	2.628 inches
Beta	0.657
Plate thickness	1/16 inch
Tads	Flange

A plot of air density as a function of dry bulb temperature and percent humidity is shown in Figure A-2. This Figure is used to obtain densities for air rate correction factor and velocity calculations. Figure A-3 shows calculated air velocity at flood point (at actual operating conditions) as a function of orifice pressure drop. Note that variations in velocity over the range of temperature and pressure experienced is very small (0.2 ft./sec.). Therefore, velocities at demister load point were obtained directly from Figure A-3 instead of by direct calculation.

ROTURETER

One O-50 GPH (water) rotometer was satisfactory over the entire liquid flow range. Rotometer calibration curves are shown in Figures A-4 and A-5 for water, water - glycerine mixtures and heavy No.2 fuel oil.

LIQUID PUMP

Type	Eastern	moćel A-l	
Capacity	4.5 GPM	maximum at zero	pressure
	maximum	output pressure	of 11 peis.

The capacity of this pump was a major limit to the range of flow rates and fluid properties that could be studied. Liquid loadings of 25-400 #/hr-ft² were obtained and were adequate for the intent of this study, However, the nozzles were designed for liquid pressures of 20-40 psig. It was therefore not possible to study viscosities of water - glycerine mixtures above 12 centipoise. At higher viscosities no adequate spray

(45)

could be obtained.

TEST DEMISTERS

York style 931 and 421 were both studied. The 931 demister was supplied by Otto York Company. The style 421 was the same demister used by Schroeder⁽¹⁾. The 421 demister also had been supplied by Otto York Company.

Demister properties used in correlation work are:

Demister	Voið	Specific	Surface,
Style	Fraction	ft^2/ft^3	
931	0.99	46	
421	0.977	110	

Both demisters were 6 inches thick with diameter of 7.45 inches. The outer surface was wrapped with nylon mesh to obtain adequate seal between lucite column and demister.

SPRAY NOZZLES

The following spray nozzles, made by Spraying Systems Co., were used:

Nozzle	Maximum Water	Rate, GPH (a)
TN-1	Less than	0.5
TN-2		0.5
TN-3		1.0

(46)

Nozzle	<u>Maximum water Rate, GPH (a)</u>
TN-6	2.5
TN-8	4.0
TN-10	5.0
TN-12	5.5
TN-14	6.0
TN-26	11.5
G-3	15.0

(a) with available pump.

PHYSICAL PROPERTY MEASUREMENTS

Densities were determined by weight of a volume of liquid in a graduated cylinder. Densities were measured during the tests and later checked in the laboratory at California Oil Company, Ferth Amboy, N.J.

Viscosities and surface tensions were determined by laboratory personnel of California Oil Company.

DATA CORRELATION

A stepwise linear regression analysis was run on the IBM 7094 Digital Computer at the Standard Oil Company of California Computer Center in San Francisco, California. (47)



FIGURE A-1 ORIFICE CALIBRATION

100

HIR RATE, #/MIN. (1)

(40)

ORIFICE PRESSURE DROP (ΔR) , INCHES WATER



AIR TEMP., OF DRY BULB

FIGURE A-2

FIGURE A-3

CALCULATED AIR VELOCITIES



APO, INCHES WATER





INDICATED FLOW RATE, GPH

いつい

FIGURE A-5

ROTOMETER CALIBRATION

0-50 GPH - U- 88



INDICATED FLOW RATE, GPH

APPENDIX B

EXPERIMENTAL DATA

TABLE B-1 EXPERIMENTAL DATA

Run No System Liquid Demist Orific Nozzle). n l Load ler No. le Diam.	- 1 - Air/Wa - 400 #/ - 421 - 2.628 - G-3	ter 'hr-ft ² In.	Run No System Liquid Demist Orific Nozzle). 1 Load er No. e Diam.	- 2 - A - 34 - 2 G	ir/Wa 80 #/ 21 .628 -3	ter hr-f	t ²
$ \underline{\triangle} P_0 \\ 0.60 \\ 0.21 \\ 0.60 \\ 0.21 \\ 0.72 \\ 0.25 \\ 0.$	$\frac{A}{29} \frac{P_{\rm D}}{0.229}$ 0.19 0.19 0.2547 0.75F 0.75F 0.75F 2.305F 1.0705F 2.305 1.075F 2.305 1.050 2.3455 1.050 2.3555 2.3555 2.3555 2.3555 2.3555 2.3555 2.35555 2.35555 2.35555 2.35555555 2.35555555555	T _W T _D 	T _{j.} 70 71 72 76 78 78 78 78	△ P ₀ 0.61 1.20 0.44 0.40 1.00 1.00 1.00 1.00 1.00 1.0	$ \Delta P_{D} $ 0.20 0.52 0.09 0.17 0.27 0.86 1.43 3.85 4.34F 4.55F	T <u>w</u> 6880903899990 7777778	T 1 981266 24 111408	TL 70 70 70 71 74 77 77 77 77 77 77 77 77 77	
Run No System Liquid Demist Orific Nozzle). 1 1 Load 2er No. 2e Diam.	- 4 - Air/We - 370 #/ - 421 - 1.789 - G-3	iter 'hr-ft ² In.	Run No Syster Liquid Demist Orific Nozzle	er No.		ir/₩a 85 #/ 21 •628 - 3	ter hr-f	t ²
$\frac{\Delta}{0.59} \frac{P_0}{1.53} \frac{1.84}{15.57} \frac{11.6}{19.959} \frac{11.6}{2222} \frac{100}{225} \frac{100}{25} 100$	△ PD 0.07 0.10 0.52 0.74 0.52 0.74 1.855 0.74 1.855 0.74 1.855 0.25 FF F	$\begin{array}{c c} \underline{T}_{\underline{W}} & \underline{T}_{\underline{D}} \\ \hline 70 & 70 \\ 67 & 67 \\ 69 & 72 \\ 74 & 84 \\ 77 & 94 \\ 78 & 100 \\ 80 & 101 \\ 80 & 100 \\ 80 & 100 \\ 80 & 99 \\ 82 & 98 \\ 84 \\ 85 & 97 \end{array}$	T <u>I.</u> 70 70 70 70 70 70 70 70 70 70 70 70 70	▲ P0 0.60 1.14 2.63 3.945 3.945 5.46 3.055 5.46 3.055 4.003	$ \frac{\triangle P_{D}}{0.22} 0.37 0.90 1.16 1.40 2.34 3.80 4.50 F 1.82 1.90 2.93 $	T69268999999999999999	T _D 77 88 104 106 105 104 105 102	T. 73 774 775 778 779 799 78 778 778 778 779 778 778	

TABLE 3-2 EXPERIMENTAL DATA

Run No System Liquid Demista Orific Nozzle	• Load er No. Ə Diam.	- 6 - Air - 385 - 931 - 2.6 - G-3	/Water #/hr-ft ² 28 In.	Run No System Liquid Demist Orific Nozzle	Load er No. e Diam.	- 8 - Air/Wa - 110 #/ - 421 - 2.628 - TN-8	ater /hr-ft ² In.
▲ P0 1.45 5.70 5.50 7.80 5.80 5.80 5.80 5.80 5.80 5.80 5.80 5	$\frac{\Delta P_{D}}{0.24}$ 0.48 1.29 1.71 2.10 0.50 1.25 1.75 0.26 0.11 2.05F 2.17F	T <u>w</u> 769 809 778 877 799 812 788 778 8778	$ \begin{array}{c} T_{D} & \underline{T}_{L} \\ 101 \\ 108 \\ 108 \\ 105 \\ 105 \\ 104 \\ 105 \\ $	▲ P ₀ 1.32 3.57 5.60 5.23 5.60 5.23 4.96 2.56	$\frac{2}{0.38}$ 1.05 4.34F 4.45F 2.40 1.55 1.75 0.98	T _M T _D 69 70 75 78 80 102 80 101 81 102 78 94 78 95	T <u>;</u> 68 68 71 72 78 79 80

Run No. System Liquid Demiste Orifice Nozzle	Load er No. e Diam.	- 9 - Air - 110 - 931 - 2.0 - TN-	:/Wate) #/h: 528 I: -8	er r-ft ²	Run No System Liquid Demist Orific Nozzle	Load er No. e Diam.	- 10 - Ai) - 41(- 93 - 2.6 - G-	r/Wat D #/h L 528 I 3	er r-ft ² n.
		TW 79 81 79 82 79 80 80 80	T _D 98 1000 972 94 99 99 99 99	T _L 90 80 81 81 81 81 81	$\frac{\sum_{\substack{p \in P_0 \\ p \in Q_2}} P_0}{\sum_{\substack{p \in Q_2 \\ p \in Q_2}} P_0} P_0$		TW 74 76 79 81 80 79	TD 852 1022 1022 104 999 99	T 76 76 77 80 81 80 80

TABLE 8-3 EXPERIMENTAL DATA

.

Run N Syste Liqui Demis Orifi Nozzl	o. m d Load ter No. ce Diam. e	- 11 - Air/Wa - 55 #/M - 931 - 2.628 - TN-3	ter 1r-ft ² In.	Run No System Liquid Demist Orific Nozzle	Load er No. e Diam.	- 12 - Ai - 55 - 93 - 2. - TN	r/wat #/hr 1 628 I - 3	er -ft ² n.
Δ^{20} 2.21 2.4.50 5.70 6.30 6.30 7.80 5.70 7.90 5.70 7.80 5.70 7.90 5.90 7.80 5.90 7.90	$\frac{2}{0.52}$ 0.52 0.60 0.82 0.91 1.00 1.62F 1.70F 0.70 0.40 0.53 0.74	$ \frac{T_{W}}{77} 95 \\ 80 107 \\ 80 107 \\ 80 101 \\ 80 101 \\ 81 101 \\ - 101 \\ - 101 \\ - 102 \\ - 104 \\ - 106 $	$ \frac{T_{L}}{80} \\ 5 80 \\ 79 \\ 77 \\ 77 \\ 79 \\ 81 \\ 82 \\ 82 \\ $	▲ PO 3557 3.677 3.677 3.677 3.667 3.667 3.667 3.667 3.667 3.667 3.667 3.667 3.667 3.667 3.667 3.667 3.667 3.667 3.667 3.667 3.667 3.65 3.667 3.65 3.667 3.65 3.65 3.65 3.65 3.65 3.65 3.65 3.65	$ \underline{A} \underline{P}_{D} \\ 0.19 \\ 0.44 \\ 0.75 \\ 0.97 \\ 1.06 \\ 1.21 \\ 1.51 \\ 1.72F \\ 1.80F \end{array} $	T <u>W</u> 814 84 790 823 -	Tn 109 1150 102 99565 -	T <u>1</u> 79 80 81 83 - 84 -
Run N Syste Liqui Demis Orifi Nozzl	o. m d Load ter No, ce Diam. e	- 13 - Air/V - 55 #/ - 421 - 2.628 - TN-6	later hr-ft ² } In.	Run No Systen Liquid Demist Orific Nozzle). 1 Load Ser No. 20 Diam.	- 14 - Ai - 40 - 42 - 2. - G-	r/Wat 0 #/h 1 628 I 3	er r-ft ² n.
$\Delta_{2,0}^{P_0}$ 1.30925885060505 2.3.5683506050 7.4	$ P_{D} 0.35 0.55 0.95 1.56 0.78 1.926 1.926 3.926 3.95 4.60F 4.60F $	$ \frac{T_{W}}{76} \qquad \frac{T_{D}}{83} \\ 80 \qquad 92 \\ 81 \qquad 96 \\ 81 \qquad 95 \\ - \qquad 94 \\ 81 \qquad 104 \\ 85 \qquad 105 \\ - \qquad 103 \\ - \qquad 103 \\ - \qquad 103 $	T1 782234 882234 8844 885555 8888 8888	▲ PO 000000000000000000000000000000000000	^ PD 0.29 0.257 7.90 1.70 7.20 1.70 7.20 1.00 0.45 1.00 0.45 1.00 0.45 0.0	Tw 694 79 79 78 78 78 78 78 78 76 75 73 73	TD 7800 100 9999 50048555	TL 737777 78 78 78 78 78 79 79 79 79 79 79

.

(56)

.

TABLE B-4 EXPERIMENTAL DATA

.

Run No. -15	Run No.	- 16
System $-$ Air/Water	System	- Air/25wt.% Glyc.
Liquid Load $-410 \#/hr/ft^2$	Liquid Load	- 390 #/hr-ft ²
Demister No. -931	Demister No.	- 931
Orifice Diam. -2.628 In.	Orifice Diam.	- 2.628 In.
Nozzle $-G-3$	Nozzle	- G-3
$\frac{\Delta P_{0}}{3.80} \frac{\Delta P_{D}}{0.78} \frac{T_{W}}{80} \frac{T_{D}}{98} \frac{T_{L}}{79} \frac{T_{L}}{1.03} \frac{T_{0}}{76} \frac{T_{1}}{85} \frac{T_{1}}{77}$ Check on Run No. 10 Data are plotted with Run no.10	$\frac{\Delta}{1.44} \frac{P_0}{0.37}$ $\frac{\Delta}{3.65} \frac{P_0}{0.59}$ $\frac{\Delta}{4.96} \frac{P_0}{1.00}$ $\frac{\Delta}{5.45} \frac{P_0}{1.00}$ $\frac{1.00}{5.45} \frac{1.43}{1.95}$ $\frac{1.43}{6.30} \frac{1.95}{1.95}$ $\frac{4.90}{1.05} \frac{1.05}{4.60}$ $\frac{1.05}{4.60} \frac{1.00}{0.98}$ $\frac{4.60}{1.00} \frac{1.00}{3.16} \frac{0.72}{0.61}$ $\frac{2.10}{2.10} \frac{0.54}{0.43}$	$ \begin{array}{ccccccccccccccccccccccccccccccccc$
Run No 17	Run No.	- 18
System - Air/33wt.%Glyc.	System	- Adr/33wt.%Alyc.
Liquid Load - 360 #/hr-ft ²	Liquid Load	- 85 #/hr-ft ²
Demister No 421	Demister No.	- 421
Orifice Diam 2.628 In.	Orifice Diam.	- 2.628
Nozzle - G-3	Nozzle	- TN-8
$\frac{\Delta P_0}{1.94} \frac{\Delta P_D}{0.95} = \frac{T_W}{10} \frac{T_D}{103} \frac{T_L}{103}$ $\frac{1.94}{3.00} \frac{4.14F}{4.14F} \frac{79}{79} \frac{98}{98} \frac{81}{98}$ $\frac{3.40}{4.50F} \frac{4.50F}{79} \frac{79}{98} \frac{81}{98}$ $\frac{2.90}{2.63} \frac{2.63}{79} \frac{79}{98} \frac{82}{98}$ $\frac{2.50}{1.27} \frac{1.27}{79} \frac{79}{98} \frac{82}{95}$ $\frac{1.72}{1.26} \frac{0.98}{0.76} \frac{79}{-95} \frac{95}{82}$	$\frac{\triangle P_0}{2.60} \frac{\triangle P_D}{1.00}$ $\frac{\triangle 2.60}{4.12} \frac{1.00}{3.26}$ $\frac{4.45F}{4.80} \frac{4.45F}{4.60F}$ $\frac{4.60F}{3.60} \frac{1.64}{1.64}$ $\frac{2.95}{1.12}$ $\frac{1.16}{0.65}$	$ \begin{array}{ccccccccccccccccccccccccccccccccc$

• .

TABLE 3-5 EXPERIMENTAL DATA

Run No.		- 19			Run No.		20		
System		- Air/33wt.%Glyc.			System		Air/41wt.%Glyc.		
Liquid Load		- 85 #/hr-ft ²			Liquid Load		107#/hr-ft ²		
Demister No.		- 931			Demister No.		931		
Crifice Diam.		- 2.628 In.			Orifice Diam.		2.628 In.		
Nozzle		- TN-8			Nozzle		TN-14		
	$\frac{\Delta P_{D}}{0.43}$ 0.43 0.58 0.77 0.98 1.01 1.20 1.82F 1.60 1.87F 1.60 1.55 1.40 1.00 0.60	T _W 79 80 80 80	T _D 100 100 100 100	FI 8555666	∆ P ₀ 3.4 5.1 6.0 7.3 9.5 9.5	$\frac{\Delta P_{D}}{0.60}$	Tw 78 78 78 78	TD 9223 	T <u></u> 87 86

Run No.		- 21			Run No.		- 22			
System		- Air/43wt.%Glyc.			System		- Air/52wt.%Glyc.			
Liquid Load		- 142 #/hr-ft ²			Liquid Load		- 135 #/hr-ft ²			
Demister No.		- 421			Demister No.		- 421			
Orifico Diam.		- 2.628 In.			Orifice Diam.		- 2.628 In.			
Nozzle		- TN-14			Nozale		- TN-14			
△ Po 1.57 5.55	$\frac{\sum_{\substack{p \\ 0.67}} P_{D}}{0.67}$ $\frac{1.25}{4.8}$ $\frac{1.80}{0.90}$ $\frac{1.00}{1.26}$ $\frac{1.26}{4.2}$	Tw 76 80 78 81 -	TD 88 98 98 98 98 - 103	T 77 78 80 82 82 84 84	$ \frac{2}{200} \frac{2}{100} \frac$	$ \frac{P_{\rm D}}{0.532} \frac{0.532}{0.5627} \frac{0.552}{0.970} \frac{0.970}{0.990} \frac{0.990}{0.990} \frac{0.990}{0.990} \frac{0.990}{0.90} \frac{0.990}{$	T 76 790 800 800 800 800 800	TD 908 100 990 100 100	T <u>11</u> 80012204556 88888888888888888888888888888888888	

.

TABLE B-6 EXPERIMENTAL DATA

-

Run No System Liquid Demist Orific Nozzle	Load er No. e Diam.	- 23 - A1: - 120 - 93: - 2.0 - TN-	r/57 0 #/ 1 528 -14	wt.%Glyc. hr-ft2 In.	Run No System Liquid Demist Orific Nozzle	• Load er No. e Diam.	- 24 - Air, - 40 - 931 - 2.6 - TN-6	/65w ∉/hr. 28 I: 6	t.%Alyc. _ft2 n.
<u>P</u> 0 2.01 2.01 5.50 0.1 5.50 0.1 4.2 3.3 2.	▲ PD 0.25 0.25 0.25 1.10 1.455 FF 2.20 0.50		FD 999999	E13 8555555 8888 887 1	<u>A</u> 20 247906206155710 1087052710	△ 5555 555 555	79 	FD 1995	T ₁ 84 84 84 85 85 85
Run No System Liquid Demist Orific Nozzle	Load er No. e Diam.	- 25 - Ai: - 55 - 93 - 2.0 - TN	r/65 #/h 1 628 -6	vt.%Glyc. r-It² In.	Run Nö Syston Liquid Demist Orific Nozzle	Load er No. e Diam.	• 26 • 1455 • 255 • 255 • 255	/65% _%/b) 28 I:	t.∬2]yo. r-1t2 a.
$\frac{\Delta}{1.87} \frac{P_0}{1.87}$	$\Delta_{0.27}^{20}$	T <u>V</u> 80 81 89 79 81	TD 8663228 888888	T 76 78 799 799 80	$\frac{\sum_{i=0}^{2}}{\sum_{i=0}^{2}}$	$\frac{\Delta P_{D}}{0.25}$ 0.25 0.66 0.86 1.07 1.27 1.50	T. 80 80 80 81 -	ED 867 888 888 888	T 84 84 84 86 88 88 88 88

0.95 1.40 1.95 2.10F 1.84 1.45 1.322 0.83 1.23 1.57 1.50 1.80 2.20F 1.60 0.78 81 81 79 79 80 88 85 81 80 **6**19 -81 85 86 81 82 81 **ã**0 **-**79 e tø **w**.7 turo. 62 82 87

(59)

-86 85

-

85

-88

ante

88

TABLE 3-7 EXPERIMENTAL DATA

.

Run N Syste Liqui Demis Orifi Nozzl	o. m d Load ter No. ce Diam. e	- 27 - Ai: - 23(- 42) - 2.(- G-	r/68v D #/r 1 528] 3	/t.%Clyc. hr-ft ² In.	Run No System Liquid DEmist Orific Nozzle	o. n 1 Load ter No. te Diam.	- 28 - Ai: - 16: - 42: - 2.6 - TN-	r/68w 2 ∦/h 1 628 I -26	t.%Glyc. ir-ft ² n.
△ [₽] 0 1,70 2,55 2,50 2,50 2,50 2,50 2,50 2,50 2,5	$\frac{\Delta P_{\rm D}}{0.57}$ 0.57 0.87 5.15 1.75 2.20 5.20 1.10 0.45	T 77 79 80	T _D 80 - 84 - 82	T <u>L</u> 87 86 87 87 88 88	<u> </u>	△ PD 0.60 1.00 1.50 1.50 1.50 1.20 0.90	T _W 79 80 80	TD 5555	T 84 86 88
Run No Syster Liquid Demist Orific Nozzle	o. m d Load ter No. ce Diam. e	- 30 - Aix - 1421 - 421 - 2.6	c/38w 5 #/b 528 I	t.%Glyc. r-ft2	Run No System Liquid Demist Orific Nozzle). Load er No. e Diam.	- 31 - 21 - 280 - 420 - 420	/Hvy) #/h 528 I 3	. No. 2 Oil r-It2 n.
<u>P</u> 0 ³ ,5559245715 ³ ,12,23544 ³ ,12,2592 ³ ,12,259 ³ ,12,259 ³ ,12,259 ³ ,12,259 ³ ,12,259 ³ ,12,259 ³ ,12,259 ⁴ ,15 ⁴ ,15	A PD 4 5 5 5 5 0 1 3 5 0 1 3 5 5 5 0 1 5 5 5 5 5 5 5 5 5 5 5 5 5	TH 78 78 79 80	Ξ <u>D</u> 79 79 85 86	T <u>I.</u> - 81.	△ ^E 0 0.85 1.955 1.555 0.25 1.555 1.555 1.555 1.555 1.555 1.555	$ \frac{2}{0.52} \frac{3}{0.52} \frac{2.10}{0.70} \frac{0.35}{0.17} \frac{0.65}{1.21} $	72 75 77	9 <u>0</u> 7746 7746 1185	

TABLE 8-8 EXPERIMENTAL DATA

Run No 32 System - Air/Hvy. No.2 Oil Liquid Load - 270 #/hr-ft2 Demister No, - 931 Orifice Diam 2.628 In. Nozzle - G-3					Run No System Liquid Demist Orific Nozzle). h l Load ler No. e Diam.	- 33 - Ai - 25 - 42 - 2.	r/Hv #/h 628 -3	y.No.2 r-ft ²	Gil
$\frac{\Delta}{2.00} = \frac{2}{1.00}$ $\frac{1.00}{2.00}$ $\frac{4.00}{2.00}$ $\frac{4.00}{2.00}$ $\frac{4.00}{2.00}$ $\frac{5.00}{2.00}$ $\frac{5.00}{2.00}$ $\frac{5.00}{2.00}$ $\frac{5.00}{2.00}$	▲ PD 0.1655 0.1655 1.135 1.135 1.135 1.135 1.135 1.25 1.25	T _W 82	TD 90 95 	F 92 92 92	<u>P</u> 0 P0 P0 P0 P0 P0 P0 P0	△ ^D D 0,372 0,382 0,375 1,755 F 0,000 1,22 1,255 F 0,052 0,555 F 1,000 0,52 0,555 F	TU 70 81 82 188 188 1888 1888	TD 7849011733	Fi 4 6 55 5 55 1 5	
Run No 34 System - Air/Hvy. No.2 Oil Liquid Load - 95 #/hr-ft ² Demister No 421 Crifice Diam 2.628 In. Nozzle - TN-14					Run No System Liquid Demist Orific Nozzle	Load er No. e Diam.	- 35 - Ai - 95 - 93 - 2.	r/Hv ∦/h 1 628 -14	y.No.2 r-ft ² In.	Oil
$\frac{\sum_{n=0}^{\infty} P_0}{1.0}$ 1.0 2.0 2.0 2.25 2.4 2.5 1.8 2.3	△ PD 0.33 0.70 2.20 2.20 1.20 1.65 5.5 1.00	T _V 	TD 82 908 - 824 - 824 - 824	T <u>I.</u> 91 91 - 91 91	Po Po	△ PD 0.25 0.47 0.630 1.200 0.45 0.45 0.45 0.28 0.70 0.80	T. 80 80 83 87	TD - 66 87 - 92 - 81 -	T 91 90 90	

.

(61)

TABLE B-9 EXPERIMENTAL DATA

	36	
	Air/Hvy. No ₈ 2	Oil
1877	$107 \#/hr-ft^2$	
-	421	
	2.628 In.	
-	TN-14	
		- 36 - Air/Hvy. No. 2 - 107 #/hr-ft ² - 421 - 2.628 In. - TN-14

.

.

ΔP_0	ΔP_{D}	$\underline{\mathbf{T}}_{\mathcal{M}}$	\underline{T}_{D}	$\underline{T}_{\underline{I}}$
0.70	0.22	77	84	81
1.3	0.40	-	63	495.7
2.35	0.86	83	87	930
2.75	1.25	85	88	82
3.00	2.15	85	89	86
3.4	2.35F	83	89	89
5.0	2.10		-	-
2.7	1.20		-	
2.0	0.80	NBIO	-	-
1.2	0.50	-		-

(62)
EXPERIMENTAL DEFERMINATION OF DEMISTER

、シンノ

LOAD POINT AND FLOOD POINT

RUN NO. 1



AP, IN. WATER

EXPERIMENTAL DETERMINATION OF DEMISTER LOAD POINT AND FLOOD POINT

RUN NO. 2



APO, IN. WATER

RUN NO.4



DPo, IN. WATER

(65)

RUN NO. 5



APO, IN. WATER

LOAD POINT AND FLOOD POINT

RUN No. 6



A Po, IN. WATER

(67)

EXPERIMENTAL DETERMINATION OF DEMISTER

LOAD POINT AND FLOOD POINT

RUN NO. 8

.



A Po, IN. WATER

1001

RUN No. 9



APO, IN. WATER

(69)

EXPERIMENTAL DETERMINATION OF DEMISTER

LOAD POINT AND FLOOD POINT

RUN NO. 10 \$ 15



APO, IN. WATER

(70)

RUN No. 11



APO, IN. WATER

(71)

RUN NO. 12 (REPEAT OF RUN NO. 11)



APO, IN. WATER

(72)

EXPERIMENTAL DETERMINATION OF DEMISTER

LOAD POINT AND FLOOD POINT

RUN NO. 13.



APO, IN. WATER

(73)

EXPERIMENTAL DETERMINATION OF DEMISTER

LOAD POINT AND FLOOD POINT

RUN NO. 14



APO, IN. WATER

(74)

(75)

EXPERIMENTAL DEFERMINATION OF DEMISTER LOAD POINT AND FLOOD POINT

RUN No. 16



APO, IN. WATER

FIGURE 3-14

EXPERIMENTAL DETERMINATION OF DEMISTER

LOAD POINT AND FLOOD POINT

RUN NO. 17



0.1

APo, IN. WATER

(76)

10

(77)

RUN NO. 18



A Po, IN. WATER

EXPERIMENTAL DETERMINATION OF DEMISTER LOAD POINT AND FLOOD POINT

RUN No. 19



AP., IN. WATER

(78)

(79)

RUN No. 20



AP., IN WATER

(00)

RUN No. 21



APO, IN. WATER

(ST)

RUN No. 22



DPO, IN WATER

RUN No. 23



A Po, IN. WATER

1021



RUN NO. 24



A Po, IN. WATER

RUN No.25



APO, IN. WATER

(04)

EXPERIMENTAL DETERMINATION OF DEMISTER LOAD POINT AND FLOOD POINT

RUN No. 26



AP., IN. WATER

(85)

RUN NO. 27



APO, IN. WATER

(86)





APO, IN. WATER



A Po, IN. WATER



(89)

RUN No. 31



APO, IN. WATER

EXPERIMENTAL DETERMINATION OF DEMISTER LOAD POINT AND FLOOD POINT

RUN NO.32



AP., IN. WATER

(90)

RUN NO.33



A Po, IN. WATER

(91)

EXPERIMENTAL DETERMINATION OF DEMISTER LOAD POINT AND FLOOD POINT

(92)

RUN NO.34



APO, IN. WATER

0,1

(93)

RUN No.35



APo, IN. WATER



EXPERIMENTAL DETERMINATION OF DEMISTER LOAD POINT AND FLOOD POINT

RUN NO.36



AP., IN. WATER

0.1

APPENDIX C

PHYSICAL PROPERTIES OF TEST LIQUIDS



TIME ON SAMPLE, MINUTES

FIGURE C-1

FIGURE C-2 VISCOSITY OF GLYCERINE-WATER MIXTURES



APPENDIX D

DATA CORRELATION
TABLE D-1 CALCULATED DATA FOR FIGURES 7 AND 8

	- [0 0.	•		GL a RG	$v_{\rm F}^{2} a e_{\rm G} (\mu_{\rm L})^{0.2}$
RUN <u>NO.</u>	V Re	AL.	K F	G V RL	32.2E ³ CL
1	29.6	0.33	0.38	0.00470	0.515*
2	29.6	0.34	0.37	0.00458	0.495*
5	29.6	0.37	0.41	0.00424	0.591*
6	29.6	0.31	0.52	0.00326	0.395
8	29.6	0.42	0.45	0.00108	0.748*
9	29.6	0.48	0.53	0.000910	0.411
10	29.6	0.47	0.48	0.00376	0.337
11	29.6	0.45	0.55	0.000447	0.432
12	29.6	0.44	0.56	0.000438	0.446
13	29.6	0.41	0.48	0.000515	0.795*
14	29.6	0.35	0.36	0.00490	0.478*
16	30.4	0.43	0.48	0.00340	0.382
17	30.7	0.30	0.32	0.00464	0.462#
18	30.7	0.35	0.38	0.000923	0.622#
19	30.8	0.39	0.49	0.000717	0.410
20	30.9	0.47	0.55	0.000788	0.547
21	31.1	0.35	0.39	0.00145	0.721*
22	31.4	0.36	0.38	0.00140	0.725*
23	31.6	0.47	0.53	0.000880	0.600
24	31.9	0.46	0.54	0.000285	0.683
25	31.7	0.43	0.55	0.000388	0.713
26	31.8	0.45	0.50	0.00203	0.583
27	32.0	0.29	0.31	0.00285	0.568*
28	32.0	0.31	0.33	0.00185	0.672*
30	30.8	0.33	0.41	0.00145	0.747*
31	27.1	0.26	0.28	0.00525	0.385*
32	27.1	0.34	0.40	0.00356	0.310
33	27.1	0.41	0.45	0.000292	0.976 *
34	27.1	0.31	0.34	0.00147	0.560*
35	27.1	0.41	0.47	0.00107	0.422
36	27.1	0.32	0.38	0.00149	0,681*

*Denotes 421 demister, other data are for 931 demister.

TABLE D-2 MURTESSION COEFFICIENTS FROM STEPHIDE LINEAR NEGRESSION ANALYSIS OF FLOODING VELOCITY

		REGRESSIC	DN COEFFI DARD ERRO DN COEFFI	CIENTS R (S.E. CIENTS	(R.C.)) CF			
STEP NO.		$ln(a/e^3)$	ln(44)	<u>ln(?.)</u>	<u>ln(G2</u>)	<u>ln(Y)</u>	<u>ln(e)</u>	S.E. UF CORR,
1	R.C. 5.E.			0.615 0.008	-	•	40 40	0 .193
2	R.C. S.E.	-0.308 0.046	-	0.938 0.048	-	- -	-	0.122
3	R.C. S.S.	-0.284 0.040	-	1.01 0.047	-0.079 0.025		-	0.106
4	R.C. S.E.	-0.290 0.028	-0.087 0.015	1.07 0.034	-0.106 0.017	-	-	0,072
5	R.C. S.E.	-0.302 0.029	-0.082 0.016	0.922 0.127	-0.106 0.017		0.646 0.549	0.072
6	R.C. S.E.	-0.301 0.029	- 0 .036 0.044	0.466 0.429	-0.108 0.017	0.201 0.181	1.70 1.09	0.072
7	R.C. S.E.	-0.301 0.029		0.144 0.156	-0.108 0.017	0.338 0.062	2.46 0.56	0.071
		F	DR LOADIN	IG VELOC	ITY			
1	R.C. S.E.		-	0.583 0.007	-		-	0.170
5	R.C. S.E.	-0.238 0.047	-	0.832 0.050	-	-	-	0.126
3	R.C. S.E.	- 0.217 0.044	-	0.892 0.051	-0.068 0.027		-	J . 116
4	H.C. S.E.	~0.222 0.039	-0.063 0.021	0.935 0.048	-0.088 0.025	-	-	0.102
5	R.C. S.Z.	-0.226 0.042	-0.062 0.022	0.891 0.184	-0.088 0.025	-	0.196 0.794	0.104
6	R.C. S.E.	-0.225	-0.033 0.065	0.608 0.633	-0.089	0.125	0.85 1.61	0.106

(100)

		TABLE	$\overline{D-3}$		(101)
	REGRE	SSION OUTPU	T, STEP 6	FLOUDING-	VELOCITY
`	PRED	ICTED VS ACTUAL	RESULTS		
	RUN NO.	ACTUAL	PREDICTED	DEVIATION	WEIGHT
<u> </u>		in VE VE	en VF VF	AlnVi	
<u>}</u>	. 1.	2.42480 11.3	2.40179 11.0	✓ 0.02301	1.00000
	2	2.39334 10.9	2.4021011/0		1.00000
、	<u> </u>	2.48051 12.0	2.40082110	0.040575	1.00000
2 2 3 8		2.14004 1515	2.53756 12.1	V.05320	1.00000
· · · · ·	60	2.76254 15.9	2.81154 16:16	-0.04900	1.00000
<u>ک</u>	210	2.66236 14.4	2.66527 14.4	-0.00291	1.00000
	8 11	2.79117 4.3	2.88662 18.0	-0.09545	1.0000
•	912	2.80940 16.6	2.88482 18.0	-0.07542	1.00000
	10 13	2.64688 14,2	2.60213 13:5	V 0.C4475	1.00000
•	1714	2.37304 1017	2.39860 11.0	v-0.02555	1.00000
	1216	2.68102 14.6	2.64276 14.1	0.03826	1.00000
) 	13_17	2.29556 9.9	2.3/615 1019	V -0.08059	1.00000
•	17 18	2.40130 11.7	2.52900 12.6		1.00000
\	<u>19 19</u>	2.11463 154	2 76012 1613	0.06370	1.00000
	10,20	2 50389 11 5	2.10742 16.0	0.04628	1.00000
	18.22	2.47317 119	2.47441 1.9	-0.00124	1.00000
	1923	2.82316 11.8	2.75472 15.7	0.06845	1.00000
······································	2024	2.84491 17.2	2.87034 17.6	-0.02543	1.00000
	2125	2.84955 17.3	2.82688 16.9	0.02267	1.00000
······································	22.26	2.77384 16.0	2.66045 14.3	0.11339	1.00000
	23.27	2.28544 9.9	2.39108 10.9	✓-0.10564	1.00000
	24 28	2.36462 10.6	2.42234 11.2	V-0.05772	1.00000
·	25 30	2.53052 12.6	2.47412 1.8	V 0.05640	1.00000
	20 31	2.03/32 7.7	2.14/50 6.0		1.00000
·······	28.32	2.50001 10.9	2.42001 11.5	-0.04001 V 0.1002h	1.00000
ĸ	20 24	2.23323 4.3	2.26393 A.	-0.03070	1.00000
	311.35	2.54553 12.8	2.53858 11 14	0.00695	1.00000
)	33 6	2.33020 10.3	2.25112 9,5	0.07908	1.00000
×					
· · · · ·	· · · · · · · · · · · · · · · · · · ·			<u> </u>	
		V DENOTE	S 421 DE	MISTER	
	······································		·		
8 8				· ·	
t t		· · · · · · · · · · · · · · · · · · ·			<u> </u>
				· · · · ·	
	• •				
	· · · · · · · · · · · · · · · · · · ·				
			к	,	
	·		· · · · · · · · · · · · · · · · · · ·		
۰.				· · ·	
		· · · · · · · · · · · · · · · · · · ·			······································
o b	· · · · · · · · · · · · · · · · · · ·				
Υ.Γ	· · ·		, <u></u>		
<u></u>		· · · · · · · · · · · · · · · · · · ·	·	·	

·	TABLE D-9		(102)
	REGRESSION OUTPUT, STEP 6, LO.	ADING VELOCI	TY
······································			
	PREDICTED VS ACTUAL RESULTS		NETCHT
	Pur VE VE Pur VE VE	1 kn VE	ML1011
	1 2.28238 9.8 2.29048 99 -	0.00810	1.00000
	2 2.31254 10.1 2.29145 9.9	0.02109	1.00000
	-3 5 2.39790 11.0 2.29085 9.9	0.10705	1.00000
	4 - 6 = 2.20827 9 + 2.49885 / 2.2 - 100	1.29057	1.00000
	$\frac{2}{2} \frac{2}{2} \frac{2}{2} \frac{2}{11} \frac{1}{10} \frac{1}$		1.00000
1 [*]	$D_{10} = 2.62467 + 13.9 = 2.48838 + 13.9 = 12.9$	13629	1.00000
	8 11 2.59525 13.4 2.67061 144 -	07535	1.00000
	912 2.56495 13.0 2.66975 14.4 -	1.10480	1.00000
	10 13 2.48491 12.0 2.45974 11.7	02517	1.00000
······	11 14 2.34181 10.4 2.28814 4.9	0.05367	1.00000
	T2 16 2.57261 13.1 2.48662 12.0	.08599	1.00000
	$\frac{13}{17} 2.23001 9.3 2.29013 9.9 -100000000000000000000000000000000000$).06011	1.00000
	$\frac{147}{9} = \frac{2.50005}{100} = \frac{100}{2.4105} = \frac{177}{100} = \frac{100}{2}$	1 • VD / 40	1.00000
<u> </u>	$\frac{13/9}{1620} = \frac{2.47521}{2.67115} = \frac{2.62151}{1501} = \frac{1501}{2.60170} = \frac{1500}{2.60170} = \frac{1500}{2.600} $	1.07235	1.00000
I.	10 20 2.01415 1415 2.00117 155	02079	1.00000
1	1822 2.42480 11.3 2.37972 10.8	04508	1.00000
, ·	1923 2.70136 14,9 2.59292 13,4	0.10844	1.00000
······································	2024 2.69463 14,8 2.68891 14.8).00572	1.00000
	2125 2.60269 13.5 2.65367 142 -).05098	1.00000
	2226 2.66723 14.4 2.51753 12.4).14976	1.00000
••••••••••••••••••••••••••••••••••••••	$\frac{9327}{2.21920} = \frac{2.21920}{9.2} = \frac{2.31891}{10.2} = \frac{1}{10}$	1.09970	1.00000
	$\begin{array}{cccccccccccccccccccccccccccccccccccc$	1.00344	1.00000
	$\frac{25333}{2632} = \frac{2531234}{100} \frac{100}{100} = \frac{25314430}{100} \frac{100}{100} = \frac{1000}{100} = \frac$	12459	1.00000
	277_{37} 2.29253 as 2.27889 as	01365	1.00000
· · · ·	28 33 2.41591 12 2.28565 4.8	.13027	1.00000
÷	29 34 2.12823 8.4 2.16676 8.7 -).03853	1.00000
1	39 35 2.39790 11.0 2.37191 10.7	.02599	1.00000
	31 36 2.15176 8.6 2.15617 8.6 -	3.00441	1.00000
	V DENOTES 421 DEMISTER		
	V DENOTES 421 DEMISTER		
	V DENOTES 421 DEMISTER		
	V DENOTES 421 DEMISTER		
	V DENOTES 421 DEMISTER		
L	V DENOTES 421 DEMISTER		
L	V DENOTES 421 DEMISTER		
L	V DENOTES 421 DEMISTER		
ь р	V DENOTES 421 DEMISTER		
L	V DENOTES 421 DEMISTER		
L	V DENOTES 421 DEMISTER		
L	V DENOTES 421 DEMISTER		
L	V DENOTES 421 DEMISTER		
L	V DENOTES 421 DEMISTER		
	V DENOTES 421 DEMISTER		
L	V DENOTES 421 DEMISTER		
L	V DENOTES 421 DEMISTER		
L	V DENOTES 421 DEMISTER		
L	V DENOTES 421 DEMISTER		

REFERENCES

- Schroeder, H.F., Masters Thesis, Newark College of Engineering, June, 1962.
- 2. Poppele, E.w., Masters Thesis, Newark College of Engineering, June, 1958.
- 3. York, O.H., Chemical Engineering Progress, Vol.50, 1954.
- 4. York, O.H., and Poppele, E.W., <u>Mire Mesh Mist Eliminators</u>. Chemical Engineering Progress, Vol 59, June, 1963.
- 5. Sherwood, T.K., Shipley, G.H., and Holloway, F.A.L., <u>Flooding Velocities in Packed Columns</u>, Industrial and Engineering Chemisty, Vol 30, July, 1938.
- 6. Souders, M., and Brown, G.G., Industrial and Engineering Chemistry, Vol 26, 1934.
- Perry, J.H., ed., <u>Chemical Enginees' Handbook</u>, Third Edition, pages 597-98, 840, 1017-1020, Mc Graw-Hill Book Company, Inc. 1950.
- 8. Stearns, Jackson, Johnson and Larson, Flow Measurement with Orifice Meters, D. Van Nostrand Company, Inc., 1951.

(103)